EASA	AIRWORTHINESS DIRECTIVE	
X	AD No.: 2007- 0226	
K	Date: 24 August 2	007
No person may operate requirements of that Air	an aircraft to which an Airwo	orthiness Directive applies, except in accordance with the otherwise agreed with the Authority of the State of Registry.
Type Approval Holder's Name :		Type/Model designation(s) :
ATR-GIE Avions de Transport Régional		ATR 42 and ATR 72 series
TCDS Number : E	ASA A.084	
Foreign AD : Not :	applicable	
Supersedure : EA	SA AD 2006-0186 release	ed on 03 July 2006
ATA 28	Fuel - Fuel Tank System Wiring & Sensors -Modification/Replacement Fuel Tank Safety	
Manufacturer(s):	ATR-GIE Avions de Transport Régional (formerly AEROSPATIALE- AERITALIA, AEROSPATIALE-ALENIA, AEROSPATIALE ATR-ALENIA, EADS ATR-ALENIA)	
Applicability:	All ATR 42 series airci have received ATR mo (SB) ATR42-28-0039 and	raft models, all serial numbers, except aircraft that odification 05355 in production or ATR Service Bulletin Revision 3 or any later approved revision in service
	All ATR 72 series aircraft models, all serial numbers, except aircraft that have received ATR modification 05356 in production or ATR SB ATR72-28-1019 Revision 5 or any later approved revision in service.	
Reason:	Further to the acciden published a set of new Special Federal Aviati	t of Boeing 747-131 (flight TWA 800), the FAA has v rules related to the fuel tank safety, including the on Regulation 88 (SFAR 88).
	The JAA has issued an Interim Policy JAA INT/POL 25/12, to recommend the application of a similar requirement to the National Aviation Authorities (NAA).	
	This recommendation was followed by French DGAC, which rendered the compliance to JAA INT/POL 25/12 mandatory for all ATR Aircraft.	
	Under this regulation, all holders of type certificates are required to conduct a design review of their fuel tank systems against explosion risk. It also requires the development and implementation of maintenance and inspection instructions to maintain the safety of the fuel tank system.	

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	To answer JAA INT/POL 25/12, and in accordance with SFAR 88 requirements and guideline, ATR carried out a safety review on the fuel tank systems and zones adjacent to the fuel tanks on all ATR models using relevant safety assessment methods of JAR 25.1309.
	As a result of this safety review, ATR developed for ATR 42 the modification 05355 (SB ATR42-28-0039), and for ATR 72 the modification 05356 (SB ATR72-28-1019). Those modifications consist in the installation of fuses adapters on wiring entering the fuel tanks and current limitation devices. For ATR 72 aircraft, the modification also requires replacement of the high level sensors with new sensors having shorter harness.
	Embodiment of these modifications or Service bulletins provides full compliance with JAR 25.1309 and eliminates the unsafe conditions in accordance with SFAR 88 and AC 25-981-1B guidelines.
	EASA Airworthiness Directive (AD) 2006-0186 was issued to mandate compliance with SB ATR42-28-0039 Revision 1 and SB ATR72-28-1019 Revision 1. Missing components or discrepancies depending on aircraft variants were evidenced during its application. Consequently, both SB's were revised to correct these situations.
	The present AD, which supersedes AD 2006-0186, requires application of SB ATR42-28-0039 Revision 3 and ATR72-28-1019 Revisions 5 as well as some additional work for aircraft to which a previous revision of those SBs has been applied.
Effective Date:	07 September 2007
Compliance:	<ul> <li>Required before 31December 2009, unless accomplished previously.</li> <li>1. For all ATR aircraft NOT ALREADY MODIFIED per any previous revisions of SB ATR42-28-0039 or ATR72-28-1019. Modify the fuel system, in accordance with instructions of SB ATR42-28-0039 Revision 3 for ATR 42 series aircraft or SB ATR72-28-1019 Revision 5 for ATR 72 series aircraft.</li> </ul>
	2. For ATR 42 aircraft ALREADY MODIFIED per a previous revision of SB ATR42-28-0039.
	<ul> <li>(a) On aircraft in compliance with SB ATR42-28-0039 Revision 1.</li> <li>Install all of the components delivered in the kit S280039A04.</li> <li>Perform additional works as detailed in the Transmittal Sheet of SB ATR42-28-0039 Revision 3.</li> </ul>
	<ul> <li>(b) On aircraft in compliance with SB ATR42-28-0039 Revision 2.</li> <li>Perform additional works as detailed in the Transmittal Sheet of SB ATR42-28-0039 Revision 3.</li> </ul>
	3. For ATR 72 aircraft ALREADY MODIFIED per a previous revision of SB ATR72-28-1019.
	(a) On aircraft in compliance with SB ATR72-28-1019 Revision 1.
	<ul> <li>Install all of the components delivered in the kit S281019A03 and in the kit S281019A04.</li> <li>Perform additional works as detailed in the Transmittal Sheet of SB ATR72-28-1019 Revision 4.</li> </ul>
	<ul> <li>Verify the backshell for correct installation as detailed in the Transmittal Sheet of SB ATR72-28-1019 Revision 5</li> </ul>

	(b) On aircraft in compliance with SB ATR72-28-1019 Revision 2.		
	<ul> <li>Install all of the components delivered in the kit S281019A04.</li> <li>Perform additional works as detailed in the Transmittal Sheet of SB ATR72-28-1019 Revision 4.</li> </ul>		
in the second	<ul> <li>Verify the backshell for correct installation as detailed in the Transmittal Sheet of SB ATR72-28-1019 Revision 5</li> </ul>		
	(c) On aircraft in compliance with SB ATR72-28-1019 Revision 3.		
	<ul> <li>Perform additional works as detailed in the Transmittal Sheet of SB ATR72-28-1019 Revision 4.</li> </ul>		
	<ul> <li>Verify the backshell for correct installation as detailed in the Transmittal Sheet of SB ATR72-28-1019 Revision 5</li> </ul>		
	(d) On aircraft in compliance with SB ATR72-28-1019 at Revision 4.		
	<ul> <li>Verify the backshell for correct installation as detailed in the Transmittal Sheet of SB ATR72-28-1019 Revision 5</li> </ul>		
Ref. Publications:	Service Bulletin ATR42-28-0039 Revision 3		
	Service Bulletin ATR72-28-1019 Revision 5		
	or later EASA approved revisions.		
Remarks :	<ol> <li>If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD.</li> </ol>		
	<ol> <li>This AD was posted on 21 June 2007 as PAD 07-102 for consultation until 19 July 2007. No comments were received during the consultation period.</li> </ol>		
	<ol> <li>Enquiries regarding this Airworthiness Directive should be referred to the AD Focal Point - Certification Directorate, EASA.</li> </ol>		
	E-mail: ADs@easa.europa.eu		
	<ol> <li>For any questions concerning the technical content of the requirements in this AD, please contact:</li> </ol>		
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