NTSB ID: NYC95FA221	Aircraft Registration Number: N22BX
Occurrence Date: 09/10/1995	Most Critical Injury: Fatal
Occurrence Type: Accident	Investigated By: NTSB

Air Medical Transport Flight: No

Location/Time

Nearest City/Place
AKRON

State
OH

44306

Local Time
EDT

Time Zone
EDT

Distance From Landing Facility:

Aircraft Information Summary

Aircraft Manufacturer Model/Series Type of Aircraft

ROBINSON R-22 BETA /R-22 BETA Helicopter

Revenue Sightseeing Flight: No

#### Narrative

Brief narrative statement of facts, conditions and circumstances pertinent to the accident/incident:

HISTORY OF FLIGHT

On September 10, 1995, at 1750 eastern daylight time, a Robinson R-22 Beta, N22BX, was destroyed when it impacted the ground after takeoff at the Akron-Fulton Airport, Akron, Ohio. The private pilot was fatally injured. Visual meteorological conditions prevailed, for the personal flight that originated from Akron-Canton, Ohio, about 1735. No flight plan had been filed for the flight conducted under 14 CFR Part 91.

The pilot departed the Akron-Canton traffic pattern in N22BX, for the Akron-Fulton Airport (AKR). According to witnesses near AKR, the helicopter was observed hovering on the airport before takeoff. The helicopter departed to the north and climbed to about 200 feet, where it was observed to "wobble."

A police officer who witnessed the accident stated:

...I heard what sounded like an aircraft engine suddenly running at full throttle. Looking up, I observed a small helicopter two to three hundred feet off the ground. The helicopter was at an odd attitude, with the tail at a down angle and wobbling back and forth. While I watched, the craft pitched sideways to the right and began to spin out of control, with the aircraft body revolving 360 degrees about 4 times, before impact. The tail rotor appeared to be spinning slower than normal as the blades were visible instead of the usual blur. Impact was in a bulldozed field...

Another police officer who witnessed the accident stated:

...I saw the helicopter come across...It was swaying from left to right. It then made a U-Turn, and started to head back towards the airport. At that point it began to spin in circles and came straight down...

The accident occurred during the hours of daylight about 41 degrees, 02 minutes north latitude, and 81 degrees, 28 minutes west longitude.

#### PILOT INFORMATION

The pilot, Mr. Gerald P. Nye, held a Private Pilot Certificate with ratings for airplane single engine land, rotorcraft helicopter, and instrument airplane.

His most recent Federal Aviation Administration (FAA) Third Class Medical Certificate was issued on October 19, 1993.

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Narrative (Continued)

Mr. Nye's pilot log books revealed that he had about 2,575 hours of total flight experience, of which 417 hours were in helicopters. According to the log books, he had not logged any helicopter flight hours between April 1991, and August 30, 1995. He logged a total of 10 hours of dual instruction in the R-22, on August 30, and September 1, 1995. His estimated total flight experience in the R-22 was 13 hours, which included 3 hours that was logged on the helicopter's hobbs meter, between August 11, 1995, and the accident.

#### WRECKAGE INFORMATION

The helicopter wreckage was examined at the accident site on September 11 and 12, 1995. The examination revealed that all major components of the helicopter were accounted for at the scene. The helicopter came to rest upright, on a magnetic bearing of 195 degrees, on a level dirt area about 400 feet north of runway 27.

Examination of the wreckage revealed that the lower right, aft side of the fuselage, was crushed upward. The tailboom was buckled upward at the fuselage attaching point, and about 4 feet forward of the tail rotor gear box. The examination further revealed no pre-impact failure of helicopter systems or structures.

Control continuity was confirmed from the pilot's flight controls, to the cyclic, collective and tail rotor. Drive train continuity was confirmed from the engine, through the transmission, to the main rotor. Continuity was also confirmed to the tail rotor drive shaft; however, the shaft was observed to be buckled and separated near the tail rotor gear box. There were no rotational scratches on the inside tailboom paint, where the shaft end was laying against the tailboom.

The main rotor blades displayed symmetrical upward coning, with corresponding chord wise wrinkling. There was no leading edge damage or marks on the rotor blades. A few chord wise scratches were observed on the yellow paint, on the lower side of both blades tips. Yellow paint was observed on the white tailboom, in the vicinity of the crushed area, where the tail rotor drive shaft was separated inside the boom.

One tail rotor blade was bent over at a 90 degree angle, a few inches from the hub. Neither of the tail rotor blades displayed any chord wise scratches nor leading edge damage.

The main fuel tank was bulged outward, ruptured, and absent of fuel. The airframe fuel filter contained a few ounces of fuel. The fuel sample was tested with water finding paste and found to be absent of water.

The engine cooling fan was observed to be crushed on one side. No rotational scratches were observed. The engine was removed from the wreckage for further examination.

#### MEDICAL AND PATHOLOGICAL INFORMATION

An autopsy was performed on Mr. Gerald Nye, on September 11, 1995, by Dr. William A. Cox, Coroner of Summit County, Ohio.

Toxicological testing was conducted by the FAA Toxicology Accident Research Laboratory, Oklahoma City, Oklahoma.

#### TESTS AND RESEARCH

The engine was examined and test run at the Textron-Lycoming Facility, Williamsport, Pennsylvania, on October 31, 1995. Present during the examination were representatives from Lycoming, and the National Transportation Safety Board.

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Narrative (Continued)

The examination revealed that the timing of the left and right magnetos were set 4 and 5 degrees late, respectively. The carburetor fuel inlet finger screen, and the oil sump screen, were absent of debris. Due to impact damage, components replaced included the number two cylinder intake sump tube, number two and four intake pipe tubes, right magneto wiring harness, starter motor, and the carburetor pump inlet strainer housing (dome only). The engine was placed on a test stand, started, and ran within normal limits for about 30 minutes.

#### ADDITIONAL INFORMATION

The pilot purchased the helicopter on August 11, 1995, and helped ferry it to the Akron-Canton Airport. The pilot received the mandatory R-22 awareness training, required by SFAR 73, in West Mifflin, Pennsylvania, on August 31, and September 1, 1995. The training was not a qualification course, and the pilot did not receive a "sign-off" in the R-22. In a written statement, one of the flight instructors who flew with the pilot stated:

the fact that he had to take any training since he was already a rated helicopter pilot...he told me that he had 'a lot of time in the R22,' and had recently purchased one...[He] appeared knowledgeable about the operation of helicopters during the preflight briefing, but prior to liftoff, [he] asked me to 'stay close on the controls' since he did not have a lot of R22 time...By the end of the session [he] performed the awareness maneuvers satisfactorily...During the post flight briefing...I also discussed the differences between the R22 and the other helicopters that he had flown (especially in respect to rotor inertia), and recommended that when he returned to Akron he should seek additional training in the R22, especially since he did not appear from his logbooks to have much recent helicopter time. (Although [he] claimed he was current in helicopters, the last logbook he could produce showed no helicopter time since 1991)...As [he] was preparing to leave, [the flight school owner] strongly recommended that he receive more training at his home base...

In a written statement, the owner of the flight school stated:

...[The pilot] requested and received the ten hours of dual instruction required by SFAR 73...We trained him without the governor turned on, since he stated that his aircraft was not equipped. I recommended strongly that he have a governor installed...I recommended further that he seek out a local instructor, or return to us for more training prior to operating his aircraft...

The pilot's log books revealed that his helicopter flight experience was as follows:

Type Helicopter Total Hours Bell 47------355
Enstrom F-28-----49 Robinson R-22-----13

The pilot's log books also revealed a statement by a flight instructor that stated, "Needs additional work on basic aircraft control, settling with power, approach terminations, RPM-manual coordination." The pilot's total helicopter flight experience at that time was about 370 hours.

The helicopter, N22BX, was not equipped with an RPM Governor. According to the Robinson R-22 systems description:

The governor senses engine RPM changes and applies corrective input forces to the throttle; when RPM is low, it tends to increase the throttle and vice versa...The governor is designed to assist the pilot in controlling the RPM in the normal operating range...

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Narrative (Continued)

A review of the Robinson R-22 awareness training, revealed that of the 7 segments, 1 segment dealt with "Low RPM Rotor Stall," and another dealt with "Energy Management and Rotor RPM Decay."

A Robinson Helicopter Safety Notice, SN-10, discussed Low Rotor Stall. The Safety Notice stated:

...every pilot must have his reflexes conditioned so he will instantly add throttle and lower collective to maintain RPM in any emergency...Power available from the engine is directly proportional to RPM. If the RPM drops 10%, there is 10% less power. With less power, the helicopter will start to settle, and if the collective is raised to stop it from settling, the RPM will be pulled down even lower, causing the ship to settle even faster. If the pilot not only fails to lower collective, but instead pulls up on the collective to keep the ship from going down, the rotor will stall almost immediately. When it stalls, the blades will either "blow back" and cut off the tail cone or it will just stop flying, allowing the helicopter to fall at an extreme rate...

The helicopter wreckage was released on September 12, 1995, to the AKR airport manager. The wreckage and engine were released to Dr. Richard Simmons, the owners son, on October 31, 1995.

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AVIATION Occurrence Type: Accident										
Landing Facility/Approach Information										
Airport Name	port ID:	Airport Elevation	Run	way Used	Runwa	y Lengtl	h Rı	unway Width		
AKRON-FULTON INTL AKR			1068 Ft. MSL	_ 0						
Runway Surface Type:	•			·				·		
Runway Surface Condition:										
Approach/Arrival Flown:										
VFR Approach/Landing:										
Aircraft Information										
Aircraft Manufacturer ROBINSON		Model/ R-22		BETA			Serial I 1050	Number		
Airworthiness Certificate(s): Normal										
Landing Gear Type: Skid										
Amateur Built Acft? No Number of Seats							Numbe	r of Engir	nes: 1	
9 71						ated Power: 60 HP				
- Aircraft Inspection Information										
Type of Last Inspection Date of Last Inspection Time Since Last Inspection							Airframe Total Time			
100 Hour	0	6/1995				49 Ho	ours		2831 Hours	
- Emergency Locator Transmitter (ELT) Information										
ELT Installed?/Type No ELT Operated? ELT Aided in Locating Accident Site?										
Owner/Operator Information										
Registered Aircraft Owner		Street A	ddress PO BOX 4393	3						
GERALD P. NYE	City							Zip Code		
		Street A	AKRON					ОН	44321	
Operator of Aircraft		Olloctive	PO BOX 4393	3						
GERALD P. NYE  City  AKRON  OH						State OH	Zip Code 44321			
Operator Does Business As:	Operator Does Business As: Operator Designator Code:									
- Type of U.S. Certificate(s) Held: None				·						
Air Carrier Operating Certificate(s):										
Operating Certificate:			Operator Certific	cate:						
Regulation Flight Conducted Under: Part 91: Gene	ral Aviation	1	•							
Type of Flight Operation Conducted: Personal										
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First Pilot	Information			•										
Name City										Sta	ate	Date o	of Birth	Age
On File On							File On File On File					ïle	61	
Sex: M	Seat Occupied	: Right	Occ	cupational Pi	lot? Retire	d			С	ertifica	ate Numb	er: Or	n File	•
Certificate(s): Private														
Airplane Rating(s): Single-engine Land														
Rotorcraft/0	Glider/LTA: Helio	copter												
Instrument	Rating(s): Airpl	ane												
Instructor F	Rating(s): None	е												
Current Bie	nnial Flight Revie	ew?												
Medical Ce	rt.: Class 3	Medica	al Cert. Status	s: Valid Me	dicalw/ wa	aivers/l	im.		Date of	Last M	1edical E	xam: 1	10/1993	
		· ·												
- Flight Tim	ne Matrix	All A/C	This Make and Model	Airplane Single Engine	Airplane Mult-Engine	Night Instru Actual		Instrument Simulated		Rotorcraft		Glider	Lighter Than Air	
Total Time		2575	13	2158			405	60	)	82	41	7		
Pilot In Con	nmand(PIC)	2511	3	2134							37	'6		
Instructor												+		
Instruction			40	0.1										
Last 90 Day		37 14	13 13	24 1								3		
Last 30 Day		14	13	1		+			+	-		3		
Seatbelt Us		Shou	lder Harness	Used? Yes			Toxicology Performed? Yes Second					cond F	Pilot? No	<u> </u>
Flight Plan/Itinerary														
	ght Plan Filed: No	one												
Departure Point							State Airport Id			ort Identifier Depar			ime	Time Zone
CANTON							ОН САК		AK 1		1735		EDT	
Destination State Airport Identifier														
Same as Accident/Incident Location  AKR														
Type of Cle	earance: None							•			•			
Type of Air	space: Class	E												
Weather	Information													
Source of \	Wx Information:													
	No rec	ord of briefi	ng											
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AVIATION			Occ	Occurrence Type: Accident									
Weather	Information												
WOF ID	Observation Time	Time Zone	WOF	WOF Elevation WOF Distance From Accid			lent Site	From Accident S	iite				
CAK	1820	EDT	12	228 Ft. MSL					7 NM			180 Deg. Mag.	
	st Cloud Condition: Clear					0 Ft. AGL Condition of Light: Day							
Lowest Ce	eiling: None			0 Ft. AGL		Visibility:		10	SM Altimeter: 30.00			"Hg	
Temperatu		Dew Point:		4 °C We	ather	Condition	s at Accid	lent S	ite: Visual (	Cond	litions		-
Wind Direc	ction: 10	Wind Sp	eed: 8			Wind Gu	usts:						
Visibility (F	RVR): 0 Ft.	Visibility	(RVV)	0 SM	丌								
Precip and	d/or Obscuration:												
Accident	Information												
Aircraft Damage: Destroyed Aircraft Fire: None					ne				Aircraft Exp	olosio	n None		
- Injury Su	mmary Matrix	Fatal	Serious	Minor	Nor	ne T	OTAL						
First Pi	ilot	1					1						
Second	d Pilot												
Studen	nt Pilot												
Flight I	nstructor												
Check	Pilot												
Flight E	Engineer												
Cabin A	Attendants												
Other C	Crew												
Passer	ngers												
- TOTAL A	ABOARD -	1					1						
Other C	Ground	0	0	0			0						
- GRANE	D TOTAL -	1	0	0			1						

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	ormation

Investigator-In-Charge (IIC)

ROBERT L. PEARCE

Additional Persons Participating in This Accident/Incident Investigation:

JAMES DAVIDSON CLEVELAND, OH

SHERWOOD BRESLER TORRANCE, CA