Quantification of Back-End Nuclear Fuel Cycle Metrics Uncertainties Due to Cross Sections

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A Thesis Submitted to the Graduate Faculty of North Carolina State University in Partial Fulfillment of the Requirements for the Degree of Master of Science in Nuclear Engineering

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Abstract

This work examines uncertainties in the back end fuel cycle metrics of isotopic composition, decay heat, radioactivity, and radiotoxicity. Most advanced fuel cycle scenarios, including the ones represented in this work, are limited by one or more of these metrics, so that quantification of them becomes of great importance in order to optimize or select one of these scenarios. Uncertainty quantification, in this work, is performed by propagating cross-section covariance data, and later number density covariance data, through a reactor physics and depletion code sequence. Propagation of uncertainty is performed primarily via the Efficient Subspace Method (ESM). ESM decomposes the covariance data into singular pairs and perturbs input data along independent directions of the uncertainty and only for the most significant values of that uncertainty. Results of these perturbations being collected, ESM directly calculates the covariance of the observed output posteriori. By exploiting the rank deficient nature of the uncertainty data, ESM works more efficiently than traditional stochastic sampling, but is shown to produce equivalent results. ESM is beneficial for very detailed models with large amounts of input data that make stochastic sampling impractical.

In this study various fuel cycle scenarios are examined. Simplified, representative models of pressurized water reactor (PWR) and boiling water reactor (BWR) fuels composed of both uranium oxide and mixed oxides are examined. These simple models are intended to give a representation of the uncertainty that can be associated with open uranium oxide fuel

cycles and closed mixed oxide fuel cycles. The simplified models also serve as a demonstration to show that ESM and stochastic sampling produce equivalent results, because these models require minimum computer resources and have amounts of input data small enough such that either method can be quickly implemented and a numerical experiment performed. The simplified models are followed by more rigorous reactor physics and depletion models showing a PWR uranium oxide fuel and various metal fast reactor fuels composed of transuranics. The more rigorous models include multi-group cross sections, multiple burnup steps, neutron transport calculations to update cross sections, and multi-scale multi-physics code sequences to simulate a complete fuel lifetime. Finally, the fast reactor and PWR fuels are combined in a closed fast reactor recycle fuel cycle, and uncertainties on the resulting equilibrium cycle examined.

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Nomenclature

ALWR – Advanced Light Water Reactor

BOC – Beginning of Cycle

BOL – Beginning of Life

BWR – Boiling Water Reactor

BU - Burnup

CR - Conversion Ratio

ENDF – Evaluated Nuclear Data File

EOC – End of cycle

EOL – End of Life

ESM – Efficient Subspace Method(s)

FP – Fission Product(s)

FR – Fast Reactor

GWD – Giga Watt Days

LWR – Light Water Reactor

MOX – Mixed Oxide

MTHM – Metric Ton Heavy Metal

MTIHM – Metric Ton Initial Heavy Metal

MTU – Metric Ton Uranium

PWR – Pressurized Water Reactor

TRU – Transuranic(s)

UOX – Uranium Oxide

w/o - Weight Percent

QUANTIFICATION OF BACK-END NUCLEAR FUEL CYCLE METRICS UNCERTAINTIES DUE TO CROSS SECTIONS

1. Introduction

1.1. Importance to the Nuclear Fuel Cycle

Over the next several years, policy makers will be assessing the deployment of various components of the nuclear fuel cycle, e.g. the Yucca Mountain repository, reprocessing plants, new reactors, etc. This research will be conducted in conjunction with the SINEMA (Simulation Institute for Nuclear Energy Modeling and Analysis) project headed by Idaho National Laboratory, which aims to produce a computational tool to be provided to policy makers for the assessment and comparison of various fuel cycle scenarios [1]. The objective of this work is to develop uncertainty propagation techniques to assess the affect of certain design and operation parameters on back-end fuel cycle metrics that are of key importance in various fuel cycle scenarios. Comparing two fuel cycles might be irrelevant if the uncertainty in a key metric between them overlaps.

Key metrics will hereinafter be defined as anything that is a limiting factor for the technology or facilities which are deployed in the current nuclear fuel cycle or may be deployed in future advanced fuel cycle scenarios. The nearest future deployment seems to be the spent fuel repository to be located at Yucca Mountain, Nevada. The repository's capacity is currently limited by the heat produced by the decay of the spent fuel such that the temperature between the repository tunnels remains below the local boiling temperature of water. Heat load is dominated by fission products in the first 1500 years, when peak heat production occurs and by minor actinides thereinafter [2][3]. In the very distant future the waste packages are assumed to fail and the metric of concern is then what material is released, i.e. isotopic inventory and the radiotoxicity of the material released to the biosphere. It has been suggested that implementing a so-called advanced fuel cycle that

includes reprocessing of spent nuclear fuel could extend the lifetime of the repository by reusing fissile material and reclassifying inert material that would otherwise fill the repository quickly in the once through fuel cycle [4]. A good example is that greater than 95% of spent uranium oxide fuel is U-238, a low level waste that is safe enough to store somewhere other than the repository if separated out [5]. When considering the reprocessing of spent fuel for a mixed oxide fuel for a light water reactor, or an actinide fuel for a fast reactor, the concerns become radioactivity of the fuel, which facilities must contain, and the inventory of material which can be extracted from the fuel at the time of separation. Convenience of physical properties requires only the examination of the uncertainty that arises in isotopic inventories since heat, radioactivity, and radiotoxicity are linearly proportional to mass. Apart from the significant economical and political challenges of implementing advanced fuel cycles or operating a repository at all, e.g. high cost of reprocessing and poor public opinion [6], the nature of engineering requires designs to be built around safety margins which are limited by the metrics discussed above. Reducing uncertainty not only allows for a better evaluation of fuel cycles but also more economical and efficient designs of the associated infrastructure.

1.2. Cross Sections and Uncertainty

Reaction cross-sections, as part of Evaluated Nuclear Data File (ENDF) [7], provide a large amount of information that is essential to any nuclear calculation, e.g. the models that predict the behavior and operation of nuclear reactors and the resulting spent fuel. Since the aim of this work is to develop a generalized uncertainty propagation technique for nuclear models that demand large input data sets and produce large output parameter data sets, cross-sections' uncertainty is the source that will be examined. The uncertainty cross sections

contribute to the output parameters of discharged isotopic masses, decay heat load, radioactivity and radiotoxicity are the back-end nuclear fuel cycle metrics that are analyzed herein. Since the evaluation of these data is continuously being updated, emphasis is placed on their uncertainties – variances and covariances – with that data also made available in conjunction with the cross-sections themselves. The problem is augmented by the complex nature of cross-sections, measured as a function of the kinetic energy of the neutrons that are causing the reactions. Homogenization, or the averaging of a cross-section over a fixed energy range and/or spatial region, is often implemented to reduce the computational burden. The level of homogenization varies depending upon the application, ranging from hundreds of pieces of data for simple depletion, to millions of data pieces for precise in-core calculations. Considering that every material charged to, or created in, a reactor has many cross-sections for many different reaction types, even when represented by only one energy group, spatially homogenized over the entire core, the volume of uncertainty data is still large and propagating its affect on various metrics is a daunting task.

1.3. Review of Uncertainty Propagation Techniques

Uncertainty data allows uncertainty models to be applied and propagated through crucial parameters for evaluating the design system in question, such as reactor operation and the nuclear fuel cycle as a whole. "Propagating uncertainties is a non-trivial task because of the computational complexity often associated with the various modeling stages of the fuel cycles, and the size and type of different sources of uncertainties." [8] It is also beneficial to recognize that modeling uncertainties can be introduced through the numerical approximations that are typically found in models of complex systems, but for this study the

focus is on those sources of uncertainty that are inputs to the model, particularly crosssections input to a nuclear physics model.

The most basic analytical method is to perturb an input by some value and observe how the output is affected. While this approach efficiently arrives at a direct sensitivity of a model to an individual parameter, the investigator will usually only examine a few parameters due to the time requirements. Case in point is the work of E. Schneider [10] who introduced set perturbations into a few key cross sections and modeled the response of discharge isotopics to those perturbations. When considering huge volumes of input data such as thousands of cross-sections coupled with long CPU run-times of complex, multiphysic models, this method is very tedious and time consuming.

The classic approach to the uncertainty analysis of nuclear systems is the use of adjoint solutions that arrive at the sensitivity of a metric to all input parameters [9]. While the change in the metric to any change in that particular parameter is now known, the drawback is that m metrics will require m adjoint solutions [11]. If one follows this process to obtain sensitivity coefficients, S, for many parameters, for example S cross sections, S one arrives at a so-called sensitivity matrix \overline{S}_R . Note that here and throughout the remainder of this document, variables shown with a single bar are assumed to be vectors and variables with a double bar are assumed to be matrices. The uncertainty matrix of a metric to this set of parameters is easily obtained by multiplying the sensitivity matrix by the covariance matrix of the parameters by the transpose of the sensitivity matrix [11][12][13], producing what is sometimes called the "sandwich" equation. This classical approach has been studied and repeated, and consistently yields reliable and verified results. The work of S H. Aliberti, et. al. uses this approach to evaluate the uncertainty of reactor and fuel cycle parameters, e.g.

reactivity, decay heat, etc., in regards to cross-sections and is a valuable source with which to compare the results of this work.

The process can be very time consuming from a computational viewpoint because, every metric must have an adjoint solution and a set of sensitivity parameters evaluated. H. Abdel-Khalik of North Carolina State University has recently developed the Efficient Subspace Method (ESM) which approximates the behavior of a large, rank deficient matrix, such as the cross-section covariance matrix or the sensitivity matrix, in an effort to make computations more efficient [14]. ESM works most efficiently when the input data and the number of metrics of interest to be observed are both large. ESM also requires the problem to be illconditioned, as are many complex system problems. ESM can be implemented in existing models, but requires linear algebra operations to be applied via pre- and post- processors. In addition to its use for propagating cross-sections uncertainties, ESM has been harnessed for performing adaptive simulation of reactor core calculations. Adaptive simulation is an inverse theory approach that adjusts cross-sections to enhance the agreement between the measured and code-predicted core observables of interest, e.g. core power distribution, and core reactivity. Adaptive simulation is currently the focus of various research projects at NC State.

Another method of uncertainty propagation is the so-called forward perturbation method, which can either be deterministic or stochastic in nature [11]. The deterministic approach works best when the input data field is small because this method determines sensitivity by input data perturbation one piece at a time [15]. Because the input data set for cross-sections can be very large, this approach was not considered. Alternatively, the stochastic approach can be confidently used for a larger input data set and works well when

the amount of output data is large [15]. This method uses a Monte Carlo sampling (random or Latin-Hypercube [16]) of the total input data skewed by the data probability distributions. Many samples of inputs are run with existing models and probability distributions of output are determined directly from the results [11][15]. A study of the convergence of the distributions is often necessary to determine the number of samples needed to assure confidence in a specific problem.

Directly sampling the probability distribution is ideal only when the input parameters are independent [15]. In the case of covariance data for a large number of parameters input to a complex multi-scale model, two main issues arise: 1) in using random input samples to calculate outputs that are functions of many variables, some sample sets could be linearly dependent, i.e. the output could be approximated by a linear combination of previous samples, increasing the number of samples required because essentially the same sample is being repeated, and 2) covariance is defined as the expected variance of one random variable with respect to another random variable [17], which means that the probability distributions of input parameters are correlated, and that simply sampling a distribution of one parameter does not take into account its variance due to another. In the realm of linear algebra, covariances exist as the off diagonal elements of the covariance matrix and variances are the diagonal elements.

If the model is linear, both of these issues can be avoided by a single adjustment to the forward method. To account for correlations and to ensure that each set of samples is linearly independent, the covariance matrix is processed by singular value decomposition into eigenvalues and corresponding eigenvectors. The eigenvalues which are derived from the covariance matrix are used in the probability distributions and the eigenvectors, by

definition, are linearly independent. The samples used as input are a combination of samples from each of the eigen-pairs where the square root of the eigenvalue is the standard deviation of the sample [18]. When H. Kawano, et. al., used this procedure, it was applied only to the multi-group covariance matrix of the Pu-239 fission cross section. The resulting affects on criticality were subsequently examined and compared to a benchmark experiment. While fission of Pu-239 is very important for both uranium and mixed oxide fuels, as well as nuclear weapons, it is still just one reaction among many. When the covariance matrix is very sparse, this method yields another benefit for the analysis in that the eigenvectors of a sparse matrix will contain one element that is very close to 1 and the other elements will be very small. Thus, if one perturbs along only a single eigenvector at a time, the perturbation can be traced back to a single cross-section since one would have received the majority of the perturbation along that eigenvector. This is used to determine which cross-sections contribute most to the resulting uncertainty in the output.

In reviewing the methods available, both ESM and stochastic forward perturbation using the eigen-pair approach and random sampling show promise for such a problem as set forth in this work. It will be shown later that the models used in this work are nearly linear and converge after a reasonable number of samples to justify using either approach. In development of the propagation techniques in this work, the stochastic perturbation approach was used on simple LWR fuel models, namely uranium and mixed oxide fuels. Due to the fast execution time of simplified and somewhat crude models of these fuels and the linear algebra processors required for ESM, both of which will be addressed in Section 2, this appeared a prudent choice. The benefit of this simplified model is that one can compare the traditional stochastic method to the newly developed ESM. A validation experiment for

ESM, implemented within the simple model, shows that both methods produced equivalent results but that the stochastic method required less mathematical manipulation. When a much more detailed realistic fuel model is needed, however, e.g. many burnup steps in a neutron transport model using multi-group cross section data, stochastic methods become impractical and the use of the ESM becomes necessary. Such a model is the standard in practical fuel analysis and is also needed when the simplified models failed to provide needed resolution and linearity when examining fast reactor fuels. Due to the fact that the two approaches were determined to be equivalent, the move to this method, was made with confidence.

1.4. Overview of Computational Modeling Software

As already stated, the techniques developed in this work are implemented in preexisting fuel cycle models. Computational modeling programs are cornerstones of the
nuclear industry since full scale experiments are often not a pursuable approach. The preexisting models chosen for this study are the SCALE 5.0 software package available from
Oak Ridge National Laboratory, specifically, the ORIGEN depletion code, the SAS2H
sequence, and the TRITON sequence; and, the REBUS 3.0 code from Argonne National
Laboratory. Qualifications of the SCALE package include verified and validated models of
benchmark experiments augmented by package popularity, user-friendliness, and convenient
technical support from the developer [19]. Furthermore, the SCALE package also includes a
pre-formatted 44-group library containing variance and covariance information for a number
of key reactions types and isotopes. The REBUS model, which has also been verified and
validated, has a somewhat more difficult input structure, but was specifically designed for
fast reactor models [20].

ORIGEN is a time-dependent point-depletion analysis code that can track changes in concentrations of a large number of isotopes due to nuclear transmutation and radioactive decay. The program uses the matrix exponential expansion method to solve the Bateman depletion equations for any number of discrete points in time. ORIGEN can model nuclear fuel at various stages during the fuel cycle, including irradiation, storage, transportation, etc. ORIGEN operates with various library formats, the two most common being a card image library and a binary working library. The three-group card image library must be supplied by the user in the required format and include a corresponding three group flux spectrum in the ORIGEN input deck. SCALE is distributed with a three-group card image library, and its corresponding flux spectrum that is representative of a typical light water reactor. The typical flux spectrum is also available in 44-group and 238-group representations. The library type most often used is an AMPX formatted binary library. The master libraries containing basic ENDF data in 1-, 3-, 44-, and 238- groups are included with the SCALE package. Because SCALE is a multi-physics program, there are drivers and programs that can update the master library to create a problem-specific working library that is usable in ORIGEN. When ORIGEN uses a binary library, the cross sections applied are in either onegroup or three-group values that are representative of the specific problem that the working library was created for. This allows ORIGEN to execute very quickly and elminates the need to input a specially formatted card-image library or a fuel specific flux spectrum, which is already accounted for in the new cross-sections. [21]

The SAS2H sequence uses various codes within SCALE to produce a detailed model of a fuel assembly. SAS2H is a coupled one-dimensional depletion and shielding analysis sequence. SAS2H is designed to create a 1-D model of a specific fuel type and then track

various parameters -- reactivity, isotopics, dose rates in storage, etc. -- through the life of the fuel. The user supplies a fuel composition, geometry, power and decay history, and optionally, a storage cask description for disposal dose analysis. Problem specific, burnup dependent cross-sections are derived using two separate lattice cell models in a pseudo 2-D model that utilizes 1-D neutron transport modeling. The process also produces problem-dependent flux spectra in the same number and ranges of groups as the master library input to SAS2H. SAS2H uses the ORIGEN code to do all of its depletion analysis both for the incore depletion and out-of-core decay. While SAS2H was mainly designed to model light water reactor and research reactor fuels, it can also be used to create a crude fast reactor model if given the fuel composition and geometry for such a reactor. [22]

The TRITON sequence is also another all inclusive depletion analysis, like the SAS2H routine. Unlike SAS2H, however, TRITON solves the transport equation in a 2-dimensional geometry. TRITON is particularly used for modeling single fuel assemblies or individual Wigner cells, the latter of which will be used in this work. TRITON must be given buffer region input as it does not automatically account for non-fuel holes in the lattice like SAS2H. The biggest drawback is that TRITON was developed intentionally for commercial reactors whose fuel is by standard in a square lattice. While TRITON can model any number of polygon geometries within a given domain, the outer domain is forced to be rectangular, which is effective for square unit cells but lacks the resolution and proper moderator modeling abilities for other geometries, for example, a hexagonal cell for a fast reactor fuel. This work recognizes this shortcoming of the model and acknowledges that the results will not be absolutely accurate because of it. In its defense, TRITON is a much more

detailed model than SAS2H and overcomes some modeling inadequacies of SAS2H while maintaining all the analysis abilities. [23]

REBUS is used in the latter part of this study to compare the fast reactor results from TRITON, since the fast reactor models examined were created at Argonne using this code. Also, the many group cross section library and associated covariance matrix for REBUS is based upon a sodium cooled fast reactor flux spectrum, whereas the data available in the SCALE package is based upon a light water thermal reactor flux spectrum. REBUS was used since it has the unique ability to recycle fuel, using both reprocessing plants and external sources, and iteratively find some equilibrium fuel composition to meet operating parameters and cycle energy requirements, while using the available recycle feed. As used in this study, REBUS incorporates the DIF3D diffusion theory code utilizing the finite difference option. Hexagonal-z geometry for the core is modeled, with each hexagon representing a fuel assembly with homogenized cross sections employed. The drawback to REBUS is that a few-group covariance matrix did not exist a priori as it did with the SCALE package. Thanks to the work of Dr. Masood Iqbal and Dr. Hany Abdel-Khalik, a 15-group covariance matrix [24] for key reaction types and isotopes was created specifically for REBUS at North Carolina State University using the Argonne cross section processing code MC² 2 [25]. Dr. Hany Abdel-Khalik also implemented the efficient subspace method (ESM) of uncertainty propagation in REBUS.

1.5. Fuel Types and Scenarios of Interest

The most logical place to begin the analysis of uncertainty in various fuel types is to first analyze the fuel of the current reactor fleet deployed in the U.S. – low enriched uranium oxide fuel. Care is taken to select, directly or similarly from other studies, fuel types that

represent an actual equilibrium cycle fuel or a fuel for a predicted equilibrium cycle, i.e. not a specialized fuel designed for start-up cores or demonstration experiments. For the current reactor designs in this study, that fuel is a 4.5 w/o uranium oxide fuel burned to 40 GWD/MTU modeled first by the typical light water reactor information provided with SCALE, and then in both a pressurized water reactor and a boiling water reactor of various void fractions as modeled by SAS2H. To consider an advanced fuel cycle in the advent of a reprocessing infrastructure being considered in the U.S., mixed oxide and fast reactor fuels are also considered. Models include an ALWR MOX containing plutonium and uranium, a mixed oxide fuel with neptunium and americium impurities, and three fast reactor fuels, of various conversion ratios, made up of spent light water reactor fuel to burn off minor actinides. Finally, an experiment is conducted to demonstrate the effects of accumulating uncertainty in the input isotopics themselves as fuel is recycled in the fast reactor case. The former single pass fast reactor models are examined in TRITON and the latter fast reactor recycle scenario will be modeled both TRITON and REBUS. The fast reactor and its corresponding fuel types are modeled after Argonne's Advanced Burner Test Reactor [26].

2. Methodology

2.1. Use of SCALE Covariance Data

The SCALE 5.0 package is distributed with two 44-group covariance libraries, based on a light water thermal reactor flux spectrum, that contain information for approximately 700 nuclide-reaction pairs for many key isotopes. A full listing of all available data is too lengthy for this document but the reader is referred to the manual describing the library [31]. Effort is taken, through assumptions and model limits, to reduce this volume of data both to fit the input needed for models and to reduce the computational effort needed to implement the chosen uncertainty propagation technique. Unexpectedly, one of the assumptions made so the data will fit the ORIGEN code, actually expands the volume of information.

The covariance library containing information for most nuclide-reaction pairs is chosen as the data source for this work. The first reduction in data is to examine only the reactions that are important to reactor calculation for depletion analysis, and the only reactions ORIGEN uses -- neutron capture and fission. Those reactions in particular are: (n,γ) , (n,p), (n,α) , (fission), (n,2n), and (n,3n). The result is that covariance data for 701 nuclide-reaction pairs is reduced to 116 pairs by removing the cross-sections that are not of interest to depletion. For the simplified models, the perturbations are introduced into ORIGEN as the cross sections are read from the library, whereas with the more rigorous TRITON model, perturbations are made directly in the master cross section library before it is used by the code. When coupling a binary library to ORIGEN, generated by SAS2H as described later, the simplified model directly uses one-group cross-sections that are ideally representative of the specific problem. With this restriction on input data, the 44-group

neutron flux spectrum generated by SAS2H is used to collapse the 44-group covariance data to one-group values, instantly reducing the volume of data by a factor of 44 solely so that its affects can be applied directly to ORIGEN. With the exception of the typical library examined, which was prepared a priori by Oak Ridge, the flux spectra are generated for each simplified fuel/reactor examined in this work and the beginning of life total flux was chosen as a representative spectrum to be used for the collapse. This topic will be addressed again when discussing the TRITON results. For TRITON, perturbations obtained from the 44-group covariance library can be introduced directly into the 44-group master cross section library given as input to the code since both are of the same group structure. This eliminates all the pre-processing discussed above for the simplified models.

The covariance library contains data for ten materials in elemental form rather than the isotope specific reaction that ORIGEN uses. While this is of no consequence to TRITON which recognizes elemental forms and deals with them internally, the simplified models that use only ORIGEN for sampling need nuclide specific values. With this in mind, the data for those ten elements – magnesium, silicon, potassium, chromium, iron, nickel, copper, zirconium, hafnium and lead – is assumed to apply equally to isotopes of each element which are included in the cross section library. The result is the expansion of the data to a final value of 223 nuclide-reaction pairs that are considered in this work. Table 2.1 lists the nuclide-reaction pairs, using asterisks (*) to indicate data that were expanded from the elemental form and crosses (†) indicating pairs that have off-diagonal covariance data.

Nuclide	Reaction(s)	Nuclide	Reaction(s)	Nuclide	Reaction(s)
H-1	(n,γ)	Co-59	$(n,2n), (n,\gamma),$	Eu-153	(n,γ)
			(n,α)		
Li-6	(n,γ) (not used in	Ni-58*	$(n,\gamma), (n,p), (n,\alpha)$	Eu-154	(n,γ)
	ORIGEN)		(n,2n)		
Li-7	(n,γ)	Ni-59*	$(n,\gamma), (n,p), (n,\alpha)$	Eu-155	(n,γ)
			(n,2n)		
B-10 [†]	(n,p)	Ni-60*	$(n,\gamma), (n,p), (n,\alpha)$	Gd-154	(n,γ)
G 10		371 644	(n,2n)	01155	
C-12	$(n,\gamma), (n,p), (n,\alpha)$	Ni-61*	$(n,\gamma),(n,p),(n,\alpha)$	Gd-155	(n,γ)
27.14		3 T' (2 d)	(n,2n)	01156	
N-14	$(n,\gamma), (n,p), (n,\alpha)$	Ni-62*	$(n,\gamma),(n,p),(n,\alpha)$	Gd-156	(n,γ)
0.16) I. (34	(n,2n)	01157	
O-16	$(n,p),(n,\alpha)$	Ni-63*	$(n,\gamma),(n,p),(n,\alpha)$	Gd-157	(n,γ)
F-19	(201) (202) (200)	Ni-64*	(n,2n)	Hf-174*	(m v)
Г-19	$(n,\gamma), (n,p), (n,\alpha)$	111-04	$(n,\gamma), (n,p), (n,\alpha)$ (n,2n)	П1-1/4	(n,γ)
Na-23	$(n,\gamma), (n,p), (n,\alpha)$	Ni-65*	(n,2n) $(n,\gamma), (n,p), (n,\alpha)$	Hf-175*	(n,γ)
1Na-23	$(n,\gamma), (n,p), (n,\alpha)$ (n,2n)	111-03	$(n,\gamma), (n,p), (n,\alpha)$	111-173	(11,γ)
Mg-24*	(n,γ)	Ni-66*	$(n,\gamma), (n,p), (n,\alpha)$	Hf-176*	(n,γ)
1115 2 1	(11,7)	111 00	$(n,\gamma),(n,p),(n,\omega)$	111 170	(11,1)
Mg-25*	(n,γ)	Cu-63*	(n,γ)	Hf-177*	(n,γ)
Mg-26*	(n,γ)	Cu-64*	(n,γ)	Hf-178*	(n,γ)
Mg-27*	(n,γ)	Cu-65*	(n,γ)	Hf-179*	(n,γ)
Mg-28*	(n,γ)	Cu-66*	(n,γ)	Hf-180*	(n,γ)
A1-27	$(n,\gamma), (n,p), (n,\alpha)$	Cu-67*	(n,γ)	Hf-181*	(n,γ)
	(n,2n)				
Si-28*	$(n,p),(n,\alpha)$	Zr-89*	(n,γ)	Hf-182*	(n,γ)
Si-29*	$(n,p),(n,\alpha)$	Zr-90*	(n,γ)	Au-197	(n,γ)
Si-30*	$(n,p),(n,\alpha)$	Zr-91*	(n,γ)	Pb-204*	$(n,2n), (n,3n), (n,\gamma)$
Si-31*	$(n,p),(n,\alpha)$	Zr-92*	(n,γ)	Pb-205*	$(n,2n), (n,3n), (n,\gamma)$
Si-32*	$(n,p),(n,\alpha)$	Zr-93*	(n,γ)	Pb-206*	$(n,2n), (n,3n), (n,\gamma)$
K-39*	(n,γ)	Zr-94*	(n,γ)	Pb-207*	$(n,2n), (n,3n), (n,\gamma)$
K-40*	(n,γ)	Zr-95*	(n,γ)	Pb-208*	$(n,2n), (n,3n), (n,\gamma)$
K-41*	(n,γ)	Zr-96*	(n,γ)	Pb-209*	$(n,2n), (n,3n), (n,\gamma)$
K-42*	(n,γ)	Zr-97*	(n,γ)	Th-232	(n,γ) , (fission)
K-43*	(n,γ)	Mo-95	(n,γ)	U-233	(n,γ) , (fission)
Cr-50*	$(n,\gamma), (n,p), (n,\alpha)$	Tc-99	(n,γ)	U-234	(n,γ) , (fission)
	(n,2n), (n3n)				
Cr-51*	$(n,\gamma), (n,p), (n,\alpha)$	Ru-101	(n,γ)	U-235 [†]	(n,γ) , (fission)
	(n,2n), (n3n)				
Cr-52*	$(n,\gamma), (n,p), (n,\alpha)$	Rh-103	(n,γ)	U-236	(n,γ) , (fission)
G 55:	(n,2n), (n3n)	1 100		T. 606*	
Cr-53*	$(n,\gamma), (n,p), (n,\alpha)$	Ag-109	(n,γ)	U-238 [†]	(n,γ) , (fission),
	(n,2n), (n3n)				(n,2n), (n3n)

Cr-54*	$(n,\gamma), (n,p), (n,\alpha)$ (n,2n), (n3n)	In-115	(n,γ)	Np-237	(fission)
Cr-55*	$(n,\gamma), (n,p), (n,\alpha)$ (n,2n), (n3n)	Cs-133	(n,γ)	Pu-238	(n,γ) , (fission)
Mn-55	(n,2n)	Nd-143	(n,γ)	Pu-239 [†]	(n,γ) , (fission)
Fe-54*	$(n,\gamma), (n,p), (n,\alpha)$ (n,2n)	Nd-145	(n,γ)	Pu-240 [†]	(n,γ) , (fission)
Fe-55*	$(n,\gamma), (n,p), (n,\alpha)$ (n,2n)	Sm-147	(n,γ)	Pu-241 [†]	(n,γ) , (fission)
Fe-56*	$(n,\gamma), (n,p), (n,\alpha)$ (n,2n)	Sm-159	(n,γ)	Pu-242 [†]	(n,γ) , (fission), (n,2n), $(n3n)$
Fe-57*	$(n,\gamma), (n,p), (n,\alpha)$ (n,2n)	Sm-150	(n,γ)	Am-241 [†]	(n,γ) , (fission)
Fe-58*	$(n,\gamma), (n,p), (n,\alpha)$ (n,2n)	Sm-151	(n,γ)	Am-243	(n,γ) , (fission)
Fe-59*	$(n,\gamma), (n,p), (n,\alpha)$ (n,2n)	Sm-152	(n,γ)		

Table 2.1: Listing of Considered Nuclides and Reactions in SCALE Library.

2.2. Use of REBUS Covariance Data

The 15-group covariance library, developed for a sodium cooled fast reactor, pertains explicitly to the 15-group cross section files which are used in the REBUS model. Due to their specific nature, and the fact that they will be directly used to create perturbations in the REBUS fast reactor models, no modification or simplifications are necessary. The 101 reaction types for twenty isotopes included in this library are available from Brookhaven National Laboratory [24]. The nuclides and reactions represented appear in Table 2.2.

Nuclide	Reaction(s)	Nuclide	Reaction(s)	Nuclide	Reaction(s)
Cr-52	(n,γ), (n,el), (n,n')	Np-237	(n,γ), (fission), (n,el)	Am- 242m	(n,γ), (fission), (n,el)
	(n,2n)		(n,2n), (n,n'),(v- bar)		(n,2n), (n,n'),(v- bar)
Fe-56*	$(n,\gamma), (n,el), (n,n')$	Pu-238	(n,γ), (fission), (n,el)	Am- 243	(n,γ) , (fission), (n,el)
	(n,2n)		(n,2n), (n,n')		(n,2n), (n,n'),(v- bar)
Ni-58*	(n,γ), (n,el), (n,n')	Pu-239	(n,γ), (fission), (n,el)	Cm-242	(n,γ) , (fission), (n,el)
	(n,2n)		(n,2n), (n,n'),(v- bar)		(n,2n), (n,n')
U-234	(n,γ) , (fission), (n,el)	Pu-240	(n,γ) , (fission), (n,el)	Cm-243	(n,γ) , (fission), (n,el)
	(n,2n), (n,n')		(n,2n), (n,n'),(v- bar)		(n,2n), (n,n')
U-235	(v-bar)	Pu-241	(n,γ) , (fission), (n,el)	Cm-244	(n,γ) , (fission), (n,el)
			(n,2n), (n,n'),(v- bar)		(n,2n), (n,n')
U-236	(n,γ) , (fission), (n,el)	Pu-242	(n,γ) , (fission), (n,el)	Cm-245	(n,γ) , (fission), (n,el)
	(n,2n), (n,n')		(n,2n), (n,n')		(n,2n), (n,n')
U-238	(n,γ) , (fission), (n,el)	Am-241	(n,γ), (fission), (n,el)		
	(n,2n), (n,n'),(v-bar)		(n,2n), (n,n'),(v- bar)		

Table 2.2: Listing of Considered Nuclides and Reactions in REBUS Library.

2.3. Verification of Model Linearity

As stated in the introduction, before either stochastic forward perturbation or ESM methods are implemented, it is reasonable to check the linearity of the model to be used. Uncertainty propagation can be done by either a Monte Carlo sampling scheme, which can propagate all the moments of the input data, given infinite runs, to build the probability distributions, or a Moments Method, which propagates only selected moments of the distribution [11]. Assuming that the choice between the methods depends on the nature of

the probability distribution of the input data and the linearity of the model, it is reasonable to spend some time on this topic.

To illustrate this, the definition of the first and second moments are introduced as follows, assuming the probability distribution of input data y, is p(y) and p(y)dy is the probability that input data will be between y and y + dy:

$$p_1 = \int_{-\infty}^{\infty} y p(y) dy = \mu \tag{2.1}$$

$$p_{2} = \int_{-\infty}^{\infty} y^{2} p(y) dy = \sigma^{2} - \mu^{2} \Rightarrow p_{2} - \mu^{2} = \int_{-\infty}^{\infty} (y - \mu)^{2} p(y) dy = \sigma^{2}$$
 (2.2)

where μ is the average of all possible values of the input data, i.e. the mean, and σ^2 describes the average of the squared differences between all possible values and the mean, i.e. the variance. Higher order moments exist with physical and statistical meaning, but since they will not be addressed in this work, further explanation is available in the references [11][17]. These moments are what define a probability distribution and the nature of the distribution determines how many moments are needed for its reconstruction.

The Gaussian distribution depicted in Figure 2.1 is characterized by only the first two moments, mean and variance.

$$p_{Gauss}(y) = \frac{1}{\sigma\sqrt{2\pi}} \exp\left(-\frac{(y-\mu)^2}{2\sigma^2}\right)$$
 (2.3)

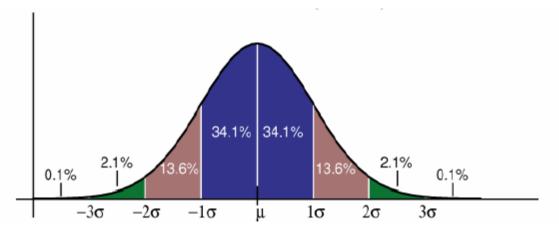


Figure 2.1: Gaussian Distribution

Further, rigorous mathematical proof shows that if a model is linear, a Gaussian input will produce a Gaussian output [23]. The first moment of the output corresponds to the reference output values calculated based on the mean input values. The second moment, variance, is obtained by re-running the model with input data perturbed by an amount proportional to the standard deviation. Most input cross-sections lack information about the second moment and no ENDF library contains information about higher order moments. For this reason it is often assumed cross-sections are normally distributed given a lack of higher order moments. Along with that assumption, this work is based on the observation that the model used, i.e. ORIGEN, which shall be discussed in detail later, is nearly linear over the range of uncertainties of interest.

To study the linearity of the model the following study is conducted. Let the model be defined by an operator, Ω ,:

$$\overline{y_0} = \Omega(\overline{\sigma_0}) \tag{2.4}$$

where $\overline{\sigma_0}$ is a vector of input cross-sections i=1,...N where subscript 0 denotes reference values, and $\overline{y_0}$ are calculated isotopics. The model Ω is judged linear around $\overline{\sigma_0}$ if it satisfies the condition:

Given arbitrary perturbations
$$\delta \overline{\sigma}_{i}$$
, $i = 1 \rightarrow N$
Calculate $\delta \overline{y}_{i} = \Omega(\overline{\sigma}_{0} + \delta \overline{\sigma}_{i}) - \Omega(\overline{\sigma}_{0})$, for $i = 1 \rightarrow N$
Then, $\Rightarrow \Omega(\overline{\sigma}_{0} + \sum_{i=1}^{N} a_{i} \delta \overline{\sigma}_{i}) - \Omega(\overline{\sigma}_{0}) = \sum_{i}^{N} a_{i} \delta \overline{y}_{i}$
for arbitrary a_{i}

$$(2.5)$$

The physical interpretation is for every cross-section perturbation, $\delta \overline{\sigma_i}$, the corresponding affect $\delta \overline{y_i}$ is obtained by running the code with the reference value and then again with cross-sections perturbed. The code can be run N times with each execution corresponding to a random cross-section perturbation, and then run with cross-sections perturbed by a linear combination of the previous N perturbations, i.e.

$$\overline{\sigma} = \overline{\sigma_0} + \sum_{i=1}^{N} a_i \delta \overline{\sigma_i}$$
 (2.6)

where a_i are arbitrary weights. If the model is linear, the perturbations should be approximately given by a linear combination of the original perturbations $\delta \overline{y_i}$, i.e.

$$\Omega\left(\sigma_{0} + \sum_{i=1}^{N} a_{i} \delta \overline{\sigma_{i}}\right) - \Omega(\sigma_{0}) \approx \sum_{i=1}^{N} a_{i} \delta \overline{y_{i}}$$
(2.7)

The difference between the two approaches is used to qualitatively judge model linearity. It is assumed that the weights a_i summed over N equal 1.

This qualitative approach must be applied to all generated outputs. If at any time the output is judged non-linear over a range of uncertainties, in this case within 4 standard deviations of the mean, then the outputs would no longer be Gaussian. It was so determined that the ORIGEN model is nearly linear. Appendix B includes further graphical support by showing 1) linear changes in output isotopics over a range of cross section perturbations, and 2) Gaussian output of samples given Gaussian inputs.

2.4. Implementation of Stochastic Sampling Method

A linear model allows the implementation of either ESM or stochastic perturbation.

Before discussing the implementation of the uncertainty propagation method, it is worthwhile to review the structure and origin of the covariance matrix, particularly as it exists in the SCALE library chosen as the data source for the majority of this work. The cross-section covariance matrix is given by:

$$\overline{\overline{C}}_{\sigma} = \begin{bmatrix}
Cov(\sigma_{1}, \sigma_{1}) & Cov(\sigma_{1}, \sigma_{2}) & \cdots & Cov(\sigma_{1}, \sigma_{n}) \\
Cov(\sigma_{1}, \sigma_{2}) & Cov(\sigma_{2}, \sigma_{2}) & \cdots & \vdots \\
\vdots & \cdots & \ddots & \vdots \\
Cov(\sigma_{n}, \sigma_{1}) & \cdots & \cdots & Cov(\sigma_{n}, \sigma_{n})
\end{bmatrix}$$
(2.8)

where $Cov(\sigma_i, \sigma_j)$ is the absolute covariance between cross-sections i and j and is defined by:

$$Cov(\sigma_i, \sigma_j) = \int_{-\infty}^{\infty} (\sigma_i - \sigma_{i0})(\sigma_j - \sigma_{j0}) p(\sigma_i, \sigma_j) d\sigma_i d\sigma_j$$
 (2.9)

In this notation, subscripts *i* and *j* denote isotope, energy group, and reaction type dependence. Since the absolute values of the cross-sections will change for each unique problem, it is not convenient to work with this absolute covariance data. The relative covariance matrix, in which each element is between -1 and 1, will be useful for simplifying the perturbation method shown later and can be obtained by using:

$$C_{Rij} = \frac{Cov(\sigma_i, \sigma_j)}{\sigma_i \sigma_j}$$
 (2.10)

to obtain [30]:

$$\overline{\overline{C}_{R\sigma}} = \begin{bmatrix}
C_{R11} & C_{R12} & \cdots & C_{R1n} \\
C_{R21} & C_{R22} & \cdots & \vdots \\
\vdots & \cdots & \ddots & \vdots \\
C_{Rn1} & \cdots & \cdots & C_{Rnn}
\end{bmatrix}$$
(2.11)

If given the sensitivity matrix of y with respect to σ , $\overline{S_R}$, the uncertainty in the output parameters, $\overline{\overline{C_y}}$, can be evaluated as

$$\overline{\overline{C}_{v}} = \overline{\overline{S}_{R}} \overline{\overline{C}_{R\sigma}} \overline{\overline{S}_{R}}^{T}$$
 (2.12)

In practice, these matrices or their products, are rarely directly constructed, but the effect of this product when using the forward perturbation with eigen-pair approach is evaluated as follows.

The singular value decomposition of $\overline{\overline{C_{R\sigma}}}$ is defined as:

$$\overline{\overline{C_{R\sigma}}} = \overline{\overline{W}} \overline{\overline{\Sigma_{\sigma}}} \overline{\overline{W}}^T$$
 (2.13)

where $\overline{\Sigma_{\sigma}}$ is the diagonal matrix of eigenvalues and \overline{W} the orthonormal matrix of eigenvectors where $\overline{\overline{W}} = \left[\overline{w_1}, \overline{w_2}, ..., \overline{w_n}\right]$ and $\overline{w_i}^T \overline{w_j} = 0 \leftrightarrow i \neq j$.

Since this is a stochastic forward perturbation method, a form of Monte Carlo sampling is implemented. Each sample is a perturbation of each cross section, and that perturbation, γ_i , for cross-section i, is defined as follows [18]:

$$\gamma_i = \sum_{j=1}^n \xi_j \left(\overline{w_j} \right)_i \tag{2.14}$$

where the value ξ_j is a random sample obtained from the eigenvalue Σ_{jj} having the Gaussian distribution defined as:

$$p(\xi_j)d\xi_j = \frac{1}{\sqrt{2\pi\Sigma_j}} \exp\left(\frac{-\xi_j^2}{2\Sigma_j}\right) d\xi_j$$
 (2.15)

Finally, since the covariance data in the matrix $\overline{\overline{C_{R\sigma}}}$ that was decomposed was relative data, any perturbed cross-section, σ_i , is simply:

$$\sigma_i = \sigma_{i0} (1 + \gamma_i) \tag{2.16}$$

where σ_{i0} denotes the unperturbed cross section. Perturbations are introduced thusly for all cross-sections i = 1,...,n. The matrix decomposition and creation of a set of input perturbations can be done a priori by auxiliary codes developed specifically for this purpose, thus sampling can use the model as a tool to produce perturbed results without modifying the model itself.

2.5. Implementation of the ESM

As indicated earlier, ESM methods are a favorable alternative to a stochastic forward perturbation when dealing with a large volume of input data, in this case a cross-section covariance matrix that is sparse and ill-conditioned, as required for ESM. The following section will describe the ESM method in brief but for the most detailed, rigorous, and formal definition, the reader is referred to H. Abdel-Khalik [14].

Consider n input data and m output data derived by using the model Ω . ESM states that for n inputs, at most n runs are required to fully characterize the distributions of the output, as opposed to stochastic methods which typically require a number of samples on the order of n. Define $\frac{1}{y}$ as the vector of m number densities calculated by:

$$\overline{y} = \overline{\overline{\Omega}}(\overline{\sigma}) = \overline{y_0} + \overline{\overline{\Omega}}(\overline{\sigma} - \overline{\sigma_0}) + O((\overline{\sigma} - \overline{\sigma_0})^2)$$
(2.17)

where $\overline{\sigma}$ are the *n* cross-section inputs. The second-order term can be ignored because of the linearity over the range of cross-section, and the matrix $\overline{\Omega}$, the Jacobi matrix, denotes the first derivatives of number density with respect to cross-cross section:

$$\left[\overline{\Omega} \right]_{ij} = \frac{\delta y_i}{\delta \sigma_j} \tag{2.18}$$

As stated before, the second moments of the input data are characterized by the covariance matrix, $\overline{\overline{C_\sigma}}$, which can be decomposed as:

$$\overline{\overline{C}_{\sigma}} = \overline{W} \overline{\Sigma_{\sigma}} \overline{W}^{T}$$
 (2.19)

Then the second order moments of the output data are characterized by the covariance matrix

$$\overline{\overline{C}_{v}} = \overline{\Omega} \overline{\overline{C}_{\sigma}} \overline{\Omega}^{T}$$
 (2.20)

Combing these yields:

$$\overline{\overline{C}_{y}} = \overline{\overline{\Omega}} \ \overline{\overline{W}} \ \overline{\Sigma_{\omega}} \ \overline{\overline{W}}^{T} \overline{\overline{\Omega}}^{T} = \overline{\overline{\Omega}} \ \overline{\overline{W}} \ \overline{\Sigma_{\omega}}^{1/2} \left(\overline{\overline{\Omega}} \ \overline{\overline{W}} \ \overline{\Sigma_{\omega}}^{1/2}\right)^{T}$$
(2.21)

The problem is that the matrix $\overline{\overline{\Omega}}$ is not available a priori, and in practice is rarely calculated. Stochastic methods build the values of $\overline{\overline{C_y}}$ by repeated sampling of perturbed inputs, where as ESM directly calculates $\overline{\overline{C_y}}$ by the following:

$$\overline{\overline{C}_{y}} = \overline{\overline{Y}_{\Sigma}} \overline{Y}_{\Sigma}^{T} \text{ where } \overline{\overline{Y}_{\Sigma}} = \left[\delta \overline{y}_{s1} \quad \delta \overline{y}_{s2} \quad \dots \quad \delta \overline{y}_{sr} \right]$$
 (2.22)

where r is the rank of the input data covariance matrix and the input perturbations are $\delta \overline{\sigma} = s_j \overline{w_j}$, where s_j is the square root of the jth diagonal element of $\overline{\Sigma_{\sigma}}$, and $\delta \overline{y_{sj}}$ is given by:

$$\delta \overline{y_{sj}} = \overline{\overline{\Omega}} \left(\overline{\sigma_0} + s_j \overline{w_j} \right) - \overline{\overline{\Omega}} \left(\overline{\sigma_0} \right), \quad j = 1, \dots, r$$
 (2.23)

The j^{th} perturbations are along the j^{th} singular vector of the input covariance matrix and proportional to the j^{th} singular value. When repeated r times, this procedure propagates the second moments of the input data through the model, where r is the effective rank of the

input covariance matrix $\overline{\overline{C}_\sigma}$, i.e. the number of singular values whose magnitudes are considered sufficiently large to not ignore. $\overline{\overline{C}_y}$ can now be calculated directly and, if desired, the singular value decomposition of $\overline{\overline{C}_y}$ can be obtained using $\overline{\overline{Y}_\Sigma} = \overline{\overline{U}_\Sigma}$ $\overline{\overline{S}_\Sigma}$ $\overline{\overline{V}_\Sigma}^T$:

$$\overline{\overline{C}_{v}} = \overline{\overline{Y}_{\Sigma}} \overline{\overline{Y}_{\Sigma}}^{T} = \overline{\overline{U}_{\Sigma}} \overline{\overline{S}_{\Sigma}} \overline{\overline{V}_{\Sigma}}^{T} \overline{\overline{V}_{\Sigma}} \overline{\overline{S}_{\Sigma}}^{T} \overline{\overline{U}_{\Sigma}}^{T} = \overline{\overline{U}_{\Sigma}} \overline{\overline{S}_{\Sigma}}^{2} \overline{\overline{U}_{\Sigma}}^{T}$$
(2.24)

Implementing this within a pre-existing model is not impossible but requires a non-trivial effort and a mastery of both the linear algebra involved and computer code to perform those mathematic operations. The experiment which was used to validate this method within the simplified model and compare it to stochastic sampling created the set of $\delta \overline{y}_{sj}$ by elementary matrix operations executed in a separate program, using data especially for this one case. The model was then executed r times. The data was collected into matrices by an auxiliary code and then processed by MatLab 6.5 to calculate \overline{C}_y as described. The numerical results validating ESM as equivalent to the stochastic approach are presented later.

When ESM was used in the more detailed TRITON model, which operates on the 44-group cross sections, the methodology had to be formally implemented in a usable code. Mr. Matthew Jessee created a code that performs the above decomposition of the 44-group covariance library, $\overline{C_{\sigma}}$, provided with SCALE 5.0, and creates perturbed 44-group cross-section libraries that can be fed directly to the TRITON model. Mr. Jessee was gracious enough to provide this resource and explain its use. This code performs the singular value decomposition of the 44-group covariance matrix, block by block, where a block is considered to be the square sub-matrix containing a single nuclide-reaction pair, and also all other nuclide-reaction pairs related to it by available covariance data. In most cases, this is

simply the 44 x 44 matrix for a particular nuclide and reaction combination since the covariance data is so sparse. The largest blocks occur for uranium and the transuranics, which have covariance data because their practical significance has warranted such studies. Further, those studies are of an experimental nature in which transuranics are often so dilute in the sample that their reactions are measured as ratios to the reactions of uranium, thus producing correlation data between those reactions. A simple post-processing code was then written to handle the calculation of $\overline{C_v}$.

2.6. Computational Models Employed for Each Method

As indicated in the introduction, this study will first use the SAS2H sequence and the ORIGEN depletion code in a simplified manner, and later the detailed TRITON sequence, all of which are available in the SCALE 5.0 package. The final model in this study uses the fast reactor code REBUS available from Argonne. For the simplified models, a particular fuel type and geometry are modeled using the SAS2H sequence supplied with a 44-group master library provided with the SCALE package which contains cross section data along with resonance parameters, Bondarenko data, flux spectrum information, scattering matrices, specific radioactivity and decay heat constants for each isotope, etc. [30]. Table 2.3 shows a brief summary of the different fuel types used in this study. More explicit definitions will appear later in the Results section, with a more detailed description, and SAS2H and ORIGEN input decks for each of the fuels included in Appendix A.

Fuel Type	Reactor	Enrichment	Geometry
UOX	BWR	4.5 w/o	7 x7 square lattice
UOX	PWR	4.5 w/o	17 x 17 square lattice, 25
			water holes
MOX	PWR	1.4 w/o U-235, 8 w/o Pu (65%	17 x 17 square lattice, 25
		fissile)	water holes
MOX	PWR	1.4 w/o U-235, 8 w/o Pu (65%	17 x 17 square lattice, 25
		fissile), 1 w/o Am, 1.5 w/o Np	water holes

Table 2.3: Brief Summary of Fuel Types Examined Using SAS2H + ORIGEN

SAS2H, when given the 44-group master library, produces 44-group flux spectra. The beginning of cycle fuel specific flux spectrum is used to collapse the 44-group covariance matrix to a 1-group covariance matrix for use in the stochastic sampling procedure described in Section 2.4. The result is a set of covariance data specific for the given fuel type being modeled. SAS2H also produces a transport updated 1-group, binary cross-section library on which the problem specific cross sections are now stored and will be used in the stand alone ORIGEN model. Note that the library for the simplified model accounts for only one representative burnup step across the life of the fuel. The following figures show the SAS2H flux spectra calculated for each of the models in Table 2.3 plotted with the typical LWR spectrum as well, where Figure 2.2 shows the PWR fuels for both UOX and MOX, and Figure 2.3 shows the BWR fuel at various void fractions. Unlike the later TRITON model, fluxes from SAS2H are not normalized to the same fuel specific power density, but the input power applied to all materials in the model.

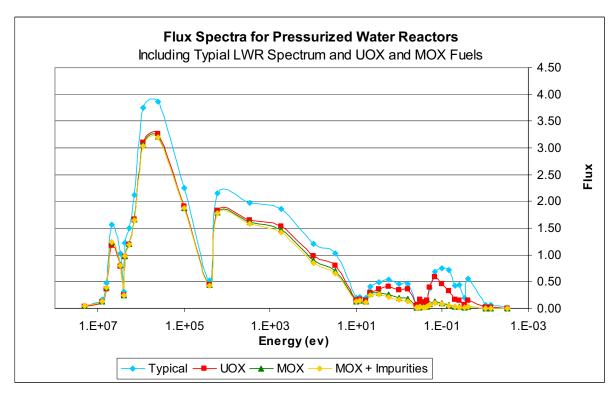


Figure 2.2: Flux Spectra for PWR Models

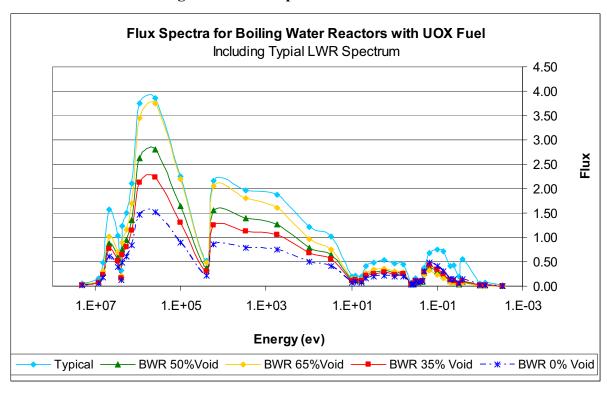


Figure 2.3: Flux Spectra for BWR Models

The stand-alone ORIGEN model is defined as 1 metric ton of the fuel modeled by SAS2H. ORIGEN depletes the model using the same power history and burnup as specified in the SAS2H model, then proceeds to decay the discharged isotopics over a series of time steps between discharge and 10,000 years, which covers the reprocessing time-frame through when waste canister failures are assumed in a waste repository. ORIGEN uses the fuel specific cross sections that were created by SAS2H for a specific fuel with a set of specific power cycle parameters. The nominal ORIGEN case, which is run with the unperturbed cross sections, creates the nominal values for the model. The nominal ORIGEN model is set to output not only the discharge isotopic masses, but also the isotopic and total values for each of the metrics of interest. Specific values for decay heat, activity, and toxicity can be obtained by dividing the metric by the mass or by directly printing the specific values from the binary library. The ORIGEN model, into which perturbations are introduced, is set to determine only the masses, since they are all that will be needed for statistical evaluation, as will be described in the next section. One-group perturbations are created a priori for the stochastic sampling method for each fuel and for the ESM comparison experiment. The perturbations, as described in the previous sections, are then introduced directly into the cross sections as they are read into ORIGEN from the binary library by interrupting the code at that point and perturbing the cross-sections that covariance data are available for. A code was written to run ORIGEN for N samples and then acquire the number densities of the tracked nuclides from each sample (see Table 2.4) and group the results by the various decay times. The nuclides tracked are chosen mainly for their contribution to decay heat or toxicity, some of which are only chemically toxic rather than a producer of non-negligible radiation. Moreover, together the tracked nuclides represent greater than 95% of heat, radioactivity, and

radiotoxicity for any time greater than 10 years for all of the fuels modeled in this work. A nominal execution of ORIGEN takes approximately eight seconds to execute on desktop PC and the sampling version that introduces perturbations takes approximately seconds to execute on the same platform.

Pb-210	Pa-231	U-238	Pu-240	Am-243	Se-79	Cs-134
Ra-226	U-234	Np-237	Pu-241	Cm-242	Sr-90	Cs-137
Ac-227	U-235	Np-239	Pu-242	Cm-244	Y-90	Ba-137m
Th-227	U-236	Pu-238	Am-241	Cm-245	Tc-99	
Th-230	U-237	Pu-239	Am-242m	C-14	I-129	

Table 2.4: Isotopes Tracked for Analysis

The TRITON model was used only for the fast reactors fuels and a validation for the pressurized water reactor fuel. For the purposes of this research, TRITON will be run as a stand-alone model using the 44-group master library as an input. The choice was made to fully exploit TRITON's detailed modeling abilities as this was the natural progression from the cruder light water reactor models already described. For each of the fast reactor fuels modeled, and a 4.5 w/o uranium oxide fuel, detailed geometries of the unit cell and fuel composition defined by volume fractions were entered into the input. The biggest difference, besides going from simple ORIGEN depletion to a rigorous 2-D transport, was burning the fuel over 25+ smaller burnup steps with both cross-section and flux spectrum updates for each step as opposed to a single, representative step employed in the ORIGEN model with only one cross section update at mid-point of that step. At the end of the TRIRON execution, the discharge number densities of the same isotopes are decayed using the same time steps as in previous models.

Fuel Type	Reactor	Enrichment	Geometry
UOX	PWR	4.5 w/o	17 x 17 square lattice, 25 water
			holes
Actinide,	FR,	59.2 w/o transuranics, 20 w/o	217 pin hexagonal lattice
metal	CR=0.25	zirconium	
Actinide,	FR,	20.6 w/o transuranics, 10 w/o	169 pin hexagonal lattice
metal	CR=0.70	zirconium	
Actinide,	FR,	16.2 w/o transuranics, 10 w/o	127 pin hexagonal lattice
metal	CR=1.05	zirconium	

Table 2.5: Brief Summary of Fuel Types Examined Using TRITON

Since TRITON takes approximately twenty to thirty seconds per burnup step to execute and has, in principle, 44 times as much input data, it is apparent that stochastic sampling is not a reasonable method to use with this model. Not only does TRITON take longer to execute, but it also uses the 44-group master cross section library as input and thus the 44-group covariance library as a source for perturbations. The principles of Monte Carlo sampling indicate that many thousands of samples would have to be run to properly propagate the uncertainties. In going from 1-group to 44-group covariance data the effective rank of the covariance matrix increased from 223 to 1938, with cutoff criteria imposed on singular values with a magnitude less than 10⁻⁶ relative to the reference cross section. So even ESM will require the execution of the code 1938 times, but this is still far less than computationally taxing than running over 3000 samples which is on the order of the input data. For each sample set, perturbations according to the ESM method are introduced directly into the 44-group master library, using a code generously provided by Mr. Jessee, and a new perturbed library is created which is subsequently input to the TRITON model. It was concluded that the primary depletion model is nearly linear from the linearity study of ORIGEN, but in order to confidently avoid non-linearities that may arise if cross sections are perturbed outside the linear region, instead of multiplying by the square root of the

eigenvalue, the perturbation is scaled by a scaling factor, SCF = 0.07, divided by the infinity norm of the eigenvector being used in a particular sample.

$$\overline{\sigma} = \overline{\sigma_0} \left(I + \frac{SCF}{\|\overline{w_i}\|_{\infty}} \overline{w_i} \right)$$
 (2.25)

The scaling factor and infinity norm are divided out in post processing, and the singular value is multiplied back into the output when computing the covariance matrix of the output.

$$\delta y_{i} = \left[\Omega \left(\overline{\sigma_{0}} + \frac{SCF}{\left\| \overline{w_{i}} \right\|_{\infty}} \overline{w_{i}} \right) - \Omega \left(\overline{\sigma_{0}} \right) \right] \frac{\left\| \overline{w_{i}} \right\|_{\infty}}{SCF}$$
(2.26)

$$\overline{\overline{C}_{y}} = \left(\overline{\overline{Y}_{\Sigma}} \Sigma^{1/2}\right) \left(\overline{\overline{Y}_{\Sigma}} \Sigma^{1/2}\right)^{T}$$
(2.27)

The covariance matrix for output isotopics directly provides the uncertainty information needed for this study, namely the standard deviation of each isotope tracked.

The ESM sampling method was similarly implemented in REBUS by Dr. Abdel-Khalik. Both TRITON and REBUS will be used for modeling recycling of the fast reactor fuel corresponding to the conversion ratio of about 0.70. As discussed before, given material feeds and reprocessing parameters, REBUS does this automatically. A external procedure is developed and implemented for TRITON to emulate this recycling methodology. That procedure is described in the following section. The recycle procedure of taking all the fast reactor transuranics and combining them with spent LWR fuel is that which is outlined in Argonne's ABTR Preconceptual Design Report [26].

2.7. TRITON Recycle Methodology

Having evaluated the resulting isotopic covariance matrices for once-through fuels in both thermal and fast reactors, our attention turns to a recycling scheme. The recycling scheme creates a transuranic fuel, made of spent LWR fuel, burns it in a fast reactor, and then recycles that fuel back into the fast reactor, making up part of the fuel mass by adding more spent LWR fuel to the mix. The transuranics recycled are Np-237, Np-239, Pu-238, Pu-239, Pu-240, Pu-241, Pu-242, Am-241, Am-242m, Am-243, Cm-242, Cm-244, and Cm-245. The first step is to create a nominal recycle case by taking mass $M_{FR}^{Recycle}$ from 1.5 year decayed fast reactor fuel and adding mass M_{LWR} from 10 yr decayed thermal reactor fuel (burnup of 33 GWD/MTU and an original enrichment of 3.3 w/o) and mass M_{DU} from a depleted uranium source to create a new fuel having the same volume loading as the original fuel, and thus approximately the same total heavy metal mass, given by:

$$M_{FR}^{BOL} = M_{FR}^{\text{Re cycle}} + M_{LWR}^{\text{Re cycle}} + M_{DU}$$
 (2.28)

Here, an assumption about the type of reprocessing is made, e.g. UREX, perfect separation, etc. In this work, all the transuranics from the spent fast reactor fuel are added back and enrichment made up by LWR spent fuel transuranics, $M_{LWR}^{Recycle}$, with the remaining mass being depleted uranium, M_{DU} . To obtain the additional equation required to solve for $M_{LWR}^{Recycle}$ and M_{DU} , it is required that the composition of M_{FR}^{BOL} be such so as to achieve the cycle energy requirement. $M_{FR}^{Recycle}$ is therefore all the mass of the recycled fuel extracted after M_{FR}^{BOL} is burnt and reprocessed. This implies the following relationship:

$$M_{FR}^{BOL} - M_{FR}^{\text{Re cycle}} = M_{LWR}^{\text{Re cycle}} + M_{DU}$$
 (2.29)

The depleted uranium is assumed to have a fixed isotopic composition of 99.8 w/o U-238 and 0.2 w/o U-235. Further, the isotopic compositions of the LWR and FR fuels are known as well, so that when the masses are combined, masses of individual isotopes add. Since the TRITON model requires input isotopics to be in w/o, M_R must be expanded in terms of its composition, completed as now explained. Treat the mass as a vector composed of isotopes

from element
$$k$$
: $\overline{M_{FR,k}^{BOL}}$. Convert the masses in $\overline{M_{FR,k}^{BOL}}$ to weight percents by $\frac{\overline{M_{FR,k}^{BOL}}}{\left\|\overline{M_{FR,k}^{BOL}}\right\|_1}$, where

the one-norm is the sum of the masses of all isotopes of element k. The last expression gives the isotopic data for element k that will be input to the TRITON model. This process is repeated for each element k. For this experiment, the discharged isotopics and end of life k-effective values from the initial fresh fuel TRITON input are taken to be the target values, despite k-effective being greater than 1. The output data from the unperturbed TRITON model are considered to be the nominal values for each recycle step. The k-effective values are not only recorded for statistical analysis, but are also used to adjust the transurance enrichment for the next recycle, in an effort to maintain the end of life k-effective value.

Next, let $\overline{C_{LWR}^{Recycle}}$ and $\overline{C_{FR}^{Recycle}}$ be the absolute isotopic covariance matrices of the once-through thermal and fast reactors, respectively, and converted, if necessary, from their relative values to units of mass. The fast reactor data are taken 1.5 years after discharge and the thermal reactor data are taken at 10 years of decay, i.e. time lapse to recycle. These matrices are 13 x 13 containing number density uncertainties for the transuranics already mentioned. These two matrices are the results of earlier work with each of these fuel types

using the TRITON model with our ESM approach to propagate uncertainties due to crosssections. We now wish to recycle the fuel characterized by these uncertainties.

The masses of each isotope of each element in each fuel stream is perturbed separately using $\overline{\overline{C_{LWR}^{\text{Re}\,\text{cycle}}}}$ and $\overline{\overline{C_{FR}^{\text{Re}\,\text{cycle}}}}$, via the ESM method. Each matrix is decomposed such that:

$$\overline{C^{\text{Re}\,cycle}} = \overline{W^{\text{Re}\,cycle}} \quad \overline{\Sigma^{\text{Re}\,cycle}} \quad \overline{\overline{W^{\text{Re}\,cycle}}}^T \tag{2.30}$$

Perturbations are introduced into the masses of the 13 isotopes in that fuel stream by scaling a singular vector by the square root of the corresponding singular value and adding this perturbation vector to the mass vector. These perturbed isotopics are then used to satisfy equation 2.29, just as the unperturbed values would be. Finally, just as in the unperturbed case, the masses of each element are converted to weight percents as required by TRITON's input structure, and a perturbed input written for the model.

Finally, uncertainty is propagated by running the model to equilibrium several times, each time choosing a subsequent singular pair to perturb with, i.e. perturbing along w_I , w_2 , etc. with each new run of the model. Experimentation revealed that only the first six singular pair perturbations needed to be run to effectively propagate the uncertainty, which is in accordance with the theory of ESM. The above procedure is repeated each time the fuel is recycled, thus creating a new $\overline{C_{FR}^{BOL}}$ and a new decomposition to perturb by, for each recycle step.

The resulting recycled fuel nuclei number density uncertainties are combined with the nuclei number density uncertainties due to cross-section uncertainties into a single uncertainty vector, $\overline{\sigma_{TOTAL}}$, where the elements of the vector denote different isotopes.

Experience showed that nuclei number density uncertainties due to cross-section uncertainties changed little unless major compositional changes are made to the fuel (see Results section). In the following, the subscript *FR* denotes recycled fuel nuclei number density uncertainties at EOL originating from the uncertainty of the material making up the fuel independent of cross section uncertainties, and the subscript *XS* denotes fuel nuclei number density uncertainties at EOL originating from cross-section uncertainties:

$$\left(\overline{\sigma_{TOTAL}}\right)_{i} = \sqrt{\left(\overline{\sigma_{XS}}\right)_{i}^{2} + \left(\overline{\sigma_{FR}}\right)_{i}^{2} + 2\left(\overline{\sigma_{XS}}\right)_{i}\left(\overline{\sigma_{FR}}\right)_{i}\left(\rho_{i}\right)}$$
(2.31)

where ρ is a correlation between the two uncertainties. Having the total uncertainty on the recycled fuel composition at equilibrium, the mass uncertainties, and in turn the key metrics uncertainties, for any given discharged fuel that will be sent to permanent disposal can be computed. Values of operational parameters (e.g. k-effective) can be collected from the model both at equilibrium and between recycle steps, and the uncertainties on those parameters computed as well.

Since the cross section originated uncertainties on EOL isotopics affect the subsequent reload isotopics, there must be correlation. However, within the scope of this work, it is assumed that $\rho=0$, because the only foreseeable method to obtain that correlation with the TRITON model is a posteriori calculation from the results of running the possible cross-section uncertainties with each of the possible recycle nuclei number density uncertainties, through equilibrium. This task is currently too computationally taxing as it would require execution time of (15 minutes per execution)($r_{XS}+1$)(r_R+1)(6+1) times, where 1 is for the nominal case and r_{XS} and r_R are the effective ranks of the covariance matrices for cross-sections and recycled isotopics, respectively.

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2.8. Statistical Analysis Performed on Results

The forward perturbation sampling process produces a large group of data for each fuel type; these raw data are in a form in which it can be processed using rudimentary statistical methods. The ESM sampling method directly calculates the covariance matrix of the isotopics in post processing, as already discussed, so much of this section applies only to the simplified ORIGEN models rather than the TRITON and REBUS models. First, the unperturbed values of the fuel sample produced by ORIGEN -- isotopics and total heat, radioactivity and radiotoxicity -- using the problem specific cross-sections provided by the SAS2H model are, by definition, the mean and most likely values for the particular model. For the thirty-three nuclides tracked, there are N samples of isotopic masses at discharge and 1, 5, 10, 50, 100, 500, 1000, 2500, 5000, and 10000 years of decay. The values at discharge and 1 year are neglected in the statistical analysis as many very-short lived isotopes present at this time contribute much of the heat load in the first 1 year or so until they die off. Thus, the thirty three nuclides tracked would not cover greater than 95% of the heat, etc. that is of interest. This is practically justified by the fact that the fuel will be closely monitored and guarded in wet storage for at least the first five years, and this work is mainly concerned with the affects on the repository and reprocessing aspects which take place later.

Basic statistics are applicable to this data because all the metrics are linearly and directly proportional to mass, therefore each have specific values. Each isotope tracked has some specific constant for heat [W/g], activity [Ci/g], and radiotoxicity [(m³ air or water to dilute to acceptable leve) / g]. The statistical process is justified by the equation:

$$\mu_{y} = \left(\frac{dy}{dx}\right)\mu_{x} \tag{2.32}$$

which states that the uncertainty in a parameter y that is dependent upon parameter x is simply the derivative of the relationship that relates the two times the uncertainty in x [28]. For all the metrics of interest, a change in the metric is simply the change in mass times the specific value for that metric. All that is needed, therefore, are the statistics of the isotopic masses which can be translated to the metrics by means of these specific values, which are given in Table 2.6.

		Heat/Gram		_
Nuclide	Ci/Gram	(W)	m³ Air/g	m³ Water/g
pb210	7.6376E+01	1.7901E-02	NA	NA
ra226	9.8912E-01	2.8565E-02	NA	NA
ac227	7.2373E+01	3.5003E-02	2.3886E+16	NA
th227	3.0749E+04	1.1235E+03	1.8412E+17	NA
th230	2.0627E-02	5.8238E-04	1.2352E+12	NA
pa231	4.7253E-02	1.4378E-03	3.9708E+12	NA
u234	6.2204E-03	1.7905E-04	3.5144E+10	1.6456E+04
u235	2.1624E-06	5.9912E-08	1.1033E+07	5.4884E+00
u236	6.4706E-05	1.7521E-06	3.3701E+08	1.6423E+02
u237	8.1658E+04	1.5809E+02	9.3110E+13	3.3604E+09
u238	3.3633E-07	8.5129E-09	1.6170E+06	8.1832E-01
np237	7.0521E-04	2.0119E-05	2.1177E+10	4.1977E+03
np239	2.3206E+05	5.8682E+02	1.3896E+14	1.0046E+10
pu238	1.7132E+01	5.6779E-01	1.1271E+15	2.1309E+08
pu239	6.2072E-02	1.9291E-03	4.4656E+12	8.3881E+05
pu240	2.2708E-01	7.0707E-03	1.6336E+13	3.0686E+06
pu241	1.0343E+02	3.2868E-03	1.4266E+14	2.6864E+07
pu242	3.9558E-03	1.1682E-04	2.6025E+11	5.1307E+04
am241	3.4309E+00	1.1448E-01	1.9718E+14	3.7091E+07
am242m	1.0481E+01	4.2370E-03	5.7904E+14	1.0760E+08
am243	1.9969E-01	6.4285E-03	1.1476E+13	2.1588E+06
cm242	3.3124E+03	1.2085E+02	1.1705E+16	2.1509E+09
cm244	8.0981E+01	2.8322E+00	2.7733E+15	5.2585E+08
cm245	1.7177E-01	5.7170E-03	1.0224E+13	1.9497E+06
c 14	4.4584E+00	1.3074E-03	1.5535E+10	NA
se 79	1.5362E-02	5.0813E-06	6.2704E+07	2.4079E+03
sr 90	1.4117E+02	1.6393E-01	1.3574E+13	2.1357E+08
tc 99	1.7114E-02	8.5821E-06	1.3370E+08	5.9217E+02
i129	1.7659E-04	8.2592E-08	1.0149E+07	1.0512E+03
cs137	8.7021E+01	9.6718E-02	2.0380E+12	6.1282E+07
ba137m	5.3801E+08	2.1138E+06	NA	5.3801E+08
y90	5.4342E+05	3.0086E+03	4.8957E+14	7.9331E+10
cs134	1.2944E+03	1.3197E+01	1.5521E+13	1.3290E+09

Table 2.6: Specific values, per nuclide, for metrics of interest.

For each isotope tracked, the sample mean, μ_m and sample standard deviation, Σ_{SD} are calculated using the equations below [28].

$$\mu_m = \frac{1}{N} \sum_{i=1}^{N} x_i \tag{2.33}$$

$$\Sigma_{SD} = \sqrt{\frac{1}{N-1} \sum_{i=1}^{N} (x_i - \mu_m)^2}$$
 (2.34)

The standard deviation of the isotopic masses is the key metric needed for the analysis. Thus it must be justified that the standard deviation calculated is from a reasonably accurate sample. Comparison is made between the sampled mean and the true mean in terms of the expected standard deviation of the mean, Σ_m , calculated by:

$$\Sigma_m = \frac{\Sigma_{SD}}{\sqrt{N}} \tag{2.35}$$

This is a modification of the Central Limit Theorem, where the value of the mean of the sample is expected to deviate from the true value for a finite number of samples [28]. If the difference between the sampled mean and the true mean is within two or fewer standard deviations of the mean, the sample can be said to be reasonable. Finally, the convergence of the mean and the standard deviation are examined, i.e. after how many samples do they reach a nearly constant value. Those values tended toward 200 for the mean and 220-250 for the standard deviation so N = 300 samples is adequate. The standard deviation of the mass translates directly into uncertainties for each of the metrics observed, i.e. +/- 5% in mass produces +/- 5% in heat, etc. Also, consider the objective of a 95% confidence interval, using the student-t distribution, the standard deviation is multiplied by 1.96 to obtain the 95% confidence interval rather than just one standard deviation which provides approximately 68% confidence [28].

Next, the affect of the uncertainty on the total values is examined. The resulting uncertainties, μ_R , are propagated by the square root of the sum of the squares [28]:

$$\mu_R = \sqrt{\sum_{i=1}^L \mu_{m,i}^2} \tag{2.36}$$

where L is the number of parameters, i.e. tracked nuclides. This equation, by definition, assumes that there is no correlation between each isotopic metric, e.g. the heat produced by plutonium does not affect the heat produced by strontium. This produces a total uncertainty in that metric as contributed by the nuclides tracked, which can be compared to the nominal value of that metric for the particular fuel type and decay time. This is done not only to see how much uncertainty is imparted to the metric by the uncertainties of these isotopes, but also to verify that in tracking the specific 33 nuclides at least 95% of the total heat, etc. for that discrete point in time is observed.

Finally, for the fast reactor fuels modeled in TRITON and REBUS, the isotopics covariance matrix was constructed using the algorithm discussed in Section 2.5. The square roots of the diagonal elements represent the standard deviations of each of the tracked nuclides, equivalent to the standard deviations calculated from stochastic sampling. Also, in the TRITON results, the sub-matrix containing the actinides Np-237 through Cm-244 is extracted to be used for uncertainty propagation when examining recycling of fast reactor fuels.

3. Numerical Results

3.1. Simplified ORIGEN Models

3.1.1. Equivalency of ESM and Stochastic Methods

The essential benefit of ESM is that it will produce the same results as stochastic sampling and will work efficiently in models where stochastic methods would be impractical to implement. To verify this numerically, both ESM and the full stochastic sampling are implemented for the simplified PWR model, which is simple enough to allow either method. The 1-group covariance matrix is decomposed and 223 perturbations (rank = 223) are created according to the formulas already discussed. Perturbations are introduced directly within ORIGEN, using the cross section library made by SAS2H for the PWR fuel. Considering the time required to implement this method versus the fast execution time of the simple stochastic model, ESM is not well suited to small, simple models. In more sophisticated models where the runtime increases greatly and stochastic methods are not practical, however, ESM becomes worth the time it takes to implement.

Table 3.1 presents the comparison of the isotopics' uncertainties predicted by each method and Table 3.2 gives the nominal discharge isotopics for this model for reference. It is clear from these results that the two methods are producing equivalent results, as expected. A further look at the simplified PWR model is included in the next section. Note here, and in subsequent sections, only the discharge isotopics and the isotopic uncertainties for each model will be presented, as number density vs. time is calculated from the decay of the discharge isotopics. A larger, generalized results table for decay heat, radioactivity,

radiotoxicity, and uncertainty contributors is available for each model, but due to their size, have been included in Appendix C and the reader is referred to that section. The tables included there appear in the order in which models are presented in the main text.

	% Uncer	tianty
Nuclide	Stochastic	ESM
pb210	1.1951	1.1477
ra226	1.3548	1.3008
ac227	0.4805	0.4575
th227	0.4805	0.4578
th230	1.4910	1.4327
pa231	0.5177	0.4957
u234	1.5787	1.5193
u235	1.3125	1.3377
u236	0.7124	0.7017
u237	2.5207	2.5509
u238	0.0775	0.0759
np237	0.6054	0.6158
np239	13.6428	13.8702
pu238	1.0497	1.0579
pu239	0.8064	0.8851
pu240	2.5656	2.7661
pu241	2.5207	2.5509
pu242	2.5983	2.6248
am241	2.5016	2.5345
am242m	2.1720	2.2293
am243	13.6428	13.8698
cm242	2.1717	2.2289
cm244	11.3402	11.6702
cm245	10.2508	10.5995
c 14	0.4523	0.4900
se 79	0.3663	0.3825
sr 90	0.3748	0.3853
tc 99	1.9704	1.8803
i129	0.4186	0.4487
cs137	0.3822	0.3986
ba137m	0.3822	0.3988
y90	0.3748	0.3857
cs134	1.1380	1.2094

Table 3.1: Comparison of isotopic uncertainties from the two methods.

	Discharg	je Isotop	oics, gran	ns / MTI	НМ
pb210	2.415E-11	np237	5.538E+02	cm244	3.215E+01
ra226	1.628E-08	np239	8.925E+01	cm245	1.208E+00
ac227	3.443E-09	pu238	1.791E+02	c 14	3.335E-03
th227	1.014E-11	pu239	5.466E+03	se 79	5.817E+00
th230	1.140E-03	pu240	1.887E+03	sr 90	6.899E+02
pa231	3.116E-04	pu241	1.575E+03	tc 99	9.577E+02
u234	1.530E+02	pu242	5.576E+02	i129	1.741E+02
u235	1.233E+04	am241	4.302E+01	cs137	1.485E+03
u236	5.524E+03	am242m	9.197E-01	ba137m	2.283E-04
u237	1.346E+01	am243	1.162E+02	y90	1.866E-01
u238	9.298E+05	cm242	1.346E+01	cs134	1.439E+02

Table 3.2: Discharge isotopics for the PWR simplified model.

3.1.2. PWR Model with UOX Fuel

The following data are for a representative UOX fuel that is burned in a pressurized water reactor. An updated cross-section library is created using SAS2H for one representative burnup step, and the resulting 1-group working library used with ORIGEN. The 44-group covariance library is collapsed to 1-group using the beginning of cycle flux spectrum for this fuel, and stochastic sampling is implemented. The UOX fuel is 4.5 w/o and burned to 40 GWD/MTU in a single cycle representative of a once-through fuel. The geometry is a 17x17 Westinghouse fuel assembly with no burnable poison elements and 25 water holes with one being an instrumentation hole; adapted from Gauld [21]. See Appendix A for a more detailed description of this model. Since the discharge isotopics and the isotopic uncertainties for this model have already been presented in the previous section, and the results table is available in Appendix C, the discussion moves on to the separation study of this fuel.

This simple experiment examines the affect of decay heat uncertainties in the process of UOX fuel separation, a key aspect of fuel reprocessing. The simple model 4.5 w/o UOX that was burned to 40 GWD/MTU is decayed in three separate cases for 5, 10 and 25 years

(Table 3.3). At each individual time the uranium, plutonium, neptunium, americium, and curium were separated out by elemental species (henceforth referred to as lumps) with assumed 100% separation efficiency. The lumps were then decayed over the 10,000 year time, regardless of the lump's separation time or decay products. The heat load of each of these lumps is compared with the heat load of the total fuel assembly over the same decay time. The same process is then repeated for the masses plus one standard deviation of the isotopic uncertainties which propagates the uncertainty associated with each separation time. Table 3.4 shows these decays heat loads for the first 1000 years and Table 3.5 shows 2,500 to 10,000 years (see the Appendix A for a more detailed description of this model). As data shows, the majority of the long term heat load resides with the decaying of actinides. If these can be burned off in some reprocessing scheme, margin to the taxing heat limits on the repository could be realized. This experiment provides some insight for the more rigorous experiment of uncertainty propagation in recycled fast reactor fuel.

Nuclide	Mass at	5 Years	Mass at	10 Years	Mass at	25 Years
	Grams	+/-	Grams	+/-	Grams	+/-
u234	1.60E+02	2.53E+00	1.67E+02	2.64E+00	1.87E+02	2.96E+00
u235	1.23E+04	1.62E+02	1.23E+04	1.62E+02	1.23E+04	1.62E+02
u236	5.53E+03	3.94E+01	5.53E+03	3.94E+01	5.53E+03	3.94E+01
u237	3.75E-05	9.45E-07	2.95E-05	7.42E-07	1.43E-05	3.60E-07
u238	9.30E+05	7.21E+02	9.30E+05	7.21E+02	9.30E+05	7.21E+02
np237	5.69E+02	3.44E+00	5.73E+02	3.47E+00	5.95E+02	3.60E+00
np239	1.00E-04	1.37E-05	1.00E-04	1.37E-05	9.99E-05	1.36E-05
pu238	1.87E+02	1.96E+00	1.80E+02	1.88E+00	1.60E+02	1.67E+00
pu239	5.56E+03	4.48E+01	5.55E+03	4.48E+01	5.55E+03	4.48E+01
pu240	1.89E+03	4.85E+01	1.90E+03	4.86E+01	1.90E+03	4.88E+01
pu241	1.24E+03	3.12E+01	9.72E+02	2.45E+01	4.71E+02	1.19E+01
pu242	5.58E+02	1.45E+01	5.58E+02	1.45E+01	5.58E+02	1.45E+01
am241	3.79E+02	9.49E+00	6.41E+02	1.60E+01	1.12E+03	2.80E+01
am242m	8.97E-01	1.95E-02	8.76E-01	1.90E-02	8.13E-01	1.77E-02
am243	1.16E+02	1.59E+01	1.16E+02	1.59E+01	1.16E+02	1.58E+01
cm242	8.07E-03	1.75E-04	2.28E-03	4.96E-05	2.12E-03	4.60E-05
cm244	2.66E+01	3.02E+00	2.20E+01	2.49E+00	1.24E+01	1.40E+00
cm245	1.21E+00	1.24E-01	1.21E+00	1.24E-01	1.21E+00	1.24E-01

Table 3.3: Masses, with uncertainty, of actinides at 3 decay times.

Fi			rs After Disc re after sepa		4.5 w/o Bur	ned for 4	0 GWD/MT	U,	
Element	50 (55) Y	ears	100 (105)	100 (105) Years		500 (505) Years		1000 (1005) Years	
	W	+/- W'	W	+/- W'	W	+/- W'	W	+/- W'	
U	0.049	0.001	0.049	0.001	0.049	0.001	0.049	0.001	
Np	0.012	0.000	0.012	0.000	0.012	0.000	0.013	0.000	
Pu	218.010	4.235	195.781	4.025	91.127	2.085	52.097	1.136	
Am	41.217	1.102	38.110	1.026	20.348	0.589	9.513	0.319	
Cm	11.264	0.310	1.823	0.050	0.186	0.005	0.179	0.005	

Fl			ars After Dis re after sepa		4.5 w/o Bu	rned for	40 GWD/M	ΓU,
Element	50 (60) Y	ears/	100 (110)	Years	500 (510) Years		1000 (1010) Years	
	W	+/- W'	W	+/- W'	W	+/- W'	W	+/- W'
U	0.050	0.001	0.050	0.001	0.050	0.001	0.051	0.001
Np	0.013	0.000	0.013	0.000	0.013	0.000	0.013	0.000
Pu	188.917	3.590	167.357	3.374	76.953	1.739	45.779	0.981
Am	68.819	1.792	63.588	1.662	33.770	0.925	15.538	0.471
Cm	9.302	1.033	1.506	0.167	0.156	0.016	0.149	0.015

,			ars After Dis re after sepa		4.5 w/o Bu	rned for	40 GWD/M	TU,	
Element	50 (75) Y	ears	100 (125)	100 (125) Years		500 (525) Years		1000 (1025) Years	
	W	+/- W'	W	+/- W'	W	+/- W'	W	+/- W'	
U	0.054	0.001	0.054	0.001	0.054	0.001	0.054	0.001	
Np	0.013	0.000	0.013	0.000	0.013	0.000	0.013	0.000	
Pu	131.727	1.327	112.187	1.425	50.156	1.033	33.859	0.677	
Am	119.336	3.064	110.214	2.835	58.356	1.542	26.584	0.747	
Cm	5.274	0.516	0.857	0.083	0.093	0.008	0.090	0.008	

	1MT of Fuel After Discharge, a 4.5 w/o FA Burned for 40 GWD/MTU, (Note: times below are after irradiation)								
	50 Ye	ars	100Ye	ars 500 Ye		ars	1000 Years		
	W	+/- W'	W	+/- W'	W	+/- W'	W	+/- W'	
FUEL	662.820	19.380	356.663	8.038	112.558	4.262	62.215	1.999	

Table 3.4: Comparison of separation at 3 times vs. no separation, first 1000 years of decay.

Fl		r 40 GWD	ears After D /MTU, (No					
Element	2500 (2505) Years		Years 5000 (5005) Years		5) Years 5000 (5005) Years		10000 (10005) Years	
	W	+/- W'	W	+/- W'				
U	0.051	0.001	0.056	0.001	0.065	0.001		
Np	0.013	0.000	0.013	0.000	0.014	0.000		
Pu	23.033	0.413	17.347	0.281	12.833	0.187		
Am	1.481	0.112	0.604	0.079	0.439	0.059		
Cm	0.154	0.004	0.119	0.003	0.071	0.002		

- 1	If Separated At 10 Years After Discharge, a 4.5 w/o FA Burned for 40 GWD/MTU, (Note: times below are after separation)							
Element	2500 (251	0) Years	5000 (501	0) Years	10000 (<i>*</i> Yea			
	W	+/- W'	W	W +/- W' W				
U	0.053	0.001	0.057	0.001	0.067	0.001		
Np	0.013	0.000	0.013	0.000	0.014	0.000		
Pu	22.472	0.399	17.343	0.280	12.833	0.187		
Am	2.029	029 0.128 0.619 0.081 0.444 0.0						
Cm	0.129	0.013	0.100	0.010	0.060	0.006		

If Separated At 25 Years After Discharge, a 4.5 w/o F Burned for 40 GWD/MTU, (Note: times below are aft separation)								
Element	2500 (252	5) Years	5000 (502	5) Years	10000 (* Yea			
	W	+/- W'	W +/- W' W +/-					
U	0.056	0.001	0.061	0.001	0.072	0.001		
Np	0.013	0.000	0.014	0.000	0.015	0.000		
Pu	21.410	0.372	17.332	0.280	12.828	0.186		
Am	3.037	0.037 0.151 0.647 0.081 0.456 0.05						
Cm	0.078	0.007	0.061	0.005	0.036	0.003		

	1MT of Fuel After Discharge, a 4.5 w/o FA Burned for 40 GWD/MTU, (Note: times below are after irradiation)						
	2500 Years 5000 Years			10000	Years		
	W	+/- W'	W +/- W' W +/- W'				
FUEL	24.808	0.594	18.184	0.446	13.460	0.281	

Table 3.5: Comparison of separation at 3 times vs. no separation, 2500 – 10,000 years of decay.

3.1.3. Typical LWR with UOX Fuel

The SCALE package comes with a prepared "test" card-image cross-section library that is ideally representative of a typical LWR. The following data depict a uranium oxide fuel depleted, with stochastic sampling, in ORIGEN using this typical LWR cross section library provided with SCALE, and the typical LWR flux spectrum (also provided) to collapse the 44-group covariance library. The UOX fuel is 4.5 w/o and burned to 40 GWD/MTU in a single representative burnup step divided into depletion intervals by default in ORIGEN (see the Appendix A for a more detailed description of these models). While in terms of isotopics, the model produced similar results to the PWR model as expected (Table 3.6), the uncertainties using the typical flux spectrum tended to over predict those obtained from the PWR model (Table 3.7).

	Discharge Isotopics, grams / MTHM							
pb210	2.113E-11	np237	4.696E+02	cm244	1.930E+01			
ra226	1.641E-08	np239	8.942E+01	cm245	7.951E-01			
ac227	3.114E-09	pu238	1.343E+02	c 14	3.343E-03			
th227	8.370E-12	pu239	4.863E+03	se 79	5.895E+00			
th230	1.164E-03	pu240	2.193E+03	sr 90	7.127E+02			
pa231	3.159E-04	pu241	1.245E+03	tc 99	9.754E+02			
u234	1.049E+04	pu242	4.435E+02	i129	1.690E+02			
u235	1.547E+02	am241	3.190E+01	cs137	1.488E+03			
u236	5.641E+03	am242m	5.701E-01	ba137m	2.285E-04			
u237	1.202E+01	am243	8.028E+01	y90	1.939E-01			
u238	9.323E+05	cm242	1.000E+01	cs134	1.506E+02			

Table 3.6: Discharge isotopics for typical LWR simplified model.

	Isotopic Uncertainties (% St. Dev.)								
Nuclide	Uncertainty	Nuclide	Uncertainty	Nuclide	Uncertainty				
pb210	1.1518	np237	1.0320	cm244	11.4636				
ra226	1.3188	np239	12.6984	cm245	11.2314				
ac227	0.8004	pu238	2.4674	c 14	0.6833				
th227	0.8007	pu239	2.2019	se 79	0.4147				
th230	1.4124	pu240	2.8370	sr 90	0.7126				
pa231	1.3595	pu241	3.7081	tc 99	1.6437				
u234	1.4532	pu242	3.9948	i129	0.5690				
u235	6.1838	am241	3.6751	cs137	0.3347				
u236	1.3642	am242m	3.1688	ba137m	0.3347				
u237	3.7081	am243	12.6985	y 90	0.7126				
u238	0.0894	cm242	3.1693	cs134	1.9556				

Table 3.7: Isotopics uncertainties for typical LWR simplified model.

3.1.4. BWR Models with UOX Fuel

The following data are for a representative UOX fuel burned in a boiling water reactor. An updated cross-section library is created using SAS2H for one representative burnup step, and the 1-group working library used with ORIGEN. The 44-group covariance library is collapsed to 1-group using the beginning of cycle flux spectrum for this fuel, and stochastic sampling is implemented. The UOX fuel is 4.5 w/o and burned to 40 GWD/MTU in a single cycle representative of a once-through fuel. The geometry is a 7x7 General Electric fuel assembly homogenized to 4.5 w/o, with no burnable poison elements; adapted from Hermann [32]. The experiment is repeated for void fractions of 0%, 35%, 50%, and 65% by modifying the average density of the coolant (see the Appendix A for a more detailed description of this model). Table 3.8 – Table 3.11 show the discharge isotopics for each of the voids, and Table 3.12 gives a listing of the isotopics uncertainties for each void. The key observation to take away from these data, as will be stressed again later, that for UOX fuels in a LWR (be it PWR or BWR), the uncertainties are on the same order of magnitude. Across the range of voids, uncertainties do change by a factor of 1.1 to 1.7. The

top of the fuel is typically at 70-80% void while the bottom is always at 0% void, resulting in an average operating void in the 40% - 50% range. This indicates that the isotopics and their uncertainties will be a function of not only burnup but also void history, both dependent upon not only the fuel assembly, but axial position within the assembly. The BWR results in that voided region are similar to the PWR results. In fact the PWR uncertainties fall within the uncertainties of the 50% void and 65% BWR void results.

	Discharge Isotopics, grams / MTHM							
pb210	1.338E-11	np237	3.433E+02	cm244	9.658E+00			
ra226	1.793E-08	np239	8.156E+01	cm245	1.883E-01			
ac227	1.680E-09	pu238	8.692E+01	c 14	3.165E-03			
th227	4.465E-12	pu239	3.294E+03	se 79	5.966E+00			
th230	1.380E-03	pu240	1.848E+03	sr 90	7.468E+02			
pa231	1.941E-04	pu241	8.414E+02	tc 99	1.001E+03			
u234	1.675E+02	pu242	4.334E+02	i129	1.602E+02			
u235	8.102E+03	am241	1.936E+01	cs137	1.491E+03			
u236	5.706E+03	am242m	3.008E-01	ba137m	2.290E-04			
u237	9.736E+00	am243	5.414E+01	y90	2.063E-01			
u238	9.370E+05	cm242	7.232E+00	cs134	1.141E+02			

Table 3.8: Discharge isotopics for BWR fuel burned at 0% void.

	Discharge Isotopics, grams / MTHM							
pb210	1.692E-11	np237	4.265E+02	cm244	1.668E+01			
ra226	1.728E-08	np239	8.489E+01	cm245	4.522E-01			
ac227	2.391E-09	pu238	1.202E+02	c 14	3.247E-03			
th227	6.618E-12	pu239	4.161E+03	se 79	5.911E+00			
th230	1.274E-03	pu240	1.922E+03	sr 90	7.234E+02			
pa231	2.459E-04	pu241	1.112E+03	tc 99	9.833E+02			
u234	1.613E+02	pu242	4.859E+02	i129	1.663E+02			
u235	9.843E+03	am241	2.770E+01	cs137	1.489E+03			
u236	5.640E+03	am242m	4.901E-01	ba137m	2.288E-04			
u237	1.124E+01	am243	7.594E+01	y90	1.977E-01			
u238	9.340E+05	cm242	9.467E+00	cs134	1.268E+02			

Table 3.9: Discharge isotopics for BWR fuel burned at 35% void.

	Discharge Isotopics, grams / MTHM							
pb210	2.010E-11	np237	4.842E+02	cm244	2.278E+01			
ra226	1.671E-08	np239	8.780E+01	cm245	7.550E-01			
ac227	2.958E-09	pu238	1.474E+02	c 14	3.306E-03			
th227	8.384E-12	pu239	4.905E+03	se 79	5.867E+00			
th230	1.195E-03	pu240	1.990E+03	sr 90	7.060E+02			
pa231	2.839E-04	pu241	1.334E+03	tc 99	9.685E+02			
u234	1.564E+02	pu242	5.166E+02	i129	1.705E+02			
u235	1.109E+04	am241	3.478E+01	cs137	1.487E+03			
u236	5.621E+03	am242m	6.776E-01	ba137m	2.286E-04			
u237	1.223E+01	am243	9.138E+01	y90	1.918E-01			
u238	9.317E+05	cm242	1.122E+01	cs134	1.359E+02			

Table 3.10: Discharge isotopics for BWR fuel burned at 50% void.

	Discharge Isotopics, grams / MTHM							
pb210	2.601E-11	np237	5.685E+02	cm244	3.264E+01			
ra226	1.570E-08	np239	9.288E+01	cm245	1.398E+00			
ac227	3.847E-09	pu238	1.934E+02	c 14	3.392E-03			
th227	1.126E-11	pu239	6.262E+03	se 79	5.797E+00			
th230	1.073E-03	pu240	2.119E+03	sr 90	6.803E+02			
pa231	3.400E-04	pu241	1.704E+03	tc 99	9.429E+02			
u234	1.487E+02	pu242	5.478E+02	i129	1.764E+02			
u235	1.290E+04	am241	4.697E+01	cs137	1.483E+03			
u236	5.638E+03	am242m	1.058E+00	ba137m	2.281E-04			
u237	1.363E+01	am243	1.118E+02	y90	1.837E-01			
u238	9.280E+05	cm242	1.392E+01	cs134	1.488E+02			

Table 3.11: Discharge isotopics for BWR fuel burned at 65% void.

	Isotopic	Uncertainties	s (% St. Dev.)	
Nuclide	-		ertainty	
Nuclide	0% Void	35% Void	50% Void	65% Void
pb210	0.798	1.056	1.126	1.388
ra226	0.902	1.193	1.274	1.575
ac227	0.335	0.429	0.477	0.548
th227	0.335	0.429	0.477	0.548
th230	1.022	1.335	1.415	1.725
pa231	0.424	0.504	0.570	0.594
u234	1.146	1.462	1.526	1.809
u235	1.360	1.472	1.757	1.677
u236	0.481	0.608	0.731	0.839
u237	2.105	2.390	2.492	2.479
u238	0.055	0.065	0.069	0.079
np237	0.541	0.593	0.638	0.697
np239	8.910	11.086	12.685	14.483
pu238	1.102	1.101	1.147	1.172
pu239	0.817	0.867	0.874	0.920
pu240	2.157	2.381	2.637	2.536
pu241	2.105	2.390	2.492	2.479
pu242	2.011	2.252	2.437	2.531
am241	2.089	2.372	2.474	2.462
am242m	1.807	2.071	2.156	2.154
am243	8.910	11.086	12.685	14.483
cm242	1.806	2.070	2.155	2.153
cm244	7.416	9.233	10.618	12.041
cm245	6.816	8.431	9.687	10.851
c 14	0.392	0.441	0.483	0.544
se 79	0.316	0.363	0.408	0.465
sr 90	0.322	0.381	0.436	0.496
tc 99	1.309	1.609	1.864	2.238
i129	0.362	0.408	0.444	0.502
cs137	0.326	0.374	0.413	0.473
ba137m	0.326	0.374	0.413	0.473
y 90	0.322	0.381	0.436	0.496
cs134	0.962	1.063	1.166	1.305

Table 3.12: Isotopics uncertainties for BWR models.

3.1.5. PWR Models with MOX Fuels

The following data is for two representative MOX fuels burned in a pressurized water reactor, adapted from fuel compositions in Bathke [4]. An updated cross-section library is created using SAS2H for one representative burnup step and the 1-group binary library used

with ORIGEN. The 44-group covariance library is collapsed to 1-group using the beginning of cycle flux spectrum for each fuel, and stochastic sampling is implemented. The first, clean MOX fuel is 91.903 w/o uranium with the following composition: 1.40 w/o U-235, 98.572 w/o U-238, 0.028 w/o U-234, and 8.097 w/o plutonium. The plutonium has the composition: 1.655 w/o Pu-238, 61.751 w/o Pu-239, 24.701 w/o Pu-240, 3.248 w/o Pu-241, 8.645 w/o Pu-242. The second MOX fuel is representative of imperfect separation techniques and includes only 89.403 w/o uranium and impurities of 1 w/o Np-237 and 1.5 w/o Am-241. The geometry of both fuels is that of a 17x17 Westinghouse-type fuel assembly with 25 water holes and they are each burned to 50 GWD/MTHM (see the Appendix A for a more detailed description of these models). Table 3.13 and Table 3.14 show the discharge isotopics for the two MOX fuels, and Table 3.15 presents the isotopic uncertainties for the two fuels. It can be seen from these data that making a significant change in the fuel composition will result in a change in the uncertainties.

	Discharge Isotopics, grams / MTHM							
pb210	1.310E-10	np237	3.290E+02	cm244	9.958E+02			
ra226	1.840E-08	np239	9.836E+01	cm245	9.472E+01			
ac227	6.215E-09	pu238	4.530E+03	c 14	5.611E-03			
th227	3.415E-11	pu239	2.384E+04	se 79	6.108E+00			
th230	1.292E-03	pu240	1.549E+04	sr 90	4.569E+02			
pa231	4.894E-04	pu241	9.475E+03	tc 99	1.127E+03			
u234	2.299E+02	pu242	6.822E+03	i129	3.038E+02			
u235	7.081E+03	am241	6.046E+02	cs137	1.864E+03			
u236	1.503E+03	am242m	1.961E+01	ba137m	2.872E-04			
u237	7.047E+00	am243	1.851E+03	y90	1.218E-01			
u238	8.759E+05	cm242	1.355E+02	cs134	2.055E+02			

Table 3.13: Discharge isotopics for the "clean" MOX fuel.

	Discharge Isotopics, grams / MTHM								
pb210	1.442E-09	np237	5.422E+03	cm244	1.075E+03				
ra226	1.928E-08	np239	9.590E+01	cm245	1.075E+02				
ac227	6.988E-09	pu238	1.328E+04	c 14	5.300E-03				
th227	2.498E-10	pu239	2.816E+04	se 79	6.053E+00				
th230	1.441E-03	pu240	1.593E+04	sr 90	4.491E+02				
pa231	5.488E-04	pu241	9.545E+03	tc 99	1.105E+03				
u234	3.133E+02	pu242	7.792E+03	i129	3.039E+02				
u235	7.406E+03	am241	4.880E+03	cs137	1.843E+03				
u236	1.501E+03	am242m	2.003E+02	ba137m	2.842E-04				
u237	6.927E+00	am243	2.130E+03	y90	1.196E-01				
u238	8.501E+05	cm242	1.304E+03	cs134	1.993E+02				

Table 3.14: Discharge isotopics for MOX fuel with impurities.

Isotopic Uncertainties (% St. Dev.)		
Nuclide	Uncertainty	
	"Clean MOX	MOX w/ Impurities
pb210	1.0641	0.6387
ra226	1.2235	0.8533
ac227	0.6231	0.6575
th227	0.6231	0.6575
th230	1.1831	0.7945
pa231	0.6028	0.6460
u234	1.1110	0.7477
u235	1.3668	1.3539
u236	1.0367	1.0588
u237	1.9287	1.8627
u238	0.0918	0.0963
np237	0.7904	0.3727
np239	24.9776	23.0736
pu238	1.3121	0.7914
pu239	1.3840	1.3535
pu240	2.3612	2.2987
pu241	1.9287	1.8627
pu242	2.5902	2.3931
am241	1.8381	1.1829
am242m	1.2130	1.1621
am243	24.9775	23.0737
cm242	1.2128	1.1618
cm244	17.6375	16.1298
cm245	14.4246	13.1267
c 14	0.8374	0.9462
se 79	0.7336	0.8164
sr 90	0.6899	0.7637
tc 99	2.5609	2.9419
i129	0.7641	0.8277
cs137	0.7591	0.8189
ba137m	0.7591	0.8189
y 90	0.6899	0.7637
cs134	1.6727	1.6825

Table 3.15: Isotopic uncertainties for MOX fuels.

3.1.6. Comparison of Results, Simplified ORIGEN Models

In the following pages, the two most commonly examined metrics for repository performance and reprocessing are graphically compared: decay heat and radioactivity, spanning 10 to 10,000 years of decay time. These data are depicted as plots of the isotopic

uncertainty information from the simplified ORIGEN models as propagated to these two key metrics. For additional information about each model, please refer to the previous sections. Attention is again drawn to the observation that, by experience, it is shown that major changes in the fuel composition, e.g. from UOX to MOX, cause changes to the distribution of isotopic uncertainties. Comparing the PWR UOX and PWR MOX fuels, which have the same geometry and are burned in the same reactor type, exemplifies this by showing a drop in uncertainties for plutonium isotopes, increase in americium and curium isotopic uncertainties, and fission product uncertainties more than doubling. This is quite dissimilar from when the BWR (in the 50-65% void region) and PWR results are compared. Even with two different reactor types and two different geometries, those results are similar due to their fuel type and burnup.

Following the order in which the models were presented, the first comparison made is of the ORIGEN model representing a PWR, using both the stochastic method and the ESM approach for the same model. The nominal information is the same for each since it is the same model, only analyzed with different uncertainty analysis methods. Both the decay heat uncertainty (Figure 3.1) and the radioactivity uncertainty (Figure 3.2) were nearly identical between the two models, showing graphically that ESM yields the same results as forward perturbation.

Next, comparison is made between the metrics for the PWR model using the SAS2H updated flux spectrum and cross sections versus using the typical LWR spectrum and cross sections provided with the SCALE package for decay heat (Figure 3.3) and for radioactivity (Figure 3.4). While nominal decay performance was nearly the same, indicating similar

isotopics results, the typical values tended to over-estimate decay heat uncertainty and underestimate radioactivity uncertainty after the first one hundred years.

The BWR models at various void fractions are compared next. In terms of nominal values, the different voids produced nearly the same short term decay heat (Figure 3.5) and radioactivity (Figure 3.6), but the variance can be seen in the later decay times. This variance in longer term performance is due to slightly different actinide buildup, which can be seen in the discharge isotopics presented in Section 3.1.4. The uncertainties which are shown to vary between different voids models do so due to a different flux spectrum used to collapse the covariance matrix with each different void. It is also observed that the magnitude of the relative uncertainties for the BWR is similar to that of the PWR, which uses the same UOX fuel. One notices a trend in the UOX fuels, that as time increases, the uncertainty tends to increase, typically around 100-500 years. This is due to low uncertainty, high contributing fission products decaying away, leaving higher uncertainty, long lived actinides to decay.

Finally, the PWR model with UOX fuel is compared to the two PWR models using MOX fuels – both clean and with impurities – that were examined in this study. MOX fuels maintain higher heat load (Figure 3.7) and radioactivity (Figure 3.8) for a longer span of time than the PWR fuel due to the build up of long-lived actinides in the MOX fuel. Also, the MOX uncertainty is higher than that of the PWR UOX fuel due to the presence of higher quantities of actinides initially, longer burnup, and different operating conditions.

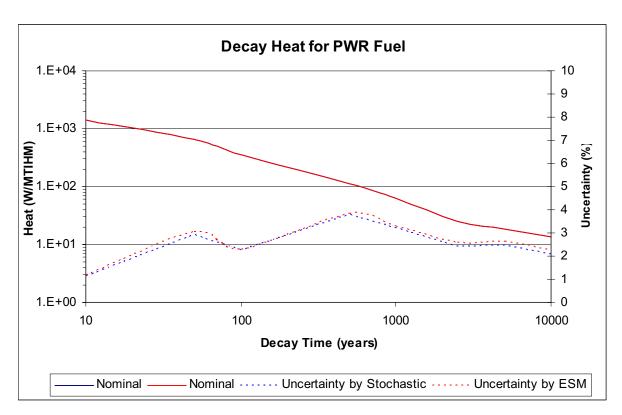


Figure 3.1: Decay heat comparison of stochastic and ESM sampling methods.

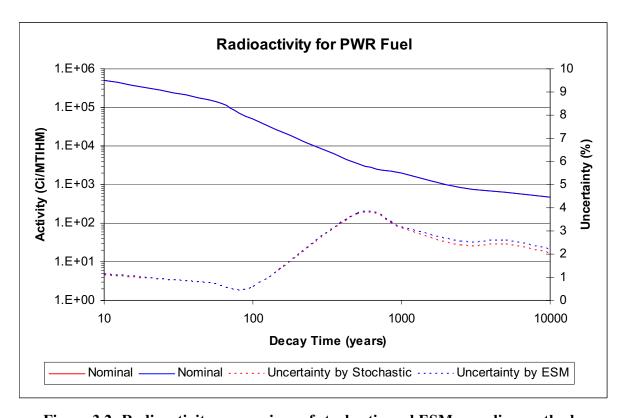


Figure 3.2: Radioactivity comparison of stochastic and ESM sampling methods.

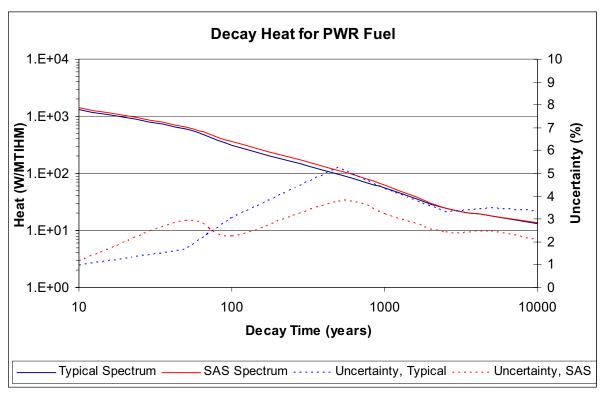


Figure 3.3: Decay heat comparison of SCALE provided data and SAS2H updated data.

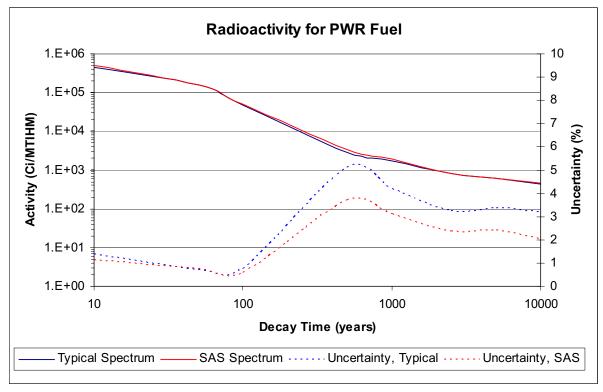


Figure 3.4: Radioactivity, SCALE provided data and SAS2H updated data.

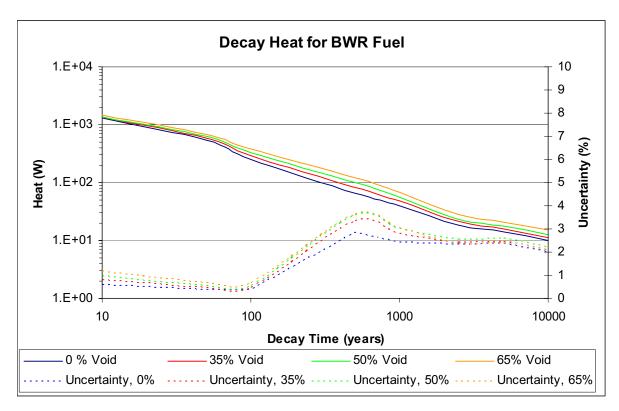


Figure 3.5: Decay heat comparison for BWR fuels.

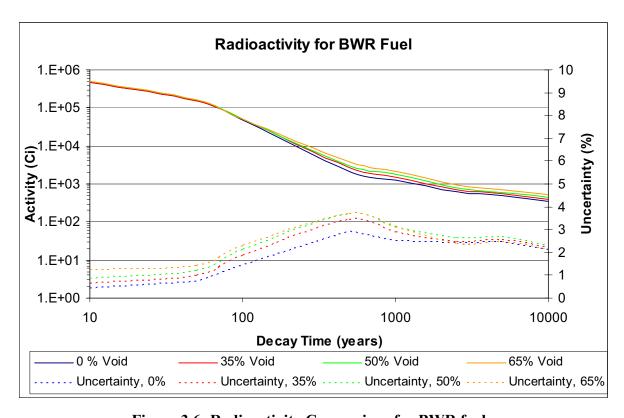


Figure 3.6: Radioactivity Comparison for BWR fuels.

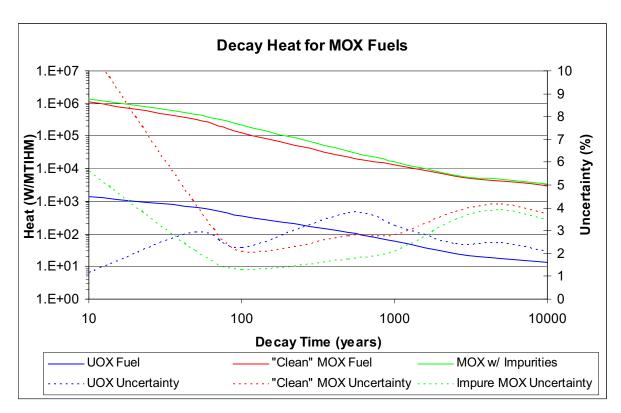


Figure 3.7: Decay heat comparison of UOX and MOX fuels.

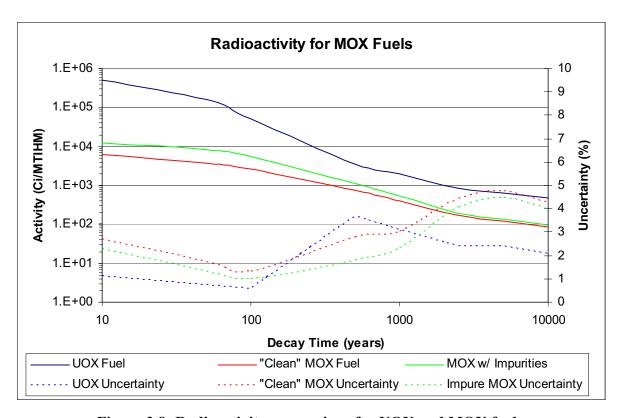


Figure 3.8: Radioactivity comparison for UOX and MOX fuels.

3.1.7. A Brief Experiment with Operational Uncertainties

Since cross section uncertainties have shown almost negligible affects on the metrics of interest that have been examined in this study, whether operational parameters could outweigh these small cross section induced uncertainties was examined. A simple experiment was performed to test this hypothesis. The simplified, unperturbed PWR model in ORIGEN as discussed in Section 3.1.2 was depleted adjusting the following parameters: +/- 2 GWD/MTU burnup (a reasonable measurement uncertainty, [33]), and varied power history between 90 and 105% of full power over the life of the fuel. Data are presented for several key isotopes that are primary contributors to decay heat (Figure 3.9) and radioactivity (Figure 3.10), taken at a time of one hundred years after decay. In examining the following figures, operational uncertainties can clearly be seen to have a much greater impact on the metrics of interest than cross-section uncertainty.

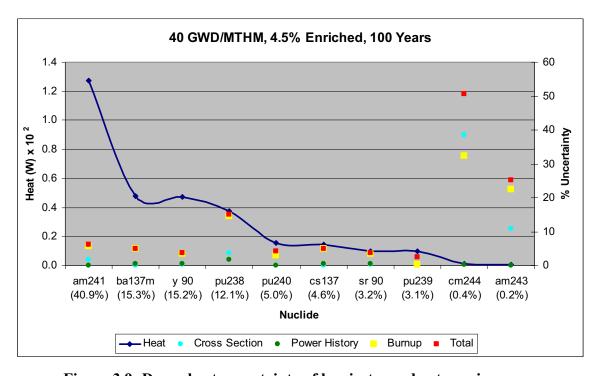


Figure 3.9: Decay heat uncertainty of key isotopes due to various sources.

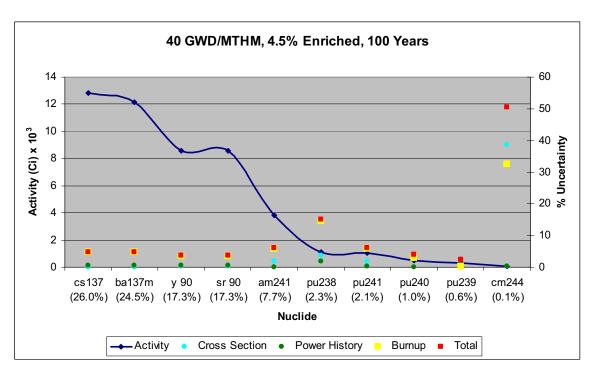


Figure 3.10: Radioactivity uncertainty in key isotopes due to various sources.

3.2. TRITON Models

3.2.1. PWR Model in TRITON with UOX Fuel

The choice to move the analysis to using TRITON for the remaining fuel types has been discussed previously. The PWR model was reconstructed in the more rigorous TRITON sequence to 1) validate the uncertainty propagation approach of using ESM in TRITON, 2) compare to the SAS2H/ORIGEN model, 3) use a more detailed and finer model which is a procedure closer to standard fuel analysis, and 4) provide an isotopic covariance matrix for the spent PWR fuel that will be used in the recycling experiment. If TRITON, using only the Wigner cell, is given the same geometry, isotopics, and power history -- using one burnup step -- as SAS2H it will produce equivalent results (Table 3.16). Results are not identical due to different transport solutions, but the differences are statistically insignificant. Note that if more burnup steps are used or the buffer region equivalent to water holes is neglected in TRITON the results are no longer equivalent (Table 3.16). The model, using 4.5 w/o UOX fuel, was first burned to 40 GWD/MTU to match the SAS2H model, and then the burnup was extended to 48 GWD/MTU so as to provide a more realistic end of life keffective value. The 40 GWD model used 26 burnup steps while the 48 GWD model had 30. The discharge isotopics of the 48 GWD/MTU model are given in Table 3.18, and it is these values to which uncertainty is propagated in later discussion and comparison. For completeness, the 40 GWD/MTU discharge isotopics are presented in Table 3.17 and the reader can see how an additional 8 GWD/MTU changes the fuel composition.

	Number Density at 10 Years Decay, (g)						
	ORIGEN with 1	TRITON		TRITON			
	SAS2H	with	TRITON with 1 BU	with			
			Step without				
Nuclide	Updated BU Step	1 BU Step	Buffer	4 BU Steps			
pb210	5.054E-10	5.369E-10	5.193E-10	5.357E-10			
ra226	3.165E-07	3.369E-07	3.239E-07	3.353E-07			
ac227	6.941E-08	7.115E-08	7.796E-08	7.326E-08			
th227	1.613E-10	1.653E-10	1.812E-10	1.702E-10			
th230	5.585E-03	5.951E-03	5.766E-03	5.921E-03			
pa231	4.323E-04	4.412E-04	4.806E-04	4.537E-04			
u234	1.675E+02	1.781E+02	1.754E+02	1.771E+02			
u235	1.234E+04	1.252E+04	1.315E+04	1.252E+04			
u236	5.526E+03	5.526E+03	5.570E+03	5.555E+03			
u237	2.945E-05	2.978E-05	3.463E-05	2.603E-05			
u238	9.298E+05	9.293E+05	9.265E+05	9.292E+05			
np237	5.730E+02	5.668E+02	6.277E+02	5.489E+02			
np239	1.001E-04	1.012E-04	1.187E-04	9.312E-05			
pu238	1.796E+02	1.771E+02	2.088E+02	1.725E+02			
pu239	5.555E+03	5.619E+03	6.484E+03	5.796E+03			
pu240	1.895E+03	1.899E+03	2.002E+03	2.278E+03			
pu241	9.717E+02	9.824E+02	1.143E+03	8.589E+02			
pu242	5.576E+02	5.571E+02	5.976E+02	5.123E+02			
am241	6.407E+02	6.533E+02	7.596E+02	5.690E+02			
am242m	8.756E-01	9.965E-01	1.237E+00	7.995E-01			
am243	1.163E+02	1.176E+02	1.380E+02	1.082E+02			
cm242	2.284E-03	2.599E-03	3.226E-03	2.086E-03			
cm244	2.200E+01	2.204E+01	2.826E+01	2.087E+01			
cm245	1.207E+00	1.218E+00	1.773E+00	1.213E+00			
c 14	3.331E-03	3.327E-03	3.461E-03	3.324E-03			
se 79	5.817E+00	5.804E+00	5.878E+00	5.796E+00			
sr 90	5.393E+02	5.363E+02	5.342E+02	5.366E+02			
tc 99	9.622E+02	9.603E+02	9.705E+02	9.585E+02			
i129	1.760E+02	1.762E+02	1.839E+02	1.755E+02			
cs137	1.179E+03	1.175E+03	1.199E+03	1.173E+03			
ba137m	1.801E-04	1.794E-04	1.831E-04	1.791E-04			
y90	1.401E-01	1.394E-01	1.388E-01	1.394E-01			
cs134	4.990E+00	4.749E+00	5.162E+00	4.751E+00			

Table 3.16: Comparison of isotopics between models.

65

	Discharge Isotopics, grams / MTHM					
pb210	3.068E-11	np237	5.327E+02	cm244	2.989E+01	
ra226	2.051E-08	np239	8.249E+01	cm245	1.188E+00	
ac227	4.409E-09	pu238	1.743E+02	c 14	3.337E-03	
th227	1.286E-11	pu239	5.745E+03	se 79	5.807E+00	
th230	1.311E-03	pu240	2.342E+03	sr 90	6.853E+02	
pa231	3.637E-04	pu241	1.362E+03	tc 99	9.752E+02	
u234	1.630E+02	pu242	5.071E+02	i129	1.741E+02	
u235	1.248E+04	am241	4.344E+01	cs137	1.476E+03	
u236	5.563E+03	am242m	8.884E-01	ba137m	2.268E-04	
u237	1.108E+01	am243	1.062E+02	y90	1.844E-01	
u238	9.291E+05	cm242	1.277E+01	cs134	1.346E+02	

Table 3.17: 40 GWD/MTU Discharge Isotopics

	Discharge Isotopics, grams / MTHM					
pb210	5.772E-11	np237	6.646E+02	cm244	6.780E+01	
ra226	2.667E-08	np239	8.826E+01	cm245	3.088E+00	
ac227	5.433E-09	pu238	2.678E+02	c 14	4.127E-03	
th227	1.827E-11	pu239	5.746E+03	se 79	6.827E+00	
th230	1.370E-03	pu240	2.716E+03	sr 90	7.807E+02	
pa231	4.338E-04	pu241	1.566E+03	tc 99	1.135E+03	
u234	1.436E+02	pu242	7.621E+02	i129	2.141E+02	
u235	8.934E+03	am241	5.407E+01	cs137	1.757E+03	
u236	5.989E+03	am242m	1.111E+00	ba137m	2.700E-04	
u237	1.240E+01	am243	1.883E+02	y90	2.112E-01	
u238	9.228E+05	cm242	1.912E+01	cs134	1.824E+02	

Table 3.18: 48 GWD/MTU Discharge Isotopics

The ESM approach was implemented in TRITON using the eigenvalues and eigenvectors of the 44-group covariance matrix for selected reactions to perturb the 44-group master cross section library input to the code. A total of 1938 samples were run as this number was the effective rank of this covariance matrix obtained by limiting perturbations to those corresponding to eigenvalues greater than 10⁻⁶ relative to the reference cross section. Results for uncertainty of the 33 tracked nuclides are presented in Table 3.19, along with the uncertainty results from the simple ORIGEN 1-group model using the direct perturbation method. Appendix C present further details.

	% Standard Deviation				
Nuclide	Deviation TRITON ORIGEN				
pb210	0.152	1.195			
ra226	0.185	1.355			
ac227	0.231	0.480			
th227	0.233	0.480			
th230	0.185	1.491			
pa231	0.253	0.518			
u234	0.173	1.579			
u235	0.583	1.312			
u236	1.048	0.712			
u237	2.053	2.521			
u238	0.015	0.078			
np237	1.602	0.605			
np239	3.910	13.643			
pu238	1.751	1.050			
pu239	1.045	0.806			
pu240	2.490	2.566			
pu241	2.053	2.521			
pu242	3.907	2.598			
am241	2.053	2.502			
am242m	2.104	2.172			
am243	3.908	13.643			
cm242	2.107	2.172			
cm244	4.345	11.340			
cm245	4.680	10.251			
c 14	0.379	0.452			
se 79	0.088	0.366			
sr 90	0.117	0.375			
tc 99	0.075	1.970			
i129	0.241	0.419			
cs137	0.032	0.382			
ba137m	0.032	0.382			
y 90	0.107	0.375			
cs134	0.300	1.138			

Table 3.19: Resulting standard deviations from both models.

At first glance, one is likely to say that the models are not equivalent. Rigorous and thorough search for the difference, however, yielded an explanation of these differences.

Both a primary and a secondary reason for the differences were discovered. Discussed first will be the secondary cause, as it has more tangible data. Consider carefully what terms of

the uncertainty are being propagated in each model. Let the result, \overline{y} , of a perturbed model be defined as:

$$\overline{y} = \overline{\Omega} \left| \overline{\sigma_0}^T \overline{\phi_0} + \overline{\Delta \sigma}^T \overline{\phi_0} + \overline{\sigma_0}^T \overline{\Delta \phi} + \overline{\Delta \sigma}^T \overline{\Delta \phi} \right|$$
(3.1)

where $\|\overline{\phi_0} + \overline{\Delta \phi}\|_1 = 1$. $\overline{\sigma_0}^T \overline{\phi_0}$ is the cross sections collapsed using the BOL, unchanged flux, $\overline{\Delta \sigma}^T \overline{\phi_0}$ is the perturbation added by a cross section perturbations, $\overline{\sigma_0}^T \overline{\Delta \phi}$ is the perturbation added by an updated burnup dependent flux spectrum, and $\overline{\Delta\sigma}^{\scriptscriptstyle T}\overline{\Delta\phi}$ is the second order perturbation added by both sources, which is assumed to be negligible. A nominal run of ORIGEN inputs $\overline{\sigma_0}^{^T}\overline{\phi_0}$. When making perturbations only in the ORIGEN input via the collapsed covariance data only the term $\overline{\Delta \sigma}^{\scriptscriptstyle T} \overline{\phi_0}$ is captured. When using a sequence such as TRITON that performs a transport update of flux and applies it to the master library, however, the term $\overline{\sigma_0}^{\scriptscriptstyle T} \overline{\Delta \phi}$ is also captured. Table 3.20 contains 44-group fluxes that have been normalized to the 1-group BOL flux, i.e. the sum of the BOL column of fluxes is equal to 1. As can be seen in Table 3.20, as fuel depletes with burnup, the groups from (0.1 - 3.0)ev tend to have the group fluxes decrease in value. Even though the flux is normalized by ORIGEN to keep the same power density, the overall shape of the flux changes. The energy spectra shift causes the term $\overline{\sigma_0}^T \overline{\Delta \phi}$ to be negative and reduces the overall uncertainty. A numerical experiment of this is shown in Table 3.21where the 44-group variance was collapsed using both averaged and burnup averaged flux changes from Table 3.20. Since the 1-group variances are collapsed from the 44-group using the flux, this directly affects the magnitude of the 1-group covariance values. More explicitly, the change in the 1-group covariance, $\overline{\Delta COV(\sigma)}$, due to a change in the 44-group flux, $\overline{\Delta \phi}$, is:

$$\overline{\Delta COV(\sigma)} = \overline{\phi_0}^T \overline{\overline{COV(\overline{\sigma})}} \ \overline{\Delta \phi} + \overline{\Delta \phi}^T \overline{\overline{COV(\overline{\sigma})}} \ \overline{\phi_0}$$
 (3.2)

where $\|\overline{\phi_0} + \overline{\Delta \phi}\|_1 = 1$. As shown in Table 3.22, the reduced covariance data cause output uncertainties to be reduced. These affects can account for at most about 20% of the difference between the two models.

ENEDOV	BOL	FI	ux at Various Bu	rnups (MWD/MTI	J)
ENERGY (ev)	FLUX	2000	14000	26000	38000
2.00E+07	1.5651E-03	1.5763E-03	1.6394E-03	1.6954E-03	1.7454E-03
8.19E+06	4.4291E-03	4.4524E-03	4.5744E-03	4.6727E-03	4.7564E-03
6.43E+06	1.3934E-02	1.3987E-02	1.4250E-02	1.4445E-02	1.4606E-02
4.80E+06	4.6375E-02	4.6501E-02	4.7090E-02	4.7490E-02	4.7804E-02
3.00E+06	3.0549E-02	3.0614E-02	3.0917E-02	3.1119E-02	3.1279E-02
2.48E+06	9.2170E-03	9.2356E-03	9.3223E-03	9.3810E-03	9.4283E-03
2.35E+06	3.7375E-02	3.7434E-02	3.7707E-02	3.7891E-02	3.8038E-02
1.85E+06	4.6158E-02	4.6211E-02	4.6462E-02	4.6639E-02	4.6788E-02
1.40E+06	6.4489E-02	6.4525E-02	6.4697E-02	6.4832E-02	6.4951E-02
9.00E+05	1.1652E-01	1.1653E-01	1.1664E-01	1.1676E-01	1.1687E-01
4.00E+05	1.1869E-01	1.1872E-01	1.1885E-01	1.1900E-01	1.1915E-01
1.00E+05	6.7525E-02	6.7549E-02	6.7674E-02	6.7801E-02	6.7913E-02
2.50E+04	1.5345E-02	1.5351E-02	1.5386E-02	1.5420E-02	1.5450E-02
1.70E+04	6.3885E-02	6.3925E-02	6.4125E-02	6.4312E-02	6.4473E-02
3.00E+03	5.7312E-02	5.7371E-02	5.7669E-02	5.7936E-02	5.8161E-02
5.50E+02	5.3043E-02	5.3131E-02	5.3575E-02	5.3974E-02	5.4312E-02
1.00E+02	3.3399E-02	3.3474E-02	3.3846E-02	3.4203E-02	3.4521E-02
3.00E+01	2.7654E-02	2.7687E-02	2.7868E-02	2.8123E-02	2.8389E-02
1.00E+01	5.0899E-03	5.1134E-03	5.2116E-03	5.2966E-03	5.3718E-03
8.10E+00	5.0429E-03	5.0502E-03	5.0835E-03	5.1301E-03	5.1791E-03
6.00E+00	5.2123E-03	5.1352E-03	4.7772E-03	4.5935E-03	4.4920E-03
4.75E+00	1.1050E-02	1.1047E-02	1.1012E-02	1.1009E-02	1.1022E-02
3.00E+00	1.3506E-02	1.3511E-02	1.3504E-02	1.3459E-02	1.3353E-02
1.77E+00	1.4724E-02	1.4636E-02	1.3081E-02	1.2225E-02	1.1719E-02
1.00E+00	1.2651E-02	1.2620E-02	1.1948E-02	1.1464E-02	1.1130E-02
6.25E-01	1.2905E-02	1.2795E-02	1.2113E-02	1.1786E-02	1.1611E-02
4.00E-01	1.9535E-03	1.9051E-03	1.7194E-03	1.6557E-03	1.6395E-03
3.75E-01	2.1397E-03	2.0608E-03	1.7892E-03	1.7019E-03	1.6831E-03
3.50E-01	2.3695E-03	2.2414E-03	1.8389E-03	1.7169E-03	1.6920E-03
3.25E-01	5.7177E-03	5.2816E-03	4.0465E-03	3.6950E-03	3.6285E-03
2.75E-01	3.6375E-03	3.3722E-03	2.6589E-03	2.4553E-03	2.4277E-03
2.50E-01	4.4406E-03	4.1571E-03	3.4519E-03	3.2579E-03	3.2562E-03
2.25E-01	5.4824E-03	5.1577E-03	4.4577E-03	4.2966E-03	4.3487E-03
2.00E-01	1.5222E-02	1.4320E-02	1.2882E-02	1.2742E-02	1.3134E-02
1.50E-01	2.2839E-02	2.1257E-02	1.9756E-02	2.0051E-02	2.1107E-02
1.00E-01	1.7676E-02	1.6305E-02	1.5454E-02	1.5962E-02	1.7058E-02
7.00E-02	1.2312E-02	1.1381E-02	1.0939E-02	1.1434E-02	1.2344E-02
5.00E-02	5.7830E-03	5.3621E-03	5.2021E-03	5.4819E-03	5.9593E-03
4.00E-02	5.1278E-03	4.7654E-03	4.6568E-03	4.9395E-03	5.4003E-03
3.00E-02	2.0759E-03	1.9324E-03	1.8998E-03	2.0264E-03	2.2263E-03
2.53E-02	4.7235E-03	4.4056E-03	4.3795E-03	4.7204E-03	5.2342E-03
1.00E-02	4.0214E-04	3.7567E-04	3.7910E-04	4.1461E-04	4.6586E-04
7.50E-03	4.0006E-04	3.7390E-04	3.8150E-04	4.2187E-04	4.7886E-04
3.00E-03	5.5627E-05	5.1989E-05	5.4307E-05	6.1555E-05	7.1549E-05

Table 3.20: Change in flux as burnup increases.

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	Reaction	Relative Variance	Relative Variance	Relative Variance
			for Averaged	for BU Averaged
Nuclide	Туре	for BOL Flux	Flux	Flux
92235	(Fission)	2.479E-05	2.327E-05	2.341E-05
92235	(n,γ)	1.343E-03	1.332E-03	1.335E-03
92236	(Fission)	3.619E-04	3.413E-04	3.373E-04
92236	(n,γ)	2.469E-03	2.271E-03	2.233E-03
92238	(n,2n)	7.371E-03	8.136E-03	8.366E-03
92238	(n,3n)	1.442E-02	1.627E-02	1.684E-02
92238	(Fission)	4.245E-04	4.365E-04	4.398E-04
92238	(n,γ)	9.982E-04	1.024E-03	1.033E-03
93237	(Fission)	8.608E-03	8.764E-03	8.807E-03
94238	(Fission)	8.105E-03	8.151E-03	8.205E-03
94238	(n,γ)	8.151E-04	7.013E-04	7.182E-04
94239	(Fission)	6.927E-05	4.575E-05	4.275E-05
94239	(n,γ)	6.489E-04	3.999E-04	3.659E-04
94240	(Fission)	2.087E-03	1.962E-03	1.928E-03
94240	(n,γ)	1.008E-03	7.567E-04	6.873E-04
94241	(Fission)	6.947E-05	5.915E-05	5.859E-05
94241	(n,γ)	1.668E-03	1.233E-03	1.210E-03
94242	(n,2n)	5.685E-02	6.301E-02	6.487E-02
94242	(n,3n)	2.474E-01	2.791E-01	2.889E-01
94242	(Fission)	1.529E-03	1.562E-03	1.571E-03
94242	(n,γ)	5.200E-03	5.168E-03	5.147E-03
95241	(n,2n)	9.176E-01	1.023E+00	1.056E+00
95241	(n,3n)	9.938E-01	1.121E+00	1.160E+00
95241	(Fission)	4.635E-04	4.583E-04	4.573E-04
95241	(n,γ)	5.538E-05	4.694E-05	4.486E-05
95243	(Fission)	3.918E-03	3.987E-03	4.007E-03
95243	(n,γ)	4.273E-01	3.207E-01	2.913E-01

Table 3.21: Change in 1-group cross section uncertainty due to flux change.

		% Standa	ard Deviation	
	ORIGEN	ORIGEN	ORIGEN - BU	TRITON
Nuclide		Averaged	Averaged	
pb210	1.195	1.115	1.106	0.152
ra226	1.355	1.262	1.252	0.185
ac227	0.480	0.453	0.454	0.231
th227	0.480	0.453	0.454	0.233
th230	1.491	1.394	1.384	0.185
pa231	0.518	0.490	0.494	0.253
u234	1.579	1.484	1.474	0.173
u235	1.312	1.316	1.339	0.583
u236	0.712	0.622	0.642	1.048
u237	2.521	2.154	2.248	2.053
u238	0.078	0.077	0.077	0.015
np237	0.605	0.566	0.544	1.602
np239	13.643	11.240	10.889	3.910
pu238	1.050	1.014	0.955	1.751
pu239	0.806	0.651	0.614	1.045
pu240	2.566	2.263	2.298	2.490
pu241	2.521	2.154	2.248	2.053
pu242	2.598	2.450	2.426	3.907
am241	2.502	2.137	2.233	2.053
am242m	2.172	1.855	1.968	2.104
am243	13.643	11.240	10.889	3.908
cm242	2.172	1.854	1.968	2.107
cm244	11.340	9.238	9.015	4.345
cm245	10.251	8.297	8.138	4.680
c 14	0.452	0.446	0.411	0.379
se 79	0.366	0.355	0.344	0.088
sr 90	0.375	0.362	0.359	0.117
tc 99	1.970	1.903	1.847	0.075
i129	0.419	0.413	0.388	0.241
cs137	0.382	0.371	0.354	0.032
ba137m	0.382	0.371	0.354	0.032
y 90	0.375	0.362	0.359	0.107
cs134	1.138	1.204	1.105	0.300

Table 3.22: Comparing discharge isotopic uncertainties for various flux updates.

The primary cause is due to the intrinsic methodology within TRITON itself, making this cause less tangible than the previous data. When working with the simplified ORIGEN models, the cross section library used and perturbed was a working library that had already been updated by SAS2H. What occurs in updating is that the reference cross sections of the

master library are subjected to resonance self-shielding analysis and updated based on those procedures. Consider Table 3.23 which shows the infinitely dilute, 44-group Am-243 capture cross section and the same 44-group cross section after resonance treatment is applied. Notice in the fast region, for example, some cross sections have changed by up to two orders of magnitude, while thermal energy groups show almost no change. The cross sections in the working library used by the lattice physics codes consisted of two parts: the reference component and the resonance self-shielded component in the resolved resonance energy range. When introducing perturbations into the cross sections in the master library, only the reference cross sections are perturbed. In the thermal energy range and unresolved resonances energy ranges, this perturbation is picked up because cross sections in these energy ranges were not considered in the resonance self-shielding analysis. However in the resonance regions, perturbations in the reference cross sections are easily overwhelmed by the magnitude of the resonance updates. This accounts for the smaller values of uncertainty seen in the TRITON models. This in turn forces the assumption that the resonances are not perturbed at all which means they are assumed to be perfectly known. This assumption can lead to under-estimated uncertainties, e.g. plutonium is highly affected by low-lying resonances in the U-238 absorption cross section. Further study into this matter was beyond the scope of this work.

Energy	Master Cross Section	Treated Cross Section
2.00E+07	1.10580996E-02	2.51107000E+00
8.19E+06	1.15339998E-02	2.44640000E+00
6.43E+06	1.23779997E-02	1.80170000E+00
4.80E+06	1.90602001E-02	1.58490000E+00
3.00E+06	3.09389997E-02	1.57700000E+00
2.48E+06	3.72469984E-02	1.57860000E+00
2.35E+06	4.79950011E-02	1.66690000E+00
1.85E+06	7.02812970E-02	1.63750000E+00
1.40E+06	1.07110001E-01	1.45130000E+00
9.00E+05	2.26100996E-01	4.88705000E-01
4.00E+05	5.77825010E-01	5.96831000E-01
1.00E+05	1.56948996E+00	1.58408000E+00
2.50E+04	2.11269999E+00	2.12890000E+00
1.70E+04	3.18284011E+00	3.19814000E+00
3.00E+03	8.80949974E+00	8.81098000E+00
5.50E+02	2.44122009E+01	2.44834000E+01
1.00E+02	4.30122986E+01	4.36353000E+01
3.00E+01	1.03167999E+02	1.02405000E+02
1.00E+01	3.96268997E+01	4.01052000E+01
8.10E+00	2.73347992E+02	2.98206000E+02
6.00E+00	1.06544998E+02	1.05959000E+02
4.75E+00	1.19114998E+02	1.19325000E+02
3.00E+00	9.09586029E+01	1.03057000E+02
1.77E+00	2.30728003E+03	2.27778000E+03
1.00E+00	1.05037003E+02	1.04765000E+02
6.25E-01	4.81402016E+01	4.75895000E+01
4.00E-01	5.20870018E+01	5.45036000E+01
3.75E-01	3.81459999E+01	3.93748000E+01
3.50E-01	3.39099998E+01	3.42099000E+01
3.25E-01	3.19234009E+01	3.19722000E+01
2.75E-01	3.14440002E+01	3.14953000E+01
2.50E-01	3.16469994E+01	3.16905000E+01
2.25E-01	3.21749992E+01	3.21896000E+01
2.00E-01	3.37787018E+01	3.34703000E+01
1.50E-01 1.00E-01	3.78905983E+01	3.79820000E+01
	4.37220001E+01	4.38877000E+01 5.06487000E+01
7.00E-02	5.05588989E+01	
5.00E-02 4.00E-02	5.72869987E+01 6.43789978E+01	5.72870000E+01 6.43790000E+01
3.00E-02	7.17419968E+01	7.17420000E+01
2.53E-02	9.00500031E+01	9.00490000E+01
1.00E-02	1.25260002E+01	9.00490000E+01 1.25260000E+02
7.50E-02	1.60410004E+02	1.60868000E+02
3.00E-03	2.79713013E+02	2.85547000E+02
J.00E-03	4.131 IOU IOETUZ	∠.0JJ41UUU⊑±UZ

Table 3.23: Change in Am-243 capture cross section due to resonance treatment.

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3.2.2. Fast Reactor Models with Transuranic Fuels

Before discussing the fast reactor models, it is important to take an objective look at the capabilities of TRITON, and to recognize how these may limit the results. TRITON is part of SCALE, which, for the most part was designed with current LWR's in mind. The most restrictive issue is with TRITON's cell domain restrictions. While any polygon can be modeled inside the cell, the cell itself is required to be rectangular, and the remaining space must be filled with moderator. For BWR and PWR square assemblies this is fine – one can model the exact dimensions of either the Wigner cell or the entire assembly. However if one tries to input a hexagonal cell, like the ones in the following models, the exact dimensions cannot be modeled. Essentially one ends up with a hexagonal peg in a square hole, which is filled with additional coolant, which yields an over moderated cell, which in turn affects flux which, in turn affects isotopic depletion. Another restriction is that we are using the 44group cross-section library which has a corresponding 44-group covariance library. These cross-sections were generated for a thermal reactor, i.e. about 50% of the data is in the thermal groups. There are only a few, broad fast energy groups, whose cross-sections are the reactions which drive the fast reactor. So, this over moderated cell and lack of fine data in the region where most reactions occur forces one to question the results obtained using this method. Further, it was early noted that the uncertainty of the resonances could not be treated. While the results do clearly demonstrate the methodology developed in this work, the actual numerical values can only be taken as plausible, rather than absolutely accurate. To provide comparison, Argonne's REBUS fast reactor code was used to further examine one of the fast reactor models. The fuel design corresponding to a conversion ratio of 0.70 is

modeled in REBUS and those results are presented in a subsequent section following the TRITON models.

Three distinctly different fast reactor models were examined in TRITON. Physically, they differ in terms of composition and operating conditions, but those parameters cause a difference in another key property of the fuel – conversion ratio or CR. For this study, the fuels examined have CRs equal to 0.25, 0.70 and 1.05, the latter being a so called breeder reactor and the two former being burner reactors. A description of basic composition (Table 3.24) and geometry and operating conditions (Table 3.25) of each is presented below, but the reader is referred to Appendix A where more detailed model data are provided. Table 3.26 through Table 3.28 show the discharge isotopics and Table 3.29 presents the isotopic relative standard deviations for each fuel type, in order of ascending conversion ratio. The reader will observe that U-235 content does not monotonically change between conversion ratios as do other isotopics. This anomalous behavior is noted but the source could not be identified within this work. The reader will also observe that uncertainty increases with conversion ratio. Increasing conversion ratio requires increasing uranium content and the relative fissile fraction of TRU. In observing the values in the SCALE covariance library, the largest sources of uncertainty are the fission and absorption reactions of the fissile minor actinides, which are often correlated to U-235. It follows that the increase in uranium and fissile TRU faction serve to magnify these uncertainties, thus uncertainty should increase as conversion ratio increases, which is the observed behavior.

	Weight Percent in TRU		
Nuclide / Conversion Ratio:	0.25	0.7	1.05
Np-237	18.635	7.334	9.907
Pu-238	0.855	1.253	0.000
Pu-239	32.764	48.058	72.150
Pu-240	14.983	21.973	4.469
Pu-241	4.936	7.241	0.250
Pu-242	2.956	4.335	0.000
Am-241	20.579	8.100	10.941
Am-242m	0.041	0.016	0.022
Am-243	3.565	1.403	1.895
Cm-244	0.689	0.271	0.366
Cm-245	0.041	0.016	0.022
Fissile Fraction, %	37.7	55.30	72.40
TRU Enrichment, %	59.2	20.6	16.2
Zr w/o	20	10	10
Depleted U, w/o	20.8	69.4	73.8

Table 3.24: Fuel composition data for fast reactor models.

Conversion Ration	0.25	0.70	1.05
Specific Power of active core,			
MW/MT	114.8	47.7	41.2
Discharge Burnup, GWD.MT	94.3	78.4	67.7
Height, cm	80	80	80
Number of pins per assembly	217	169	127
Assembly lattice pitch, cm	14.834	14.834	14.834
Inter-assembly gap, mm	4.45	4.0	4.0
Duct thickness, mm	4.45	3.0	3.0
Pin pitch-to-diameter ratio	1.29	1.11	1.10
Cladding thickness, mm	0.75	0.41	0.41

Table 3.25: Fuel geometry and power data for fast reactor models.

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	Discharge Isotopics, grams / MTHM					
pb210	4.950E-10	np237	9.563E+03	cm244	1.268E+04	
ra226	4.002E-09	np239	1.147E+02	cm245	3.075E+03	
ac227	1.967E-09	pu238	2.691E+04	c 14	9.454E-03	
th227	2.759E-11	pu239	1.367E+05	se 79	1.023E+01	
th230	4.585E-04	pu240	1.934E+05	sr 90	7.015E+02	
pa231	4.248E-04	pu241	3.548E+04	tc 99	2.454E+03	
u234	4.204E+02	pu242	5.692E+04	i129	5.747E+02	
u235	5.135E+02	am241	2.383E+04	cs137	3.582E+03	
u236	1.228E+02	am242m	1.908E+03	ba137m	5.560E-04	
u237	9.338E-01	am243	1.807E+04	y90	2.008E-01	
u238	3.789E+05	cm242	2.045E+03	cs134	1.694E+02	

Table 3.26: Discharge isotopics for fast reactor fuel of CR = 0.25.

	Discharge Isotopics, grams / MTHM					
pb210	3.200E-10	np237	1.895E+03	cm244	2.249E+03	
ra226	3.978E-09	np239	1.288E+02	cm245	5.264E+02	
ac227	1.573E-09	pu238	5.555E+03	c 14	9.724E-03	
th227	1.118E-11	pu239	9.083E+04	se 79	8.726E+00	
th230	2.281E-04	pu240	6.292E+04	sr 90	6.166E+02	
pa231	2.595E-04	pu241	9.851E+03	tc 99	2.040E+03	
u234	1.632E+02	pu242	1.066E+04	i129	4.561E+02	
u235	6.758E+02	am241	4.796E+03	cs137	2.878E+03	
u236	2.213E+02	am242m	3.738E+02	ba137m	4.465E-04	
u237	1.576E+00	am243	3.234E+03	y90	1.751E-01	
u238	7.246E+05	cm242	3.251E+02	cs134	1.704E+02	

Table 3.27: Discharge isotopics for fast reactor fuel of CR = 0.70

	Discharge Isotopics, grams / MTHM					
pb210	8.775E-11	np237	5.698E+02	cm244	2.776E+02	
ra226	1.135E-09	np239	1.438E+02	cm245	6.964E+01	
ac227	4.121E-10	pu238	1.370E+03	c 14	1.071E-02	
th227	2.723E-12	pu239	8.565E+04	se 79	8.740E+00	
th230	5.873E-05	pu240	3.916E+04	sr 90	6.242E+02	
pa231	6.911E-05	pu241	5.607E+03	tc 99	2.013E+03	
u234	4.292E+01	pu242	2.129E+03	i129	4.408E+02	
u235	5.982E+02	am241	1.414E+03	cs137	2.814E+03	
u236	2.533E+02	am242m	9.679E+01	ba137m	4.372E-04	
u237	1.674E+00	am243	4.754E+02	y90	1.776E-01	
u238	7.826E+05	cm242	9.319E+01	cs134	1.890E+02	

Table 3.28: Discharge isotopics for fast reactor fuel of CR = 1.05.

	Conversion Ratio			
Nuclide	0.25	0.7	1.05	
pb210	2.190	2.668	3.029	
ra225	2.770	3.760	4.214	
ac227	1.229	2.415	2.992	
th227	1.227	2.415	2.990	
th230	2.935	3.999	4.431	
pa231	1.222	2.384	2.900	
u234	3.129	4.317	4.723	
u235	1.490	2.587	3.267	
u236	1.795	4.793	5.535	
u237	2.764	3.832	4.247	
u238	0.205	0.230	0.255	
np237	1.114	1.680	2.828	
np239	10.752	15.475	15.989	
pu238	3.371	4.824	5.204	
pu239	1.468	2.406	2.800	
pu240	0.906	1.992	3.066	
pu241	2.764	3.833	4.246	
pu242	0.548	1.150	3.141	
am241	1.341	2.369	3.271	
am242m	1.434	1.836	1.972	
am243	10.752	15.475	15.989	
cm242	1.434	1.837	1.972	
cm244	13.025	16.925	19.875	
cm245	4.344	9.295	11.948	
c 14	1.655	0.962	0.708	
se 79	1.137	0.627	0.544	
sr 90	0.348	0.180	0.159	
tc 99	0.319	0.472	0.554	
i129	0.412	0.460	0.494	
cs137	0.094	0.098	0.099	
ba137m	0.094	0.092	0.099	
y 90	0.348	0.181	0.158	
cs134	2.553	2.250	2.214	

Table 3.29: Relative isotopic uncertainties for fast reactor fuels.

3.2.3. Fast Reactor Model with Transuranic Fuel and Recycling

The principles of the recycling methodology have already been discussed, and it suffices to say that the model used in this experiment is the same as the CR=0.70 model discussed in the previous section, save for the fact that the model burnup will be adjusted to give the end of cycle burnup of 41.4 GWD/MTHM rather then the end of life burnup

modeled in the previous section. This is done so that the target k-effective will the one for end of cycle, which is what would be real world objective of the reactor operator and fuel designer. Also, as indicated previously, the method of choice is the UREX process in which transuranics are separated as a stream and combined with depleted uranium to make up the new recycled fuel; however one could use any separation scheme with the methodology described. The model was run to an equilibrium state at 6 recycles and, in addition to typical uncertainty data, the beginning of cycle and end of cycle k-effective values, and their uncertainties, were also collected in output data. The values for end of cycle k-effective were also used to adjust transuranics loading with each recycle to maintain cycle energy production. Since the composition is defined by elemental weight percents and volume fractions, those are compared in Table 3.30. Table 3.31 gives the discharge isotopics of the equilibrium model and Table 3.32 shows discharge isotopics uncertainties originating due to cross sections, recycled isotopics, and total combined uncertainties. As can easily be seen from just the isotopics uncertainties, recycled isotopics originated uncertainties add to the cross-sections originated uncertainties to give almost a two-fold increase in total discharge isotopics uncertainties. A simple study (not shown) was conducted where it is assumed that the recycled fast reactor composition is known. When only the uncertainties of the LWR fuel were applied, this being a small fraction of the fuel in this particular TRITON model and possessing small uncertainties, the uncertainties originating from recycled isotopics in this case were negligible, (which is why they were not presented). In reality the reprocessing engineer knows fairly accurately the composition of the spent fast reactor and LWR fuels via performing mass spectroscopy. Knowing these compositions, the mass fractions of recycled fast reactor fuel, thermal reactor fuel, and depleted uranium would be altered to assure the

target EOC k-effective value is predicted to be achieved based upon cross section values which have inaccuracies.

Fuel Composition Properties						
		No Recycle	Equilibirum			
Uranium Volume Fraction:		0.5682	0.5594			
U-235	(w/o)	0.200	0.200			
U-238	(w/o)	99.800	99.800			
Neptunium Volu	me Fraction:	0.0023	0.001			
Np-237	(w/o)	100.000	100.000			
Plutonium Volur	ne Fraction:	0.1283	0.1380			
Pu-238	(w/o)	1.512	2.972			
Pu-239	(w/o)	57.999	47.048			
Pu-240	(w/o)	26.518	37.937			
Pu-241	(w/o)	8.739	5.715			
Pu-242	(w/o)	5.232	6.929			
Americium Volu	me Fraction:	0.0109	0.0108			
Am-241	(w/o)	85.092	63.995			
Am-242m	(w/o)	0.168	3.574			
Am-243	(w/o)	14.740	32.431			
Curium Volume Fraction:		0.0022	0.0031			
Cm-242	(w/o)	0.000	1.170			
Cm-244	(w/o)	94.444	76.878			
Cm-245	(w/o)	5.556	21.952			
Cm-246	(w/o)	0.000	1.170			

Table 3.30: Comparison of fuel composition properties for once through and recycled fuel.

Discharge Isotopics, grams / MTHM					
pb210	7.407E-11	np237	1.172E+03	cm244	2.593E+03
ra226	1.074E-09	np239	1.264E+02	cm245	7.121E+02
ac227	5.675E-10	pu238	5.574E+03	c 14	4.817E-03
th227	5.309E-12	pu239	8.999E+04	se 79	4.781E+00
th230	1.214E-04	pu240	7.586E+04	sr 90	3.318E+02
pa231	1.061E-04	pu241	1.201E+04	tc 99	1.093E+03
u234	9.651E+01	pu242	1.257E+04	i129	2.591E+02
u235	9.771E+02	am241	5.878E+03	cs137	1.560E+03
u236	1.451E+02	am242m	3.963E+02	ba137m	2.418E-04
u237	1.475E+00	am243	3.506E+03	y90	9.372E-02
u238	7.452E+05	cm242	3.720E+02	cs134	5.866E+01

Table 3.31: Discharge isotopics for equilibrium recycled fuel.

	Recycled Isotopics Orignated	Cross Sections	
Nuclide	(At Equilibrium)	Orignated	Total
pb210	12.450	2.668	12.732
ra225	12.931	3.760	13.466
ac227	14.025	2.415	14.231
th227	14.026	2.415	14.232
th230	12.681	3.999	13.296
pa231	13.465	2.384	13.674
u234	12.364	4.317	13.096
u235	1.121	2.587	2.819
u236	3.189	4.793	5.757
u237	5.460	3.832	6.671
u238	0.120	0.230	0.260
np237	11.311	1.680	11.435
np239	44.021	15.475	46.662
pu238	11.931	4.824	12.869
pu239	4.296	2.406	4.924
pu240	5.107	1.992	5.482
pu241	5.460	3.833	6.671
pu242	6.325	1.150	6.429
am241	6.846	2.369	7.244
am242m	9.100	1.836	9.284
am243	44.021	15.475	46.662
cm242	9.100	1.837	9.284
cm244	9.202	16.925	19.265
cm245	12.163	9.295	15.308
c 14	2.886	0.962	3.042
se 79	1.415	0.627	1.548
sr 90	0.369	0.180	0.410
tc 99	0.204	0.472	0.514
i129	0.719	0.460	0.854
cs137	0.106	0.098	0.144
ba137m	0.106	0.092	0.140
y 90	0.369	0.181	0.410
cs134	1.932	2.250	2.966

Table 3.32: Discharged isotopics uncertainties originating from recycled isotopics and cross sections sources of uncertainty.

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3.2.4. Comparison of Results, TRITON Models

As with the simplified models, the two most commonly examined metrics for repository performance or reprocessing – decay heat and radioactivity spanning 10 to 10,000 years of decay time, are graphically presented. These data are plots of the numerical values of the metrics, propagated from number density uncertainties presented in the previous sections for the TRITON models, including once through and the recycling method.

The first plots compare the uncertainty between the PWR models for the simplified ORIGEN method and the TRITON method (decay heat in Figure 3.11 and radioactivity in Figure 3.12). While, in terms of isotopics, the two methods deliver different uncertainties for reasons already discussed, as can clearly be seen, the long term affect on the metrics of interest is generally the same. This occurs mainly because the uncertainty on the long term heat contributors, e.g. plutonium, is on the same order of magnitude between the two models.

Next, the fast reactor models, three different compositions and conversion ratios, are compared with each other for the no recycle case. Since these are three different fuel types with different operating and composition parameters, comparison just provides a look at the three possibilities. While decay heat (Figure 3.13) and radioactivity (Figure 3.14) were higher for the low conversion ratio fuel, its long term uncertainty was the lowest. This is an interesting consideration for planning a fuel scenario regarding what one wants to dispose of and what one wants to recycle.

Finally, comparison is drawn between once-through faster reactor fuel and recycled fast reactor fuel, adding to it a stream of spent light water reactor fuel. As one would expect, uncertainty in both decay heat (Figure 3.15) and radioactivity (Figure 3.16) is higher in the

recycled fuel than in the once-through, and these figures give a visual comparison of that difference. In general, recycling nearly doubled the amount of uncertainty on the metrics examined, though the nominal long term performance is nearly identical for the two cases. Also, the k-effective study shows that the uncertainties originating from cross sections are the greatest contributor to k-effective uncertainty in that method (Figure 3.18), but uncertainties originating from recycled isotopic uncertainties scheme (recycling uncertainties alone in Figure 3.17) add a noticeable increase to that uncertainty (Figure 3.18). As can be seen from these figures, uncertainty from recycling seems to increase to some saturation as equilibrium is reached.

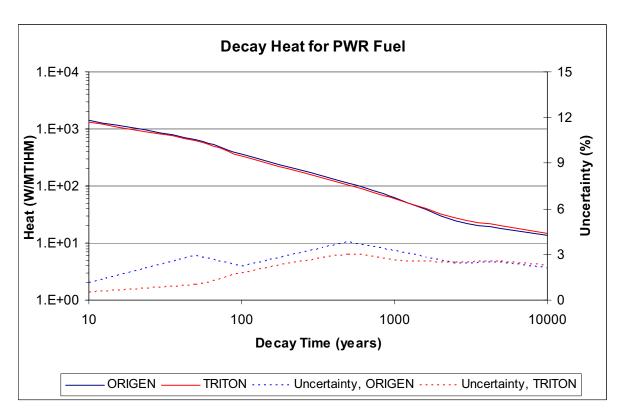


Figure 3.11: Decay heat comparison of simple ORIGEN and TRITON models.

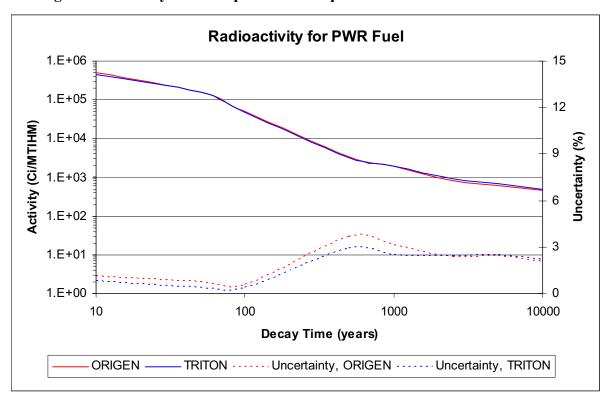


Figure 3.12: Radioactivity comparison of simple ORIGEN and TRITON models.

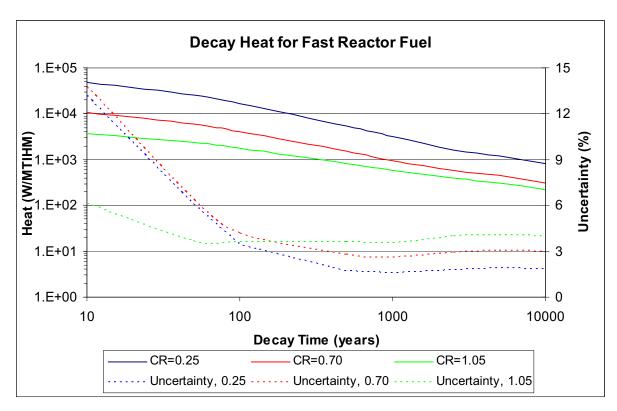


Figure 3.13: Decay heat comparison of three fast reactor fuels.

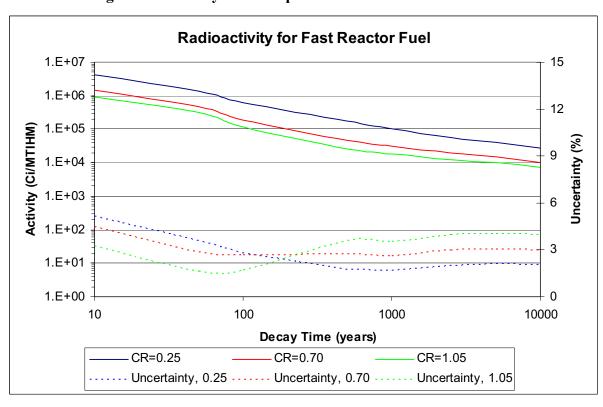


Figure 3.14: Radioactivity comparison for three fast reactor fuels.

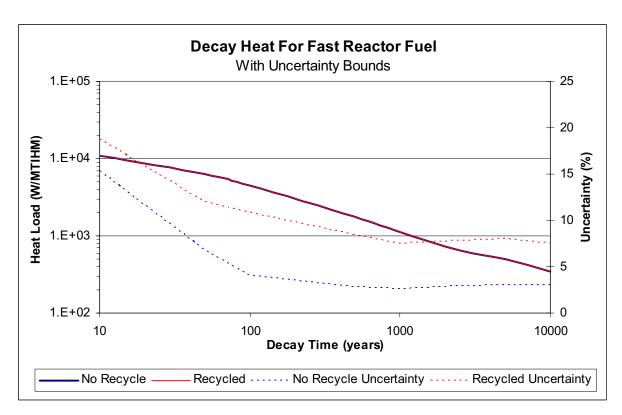


Figure 3.15: Decay heat comparison of once through and recycled fast reactor fuels.

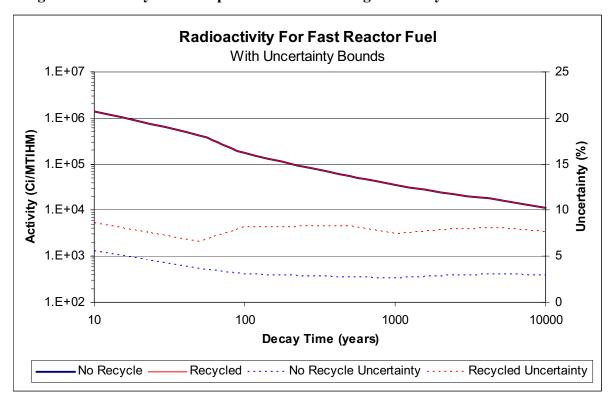


Figure 3.16: Radioactivity comparison of once through and recycled fast reactor fuels.

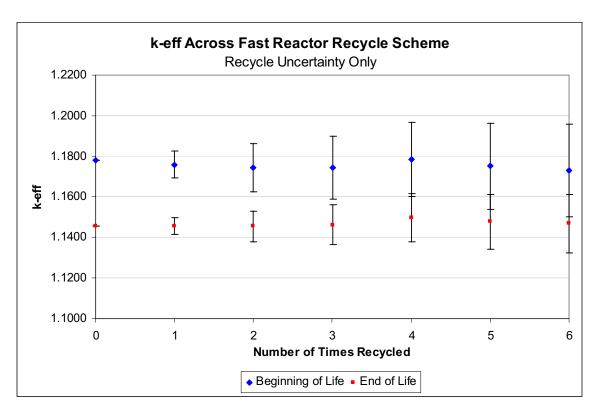


Figure 3.17: k-effective uncertainty due to recycled isotopics uncertainties only.

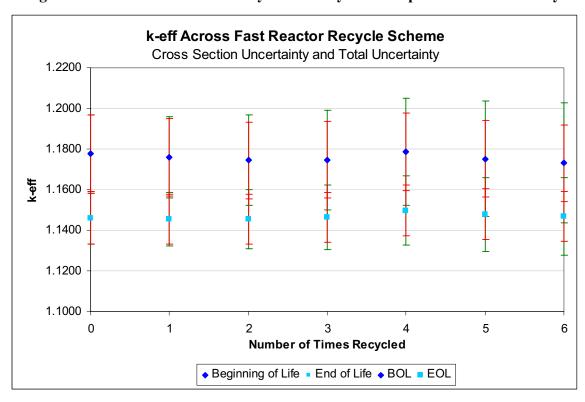


Figure 3.18: k-effective uncertainty due to cross sections and recycled isotopics uncertainties.

3.3. REBUS Fast Reactor Equilibrium Model

With the TRITON sequence producing questionable results for the fast reactor models examined, REBUS was used to examine a fast reactor fuel with uncertainty. The code REBUS, developed by Argonne National Laboratory, is similar to TRITON in that it takes a fuel composition, simulates operating a reactor with that fuel and performs depletion analysis and returns reactor operating parameters (e.g. burnup, k-effective, etc.) REBUS's capability to automatically perform a recycling analysis, brining the fuel to an equilibrium recycling scenario, is exploited here unlike in TRITON where this process was done externally. Further REBUS was able to easily model a 1/3 reactor core with all the heterogeneities as opposed to TRITON's one smeared cell, implying a more reliable model. One draw-back of REBUS is that it can only specifically track the actinide number densities, unlike the SCALE codes which track almost every isotope. For this reason, information from REBUS about amounts of fission and decay products present in the spent fuel is not available. The REBUS k-effective results are closer to expected values than those of the TRITON model, implying a more likely flux spectrum, and thus more likely isotopic composition and fuel depletion. The k-effective for EOC, along with EOC conversion ratio, are provided in Table 3.33 with uncertainty. All results are for the equilibrium composition.

Cross-section uncertainty propagation, using ESM, was implemented in the REBUS code by Dr. Hany Abdel-Khalik. Developed as part of the work reported here was an equilibrium model in REBUS for the fast reactor with a conversion ratio of ~0.77, using the same recycling specifications given for the recycle scenario using TRITON. The model was executed using Dr. Abdel-Khalik's modified version of REBUS. Table 3.34 gives the discharge isotopics, here EOC core composition normalized to 1 MTHM, and Table 3.35

presents the isotopic uncertainties. Figure 3.19 - Figure 3.20 show the heat load and radioactivity, respectively, with the uncertainty for the REBUS model compared to the TRITON recycle model. Uncertainties on the LWR recycled isotopics were considered, accomplished by perturbing the LWR isotopics in each input deck via the ESM approach and then determining the uncertainties produced by these, which were very small. The results presented include these perturbations as well as the uncertainty induced by cross section uncertainties. A drawback in REBUS is that correlations between the cross section induced uncertainties and the recycled LWR isotopics induced uncertainties must be assumed to be zero. That is the cross section uncertainties that lead to producing the recycled isotopics uncertainties during LWR operations are not consistently carried forward to the REBUS model of FR operations. To do this, a singlet set of cross sections and their perturbations would need to be employed by the models representing LWR and FR operations. In examining the results, note that these values are much closer to those indicated by the ABTR report [26], in isotopics as well as conversion ratio and k-effective, than are the TRITON results. As discussed before, uncertainties are available only for the actinides and are considerable higher than those predicted by TRITON. With the exception of the initial decay heat uncertainty, the REBUS results had uncertainties more than twice as high as the TRITON model. The higher results are due to a different cross section library, covariance library, and model – all specialized for the fast reactor. The assumptions forced by the resonance treatment in TRITON were recognized to be missing uncertainty components, thus the uncertainties likely underestimated. The REBUS 15-group structure has very little dependence on thermal energies, with 14 of the 15 groups spanning fast and resonance

energies. Since the structure was designed for the fast reactor, the associated uncertainties are more indicative of the fast system and less restricted by the issues in TRITON.

Operational Parameter	Nominal Value	Uncertainty (%)	
EOC k-effective	0.99925	0.2180	
EOC Core Conversion Ratio	0.7695	1.7147	

Table 3.33: Operating Parameters for REBUS model.

	Discharge Isotopics, grams / MTHM						
pb210		np237	3.644E+03	cm244	1.836E+03		
ra226		np239		cm245	4.324E+02		
ac227		pu238	6.074E+03	c 14			
th227		pu239	1.030E+05	se 79			
th230		pu240	6.228E+04	sr 90			
pa231	1	pu241	8.259E+03	tc 99			
u234	1.088E+02	pu242	1.230E+04	i129			
u235	1.213E+03	am241	7.936E+03	cs137			
u236	9.393E+01	am242m	5.665E+02	ba137m			
u237		am243	3.968E+03	y90			
u238	7.878E+05	cm242	2.355E+02	cs134			

Table 3.34: Discharge Isotopics for REBUS model.

I:	Isotopics Uncertainties, grams / MTHM					
pb210		np237	7.344	cm244	20.387	
ra226		np239		cm245	38.508	
ac227		pu238	18.616	c 14		
th227		pu239	1.336	se 79		
th230		pu240	7.463	sr 90		
pa231		pu241	10.570	tc 99		
u234	18.383	pu242	18.769	i129		
u235	0.809	am241	10.163	cs137		
u236	2.034	am242m	14.836	ba137m		
u237		am243	17.986	y90		
u238	0.950	cm242	8.856	cs134		

Table 3.35: Isotopc Uncertainties for REBUS model.

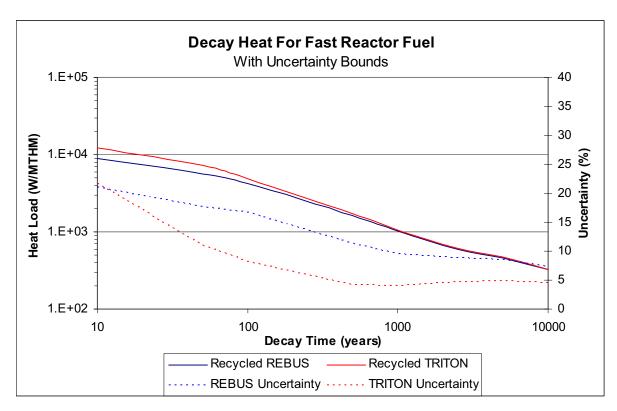


Figure 3.19: Decay heat comparison of REBUS and TRITON Models.

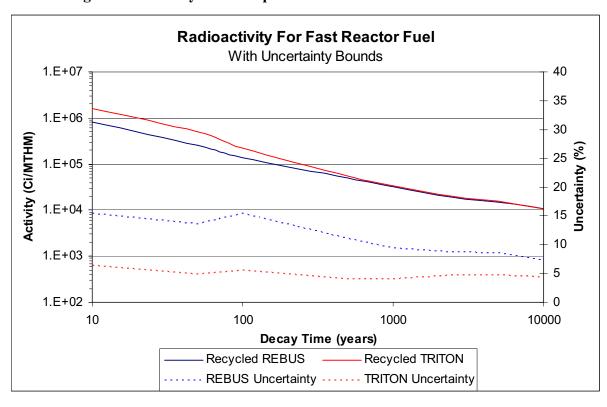


Figure 3.20: Radioactivity comparison of REBUS and TRITON Models.

4. Discussion and Conclusions

4.1. Discussion of the Use of ESM in this Study

The Efficient Subspace Method (ESM) has been demonstrated to produce results equivalent to those of traditional stochastic sampling methods. While this provides an alternative to these methods in any case where stochastic sampling could be used, it is most beneficial in models where stochastic methods would not be practical. For the case of the TRITON and REBUS models, where perturbed input data numbered in thousands, stochastic sampling would have taken at least twice as long as using ESM. In most multi-scale, multiphysics codes, such as the TRITON sequence in SCALE, input data do number in the thousands, if not orders of magnitude higher. For example, the core simulator FORMOSA developed at North Carolina State University has millions of input data, and, if perturbed, would have thousands of millions of perturbations. With growing reliance on computer simulation in many industries, including nuclear, these large, complex models are becoming more popular and necessary. Thus, having a method to quickly and efficiently propagate uncertainties becomes a much desired capability. Since one goal of this study was to demonstrate the capability of ESM to perform in this manner, it can be concluded that ESM can be successfully used on large, complex models while producing results that are equivalent to traditional sampling methods.

4.2. Discussion Concerning the Results of the Models

The other objective of this study was to determine how cross section uncertainties affect back-end fuel cycle metrics such as decay heat, radioactivity, and radiotoxicity.

Essentially, this implies determining how cross section uncertainty affects the composition of

what is discharged from the reactor, since all other metrics are directly proportional to isotopic masses. For the UOX fuels, in both the simplified and detailed models, uncertainty on these metrics was 1-2 % for the first 100 years and then 3-5% thereafter. Short term uncertainty is dominated by low-uncertainty fission products that make up the majority of the heat load and activity in the first 100 years. The constant uncertainty in the long term is due to a few long lived actinides, mainly plutonium and americium isotopes. When looking at the simplified MOX fuels, the uncertainty increases in the short term, especially for the decay heat, due to uncertainty in fission products and short lived actinides caused by the uncertainties on the fission and absorption cross sections of the minor actinides, which are now present in greater quantities. However, the long term uncertainty is still about 5%, stemming from long lived actinides with similar uncertainties to those for the same isotopes in the UOX fuel.

Considerable differences between the short term uncertainties for the PWR models were discovered between the simplified ORIGEN model and the TRITON model. An evaluation concluded this was the result of the resonance treatment applied in the TRITON sequence. In the simple ORIGEN models, prepared 1-group cross sections were perturbed just before the depletion calculation. However in TRITON, only the reference 44-group cross sections were perturbed, and the resonance component in the resolved resonance regions missed, which diminished the affect of the perturbations. Further study on this topic was beyond the scope of the current work.

The fast reactor TRITON models, though the short-comings of the model are recognized, showed considerably higher values for heat load and radioactivity than those of the PWR model. This was expected because of the nature of the fuel, that being a large

weight percent of transuranics which contribute to the longevity and magnitude of these metrics. The increasing activity and decay heat with respect to decreasing conversion ratio is the result of increasing the transuranic enrichment and extending the burnup to reach the clad fluence limit, therefore building up more of products contributing to these metrics. The increasing uncertainty with increasing conversion ratio results from the increase of uranium content and fissile fraction of the TRU, which have large, highly correlated uncertainties. Still the over all long term uncertainty of these fuels is about 3-5 %.

The TRITON recycling models show the expected result of increasing uncertainty as fuel is recycled to an equilibrium. While this process nearly doubles the uncertainty, long term uncertainties are still on the order of 5% but short term uncertainties could increase from 10 to 20%, which would result in more conservative handling and processing in the near term. The main concern of recycling would appear to be the uncertain compositions being used as fuel for the reactor, but this is easily eliminated by measuring the isotopic masses prior to refabrication. Recycling isotopics uncertainties add a non-negligible amount to the already present cross-section uncertainty on k-effective values.

The REBUS equilibrium recycling model was nominally comparable to the TRITON model in terms of long term heat load and radioactivity. However the nominal REBUS results are regarded as more indicative of the properties of the spent fuel in consideration, since that model very nearly matched the one in the Argonne report after which it was designed, both in isotopics and operating conditions. The uncertainty displayed by the REBUS model was, however, considerable different than that of the TRITON model, in some instances up to 7 times greater. This indicates that, as stated in the discussion on resonance treatment, that the TRITON model was underestimating the uncertainty on the

isotopics, which propagated to the metrics examined. Also the 11% uncertainty on the conversion ratio implies that reactor operation may or may not destroy the desired amount of actinides, which is essentially the job of this type of reactor. So, for this equilibrium fuel cycle, judging from the REBUS results, the fuel that is discharged from the reactor needs to be carefully measured in terms of isotopics before refabrication or would have to be treated very conservatively if immediately disposed of. Thankfully there would not be much of it to dispose of if widespread use of fast reactors destroyed much of the spent LWR fuel.

If one is simply disposing of UOX or even MOX fuel that was burned in a LWR, uncertainties due to cross sections seem to be of little concern. This is especially true when one considers the highly restrictive geological uncertainties, waste package material uncertainties, and even reactor operation uncertainties, which can easily overwhelm the low uncertainties seen for these models. However, when one considers reprocessing, cross-section uncertainties become more important. High short term uncertainties on decay heat and activity would result in more conservative handling of fuels for reprocessing. Isotopic uncertainties for fuel mainly impact composition changes during irradiation and design of the repository, since isotopics can be measured prior to refabrication. The decrease in repository loading due to fast reactor operation likely will outweigh the uncertainty associated with that fuel, but further study on such margins is needed. Concluding from these results, cross section uncertainty would need to be reduced to better the operation of fast reactors and dispose of their reprocessed fuel.

4.3. Recommendations and Future Work

The immediate future work is to develop the uncertainty propagation methodology in this work into a model that is usable within SINEMA's primary software, GENIUS. In

keeping with the GENIUS format, this will be a series of tables from which to interpolate data concerning uncertainties, given a fuel composition and discharge burnup, for each reactor type. Further development should be implemented in the TRITON sequence to improve the treatment of resonance uncertainties.

Beyond the cross section uncertainties propagated into the isotopics, which is all that was considered in this work, reprocessing includes chemical reaction rates and uncertainties, plant efficiencies, separation time, storage and transport time, etc. All of these can play a role in the composition of the fuel that is fabricated. Also, as shown in a very simple experiment for an LWR, operational uncertainties can add much more to the uncertainties seen in the discharged isotopics than the cross sections do, so they warrant further study. Finally, the metrics considered herein were rather simplistic metrics, directly related to isotopics masses and instantaneous in nature. The uncertainty margins on more thorough, integral metrics (e.g. long-term heat integral, or dose calculations) would also be of interest, especially for disposal purposes. For the recycling cases, further study should be conducted to include correlations between recycled isotopics uncertainties and cross sections uncertainties, which were assumed uncorrelated in this work. Finally the cut-off value for the choice of samples to run could be adjusted, as the value used in this work was a very conservative. Removing unused cross sections and working with absolute values rather than relative values could help the user to better select which samples to run. In this work, an unimportant cross section may have a high relative uncertainty and that sample would be kept when it does not actually need to be run.

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Appendix A: Fuel Models

I. The Typical LWR Model Using a UOX Fuel

For use in base cases and general examples, SCALE is distributed with a flux spectrum and three-group card image cross-section library that are representative of a typical LWR. The first sampling routine was implemented with this resource. The input library can be used in its three-group form along with the flux spectrum supplied directly to ORIGEN or the library may be collapsed to a one-group binary library. This model was taken directly from the ORIGEN users manual and changed only in the enrichment of U-235 to 4.5 w/o rather than the 3.3 w/o of the original example problem so the reader is referred to the ORIGEN users manual [21].

II. The PWR Model Using a UOX Fuel

The PWR model used in this study is based on the example provided with the SCALE 5.0 SAS2H User's Manual, that being a Westinghouse type PWR fuel assembly (Figure A.II.1). The assembly is 17 pins by 17 pins with 25 water holes and an active fuel length of 12 feet (365.76 cm) in a square pitch design. Fuel rods have a pitch of 1.25984 cm and an outside diameter of 0.83566 cm (no gap in this model). The fuel itself is UO₂ containing 461.4 kg of uranium in the proportions 4.50 w/o U-235, 95.472 w/o U-238 and 0.028 w/o U-234 and is volume fraction weighted (VF=0.90182) for the given fuel assembly based on the volume weighting method described in the SAS2H user's manual, i.e. VF depends on mass of fuel and volume of fuel assembly. Cladding is zirc2 (versus actual zirc4) and the moderator is water. Operating temperatures are 811, 570 and 570 degrees Fahrenheit for the fuel, clad and moderator respectively with the moderator density at 0.733 g/cm³. The

SAS2H model is burned at a specific power of 18.456 MW/assembly for 1000 days giving a total burnup of 40 GWD/MTU.

The corresponding ORIGEN model (Figure A.II.2), from which sampling is conducted, is one metric ton of the 4.5 w/o UOX fuel with approximately 271 kg of zirc2 cladding. The ORIGEN model is set to deplete the fuel using the binary cross section library generated from the fuel specific SAS2H model. Note: cross section perturbations are introduced directly into this binary working library during the sampling procedure. Fuel specific power density is 40 MW/MTU for a cycle length of 1000 days giving the same 40GWD/MTU burnup as the SAS2H Model. The ORIGEN model is set to "burn" the fuel for 10 equal time steps at 100% power and then print out decay isotopics at charge, discharge, 1, 5, 10, 50, 100, 500, 1000, 2500, 5000, and 10000 year times. ORIGEN also has the availability to print the decay heat, activity, and radiotoxicity for these times.

```
=sas2 parm='skipcellwt'
sas2 LWR UOX: 40 mwd/kgHM, 17*17 pin, pwr, 1 cyc
44groupndf5 latticecell
' FUEL COMPOSITION
uo2 1 0.90182 811 92234 0.028 92235 4.5 92238 95.472 end
' CLADDING
zirc2 2 1 620 end
' MODERATOR / COOLANT
                h2o \ 3 \ den=0.733 \ 1 \ 570 \ end
end comp
' FUEL-PIN GEOMETRY
squarepitch 1.25984 0.83566 1 3 0.94996 2 end
more data szf=1.2 eps=1.0-7 ptc=1.0-8 end
' ASSEMBLY AND CYCLE PARAMETERS
npin/assm=264 fuelngth=365.76 ncycles=1 nlib/cyc=1
printlevel=6
lightel=16 inplevel=1
```

Figure A.II.1: SAS2H Model, PWR, 4.5 w/o and 40 GWD/MTU

Figure A.II.1: SAS2H Model, PWR, 4.5 w/o and 40 GWD/MTU, cont.

```
=origens
0$$ a5 28 e
1$$ 1
pwr nuclear data - sample case 1
' _ _ _ _ _ _ _ _ _ _ _ _ _ _ _ _ .
'Uses SAS Updated Lib
3$$ 33 0 1 -88 a33 -88
35$$ 0
4 t.
56$$ 10 a13 50 4 3 0 1 1 e
57** a3 1-14 e
95$$ 1
pwr - 4.5% enriched u
mt of heavy metal charged to reactor
'Power specifications, 40 MW for 1000 Days
58** 10r40
60** 8i110 1000
'Nuclide identifies - charged nuclides
66$$ 1 a5 1 a9 1 e
73$$ 60120 130270 140280 140290 220460 220470 220480 220490
220500 240500 240520 240530 240540 250550 260540 260560 260570 260580
270590 280580 280600 280610 280620 280640 400900 400910 400920 400940
400960 410930 420920 420940 420950 420960 420970 420980 421000 501120
501140 501150 501160 501170 501180 501190 501200 501220 501240
922350 922380 922340
'Molar masses of charged nuclides
' - - - - - - - - - - - - - - -
74** 1.5 4.0 .607 .034 .304 .277 2.771 .204 .2 5.04 57.423 6.415 1.574 0.327 4.037 61.018 1.439 0.31 0.915 111.862
41.783 1.869 5.645 1.609 1421.122 306.725 462.239 460.074 72.5 10.258
.957 .532 .926 .958 .546 1.357 .54 .321 .219 .113
4.681 2.47 7.729 2.739 10.392 1.467 1.823 191.45 4011.75 1.13
75$$ 47r1 3r2 t
56$$ 0 10 a10 10 a14 5 a17 2 e
```

Figure A.II.2: ORIGEN Model, PWR, 4.5 w/o and 40 GWD/MTU

Figure A.II.2: ORIGEN Model, PWR, 4.5 w/o and 40 GWD/MTU, cont.

III. The BWR Model Using a UOX Fuel

The BWR model used in this study is adapted from a whole assembly SAS2H model in an Oak Ridge National Lab report concerning validation of SAS2H for BWR predictions (Figure A.III.1). The assembly is a General Electric type 7x7 fuel assembly with an active fuel length of 12.17 feet (370.84 cm) in a square pitch design. Fuel rods have a pitch of 1.875 cm and an outside diameter of 1.242 cm (no gap in this model). The model has been simplified from the mentioned report in that it has been homogenized (no burnable poisons or water holes) and has an initial enrichment of 4.5 w/o so as to match the PWR fuel in that regard. The fuel itself is UO₂ containing 190.71 kg of uranium in the proportions 4.50 w/o U-235, 95.472 w/o U-238 and 0.028 w/o U-234 and is volume fraction weighted (VF=0.5589) for the given fuel assembly based on the volume weighting method described in the SAS2H user's manual, i.e. VF depends on mass of fuel and volume of fuel assembly. Cladding is zirc2 and the moderator is water. Average operating temperatures are 840, 620 and 558 degrees Fahrenheit for the fuel, clad and moderator respectively. Void fraction for the whole assembly model is handled by choosing an average moderator density based on the

void fraction: $\rho_{average} = (\alpha)\rho_{vapor} + (1-\alpha)\rho_{liquid}$. The saturated liquid and vapor densities of water at 558 degrees Fahrenheit are 0.74178 and 0.03593 g/cm³ respectively and the void fractions used in this study were 0, 0.35, 0.50 and 0.65. The SAS2H model is burned at a specific power of 7.628 MW/assembly for 1000 days giving a total burnup of 40 GWD/MTU.

The corresponding ORIGEN model (Figure A.III.2), from which sampling is conducted, is one metric ton of the 4.5 w/o UOX fuel with approximately 271 kg of zirc2 cladding. The ORIGEN model is set to deplete the fuel using the binary cross section library generated from the fuel specific SAS2H model. Note: cross section perturbations are introduced directly into this binary working library during the sampling procedure. Fuel specific power density is 40 MW/MTU for a cycle length of 1000 days giving the same 40GWD/MTU burnup as the SAS2H Model. The ORIGEN model is set to "burn" the fuel for 10 equal time steps at 100% power and then print out decay isotopics at charge, discharge, 1, 5, 10, 50, 100, 500, 1000, 2500, 5000, and 10000 year times. ORIGEN also has the availability to print the decay heat, activity, and radiotoxicity for these times.

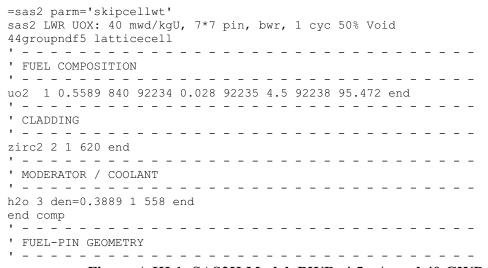


Figure A.III.1: SAS2H Model, BWR, 4.5 w/o and 40 GWD/MTU

```
squarepitch 1.875 1.242 1 3 1.430 2 end
more data szf=1.2 eps=1.0-7 ptc=1.0-8 end
' ASSEMBLY AND CYCLE PARAMETERS
npin/assm=49 fuelngth=370.84 ncycles=1 nlib/cyc=1
printlevel=6
lightel=16 inplevel=1
numins= 1 ortube= 0.61214 srtube=0.5715 facmesh=1.4 end
power=7.6284 burn=1000 down=0 end
' Light elements (kg) per assembly
c 0.05999 n 0.03377 o 62.14 al 0.04569
si 0.06586 p 0.1422 ti 0.04983 cr 2.340
mn 0.1096 fe 4.599 co 0.03344 ni 4.402
zr 100.8 nb 0.3275 mo 0.1816 sn 1.652
end
```

Figure A.III.1: SAS2H Model, BWR, 4.5 w/o and 40 GWD/MTU, cont.

```
=origens
0$$ a5 28 e
1$$ 1 1t
bwr nuclear data - sample case 1
'Uses SAS Updated Lib
3$$ 33 0 1 -88 a33 -88
2.t.
35$$ 0 4t
56$$ 10 a13 50 4 3 0 1 1 e
57** a3 1-14 e
95$$ 1 5t
bwr - 4.5% enriched u
mt of heavy metal charged to reactor
'Power specifications, 40 MW for 1000 Days
58** 10r40
60** 8i110 1000
'Nuclide identifies - charged nuclides
66$$ 1 a5 1 a9 1 e
73$$ 60120 130270 140280 140290 220460 220470 220480 220490
220500 240500 240520 240530 240540 250550 260540 260560 260570 260580
270590 280580 280600 280610 280620 280640 400900 400910 400920 400940
400960 410930 420920 420940 420950 420960 420970 420980 421000 501120
501140 501150 501160 501170 501180 501190 501200 501220 501240
922350 922380 922340
'Molar masses of charged nuclides
74** 1.5 4.0 .607 .034 .304 .277 2.771 .204 .2 5.04 57.423 6.415 1.574 0.327 4.037 61.018 1.439 0.31 0.915 111.862
41.783 1.869 5.645 1.609 1421.122 306.725 462.239 460.074 72.5 10.258
.957 .532 .926 .958 .546 1.357 .54 .321 .219 .113
```

Figure A.III.2: ORIGEN Model, BWR, 4.5 w/o and 40 GWD/MTU

```
4.681 2.47 7.729 2.739 10.392 1.467 1.823 191.45 4011.75 1.13
75$$ 47r1 3r2 t
56$$ 0 10 a10 10 a14 5 a17 2 e
57** a3 1-14 e
95$$ 1 5t
'Decay time steps in years
60** 1 5 10 50 100 500 1000 2500 5000 10000
'Decay Output Specifications
1 5z 1 2z 1 e
'65$$ 3z 1 20z 1 20z 1 e
61** 5r1-14 1+6 1+4
81$$ 2 0 26 1 e
82$$ f2 6t
56$$ f0 t
end
```

Figure A.III.2: ORIGEN Model, BWR, 4.5 w/o and 40 GWD/MTU, cont.

IV. The PWR Separation Model Using a UOX Fuel

Since reprocessing hinges on chemical separation, we are also interested in uncertainty in the separation of SNF. Since chemistry only applied to elements we look at elemental uranium, neptunium, plutonium, americium, and curium – the key actinides produced by irradiating UOX fuel and those of interest to reprocessing and repository performance. This study examined heat loads produced by the separated elements in "lumps". That is, at a given time after irradiation the key isotopics for a certain element at that time are placed into ORIGEN and decayed for the typical time steps already mentioned in previous appendices and the total heat produced from that "lump", regardless of daughters, is examined. This is equivalent to chemically separating that element, with 100% efficiency assumed, at the given time and then sitting that element away to decay. The procedure is repeated with +1 standard deviation of the isotopics as defined by the uncertainty in the PWR UOX fuel model which yields a heat +/- a heat uncertainty for each of the elements over their given decay times. A table of isotopics at 5, 10, and 25 years after irradiations is given in

Table A.IV.1. Since the ORIGEN model used in this experiment is so general and uses a well known decay model, it is not presented.

	Mass a	t 5 Years	Mass at	10 Years	Mass at 25 Years		
Nuclide	Grams	+/- Grams	Grams	+/- Grams	Grams	ms +/- Grams	
u234	1.60E+02	2.53E+00	1.67E+02	2.64E+00	1.87E+02	2.96E+00	
u235	1.23E+04	1.62E+02	1.23E+04	1.62E+02	1.23E+04	1.62E+02	
u236	5.53E+03	3.94E+01	5.53E+03	3.94E+01	5.53E+03	3.94E+01	
u237	3.75E-05	9.45E-07	2.95E-05	7.42E-07	1.43E-05	3.60E-07	
u238	9.30E+05	7.21E+02	9.30E+05	7.21E+02	9.30E+05	7.21E+02	
np237	5.69E+02	3.44E+00	5.73E+02	3.47E+00	5.95E+02	3.60E+00	
np239	1.00E-04	1.37E-05	1.00E-04	1.37E-05	9.99E-05	1.36E-05	
pu238	1.87E+02	1.96E+00	1.80E+02	1.88E+00	1.60E+02	1.67E+00	
pu239	5.56E+03	4.48E+01	5.55E+03	4.48E+01	5.55E+03	4.48E+01	
pu240	1.89E+03	4.85E+01	1.90E+03	4.86E+01	1.90E+03	4.88E+01	
pu241	1.24E+03	3.12E+01	9.72E+02	2.45E+01	4.71E+02	1.19E+01	
pu242	5.58E+02	1.45E+01	5.58E+02	1.45E+01	5.58E+02	1.45E+01	
am241	3.79E+02	9.49E+00	6.41E+02	1.60E+01	1.12E+03	2.80E+01	
am242m	8.97E-01	1.95E-02	8.76E-01	1.90E-02	8.13E-01	1.77E-02	
am243	1.16E+02	1.59E+01	1.16E+02	1.59E+01	1.16E+02	1.58E+01	
cm242	8.07E-03	1.75E-04	2.28E-03	4.96E-05	2.12E-03	4.60E-05	
cm244	2.66E+01	3.02E+00	2.20E+01	2.49E+00	1.24E+01	1.40E+00	
cm245	1.21E+00	1.24E-01	1.21E+00	1.24E-01	1.21E+00	1.24E-01	

Table A.IV.1: Isotopics of discharged UOX fuel at 5, 10, and 25 years after irradiation.

V. The PWR Model Using a MOX Fuel

For this study it was assumed that the MOX fuel would be burned in a conventional PWR and thus the SAS2H model (Figure A.V.1) for this fuel has the same geometry and operating parameters as that of the PWR UOX model described above with the exception that fuel is now burned to 50 GWD/MTHM with a specific power of 23.070 MW/assembly for 1000 days. There is still 461.4 kg of heavy metal in the fuel but it is now divided between UO₂ and PuO₂. The composition of the MOX fuel is taken from an AFCI report from Los Alamos National Lab for Fiscal Year 2003, in which the fuel composition used in this study is designated "ALWR-2." The 424.04 kg of uranium is in the proportions 1.40 w/o U-235, 98.572 w/o U-238 and 0.028 w/o U-234 and is volume fraction weighted at VF=0.82876.

The 37.36 kg of plutonium is in the proportions 1.655 w/o Pu-238, 61.751 w/o Pu-239, 24.701 w/o Pu-240, 3.248 w/o Pu-241, 8.645 w/o Pu-242 and is volume fraction weighted at VF=0.06978. Since the sited report only specified a mass of Pu in the fuel, the isotopics vector was selected to approximate the discharged plutonium isotopic proportions of UOX fuel.

The corresponding ORIGEN model (Figure A.V.2), from which sampling is conducted, is one metric ton of the MOX fuel with approximately 271 kg of zirc2 cladding. The ORIGEN model is set to deplete the fuel using the binary cross section library generated from the fuel specific SAS2H model. Note: cross section perturbations are introduced directly into this binary working library during the sampling procedure. Fuel specific power density is 50 MW/ MTHM for a cycle length of 1000 days giving the same 50GWD/ MTHM burnup as the SAS2H Model. The ORIGEN model is set to "burn" the fuel for 10 equal time steps at 100% power and then print out decay isotopics at charge, discharge, 1, 5, 10, 50, 100, 500, 1000, 2500, 5000, and 10000 year times. ORIGEN also has the availability to print the decay heat, activity, and radiotoxicity for these times.

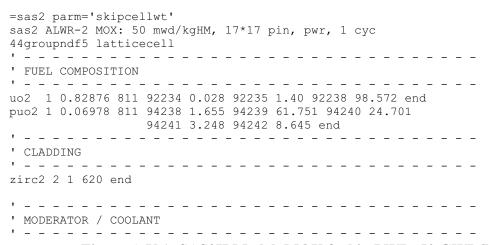


Figure A.V.1: SAS2H Model, MOX fuel in PWR, 50 GWD/MTHM

```
h2o 3 den=0.733 1 570 end
co-59 3 0 1-20 570 end
end comp
' FUEL-PIN GEOMETRY
squarepitch 1.25984 0.83566 1 3 0.94996 2 end
more data szf=1.2 eps=1.0-7 ptc=1.0-8 end
' ASSEMBLY AND CYCLE PARAMETERS
npin/assm=264 fuelngth=365.76 ncycles=1 nlib/cyc=1
printlevel=6
lightel=16 inplevel=1
numins= 1 ortube= 0.61214 srtube=0.5715 facmesh=1.4 end
power=17.3025 burn=1333.3 down=0 end
'power=23.070 burn=1000.0 down=0 end
' Light elements (kg) per assembly
c 0.05999 n 0.03377 o 62.14 al 0.04569
si 0.06586 p 0.1422 ti 0.04983 cr 2.340
mn 0.1096 fe 4.599 co 0.03344 ni 4.402
zr 100.8 nb 0.3275 mo 0.1816 sn 1.652
end
```

Figure A.V.1: SAS2H Model, MOX fuel in PWR, 50 GWD/MTHM, cont.

```
=origens
0$$ a5 28 e
1$$ 1 1t
pwr nuclear data - sample case 1
3$$ a4 -82 a11 0 0 a33 18 e
'Uses SAS Updated Lib
3$$ 33 0 1 -88 a33 -88
54$$ 5 e 2t
35$$ 0 4t
56$$ 10 a13 58 4 3 0 1 1 e
57** a3 1-14 e
95$$ 1 5t
ALWR2 Fuel
mt of heavy metal charged to reactor
'Power specifications, 50 MW for 1000 Days
58** 10r50
60** 8i100 1000
'Nuclide identifies - charged nuclides
66$$ 1 a5 1 a9 1 e
73$$ 60120 130270 140280 140290 220460 220470 220480 220490
220500 240500 240520 240530 240540 250550 260540 260560
260570 260580 270590 280580 280600 280610 280620 280640
400900 400910 400920 400940 400960 410930 420920 420940
420950 420960 420970 420980 421000 501120 501140 501150
```

Figure A.V.2: ORIGEN Model, MOX fuel in PWR, 50 GWD/MTHM

```
501160 501170 501180 501190 501200 501220 501240 922340
922380 922350 942380 942390 942400 942410 942420 932370
952410 952430
'Molar masses of charged nuclides
74** 1.5 4 0.607 0.034 0.304 0.277 2.771 0.204 0.2 5.04 57.423 6.415 1.574 0.327 4.037 61.018 1.439 0.31 0.915 111.86 41.783 1.869 5.645 1.609

    1421.1
    306.73
    462.24
    460.07
    72.5
    10.258
    0.957
    0.532

    0.926
    0.958
    0.546
    1.357
    0.54
    0.321
    0.219
    0.113

4.681 2.47 7.729 2.739 10.392 1.467 1.823 1.13
3801.84 59.56 27.38 184.93 66.11 34.93 24.58 0.00
0.00 0.00
75$$ 47r1 11r2 t
56$$ 0 10 a10 10 a14 5 a17 2 e
57** a3 1-14 e
95$$ 1 5t
'Decay time steps in years
60** 1 5 10 50 100 500 1000 2500 5000 10000
'Decay Output Specifications
65$$ 3z 1 2z 1 2z 1 5z 1 2z
1 5z 1 2z 1 e
'65$$ 3z 1 20z 1 20z 1 e
61** 5r1-14 1+6 1+4
81$$ 2 0 26 1 e
82$$ f2 6t
56$$ f0 t
end
```

Figure A.V.2: ORIGEN Model, MOX fuel in PWR, 50 GWD/MTHM, cont.

VI. The PWR Model Using a MOX Fuel with Impurities

This model has the same geometry and operating conditions as the model described in the preceding section (Figure A.VI.1). The difference is now a portion of the U-238 mass has been removed and replaced with americium, Am, and neptunium, Np, heavy metal impurities in the fuel in an effort to reflect a more realistic MOX fuel. The heavy metals have been added in the proportions 1 w/o Np-237 and 1.5 w/o Am-241 where the w/o is measured against the whole w/o of heavy metal in the fuel such that there is now 414.81 kg U, 37.36 kg Pu, 4.61 kg Np-237, and 6.92 kg Am-241. Due to the similarities in the model

the only major change is the volume fraction of the UO_2 and PuO_2 and those of Am-241 and Np-137 which are now 0.809, 0.070, 0.0096, and 0.004 respectively.

The corresponding ORIGEN model (Figure A.VI.2), from which sampling is conducted, is one metric ton of the MOX fuel with approximately 271 kg of zirc2 cladding. The ORIGEN model is set to deplete the fuel using the binary cross section library generated from the fuel specific SAS2H model. Note: cross section perturbations are introduced directly into this binary working library during the sampling procedure. Fuel specific power density is 50 MW/MTHM for a cycle length of 1000 days giving the same 50GWD/ MTHM burnup as the SAS2H Model. The ORIGEN model is set to "burn" the fuel for 10 equal time steps at 100% power and then print out decay isotopics at charge, discharge, 1, 5, 10, 50, 100, 500, 1000, 2500, 5000, and 10000 year times. ORIGEN also has the availability to print the decay heat, activity, and radiotoxicity for these times.

```
=sas2 parm='skipcellwt'
sas2 ALWR-2 MOX: 50 mwd/kgHM, 17*17 pin, pwr, 1 cyc
44groupndf5 latticecell
' FUEL COMPOSITION
uo2 1 0.82876 811 92234 0.028 92235 1.40 92238 98.572 end
puo2 1 0.06978 811 94238 1.655 94239 61.751 94240 24.701
             94241 3.248 94242 8.645 end
neptunium 1 0.0043 811 93237 100.0 end
americium 1 0.0096 811 95241 100.0 end
' CLADDING
zirc2 2 1 620 end
' MODERATOR / COOLANT
h2o 3 den=0.733 1 570 end
co-59 3 0 1-20 570 end
end comp
' FUEL-PIN GEOMETRY
squarepitch 1.25984 0.83566 1 3 0.94996 2 end
more data szf=1.2 eps=1.0-7 ptc=1.0-8 end
```

Figure A.VI.1: SAS2H Model, MOX with impurities fuel in PWR, 50 GWD/MTHM

```
'ASSEMBLY AND CYCLE PARAMETERS
'npin/assm=264 fuelngth=365.76 ncycles=1 nlib/cyc=1
printlevel=6
lightel=16 inplevel=1
numins= 1 ortube= 0.61214 srtube=0.5715 facmesh=1.4 end
power=17.3025 burn=1333.3 down=0 end
'power=23.070 burn=1000.0 down=0 end
'Light elements (kg) per assembly
'Light elements (kg) per assembly
'
c 0.05999 n 0.03377 o 62.14 al 0.04569
si 0.06586 p 0.1422 ti 0.04983 cr 2.340
mn 0.1096 fe 4.599 co 0.03344 ni 4.402
zr 100.8 nb 0.3275 mo 0.1816 sn 1.652
'end

Figure A VII 1. SAS2H Model MOV vulimovarities fuelin DNVD 50 C
```

Figure A.VI.1: SAS2H Model, MOX w/ impurities fuel in PWR, 50 GWD/MTHM, cont.

```
=origens
0$$ a5 28 e
1$$ 1 1t
pwr nuclear data - sample case 1
3$$ a4 -82 a11 0 0 a33 18 e
'Uses SAS Updated Lib
3$$ 33 0 1 -88 a33 -88
54$$ 5 e 2t
35$$ 0 4t
56$$ 10 a13 58 4 3 0 1 1 e
57** a3 1-14 e
95$$ 1 5t
ALWR2 Fuel
mt of heavy metal charged to reactor
'Power specifications, 50 MW for 1000 Days
58** 10r50
   8i100 1000
'Nuclide identifies - charged nuclides
66$$ 1 a5 1 a9 1 e
73$$ 60120 130270 140280 140290 220460 220470 220480 220490
220500 240500 240520 240530 240540 250550 260540 260560
260570 260580 270590 280580 280600 280610 280620 280640
400900 400910 400920 400940 400960 410930 420920 420940
420950 420960 420970 420980 421000 501120 501140 501150
501160 501170 501180 501190 501200 501220 501240 922340
922380 922350 942380 942390 942400 942410 942420 932370
952410 952430
'Molar masses of charged nuclides
74** 1.5 4 0.607 0.034 0.304 0.277 2.771 0.204
0.2 5.04 57.423 6.415 1.574 0.327 4.037 61.018
1.439 0.31 0.915 111.86 41.783 1.869 5.645 1.609
```

Figure A.VI.2: ORIGEN Model, MOX with impurities fuel in PWR, 50 GWD/MTHM

```
1421.1 306.73 462.24 460.07 72.5 10.258 0.957 0.532
0.926 0.958 0.546 1.357 0.54 0.321 0.219 0.113
4.681 2.47 7.729 2.739 10.392 1.467 1.823 1.13
3696.816 59.56 27.38 184.93 66.11 34.93 24.58 42.186
62.226 0.00
75$$ 47r1 11r2 t
56$$ 0 10 a10 10 a14 5 a17 2 e
57** a3 1-14 e
95$$ 1 5t
       'Decay time steps in years
60** 1 5 10 50 100 500 1000 2500 5000 10000
'Decay Output Specifications
1 5z 1 2z 1 e
'65$$ 3z 1 20z 1 20z 1 e
61** 5r1-14 1+6 1+4
81$$ 2 0 26 1 e
82$$ f2 6t
56$$ f0 t
```

Figure A.VI.2: ORIGEN Model, MOX with impurities in PWR, 50 GWD/MTHM, cont.

VII. The PWR Model Using TRITON

For purposes of conducting a comparison between using the SAS + ORIGEN scheme and using the driver program TRITON stand-alone, a TRITON model of the PWR fuel was created. The TRITON model represents a step up in the detail of modeling to a level closer to that of normal fuel analysis, in which a 2-D transport model is solved for each time step and new fluxes used to collapse a 44-group cross section library. A Wigner cell of the PWR fuel including the buffer region of extra water added by the water holes in the assembly is modeled. Two PWR models were constructed differing in the burnup steps while keeping all parameters are the same as the above model. The first model used a single burnup step the same as the cruder SAS2H model already discussed; the second, utilizing the robustness of TRITON, employed 26 burn-up steps which is closer to a realistic model of the fuel. Since

only one line in the model is different only one copy of it is presented, given here in Figure

A.VII.1

```
=t-depl
Infinite lattice depletion model for a single pincell
44groupndf
'----FUEL COMPOSITION-----
'Fuel
uo2 1 0.90182 811 92234 0.028 92235 4.5 92238 95.472
'Clad
zirc2 4 1 620 end
'Moderator
h2o 5 den=0.733 1 570 end
end comp
'----GEOMETRY-----
read celldata
latticecell squarepitch pitch=1.25984 5 fuelr=0.4178 1 cladr=0.4750 4 end
end celldata
           _____
read depletion
1 4 5
end depletion
'----POWER HISTORY------
read burndata
                              nlib=1 end 'for one burnup step
'power=40.000 burn=1000 down=0
                              nlib=1 end
power=36.53 burn=5.475 down=0
                               nlib=1
power=36.53 burn=21.9 down=0 power=36.53 burn=27.397 down=0
                               nlib=1
                                      end
power=36.53 burn=27.397 down=0
                               nlib=1 end
power=36.53 burn=27.397 down=0
                              nlib=1 end
power=36.53 burn=27.397 down=0
                              nlib=1 end
power=36.53 burn=27.397 down=0 nlib=1 end
power=36.53 burn=27.397 down=0 nlib=1 end
power=36.53 burn=27.397 down=0 nlib=1 end
power=36.53 burn=27.397 down=0 nlib=1 end
power=36.53 burn=27.397 down=0 nlib=1 end
power=36.53 burn=54.75 down=0 nlib=1 end
power=36.53 burn=54.75 down=0 nlib=1 end
power=36.53 \qquad burn=54.75 \qquad down=0 \qquad \qquad nlib=1 \quad end
                             nlib=1 end
power=36.53 burn=54.75 down=0
                              nlib=1
power=36.53 burn=54.75 down=0 power=36.53 burn=54.75 down=0
                                      end
                               nlib=1
                                      end
power=36.53 burn=54.75 down=0 nlib=1 end
power=36.53 burn=54.75 down=0 nlib=1 end
power=36.53 burn=54.75 down=0 nlib=1 end
power=36.53 burn=54.75 down=0 nlib=1 end
power=36.53 burn=54.75 down=0 nlib=1 end
power=36.53 burn=54.75 down=0 nlib=1 end
power=36.53 burn=54.75 down=0 nlib=1 end
power=36.53 burn=54.75 down=0 nlib=1 end
power=36.53 burn=54.75 down=0
                              nlib=1 end
power=36.53 burn=54.75 down=0 nlib=1 end
power=36.53 burn=54.75 down=0 nlib=1 end
                             nlib=1
power=36.53
                     down=0
           burn=54.75
                                      end
power=36.53
           burn=54.75 down=0
                               nlib=1
                                      end
end burndata
```

Figure A. VII.1: TRITON Model, PWR, 4.5 w/o and 48 GWD/MTU

```
read opus
units=gram symnuc= pb-210 ra-226 ac-227 th-227 th-230 pa-231
u-234 u-235 u-236 u-237 u-238 np-237 np-239 pu-238 pu-239 pu-240 pu-241 pu-242
am-241 am-242m am-243 cm-242 cm-244 cm-245
c-14 se-79 sr-90 tc-99 i-129 cs-137 ba-137m y-90 cs-134 end
matl=0 1 end
end opus
read model
17x17 PWR Assembly, 4.5% 40 GWD
read parm
 prtflux=no drawit=yes echo=yes
 xnlib=1 run=yes collapse=yes fillmix=5 prtmxsec=no prtbroad=yes
 sn=4 inners=10 outers=200 epsinner=1e-4 epsouter=1e-4
 epseigen=1e-5 prtmxtab=yes
end parm
·-----
read materials
 1 1 ! 4.5% enriched fuel, pin location 1 ! end
  4 1 ! clad ! end
       ! water ! end
end materials
'----WIGNER CELL DOMAIN SPECIFICATION-----
read geom
cylinder 1 0.66 0.66 0.4178 !fuel - buffer! end
cylinder 4 0.66 0.66 0.4750 !clad - buffer! end
domain 1.32 1.32 4 4
boundary 1 1 1 1
end geom
·-----
end model
'* end of newt transport model
end
=origens
'----ORIGEN DECAY ANALYSIS-----
0$$ a8 26 a11 -71 e 1t
sample case 3b
3$$ 21 0 1 -88 a33 -88
4** a4 1-35 2t
35$$ 0 4t
56$$ a13 -105 5 1 74 4 e
57** a3 1-14 e
95$$ 1 5t
sample case 3b
'Decay time steps in years
60** 1 5 10 50 100 500 1000 2500 5000 10000
'65$$ 1 20z 2q21
65$$ 3z 1 20z 1 20z 1 e
61** f1-14
81$$ 2 0 26 1 e
82$$ a10 2 6t
56$$ 2z a10 10 e 6t
56$$ f0 t
end
```

Figure A. VII.1: TRITON Model, PWR, 4.5 w/o and 40 GWD/MTU, cont.

VIII. TRITON FR Models Using Actinide Fuels

The fast reactor, FR models used in this study was created in reference to various fuel assemblies for Argonne National Lab's Advanced Burner Test Reactor (ABTR) which would use a fuel whose isotopics are based on 10 year decayed UOX that was 3.3 w/o fresh fuel and burned for 33 GWD/MTU. The metal fuel consists of depleted uranium, the transuranics: neptunium, plutonium, americium, and curium metals, and 10-20 w/o zirconium mixed into the metal. Three fuel types are analyzed that are intended to have conversion ratios of 0.25, 0.70 and 1.05 which are controlled by the TRU enrichment. Common among these assemblies are that the pins are in a triangular (hexagonal) pitch with an active fuel length of 80 cm. The composition data including volume fractions are included in Table A.VIII.1. The cladding is a material developed by Argonne and has the composition shown in Table A. VIII.2; it is namely an iron alloy. Finally, as with several other fast reactor design concepts, the coolant for this model is elemental sodium. Operating temperatures are 909, 783 and 783 degrees Fahrenheit for the fuel, clad and moderator, respectively, with the moderator density at 7.97 g/cm³. Specific powers, burnups, geometry and other important data are given in Table A. VIII.3. Note, TRITON automatically returns results in terms of 1 MTHM and the ORIGEN decay sequence is set for the standard charge, discharge, 1, 5, 10, 50, 100, 500, 1000, 2500, 5000, and 10000 year time steps. The TRITON models for CR= 1.05, 0.70, and 0.25 are presented in Figures A. VIII.1-3 respectively. For recycle sampling, consider these same models with input isotopics perturbed.

	Weight Percent in TRU				
Nuclide / Conversion Ratio:	0.25	0.7	1.05		
Np-237	18.635	7.334	9.907		
Pu-238	0.855	1.253	0.000		
Pu-239	32.764	48.058	72.150		
Pu-240	14.983	21.973	4.469		
Pu-241	4.936	7.241	0.250		
Pu-242	2.956	4.335	0.000		
Am-241	20.579	8.100	10.941		
Am-242m	0.041	0.016	0.022		
Am-243	3.565	1.403	1.895		
Cm-244	0.689	0.271	0.366		
Cm-245	0.041	0.016	0.022		
Fissile Fraction, %	37.7	55.30	72.40		
TRU Enrichment, %	59.2	20.6	16.2		
Zr w/o	20	10	10		
U-238, w/o	20.8	69.4	73.8		

Table A.VIII.1: Fast Reactor Fuel Composition Data, by conversion ratio

Cladding Composition					
Material mass/cm ³					
Iron	7.10E-02				
Nickel	4.38E-04 1.06E-02				
Chromium					
Manganese-55	4.68E-04				
Molybdenum	4.99E-04				

Table A. VIII.2: Cladding Composition Data

Conversion Ration	0.25	0.70	1.05
Specific Power of active core, MW/MT	114.8	47.7	41.2
Discharge Burnup, GWD.MT	94.3	78.4	67.7
Height, cm	80	80	80
Number of pins per assembly	217	169	127
Assembly lattice pitch, cm	14.834	14.834	14.834
Inter-assembly gap, mm	4.45	4.0	4.0
Duct thickness, mm	4.45	3.0	3.0
Pin pitch-to-diameter ratio	1.29	1.11	1.10
Cladding thickness, mm	0.75	0.41	0.41

Table A. VIII.3: Operating Conditions and Geometry Data

```
=t-depl
Infinite lattice depletion model for a single pincell, 4 cycles @1 libs/cycle.
'----FUEL COMPOSITION------
read comp
'Fuel
uranium 1 0.4300 909.0
92235 0.192
92238 99.808 end
neptunium 1 0.000377 909.0
93237 100.0 end
plutoniumalp 1 0.0549 909.0
94238 1.038
94239 72.031
94240 23.356
94241 2.249
94242 1.326 end
americium 1 0.0015 909.0
95241 75.758
95243 18.182
95601 6.061 end
curium 1 0.00014 909.0
96244 66.667
96245 33.333
96246 0.000 end
zirconium 1 0.1108 909.0 40090 51.45 40091 11.22 40094 17.38 40096 2.8
                    40092 17.15 end
sodium 2 den=4.8 1 909.0 end
'Moderator
sodium 5 den=6.15 1 783.0 end
'Clad
iron 4 0.8379 783.0 end
nickel 4 0.0048 783.0 end
chromium 4 0.1266 783.0 end
molybdenum 4 0.0078 783.0 end
'manganese 4 0.0427 783.0 end
MN-55 4 0.041 783 end
end comp
'----GEOMETRY-----
read celldata
latticecell triangpitch pitch=1.21 5 fuelr=0.4407 1 cladr=0.5845 4 end
end celldata
·-----
read depletion
-1 4 2 5
end depletion
'----POWER HISTORY-----
read burndata
power=41.2 burn=8.495 down=0 nlib=1 end power=41.2 burn=8.495 down=0 nlib=1 end power=41.2 burn=33.98 down=0 nlib=1 end
power=41.2 burn=33.98 down=0 nlib=1 end
power=41.2 burn=33.98 down=0 nlib=1 end
power=41.2 burn=84.9515 down=0 nlib=1 end
power=41.2 burn=84.9515 down=0 nlib=1 end
```

Figure A. VIII.1: TRITON Model, CR=1.05

```
      power=41.2
      burn=84.9515
      down=0
      nlib=1 end

      power=41.2
      burn=84.9515
      down=0
      nlib=1 end

      power=41.2
      burn=84.9515
      down=0
      nlib=1 end

      power=41.2
      burn=84.9515
      down=0
      nlib=1 end

power=41.2 burn=84.9515 down=0
                                               nlib=1 end
power=41.2 burn=84.9515 down=0
                                               nlib=1 end
power=41.2 burn=84.9515 down=0
                                                nlib=1 end
power=41.2 burn=84.9515 down=0
                                                nlib=1 end
power=41.2 burn=84.9515 down=0
power=41.2 burn=84.9515 down=0
                                               nlib=1 end
                                               nlib=1 end
power=41.2 burn=84.9515 down=0
                                               nlib=1 end
power=41.2 burn=84.9515 down=0
                                               nlib=1 end
power=41.2 burn=84.9515 down=0
                                                nlib=1 end
power=41.2 burn=84.9515 down=0
                                                nlib=1 end
power=41.2 burn=84.9515 down=0
                                               nlib=1 end

      power=41.2
      burn=84.9515
      down=0
      nlib=1 end

      power=41.2
      burn=84.9515
      down=0
      nlib=1 end

      power=41.2
      burn=84.9515
      down=0
      nlib=1 end

      power=41.2
      burn=84.9515
      down=0
      nlib=1 end

end burndata
read opus
units=gram symnuc= pb-210 ra-226 ac-227 th-227 th-230 pa-231
u-234 u-235 u-236 u-237 u-238 np-237 np-239 pu-238 pu-239 pu-240 pu-241 pu-242
am-241 am-242m am-243 cm-242 cm-244 cm-245
c-14 se-79 sr-90 tc-99 i-129 cs-137 ba-137m y-90 cs-134
zr-90 zr-91 zr-92 zr-94 zr-96 end
matl=0 1 2 end
end opus
read model
ABTR Assembly, CR=0.25
  prtflux=no drawit=yes echo=yes
  xnlib=1 run=yes collapse=yes fillmix=5 prtmxsec=no prtbroad=yes
  sn=4 inners=10 outers=200 epsinner=1e-4 epsouter=1e-4
  epseigen=1e-5 prtmxtab=yes
end parm
read materials
  1 1 ! fuel ! end
   2 1 ! bond - sodium ! end
   4 1 ! clad ! end
   5 1 ! sodium ! end
end materials
'-----WIGNER CELL DOMAIN SPECIFICATION------
read geom
cylinder 1 0.605 0.605 0.4407 !fuel! end
cylinder 2 0.605 0.605 0.5090 !gap! end
cylinder 4 0.605 0.605 0.5845 !clad! end
domain 1.21 1.21 3 3
boundary 1 1 1 1
end geom
end model
'* end of newt transport model
```

Figure A. VIII.1: TRITON Model, CR=1.05, cont.

```
end
=origens
'----ORIGEN DECAY ANALYSIS-----
0$$ a8 26 a11 -71 e 1t
sample case 3b
3$$ 21 0 1 -88 a33 -88
4** a4 1-35 2t
35$$ 0 4t
56$$ a13 -109 5 1 74 4 e
57** a3 1-14 e
95$$ 1 5t
sample case 3b
'Decay time steps in years
60** 1 5 10 50 100 500 1000 2500 5000 10000
'65$$ 1 20z 2q21
65$$ 3z 1 20z 1 20z 1 e
61** f1-14
81$$ 2 0 26 1 e
82$$ a10 2 6t
56$$ 2z a10 10 e 6t
56$$ f0 t
end
                Figure A. VIII.1: TRITON Model, CR=1.05, cont.
=t-depl
Infinite lattice depletion model for a single pincell, 4 cycles @1 libs/cycle.
44groupndf
'----FUEL COMPOSITION-----
'Fuel
read comp
uranium 1 0.5682 909
92235 0.192
92238 99.808 end
neptunium 1 0.00225 909
93237 100.0 end
plutoniumalp 1 0.12825 909
94238 2.895
94239 55.307
94240 31.488
94241 4.469
94242 5.841 end
americium 1 0.01085 909
95241 64.348
95243 31.304
95601 4.348 end
curium 1 0.00218 909
96244 73.913
96245 17.391
96246 8.696 end
zirconium 1 0.232 909 40090 51.45 40091 11.22 40094 17.38 40096 2.8
                  40092 17.15 end
```

Figure A. VIII.2: TRITON Model, CR=0.70

sodium 2 den=11.175 1 909 end sodium 5 den=7.652 1 783 end

```
iron 4 0.8378 783 end
nickel 4 0.0048 783 end
chromium 4 0.1266 783 end
molybdenum 4 0.0078 783 end
'manganese 4 0.0427 783 end
MN-55 4 0.041 783 end
end comp
'----GEOMETRY------
read celldata
latticecell triangpitch pitch=1.044 5 fuelr=0.4295 1 cladr=0.4705 4 end
end celldata
read depletion
-1 4 2 5
end depletion
'----POWER HISTORY-----
read burndata

      power=47.4
      burn=8.439
      down=0
      nlib=1 end

      power=47.4
      burn=42.194 down=0
      nlib=1 end

      power=47.4
      burn=42.194 down=0
      nlib=1 end

      power=47.4
      burn=42.194 down=0
      nlib=1 end

      nlib=1 end
      nlib=1 end

power=47.4 burn=42.194 down=0 nlib=1 end
power=47.4 burn=73.84 down=0
                                 nlib=1 end
power=47.4 burn=73.84 down=0
                                 nlib=1 end
power=47.4 burn=73.84 down=0
                                 nlib=1 end
power=47.4 burn=73.84 down=0 nlib=1 end
power=47.4 burn=73.84 down=0 nlib=1 end
power=47.4 burn=73.84 down=0
                                 nlib=1 end
                                 nlib=1 end
power=47.4 burn=73.84 down=0
                                  nlib=1 end
power=47.4 burn=73.84 down=0
power=47.4 burn=73.84 down=0 power=47.4 burn=73.84 down=0
                                  nlib=1 end
                                   nlib=1 end
power=47.4 burn=73.84 down=0
                                   nlib=1 end
power=47.4 burn=73.84 down=0
                                   nlib=1 end
power=47.4 burn=73.84 down=0
                                  nlib=1 end
power=47.4 burn=73.84 down=0
                                  nlib=1 end
power=47.4 burn=73.84 down=0 nlib=1 end
power=47.4 burn=73.84 down=0
                                 nlib=1 end
power=47.4 burn=73.84 down=0 nlib=1 end
power=47.4 burn=73.84 down=0 nlib=1 end
power=47.4 burn=73.84 down=0 nlib=1 end
power=47.4 burn=73.84 down=0 nlib=1 end
end burndata
read opus
units=gram symnuc= pb-210 ra-226 ac-227 th-227 th-230 pa-231
u-234 u-235 u-236 u-237 u-238 np-237 np-239 pu-238 pu-239 pu-240 pu-241 pu-242
am-241 am-242m am-243 cm-242 cm-244 cm-245
c-14 se-79 sr-90 tc-99 i-129 cs-137 ba-137m y-90 cs-134
zr-90 zr-91 zr-92 zr-94 zr-96 end
matl=0 1 end
end opus
'-----
read model
ABTR Assembly, CR=0.7
read parm
  prtflux=no drawit=yes echo=yes
  xnlib=1 run=yes collapse=yes fillmix=5 prtmxsec=no prtbroad=yes
  sn=4 inners=10 outers=200 epsinner=1e-4 epsouter=1e-4
  epseigen=1e-5 prtmxtab=yes
```

Figure A. VIII.2: TRITON Model, CR=0.70, cont.

```
end parm
read materials
 1 1 ! fuel ! end
 2 1 ! bond - sodium ! end
 4 1
       ! clad ! end
       ! sodium ! end
 5 1
end materials
'-----WIGNER CELL DOMAIN SPECIFICATION------
read geom
cylinder 1 0.531 0.531 0.3721 !fuel! end
cylinder 2 0.531 0.531 0.4295 !gap! end
cylinder 4 0.531 0.531 0.5017 !clad! end
domain 1.0623 1.0623 3 3
boundary 1 1 1 1
end geom
end model
'* end of newt transport model
end
=origens
'----ORIGEN DECAY ANALYSIS-----
0$$ a8 26 a11 -71 e 1t
sample case 3b
3$$ 21 0 1 -88 a33 -88
4** a4 1-35 2t
35$$ 0 4t
56$$ a13 -101 5 1 74 4 e
57** a3 1-14 e
95$$ 1 5t
sample case 3b
'Decay time steps in years
60** 1 5 10 50 100 500 1000 2500 5000 10000
'65$$ 1 20z 2q21
65$$ 3z 1 20z 1 20z 1 e
61** f1-14
81$$ 2 0 26 1 e
82$$ a10 2 6t
56$$ 2z a10 10 e 6t
56$$ f0 t
end
               Figure A. VIII.2: TRITON Model, CR=0.70, cont.
=t-depl
Infinite lattice depletion model for a single pincell, 4 cycles @1 libs/cycle.
44groupndf
'----FUEL COMPOSITION------
read comp
'Fuel
uranium 1 0.2368 909.0
92235 0.196
92238 99.804 end
neptunium 1 0.00785 909.0
93237 100.0 end
```

Figure A. VIII.3: TRITON Model, CR=0.25

```
plutoniumalp 1 0.2837 909.0
94238 5.304
94239 35.902
94240 40.254
94241 6.936
94242 11.605 end
americium 1 0.0435 909.0
95241 60.470
95243 35.470
95601 4.060 end
curium 1 0.01175 909.0
96242 0.806
96244 70.968
96245 18.548
96246 9.678 end
zirconium 1 0.3626 909.0 40090 51.45 40091 11.22 40094 17.38 40096 2.8
                     40092 17.15 end
sodium 2 den=7.00 1 909.0 end
'Moderator
sodium 5 den=17.73 1 783.0 end
'Clad
iron 4 0.8379 783.0 end
nickel 4 0.0048 783.0 end
chromium 4 0.1266 783.0 end
molybdenum 4 0.0078 783.0 end
'manganese 4 0.0427 783.0 end
MN-55 4 0.041 783 end
end comp
'----GEOMETRY------
read celldata
latticecell triangpitch pitch=0.7793 5 fuelr=0.2057 1 cladr=0.3613 4
end
end celldata
read depletion
-1 4 2 5
end depletion
'----POWER HISTORY------
read burndata
power=114.8 burn=2.613 down=0 nlib=1 end
power=114.8 burn=17.42 down=0 nlib=1 end
power=114.8 burn=17.42 down=0
                                     nlib=1 end
power=114.8 burn=19.60 down=0
                                     nlib=1 end
power=114.8 burn=19.60 down=0
                                     nlib=1 end
power=114.8 burn=39.20 down=0
                                     nlib=1 end
power=114.8 burn=39.20 down=0
                                     nlib=1 end
power=114.8 burn=39.20 down=0 nlib=1 end
power=114.8 burn=39.20 down=0 nlib=1 end
power=114.8 burn=39.20 down=0 nlib=1 end
power=114.8 burn=39.20 down=0 nlib=1 end power=114.8 burn=39.20 down=0 nlib=1 end power=114.8 burn=39.20 down=0 nlib=1 end power=114.8 burn=39.20 down=0 nlib=1 end power=114.8 burn=39.20 down=0 nlib=1 end power=114.8 burn=39.20 down=0 nlib=1 end power=114.8 burn=39.20 down=0 nlib=1 end power=114.8 burn=39.20 down=0 nlib=1 end
power=114.8 burn=39.20 down=0 nlib=1 end
power=114.8 burn=39.20 down=0
                                     nlib=1 end
power=114.8 burn=39.20 down=0 nlib=1 end
power=114.8 burn=39.20 down=0 nlib=1 end
```

Figure A. VIII.3: TRITON Model, CR=0.25, cont.

```
power=114.8 burn=39.20 down=0 nlib=1 end
power=114.8 burn=39.20 down=0 nlib=1 end
power=114.8 burn=39.20 down=0 nlib=1 end
power=114.8 burn=39.20 down=0
                                nlib=1 end
end burndata
read opus
units=gram symnuc= pb-210 ra-226 ac-227 th-227 th-230 pa-231
u-234 u-235 u-236 u-237 u-238 np-237 np-239 pu-238 pu-239 pu-240 pu-241 pu-242
am-241 am-242m am-243 cm-242 cm-244 cm-245
c-14 se-79 sr-90 tc-99 i-129 cs-137 ba-137m y-90 cs-134
zr-90 zr-91 zr-92 zr-94 zr-96 end
matl=0 1 2 end
end opus
'----
read model
ABTR Assembly, CR=0.25
read parm
 prtflux=no drawit=ves echo=ves
 xnlib=1 run=yes collapse=yes fillmix=5 prtmxsec=no prtbroad=yes
 sn=4 inners=10 outers=200 epsinner=1e-4 epsouter=1e-4
 epseigen=1e-5 prtmxtab=yes
end parm
read materials
  1 1 ! fuel ! end
  2 1 ! bond - sodium ! end
  4 1 ! clad ! end
  5 1 ! sodium ! end
end materials
'-----WIGNER CELL DOMAIN SPECIFICATION------
read geom
cylinder 1 0.363 0.363 0.2057 !fuel! end
cylinder 2 0.363 0.363 0.2375 !gap! end
cylinder 4 0.363 0.363 0.3613 !clad! end
domain 0.7253 0.7253 3 3
boundary 1 1 1 1
end geom
end model
'* end of newt transport model
end
=origens
'----ORIGEN DECAY ANALYSIS-----
sample case 3b
3$$ 21 0 1 -88 a33 -88
4** a4 1-35 2t
35$$ 0 4t
56$$ a13 -97 5 1 74 4 e
57** a3 1-14 e
95$$ 1 5t
sample case 3b
'Decay time steps in years
60** 1 5 10 50 100 500 1000 2500 5000 10000
'65$$ 1 20z 2q21
65$$ 3z 1 20z 1 20z 1 e
61** f1-14
81$$ 2 0 26 1 e
```

Figure A. VIII.3: TRITON Model, CR=0.25, cont.

```
82$$ a10 2 6t
56$$ 2z a10 10 e 6t
56$$ f0 t
End
```

Figure A. VIII.3: TRITON Model, CR=0.25, cont.

Since the TRITON model used for the recycling experiment is the same as the CR = 0.70 fast reactor model, it is not repeated here. The burnup is adjusted to 41.4 GWD/MTHM, which is the end of cycle core average burnup, and the input isotopics for each element are different at each recycle step.

IX.REBUS Fast Reactor Model with Actinide Fuel and Recycle

The REBUS model was set up to reproduce the fuel loading and recycle as specified by Argonne's ABTR Preconceptual Design [26] report for the medium conversion ratio core. The model is set to recycle all of the fast reactor fuel transuranics after 1.5 years of cooling, and make up the mass and reactivity by using spent LWR fuel and depleted uranium. As described in the Results section, this model more closely matched the values in the report in terms of loading and operating parameters. The complete input is given in Figure A.IX.1.

```
BLOCK=OLD
DATASET=ISOTXS
BLOCK=STP027
DATASET=A.SUMMAR
                                                              1
01
02
           Υ
                        8
03
           0
04
           4LFP35
04
           4LFP38
04
           4LFP39
           4LFP40
04
           4LFP41
           3P236I P236M P2360
0.4
04
           3C242I C242M C242O
04
           3C243I C243M C243O
04
           3C244I C244M C244O
04
           3C245I C245M C245O
           3C246I C246M C2460
           3A24MI A24MM A24MO
04
           3A242I A242M A2420
04
05
      PU2365SPEC
                    236.045761 P236I P236M P2360
```

Figure A.IX.1: REBUS Equilibrium Model, CR = 0.77.

```
242.058426 C242I C242M C242O
05
    CM2425SPEC
              243.061035 C243I C243M C2430
05
    CM2435SPEC
              244.062637 C244I C244M C244O
05
    CM2445SPEC
0.5
    CM2455SPEC 245.065247 C245I C245M C2450
0.5
    CM2465SPEC 246.066849 C246I C246M C2460
0.5
    AM242MSPEC 242.059433 A24MI A24MM A24MO
    AM2425SPEC
              242.059433 A242I A242M A2420
DATASET=A.STP027
                        0
                             0
                                  0
                                      0
                                             0
01
        0
02
         0
              0
                   0
                        1
                                       1
0.3
         1
             1
                   1
                        1
                             1
                                        1
                                             1
06
         1
01 **********************************
01 A.DIF3D: 250MWt, 12-Month
01 ***********************************
01
02
     999000999000
03
         0 0
                  0
                       0
                           50
             0 0 00
                            000 10
                                                 0
04
         0
                                      100
                                            0
05
            .000001 .0001
                           .0001
   05
                1.0E-07
                           1.0E-05
                                      1.0E-05
                                          83.33E+6
06
DATASET=A.HMG4C
01 TURN OFF HMG4C EDITS
    60000
           1 0
                       0
                            0 1
DATASET=A.NIP3
01 ***********************************
    A.NIP3: 250MWt, 12-Month
   ****************
01
01
02
              0 1 90000 90000
                                                       1
                                                            1
03
            126
04
              7
                  4
                       0
07
     TCORE ICO D ICO E ICO F ICO G ICO H
     TCORE MCO D MCO E MCO F MCO G MCO H
07
     TCORE OCO D OCO E OCO F OCO G OCO H
07
     ICORE ICO D ICO E ICO F ICO G ICO H
07
     MCORE MCO D MCO E MCO F MCO G MCO H
07
     OCORE OCO D OCO E OCO F OCO G OCO H
07
09
                             2 93.66
              3 50.24
                                            1 110.54
         Ζ
              1 127.42
                             1 144.30
                                            1 161.18
09
         Ζ
         Ζ
              1 178.07
                             1 194.95
                                            1 217.52
09
         Z
              4 280.36
                             1 300.46
                                            1 315.54
              2 345.68
09
         Ζ
  * U-20TRU-10Zr, density = 15.73 g/cc, ANL-IFR-29
         FUELI U234I
                       1.0 U235I 1.0
13
                                             U236I
          FUELI U238I
                                       1.0
13
                        1.0
                              P236I
                                            P238I
                                                       1.0
                             P240I
13
          FUELI P239I
                        1.0
                                       1.0
                                             P241I
          FUELI P242I
                            N237I
                                           A241I
13
                        1.0
                                       1.0
                                                       1.0
13
                        1.0
                                       1.0 C242I
          FUELI A24MI
                              A243I
                                                       1.0
13
          FUELI C243I
                       1.0
                             C244I
                                       1.0
```

Figure A. IX.1: REBUS Equilibrium Model, CR = 0.77, cont.

```
      FUELI C246I
      1.0
      DUMP1
      1.0
      DUMP2
      1.0

      FUELI LFP35
      1.0
      LFP38
      1.0
      LFP39
      1.0

      FUELI LFP40
      1.0
      LFP41
      1.0
      ZIRCI 1.03839E-02

13
13
13
13
            FUELO U2340
                            1.0 U2350 1.0 U2360
                                                                  1.0
13
            FUELO U2380
                             1.0 P2360
                                              1.0 P2380
                                                                  1.0
            FUELO P2390
                             1.0 P2400
13
                                              1.0 P2410
                                                                  1.0
                            A2430 1.0 C2420
1.0 C2440 1.0 C2450
1.0 DUMP1 1.0 DUMP2
1.0 LFP38 1.0 LFP39
1.0 LFP41 1.0 7770
                                               1.0 A2410
13
            FUELO P2420
                             1.0 N2370
                                                                  1.0
13
            FUELO A24MO
                                                                  1.0
13
            FUELO C2430
                                                                  1.0
           FUELO C2460
FUELO LFP35
13
                                                                  1.0
13
                                                                  1.0
13
            FUELO LFP40
                                               1.0 ZIRCO 1.03839E-02
           FUELM U234M
                            1.0 U235M
1.0 P236M
13
                                              1.0 U236M
                                                                  1.0
                                               1.0 P238M
13
            FUELM U238M
                                                                  1.0
                            1.0 P240M
1.0 N237M
13
            FUELM P239M
                                               1.0 P241M
                                                                  1.0
13
           FUELM P242M
                                               1.0 A241M
                                                                  1.0
13
                            1.0 A243M
                                              1.0 C242M
                                                                  1.0
           FUELM A24MM
13
            FUELM C243M
                            1.0 C244M
                                              1.0 C245M
                                                                  1.0
            FUELM C246M
13
                            1.0 DUMP1
                                              1.0 DUMP2
                                                                  1.0
            FUELM LFP35
                             1.0 LFP38
                                              1.0 LFP39
13
                                                                  1.0
                             1.0 LFP41
                                               1.0 ZIRCM 1.03839E-02
            FUELM LFP40
13
* Na coolant, density from Fink and Leibowitz (rho=0.850257 at 432.5 C)
            CLNTI NA23I 2.22724E-02
13
            CLNTO NA230 2.22724E-02
13
            CLNTM NA23M 2.22724E-02
13
13
            CLNTR NA23R 2.22724E-02
13
            CLNTS NA23S 2.22724E-02
  * HT9, density = 7.76 \text{ g/cc}, ASTM A826-88
            HT9 I FE I 7.10244E-02NI I 4.37911E-04CR I 1.05604E-02
13
13
            HT9 I MN55I 4.67845E-04MO I 4.99271E-04
13
            HT9 O FE O 7.10244E-02NI O 4.37911E-04CR O 1.05604E-02
            HT9 O MN550 4.67845E-04MO O 4.99271E-04
13
            HT9 M FE M 7.10244E-02NI M 4.37911E-04CR M 1.05604E-02
13
            HT9 M MN55M 4.67845E-04MO M 4.99271E-04
13
13
            HT9 R FE R 7.10244E-02NI R 4.37911E-04CR R 1.05604E-02
           HT9 R MN55R 4.67845E-04MO R 4.99271E-04
13
13
           HT9 S FE S 7.10244E-02NI S 4.37911E-04CR S 1.05604E-02
            HT9 S MN55S 4.67845E-04MO S 4.99271E-04
13
 * SS-316, density = 7.97 g/cc, /data/RA/PADB/SAMATL
            S316R FE R 5.29276E-02NI R 1.08679E-02CR R 1.07851E-02
13
13
            S316R MN55R 1.69487E-03MO R 1.45080E-03
  * B4C (natural B), density = 2.268 g/cc, 90% TD, www.azom.com
            B4CPI B-10I 0.0196760 B-11I 0.0791983 C-12I 0.0193455
  * B4C shield (Radial, 90% TD)
            B4CR B-10S 0.0196760 B-11S 0.0791983 C-12S 0.0193455
 * 5% axial swelling, 0.596% radial expansion, 0.489% axial expansion
            ICSC FUELI 0.385000 HT9 I 0.187000 CLNTI 0.300000
```

Figure A. IX.1: REBUS Equilibrium Model, CR = 0.77, cont.

```
14
                         0.385000 HT9 O 0.187000 CLNTO
           OCSC FUELO
                                                           0.300000
                        0.385000 HT9 M 0.187000 CLNTM
14
           MCSC
                 FUELM
                                                           0.300000
14
           LPSC
                S316R
                         0.3
                                          0.7
                                  CLNTR
14
           LRSC HT9 R
                        0.667897 CLNTR
                                        0.320813
14
           UPSC1 HT9 R
                         0.227980 CLNTR
                                          0.768166
14
           UPSC HT9 R
                         0.227980
                                  CLNTR
                                          0.320813
14
           USSC
                HT9 R
                         0.667897
                                          0.320813
                                  CLNTR
* pellet volume fraction for B4C with thermal expansion
14
           CRBSC HT9 I 0.263966 CLNTI 0.366908 B4CPI
                                                          0.308300
14
           CRCSC HT9 I
                        0.076960 CLNTI
                                        0.921739
14
           CRFSC HT9 I
                       0.247787 CLNTI
                                        0.748024
14
           CRPSC HT9 R
                         0.263966 CLNTR
                                          0.366908
14
           CRDSC HT9 R
                         0.247787 CLNTR
                                          0.748024
 * reflector and shield
                        0.828951 CLNTR
                                          0.157036
14
           REFSC HT9 R
14
                        0.299011 CLNTS 0.173203 B4CR 0.421138
           RS2SC HT9 S
14
           BRSC S316R
                         0.062
                                CLNTS
                                          0.938
* primary compositions
           ICPC ICSC
                         1.0
14
           OCPC
                OCSC
                         1.0
14
           MCPC
                MCSC
                         1.0
           LPPC LPSC
                         1.0
14
14
           LRPC LRSC
                         1.0
           UPPC1 UPSC1
14
                         1.0
           UPPC UPSC
14
                         1.0
           USPC USSC
14
                         1.0
14
           BRPC BRSC
                         1.0
14
           CRBPC CRBSC
                         1.0
14
                         1.0
           CRCPC CRCSC
14
           CRFPC CRFSC
                         1.0
           CRPPC CRPSC
14
                         1.0
           CRDPC CRDSC
14
                         1.0
           REFPC REFSC
14
                         1.0
14
           RS2PC RS2SC
                         1.0
15
    LPPC CR A
15
     LRPC CR B
15
     CRCPC CR C
     CRFPC CR G
15
     CRBPC CR H
15
     CRPPC CR__I
15
     CRDPC CR__J
15
15
     USPC
           CR K
15
           ICO A
     LPPC
     LPPC
           OCO A
15
15
     LPPC MCO A
15
    LRPC ICO B
15
     LRPC
           OCO B
15
     LRPC MCO B
```

Figure A. IX.1: REBUS Equilibrium Model, CR = 0.77, cont.

```
15
      ICPC ICO D ICO E ICO F ICO G ICO H
15
      OCPC OCO D OCO E OCO F OCO G OCO H
15
      MCPC MCO D MCO E MCO F MCO G MCO H
15
     UPPC1 ICO I
15
      UPPC1 OCO I
15
      UPPC1 MCO I
15
      UPPC ICO J
15
      UPPC OCO J
15
      UPPC MCO J
15
      USPC ICO K
15
      USPC OCO K
15
      USPC MCO K
15
     REFPC REFLT
15
     RS2PC SHILD
15
      BRPC BARRL
29
                 14.6850
        SECTION
                       DESCRIPTION
                       LOWER STRUCTURE/POOL
                       LOWER REFLECTOR
        D, E, F, G, H
                       ACTIVE CORE
                       FISSION-GAS PLENUM
        I,J
        K
                       UPPER STRUCTURE
                       RADIAL REFLECTOR
        REFLT
                       RADIAL SHIELD
        SHILD
        BARRL
                       CORE BARREL/POOL
    * Ring 1 = Control Rod
30
      CR A
                       0
                             0
                                  0.0
                                               50.24
                 1
      CR B
                             0
                                               93.66
30
                 1
                       0
                                  50.24
        __C
30
      CR
                 1
                       0
                             0
                                  93.66
                                               178.07
30
      CR G
                1
                       0
                             0 178.07
                                               194.95
30
                             0 194.95
      CR H
                1
                       0
                                               280.36
      CR I
                             0 280.36
                 1
30
                       0
                                               300.46
                             0
30
      CR J
                 1
                       0
                                 300.46
                                              315.54
      CR K
                 1
                       0
                             0
30
                                 315.54
                                               345.68
    * Ring 2 = Inner core
30
      ICO A
                 2
                       0
                             0
                                  0.0
                                               50.24
30
      ICO B
                             0
                                 50.24
                                               110.54
      ICO D
                 2
30
                       0
                             0
                                110.54
                                               127.42
30
      ICO E
                 2
                                 127.42
                       0
                             0
                                               144.30
                 2
30
      ICO F
                       0
                             0
                                 144.30
                                               161.18
                 2
30
      ICO G
                       0
                             0
                                 161.18
                                               178.07
                 2
                             0
30
      ICO H
                       0
                                 178.07
                                               194.95
                 2
                             0
                                 194.95
                                               217.52
30
      ICO I
                       0
30
      ICO J
                             0
                                 217.52
                                               315.54
```

Figure A.IX.1: REBUS Equilibrium Model, CR = 0. 77, cont.

30	ICO_K	2	0	0	315.54	345.68
	* Ring 3 =	Inner	/ CR/	Test		
30 30 30 30 30 30 30 30 30 30	ICO_A ICO_B ICO_D ICO_E ICO_F ICO_G ICO_H ICO_I ICO_J ICO_J ICO_K	3 3 3 3 3 3 3 3 3	0 0 0 0 0 0 0	0 0 0 0 0 0 0	0.0 50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54	50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54 345.68
30 30 30 30 30 30 30 30	CR_A CR_B CR_C CR_G CR_H CR_I CR_J CR_K	3 3 3 3 3 3 3	2 2 2 2 2 2 2 2 2	2 2 2 2 2 2 2 2	0.0 50.24 93.66 178.07 194.95 280.36 300.46 315.54	50.24 93.66 178.07 194.95 280.36 300.46 315.54 345.68
30 30 30 30 30 30 30 30 30 30	MCO_A MCO_B MCO_D MCO_E MCO_F MCO_G MCO_H MCO_I MCO_I MCO_I	3 3 3 3 3 3 3 3 3	4 4 4 4 4 4 4 4	4 4 4 4 4 4 4 4 4	0.0 50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54	50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54 345.68
30 30 30 30 30 30 30 30 30	MCO_A MCO_B MCO_D MCO_E MCO_F MCO_G MCO_H MCO_I MCO_J MCO_J MCO_K	3 3 3 3 3 3 3 3 3	12 12 12 12 12 12 12 12 12 12 12	12 12 12 12 12 12 12 12 12 12	0.0 50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54	50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54 345.68
	* Ring 4 =	Inner	/ Test	(ref	lector,so	far)
30 30 30 30 30 30 30	ICO_A ICO_B ICO_D ICO_E ICO_F ICO_G ICO_H	4 4 4 4 4 4	0 0 0 0 0 0	0 0 0 0 0	0.0 50.24 110.54 127.42 144.30 161.18 178.07	50.24 110.54 127.42 144.30 161.18 178.07 194.95

Figure A.IX.1: REBUS Equilibrium Model, CR = 0. 77, cont.

30 30 30	ICO_I ICO_K ICO_K	4 4 4	0 0 0	0 0 0	194.95 217.52 315.54	217.52 315.54 345.68
30 30 30 30 30 30 30 30 30 30	MCO_A MCO_B MCO_D MCO_E MCO_F MCO_G MCO_H MCO_I MCO_I	4 4 4 4 4 4 4 4 4	1 1 1 1 1 1 1 1	1 1 1 1 1 1 1 1 1	0.0 50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54	50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54 345.68
30	REFLT	4	4	4	0.0	345.68
	* Ring 5 =	Outer	/ CR			
30 30 30 30 30 30 30 30 30 30	OCO_A OCO_B OCO_D OCO_E OCO_F OCO_G OCO_H OCO_I OCO_J	5 5 5 5 5 5 5 5 5	0 0 0 0 0 0 0	0 0 0 0 0 0 0	0.0 50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54	50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54 345.68
30 30 30 30 30 30 30 30	CR_A CR_B CR_C CR_G CR_H CR_I CR_J CR_K	5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	3 3 3 3 3 3 3	3 3 3 3 3 3 3	0.0 50.24 93.66 178.07 194.95 280.36 300.46 315.54	50.24 93.66 178.07 194.95 280.36 300.46 315.54 345.68
30 30 30 30 30 30 30 30	CR_A CR_B CR_C CR_G CR_H CR_I CR_J CR_K	5 5 5 5 5 5 5 5 5 5 5 5 5	7 7 7 7 7 7 7	7 7 7 7 7 7 7	0.0 50.24 93.66 178.07 194.95 280.36 300.46 315.54	50.24 93.66 178.07 194.95 280.36 300.46 315.54 345.68
30 30 30 30 30 30 30	CR_A CR_B CR_C CR_G CR_H CR_I CR_J	5 5 5 5 5 5 5	23 23 23 23 23 23 23 23	23 23 23 23 23 23 23 23	0.0 50.24 93.66 178.07 194.95 280.36 300.46	50.24 93.66 178.07 194.95 280.36 300.46 315.54

Figure A.IX.1: REBUS Equilibrium Model, CR = 0. 77, cont.

30	CRK	5	23	23	315.54	345.68
	* Ring 6 =	Outer	core	and bl	anket	
30	REFLT	6	0	0	0.0	345.68
30 30 30 30 30 30 30 30	OCO_A OCO_B OCO_D OCO_E OCO_F OCO_G OCO_H OCO_I	6 6 6 6 6 6 6	3 3 3 3 3 3 3	4 4 4 4 4 4 4	0.0 50.24 110.54 127.42 144.30 161.18 178.07 194.95	50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52
30 30	0C0_1 0C0_K	6 6	3	4 4	217.52 315.54	315.54 345.68
30 30 30 30 30 30 30 30 30	OCO_A OCO_B OCO_D OCO_E OCO_F OCO_G OCO_H OCO_I OCO_J OCO_K	6 6 6 6 6 6 6	8 8 8 8 8 8 8 8	8 8 8 8 8 8 8	0.0 50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54	50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54 345.68
30 30 30 30 30 30 30 30 30	OCO_A OCO_B OCO_D OCO_E OCO_F OCO_G OCO_H OCO_I OCO_J OCO_K	6 6 6 6 6 6 6 6	29 29 29 29 29 29 29 29 29	29 29 29 29 29 29 29 29 29	0.0 50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54	50.24 110.54 127.42 144.30 161.18 178.07 194.95 217.52 315.54 345.68
30	* Ring 7 = REFLT	reflec	tor 0	0	0.0	345.68
- 0	* Ring 8 =					2 10 . 00
30	SHILD	8	0	0	0.0	345.68
30 30 30	REFLT REFLT REFLT	8 8 8	3 10 40		0.0 0.0 0.0	345.68 345.68 345.68
	* Ring 9 =	Shiel	d and	d BARRE	L	
30 30	SHILD SHILD	9	3	7 13	0.0	345.68 345.68

Figure A.IX.1: REBUS Equilibrium Model, CR = 0.77, cont.

```
30
               9 45
                        47
                               0.0
     SHILD
                                           345.68
                     2
                           2
30
     BARRL
                9
                                 0.0
                                           345.68
                                 0.0
30
     BARRL
                9
                     8
                           8
                                           345.68
                9
30
                    10
                          10
                                 0.0
                                           345.68
     BARRL
30
     BARRL
                9
                     48
                          48
                                 0.0
                                           345.68
DATASET=A.BURN
   ***************
01
                   FFTF, 250MWt, 12-Month
01
01
   *****************
01
02
           999000
                       0.001
                                   0.001
                                              0.0001
                                                                   1
03
          0
                0.0
                               0.0
                                        121.7
                                                     1.00
                                                                   0
                                                             1
               1.0000
                       0.001
                                              0.170
                                                     0.210
04
                                   1.0
06
           CPL1 0.5
09
     U-234
                1U-235
09
     U-234
                2LFP35
09
     U-234
                5DUMP1
09
     U-234
                8DUMP1
25
     U-234
                          8.978-14
                8DUMP1
09
     U-235
                1U-236
09
     U-235
                2LFP35
09
     U-235
                5U-234
09
     U-235
                8DUMP1
25
     U-235
                8DUMP1
                          3.120-17
09
     U-236
                1NP237
09
     U-236
                2LFP35
09
     U-236
                5U-235
09
     U-236
                8DUMP1
25
     U-236
                          9.379-16
                8DUMP1
09
     U-238
                1PU239
09
     U-238
                2LFP38
09
     U-238
                5NP237
09
     U-238
                8DUMP1
25
     U-238
                8DUMP1
                          4.915-18
09
     NP237
                1PU238
09
     NP237
                2LFP38
09
                             0.346U-236
                                             0.374DUMP1
                                                                0.28
     NP237
                5PU236
09
     NP237
                8DUMP1
25
     NP237
                8DUMP1
                          1.026-14
09
     PU236
                1NP237
09
     PU236
                2LFP35
09
     PU236
                5DUMP1
09
     PU236
                8DUMP1
25
     PU236
                8DUMP1
                          7.703-09
09
     PU238
                1PU239
09
     PU238
                2LFP38
09
     PU238
                5NP237
                8U-234
09
     PU238
25
     PU238
                8U-234
                          2.503-10
09
     PU239
                1PU240
09
     PU239
                2LFP39
09
     PU239
```

Figure A.IX.1: REBUS Equilibrium Model, CR = 0. 77, cont.

09 25 09	PU239 PU239 PU240 PU240	8U-235 8U-235 1PU241 2LFP40	9.109-13		
09 09 25 09	PU240 PU240 PU240 PU241 PU241	5PU239 8U-236 8U-236 1PU242 2LFP41	3.353-12		
09 09 25 09	PU241 PU241 PU241 PU242	5PU240 6AM241 6AM241 1AM243	1.494-09		
09 09 09 25	PU242 PU242 PU242 PU242	2LFP41 5PU241 8U-238 8U-238	5.833-14		
09 09 09	AM241 AM241 AM241	1CM242 2LFP41 5PU240	0.66AM242	0.20PU242	0.14
09 25 09 09	AM241 AM241 AM242 AM242	8NP237 8NP237 1AM243 2LFP41	5.081-11		
09 09 25 09	AM242 AM242 AM242 AM242	5AM241 6CM242 6CM242 7PU242	1.189-10		
25 09 25	AM242 AM242 AM242	7PU242 8PU238 8PU238	2.487-11 7.225-13		
09 09 09	AM243 AM243 AM243 AM243	1CM244 2LFP41 5AM242 8PU239	0.500PU242	0.086CM242	0.414
25 09 09 09	AM243 CM242 CM242 CM242	8PU239 1CM243 2LFP41 5AM241	2.976-12 0.99NP237	0.01	
09 25 09 09	CM242 CM242 CM243 CM243	8PU238 8PU238 1CM244 2LFP41	4.924-08		
09 09 25	CM243 CM243 CM243	5CM242 7AM243 7AM243	2.003-12		
09 25 09 09	CM243 CM243 CM244 CM244	8PU239 8PU239 1CM245 2LFP41	7.685-10		
09 09 25 09	CM244 CM244 CM244 CM245	5CM243 8PU240 8PU240 1CM246	1.213-09		
09	CM245	2LFP41		11 CD 0 55	

Figure A.IX.1: REBUS Equilibrium Model, CR = 0.77, cont.

```
09
      CM245
                  5CM244
09
      CM245
                  8PU241
25
      CM245
                  8PU241
                              2.592-12
09
      CM246
                  1DUMP2
09
      CM246
                  2LFP41
09
      CM246
                  5CM245
09
      CM246
                  8PU242
25
                  8PU242
                              4.642-12
      CM246
09
      LFP35
                  0
09
      LFP38
09
      LFP39
                  0
09
      LFP40
                  0
09
      LFP41
09
      DUMP1
09
      DUMP2
                  0
10
      U-234 U234I U234M U2340
10
      U-235 U235I U235M U2350
10
      U-236 U236I U236M U2360
10
      U-238 U238I U238M U238O
10
      NP237 N237I N237M N2370
      PU236 P236I P236M P2360
10
      PU238 P238I P238M P2380
10
      PU239 P239I P239M P2390
10
10
      PU240 P240I P240M P2400
10
      PU241 P241I P241M P2410
      PU242 P242I P242M P242O
10
10
      AM241 A241I A241M A2410
      AM242 A24MI A24MM A24MO
10
10
      AM243 A243I A243M A2430
10
      CM242 C242I C242M C242O
10
      CM243 C243I C243M C243O
10
      CM244 C244I C244M C244O
10
      CM245 C245I C245M C245O
10
      CM246 C246I C246M C246O
11
                  0
      CPL1
                        1ICSC
                                ICPC
                                            2ICSC
                                                   ICPC
      CPL1
                         3ICSC
                                ICPC
                                            4ICSC
                                                   ICPC
11
                  0
11
      CPL1
                  0
                         5ICSC
                                ICPC
                                            6ICSC
                                                   ICPC
11
      CPL1
                  0
                        7ICSC
                               ICPC
                                            8ICSC
                                                   ICPC
11
      CPL1
                  0
                        9ICSC ICPC
                                           10ICSC
                                                   ICPC
11
      CPL1
                  0
                       11ICSC
                                ICPC
                                           12ICSC
                                                   ICPC
11
      CPL1
                  0
                       13DISI
                                                   OCPC
11
      CPL2
                  0
                        10CSC
                                OCPC
                                            20CSC
11
      CPL2
                  0
                        30CSC
                                OCPC
                                            40CSC
                                                   OCPC
11
      CPL2
                  0
                        50CSC
                                OCPC
                                            60CSC
                                                   OCPC
11
      CPL2
                  0
                        70CSC
                                OCPC
                                            80CSC
                                                   OCPC
11
      CPL2
                  0
                        90CSC
                                OCPC
                                           100CSC
                                                   OCPC
                       110CSC
11
      CPL2
                  0
                                OCPC
                                           120CSC
                                                   OCPC
                       130CSC
11
      CPL2
                  0
                                OCPC
                                           140CSC
                                                   OCPC
      CPL2
                  0
                       150CSC
11
                                OCPC
11
      CPL2
                  0
                       16DISO
11
                  0
                                MCPC
      CPL3
                        1MCSC
                                            2MCSC MCPC
11
      CPL3
                         3MCSC
                               MCPC
                                            4MCSC MCPC
```

Figure A.IX.1: REBUS Equilibrium Model, CR = 0. 77, cont.

136

```
0
                    5MCSC MCPC
11
    CPL3
                                    6MCSC MCPC
             0
                    7MCSC MCPC
                                    8MCSC MCPC
11
     CPL3
              0
11
     CPL3
                    9MCSC MCPC
                                    10MCSC MCPC
              0
11
   CPL3
                   11MCSC MCPC
                                    12MCSC MCPC
11
    CPL3
              0 13DISM
12
    CPL1
                ICLOAD
                           0.0
                                           0
                                                   1.00
12
     CPL2
                OCLOAD
                           0.0
                                           0
                                                   1.25
12
                           0.0
                                                   1.13
     CPL3
                MCLOAD
                                           \cap
13
    ICLOAD
                U-234 3.64276E-02U-235 3.62721E-02U-236 3.61182E-02
13
    ICLOAD
                U-238 3.58140E-02
13
    ICLOAD
                NP237 3.59654E-02
13
    ICLOAD
                PU236 3.61181E-02PU238 3.58141E-02PU239 3.56639E-02
13
    ICLOAD
                PU240 3.55151E-02PU241 3.53674E-02PU242 3.52210E-02
                AM241 3.53673E-02AM242 3.52209E-02AM243 3.50757E-02
13
     ICLOAD
               CM242 3.52209E-02CM243 3.50757E-02CM244
13
    ICLOAD
                                                       3.49318E-02
13
   ICLOAD
               CM245 3.47888E-02CM246 3.46472E-02
13 ICLOAD
               LFP35 3.65475E-02LFP38 3.61589E-02LFP39
                                                       3.59882E-02
13
    ICLOAD
               LFP40 3.58653E-02LFP41 3.56998E-02
13
                U-234 3.64276E-02U-235 3.62721E-02U-236 3.61182E-02
    OCLOAD
                U-238 3.58140E-02
     OCLOAD
13
    OCLOAD
13
                NP237 3.59654E-02
13
    OCLOAD
                PU236 3.61181E-02PU238 3.58141E-02PU239 3.56639E-02
13
    OCLOAD
                PU240 3.55151E-02PU241 3.53674E-02PU242 3.52210E-02
                AM241 3.53673E-02AM242 3.52209E-02AM243 3.50757E-02
13
    OCLOAD
    OCLOAD
               CM242 3.52209E-02CM243 3.50757E-02CM244 3.49318E-02
13
13
    OCLOAD
               CM245 3.47888E-02CM246 3.46472E-02
                LFP35 3.65475E-02LFP38 3.61589E-02LFP39 3.59882E-02
13
     OCLOAD
    OCLOAD
13
                LFP40 3.58653E-02LFP41 3.56998E-02
   MCLOAD
               U-234 3.64276E-02U-235 3.62721E-02U-236 3.61182E-02
13
13
                U-238 3.58140E-02
    MCLOAD
13
    MCLOAD
                NP237 3.59654E-02
                PU236 3.61181E-02PU238 3.58141E-02PU239 3.56639E-02
13
    MCLOAD
                PU240 3.55151E-02PU241 3.53674E-02PU242
    MCLOAD
                                                       3.52210E-02
13
                AM241 3.53673E-02AM242 3.52209E-02AM243 3.50757E-02
13
    MCLOAD
    MCLOAD
                CM242 3.52209E-02CM243 3.50757E-02CM244 3.49318E-02
13
    MCLOAD
13
                CM245 3.47888E-02CM246 3.46472E-02
13
                LFP35 3.65475E-02LFP38 3.61589E-02LFP39 3.59882E-02
    MCLOAD
                LFP40 3.58653E-02LFP41 3.56998E-02
13
    MCLOAD
    *Reprocessing Parameters
14
     DISO 547.5
14
     DISI 547.5
14
    DISM 547.5
15
    DISO
                REPRO 1.0
                REPRI 1.0
15
     DISI
                REPRM 1.0
15
     DISM
     REPRO SFRF CLSS 180.0
    REPRI SFRF CLSS 180.0
16
16
   REPRM SFRF CLSS 180.0
                                PU238 1.0
17
                NP237 1.0
```

Figure A.IX.1: REBUS Equilibrium Model, CR = 0.77, cont.

```
PU240 1.0
                                     PU241 1.0
17
      SFRF
                                                         PU242 1.0
                   AM241 1.0
                                      AM242 1.0
17
      SFRF
                                                         AM243 1.0
17
      SFRF
                   CM242 1.0
                                     CM243 1.0
                                                         CM244 1.0
17
      SFRF
                   CM245 1.0
                                     CM246 1.0
18
                   NP237 1.0
                                     PU241 1.0
                                                         PU239 1.0
      CLSS
18
      CLSS
                   PU240 1.0
                                     PU238 1.0
                                                        PU242 1.0
18
      CLSS
                   AM241 1.0
                                     AM242 1.0
                                                        AM243 1.0
                   CM242 1.0
                                     CM243 1.0
                                                        CM244 1.0
18
      CLSS
                   CM245 1.0
                                     CM246 1.0
18
      CLSS
19
      CPL1
                   REPRI
                              1
19
      CPL2
                   REPRO
                              1
19
      CPL3
                   REPRM
                              1
    * Class - 1 : LWR-SNF
22
      ESNF
                   NP237
                          4.59900-002AM241 5.07600-002AM242
                                                                6.00000-005
22
                   PU238
                          1.34500-002PU239
                                            5.17730-001PU240
                                                                2.36650-001
      ESNF
22
                          7.80200-002PU242 4.67400-002AM243
                                                                8.80000-003
      ESNF
                   PU241
                   CM243 3.00000-005CM244 1.67000-003CM245
22
      ESNF
                                                                9.00000-005
22
      ESNF
                   CM246 1.00000-005
21
      ESNF
            SNFS
                         1.0E30
                   NP237
                         1.0
                                             1.0
                                                                1.0
18
                                      PU236
                                                         PU238
      SNFS
                   PU239
                                      PU240
                                             1.0
                                                         PU241
18
      SNFS
                          1.0
                                                                1.0
18
      SNFS
                   PU242
                                      AM241
                                             1.0
                                                         AM242
                                                                1.0
                          1.0
                   AM243
18
      SNFS
                                      CM242
                                             1.0
                                                         CM243
                          1.0
                                                                1.0
                         1.0
18
      SNFS
                   CM244
                                      CM245
                                             1.0
                                                         CM246
                              2
19
      CPL3
                   ESNF
19
                   ESNF
                              2
      CPL1
19
      CPL2
                   ESNF
                              2
    * Class - 2 : Depleted Uranium
22
                   U-238 0.998
                                      U-235 0.002
      EDU
21
      EDU
                         1.0E30
18
      SDU
                   U-234
                         0.0
                                      U-235
                                             0.0
                                                         U-236 0.0
                   U-238 0.0
18
      SDU
20
                   EDU
                              1
      CPL1
20
      CPL2
                   EDU
                              1
20
      CPL3
                   EDU
                              1
                       92 234.040945
24
      U-234
                  0
      U-235
24
                  1
                       92 235.043922
      U-236
                       92 236.045561
24
                  0
24
      U-238
                  0
                       92 238.050785
24
      NP237
                  0
                       93 237.048166
24
      PU236
                 0
                       94 236.046048
24
     PU238
                 0
                       94 238.049553
24
      PU239
                 1
                       94 239.052156
24
      PU240
                  0
                       94 240.053808
24
      PU241
                       94 241.056273
                  1
24
      PU242
                  0
                       94 242.058737
24
      AM241
                  0
                       95 241.056822
24
      AM242
                  0
                       95 242.059098
24
                  0
                       95 243.061374
      AM243
24
      CM242
                       96 242.058831
```

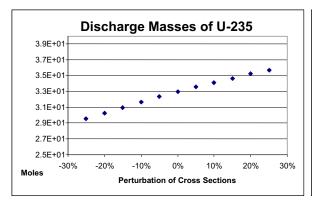
Figure A.IX.1: REBUS Equilibrium Model, CR = 0. 77, cont.

```
24
    CM243 0
                96 243.061382
                96 244.062747
24
     CM244
             0
                96 245.065484
24
     CM245
             0
24
    CM246
             0 96 246.067218
24
             0 92 233.27263
    LFP35
24
    LFP38
             0 92 235.77988
24
    LFP39
             0 94 236.89792
             0 94 237.71005
24
    LFP40
                94 238.81227
24
    LFP41
              0
                 92 232.0371
24
    DUMP1
              0
24
    DUMP2
             0
                  96 246.0672
29
    ICORE MCORE OCORE
32
        100.0
                     250.0
                            100.0
                                    121.7 3
                       0 0
34
        15
             0 0
46
    NP237
              3PU236
                        1PU238
                                  1PU239
                                             1PU240
    PU241
              1PU242
                        1AM241
                                  3AM242
                                             3AM243
46
46
    CM242
              3CM243
                        3CM244
                                   3CM245
                                             3CM246
```

Figure A.IX.1: REBUS Equilibrium Model, CR = 0.77, cont.

Appendix B: Graphical Verification of Model Linearity

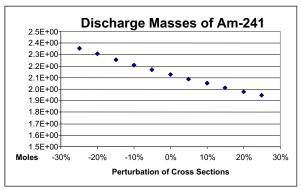
Presented below is a collection of 6 plots (Figures B.1a-1f) of selected nuclides showing that the general depletion model, namely ORIGEN, is essentially linear over the range of cross section perturbations applicable to this problem. The range used is cross sections perturbations from 0 to +/- 25% of the nominal value. Further, a corresponding set of 6 histograms, of the same nuclides, are shown representing the result of random sampling for that nuclides after implementing the foreword perturbation model (Figures B.2a-2f). The reader can see that these histograms resemble a Gaussian distribution, as the input cross sections were perturbed in a Gaussian distribution; a proof of linearity.



Discharge Masses of Pu-239 2.2E+01 2.2E+01 2.2E+01 2.1E+01 2.1E+01 2.1E+01 2.1E+01 2.1E+01 2.0E+01-Moles • 2.0E+01 2.0E+01 -30% -20% 10% 30% Perturbation of Cross Sections

Figure B.1a: U-235 Linearity

Figure B.1b: Pu-239 Linearity



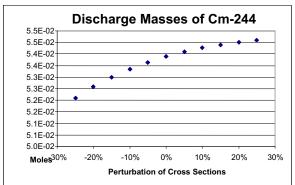
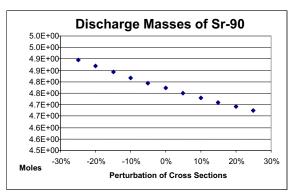


Figure B.1c: Am-241 Linearity

Figure B.1d: Cm-244 Linearity



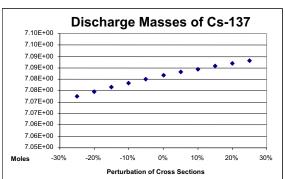


Figure B.1e: Sr-90 Linearity

Figure B.1f: Cs-137 Linearity

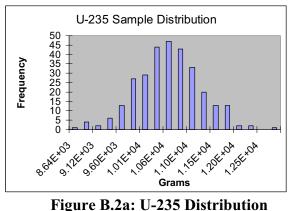


Figure B.2a: U-235 Distribution

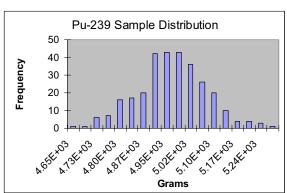


Figure B.2b: Pu-239 Distribution

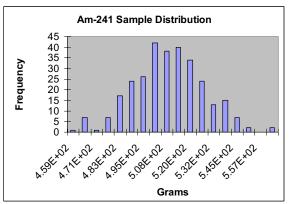


Figure B.2c: Am-241 Distribution

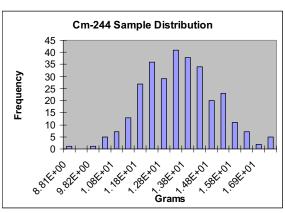


Figure B.2d: Cm-244 Distribution

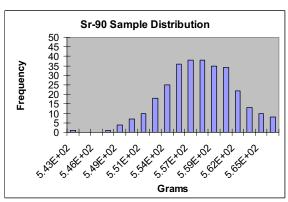


Figure B.2e: Sr-90 Distribution

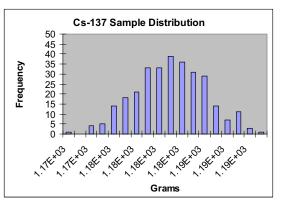


Figure B.2f: Cs-137 Distribution

Appendix C: Results Tables for Fuel Models

The following are the generalized results tables for each of the fuel models discussed in the Numerical Results section. The listing follows the order of models presented in the main body of the thesis. Since the nuclides between Pb-210 and Pa-231 were shown to contribute a negligible amount to any of the metrics, they were not included in the isotopics uncertainties lists. Tables are sub-labeled and include the absolute quantity and relative uncertainty to 95% confidence interval (1.96 standard deviations) for heat load, radioactivity, and radiotoxicity at various time steps, isotopic relative standard deviations for transuranics and fission products, main contributors to uncertainty heat load and radioactivity, and the 5 cross sections causing the most uncertainty in the model (simplified models only). The tables are somewhat different in format between the simplified ORIGEN models and the TRITON models but still follow the same principle. The table for the REBUS model follows the same format as the TRITON tables except "Charge" refers to BOC core loading and "discharge" refers to EOC core loading, both normalized to 1 MTHM.

Result	s Table for	PWR,	4.5 w/o U	OX bu	rne		
	Uncertainty f	or Key M	letrics				ic Mass tainties
	Heat			Activity		Actindes (+/- %)	
	W	+/- %	Ci	+/- %	Π,	1235	1.338
Charge	5.817E-02	N/A	2.063E+00	N/A		1236	0.702
Discharge	2.443E+06	N/A	2.325E+08	N/A	Ų	1237	2.551
1 yr	1.178E+04	N/A	2.893E+06	N/A		1238	0.076
5 yr	2.215E+03	1.152	7.205E+05	1.173	r	np237	0.616
10 yr	1.423E+03	1.166	4.982E+05	1.119	r	np239	13.870
50 yr	6.628E+02	3.037	1.603E+05	4.180	F	ou238	1.058
100 yr	3.567E+02	2.286	5.063E+04	0.601	F	ou239	0.885
500 yr	1.126E+02	3.844	3.508E+03	3.752	F	ou240	2.766
1000 yr	6.221E+01	3.280	1.966E+03	3.174	F	ou241	2.551
2500 yr	2.481E+01	2.564	8.365E+02	2.540	F	ou242	2.625
5000 yr	1.818E+01	2.642	6.274E+02	2.590	a	ım241	2.535
10000 yr	1.346E+01	2.257	4.705E+02	2.199	a	m242m	2.229
-	Ihalation H	azard	Ingestion H	lazard	a	ım243	13.870
	m³ air	+/- %	m³ water	+/- %		:m242	2.229
Charge	1.133E+13	N/A	5.380E+06	N/A	ď	:m244	11.670
Discharge	9.128E+17	N/A	6.723E+12	N/A		:m245	10.600
1 yr	6.752E+17	N/A	7.972E+11	N/A		FP (·	+/- %)
5 yr	6.074E+17	3.295	3.862E+11	1.088		: 14	0.490
10 yr	5.968E+17	2.913	3.171E+11	1.092	S	se 79	0.382
50 yr	5.184E+17	3.847	1.728E+11	2.339	5	sr 90	0.385
100 yr	4.405E+17	3.196	1.055E+11	2.512	t	c 99	1.880
500 yr	2.080E+17	3.621	3.913E+10	3.621	i	129	0.449
1000 yr	1.203E+17	3.049	2.262E+10	3.049	ď	s137	0.399
2500 yr	5.455E+16	2.610	1.026E+10	2.608	k	oa137m	0.399
5000 yr	4.133E+16	2.642	7.778E+09	2.639)	₁ 90	0.386
10000 yr	3.053E+16	2.257	5.761E+09	2.255	ď	:s134	1.209
Ma	ain Contributor Decay Heat an		•			• .	oss Sections
	Heat		Activity	v		Nuclide	Incertainty Reaction
5 yr	Cm-24		Pu241			Am-243	(n,γ)
10 yr	Cm-244, C		Pu-24			Pu-240	(n,γ)
50 yr	Cm-244, C		Pu-24			U-235	(fission)
100 yr	Am-24		Am-24			Pu-239	(fission)
500 yr	Am-24	1	Am-24	1		U-234	(n,γ)
1000 yr	Am-24	1	Am-24	1			
2500 yr	Pu-240	0	Pu-240	0			
5000 yr	Pu-240 Pu	ı - 239	Pu-240, Pt	ı-239			
10000 yr	Pu-240 Pu	ı-239	Pu-240, Pt	ı-239			

Table C.1: Results table for the PWR model using ESM for uncertainty.

Result	s Table fo	r PWR	, 4.5 w/o L	IOX bu	rned 40 GV	VD/MTU oic Mass
	Uncertainty	for Key I	Metrics			rtainties
	Heat		Activit	y		es (+/- %)
	W	+/- %	Ci	+/- %	u235	1.312
Charge	5.817E-02	N/A	2.063E+00	N/A	u236	0.712
Discharge	2.443E+06	N/A	2.325E+08	N/A	u237	2.521
1 yr	1.178E+04	N/A	2.893E+06	N/A	u238	0.078
5 yr	2.215E+03	1.113	7.205E+05	1.156	np237	0.605
10 yr	1.423E+03	1.133	4.982E+05	1.103	np239	13.643
50 yr	6.628E+02	2.935	1.603E+05	4.122	pu238	1.050
100 yr	3.567E+02	2.255	5.063E+04	0.588	pu239	0.806
500 yr	1.126E+02	3.788	3.508E+03	3.696	pu240	2.566
1000 yr	6.221E+01	3.215	1.966E+03	3.109	pu241	2.521
2500 yr	2.481E+01	2.396	8.365E+02	2.379	pu242	2.598
5000 yr	1.818E+01	2.457	6.274E+02	2.417	am241	2.502
10000 yr	1.346E+01	2.094	4.705E+02	2.049	am242m	2.172
	Inhalation I	Inhalation Hazard Ingestion Hazard			am243	13.643
	m³ air	+/- %	m³ water	+/- %	cm242	2.172
Charge	1.133E+13	N/A	5.380E+06	N/A	cm244	11.340
Discharge	9.128E+17	N/A	6.723E+12	N/A	cm245	10.251
1 yr	6.752E+17	N/A	7.972E+11	N/A	FP ((+/- %)
5 yr	6.074E+17	3.216	3.862E+11	1.059	c 14	0.452
10 yr	5.968E+17	2.846	3.171E+11	1.066	se 79	0.366
50 yr	5.184E+17	3.755	1.728E+11	2.278	sr 90	0.375
100 yr	4.405E+17	3.152	1.055E+11	2.477	tc 99	1.970
500 yr	2.080E+17	3.564	3.913E+10	3.564	i129	0.419
1000 yr	1.203E+17	2.976	2.262E+10	2.976	cs137	0.382
2500 yr	5.455E+16	2.430	1.026E+10	2.428	ba137m	0.382
5000 yr	4.133E+16	2.452	7.778E+09	2.450	y90	0.375
10000 yr	3.053E+16	2.091	5.761E+09	2.089	cs134	1.138
M	ain Contributo Decay Heat a		-			oss Sections Uncertainty
	Heat		Activit	v	Nuclide	Reaction
5 yr	Cm-24		Pu241		Am-243	(n,γ)
10 yr	Cm-244, C	s-134	Pu-24	1	Pu-240	(n,γ)
50 yr	Cm-244, C		Pu-24		U-235	(fission)
100 yr	Am-24	1	Am-24	1	Pu-239	(fission)
500 yr	Am-24		Am-24		U-234	(n,γ)
1000 yr	Am-24		Am-24			
2500 yr	Pu-24		Pu-240			
5000 yr	Pu-240 Pu		Pu-240, Pu			
10000 yr	Pu-240 Pu		Pu-240, Pu			

Table C.2: Results table for PWR fuel in the simplified model.

Results	Table for	LWR,	4.5 w/o U	OX bu	rned 40 GW	/D/MTU
	Uncertainty f	or Key M	etrics			ic Mass tainties
	Heat		Activit	y	Actinde	es (+/- %)
	W	+/- %	Ci	+/- %	u235	6.184
Charge	5.817E-02	N/A	2.063E+00	N/A	u236	1.364
Discharge	5.906E+04	N/A	2.255E+07	N/A	u237	3.708
1 yr	3.121E+03	N/A	7.167E+05	N/A	u238	0.089
5 yr	1.711E+03	1.148	5.456E+05	1.426	np237	1.032
10 yr	1.292E+03	0.966	4.494E+05	1.353	np239	12.698
50 yr	6.104E+02	1.649	1.562E+05	0.719	pu238	2.467
100 yr	3.120E+02	3.037	4.902E+04	0.730	pu239	2.202
500 yr	9.464E+01	5.269	2.925E+03	5.124	pu240	2.837
1000 yr	5.463E+01	4.340	1.720E+03	4.181	pu241	3.708
2500 yr	2.434E+01	3.304	8.046E+02	3.222	pu242	3.995
5000 yr	1.815E+01	3.485	6.092E+02	3.362	am241	3.675
10000 yr	1.309E+01	3.348	4.433E+02	3.200	am242m	3.169
	Inhalation F	lazard	Ingestion H	azard	am243	12.698
	m³ air	+/- %	m³ water	+/- %	cm242	3.169
Charge	1.133E+13	N/A	5.380E+06	N/A	cm244	11.464
Discharge	5.939E+17	N/A	1.504E+12	N/A	cm245	11.231
1 yr	4.983E+17	N/A	4.863E+11	N/A	FP (+/- %)
5 yr	4.729E+17	3.570	3.530E+11	1.135	c 14	0.683
10 yr	4.671E+17	3.337	2.960E+11	1.158	se 79	0.415
50 yr	4.163E+17	4.057	1.551E+11	2.092	sr 90	0.713
100 yr	3.581E+17	4.589	9.047E+10	3.424	tc 99	1.644
500 yr	1.776E+17	4.902	3.341E+10	4.902	i129	0.569
1000 yr	1.078E+17	3.994	2.026E+10	3.995	cs137	0.335
2500 yr	5.414E+16	3.376	1.018E+10	3.373	ba137m	0.335
5000 yr	4.149E+16	3.514	7.799E+09	3.511	y90	0.713
10000 yr	2.991E+16	3.381	5.623E+09	3.378	cs134	1.956
Ma	in Contributor	s to Unc	ertainty in		1-group Cr	oss Sections
	Decay Heat ar	nd Radioa	activity		Causing l	Uncertainty
	Heat		Activit	y	Nuclide	Reaction
5 yr	Cm-244, C		Pu241		Am-243	(n,γ)
10 yr	Cm-244, `		Pu-241	1	Pu-240	(n,γ)
50 yr	Am-24		Pu-241		U-235	(fission)
100 yr	Am-24		Am-24		Pu-239	(fission)
500 yr	Am-24		Am-24	1	U-234	(n,γ)
1000 yr	Am-24		Am-24			
2500 yr	Pu-240 Pu		Pu-240, Pu			
5000 yr	Pu-240 Pu	ı-239	Pu-240, Pu	ı-239		
10000 yr	Pu-240 Pu	ı - 239	Pu-240, Pu	ı-239		

Table C.3: Results table for typical LWR fuel simplified model.

	Results Ta		•	•	4.5 w/o UC	X
		burr	ned 40 GW	<u>/D/MTU</u>		
	Uncertainty	for Koy I	Motrics			pic Mass rtainties
	Heat	ioi Key i	Activit	v		les (+/- %)
	W	+/- %	Ci	+/- %	u235	1.360
Charge	2.063E+00	N/A	5.817E-02	N/A	u236	0.481
Discharge	2.229E+08	N/A	2.368E+06	N/A	u237	2.105
1 yr	2.767E+06	N/A	1.095E+04	N/A	u238	0.055
5 yr	6.661E+05	0.491	2.014E+03	0.617	np237	0.541
10 yr	4.602E+05	0.426	1.304E+03	0.585	np239	8.910
50 yr	1.546E+05	0.719	5.510E+02	0.368	pu238	1.102
100 yr	4.824E+04	1.430	2.536E+02	0.373	pu239	0.817
500 yr	2.098E+03	2.936	6.619E+01	2.851	pu240	2.157
1000 yr	1.256E+03	2.505	3.920E+01	2.422	pu241	2.105
2500 yr	6.293E+02	2.411	1.850E+01	2.336	pu242	2.011
5000 yr	4.829E+02	2.463	1.388E+01	2.354	am241	2.089
10000 yr	3.510E+02	2.135	9.855E+00	2.017	am242m	1.807
10000 yi	Inhalation F					8.910
			Ingestion F		am243	
	m³ air	+/- %	m³ water	+/- %	cm242	1.806
Charge	1.133E+13	N/A	5.380E+06	N/A	cm244	7.416
Discharge	5.477E+17	N/A	6.431E+12	N/A	cm245	6.816
1 yr	3.662E+17	N/A	7.115E+11	N/A		(+/- %)
5 yr	3.192E+17	1.854	3.352E+11	0.500	c 14	0.392
10 yr	3.155E+17	1.740	2.731E+11	0.510	se 79	0.316
50 yr	2.838E+17	2.243	1.326E+11	0.941	sr 90	0.322
100 yr	2.455E+17	2.555	6.998E+10	1.693	tc 99	1.309
500 yr	1.252E+17	2.752	2.355E+10	2.752	i129	0.362
1000 yr	7.806E+16	2.387	1.468E+10	2.386	cs137	0.326
2500 yr	4.122E+16	2.474	7.753E+09	2.472	ba137m	0.326
5000 yr	3.164E+16	2.484	5.956E+09	2.482	y90	0.322
10000 yr	2.239E+16	2.157	4.226E+09	2.155	cs134	0.962
M	ain Contributo Decay Heat a		-			ross Sections Uncertainty
		iiu itauli	_	.,		Reaction
5 yr	Heat Cm-244, C	s-134	Activit Pu241		Nuclide Am-243	Reaction (n,γ)
_ - J :	Cm-244,Y-9		1 424		7 111 2 70	(11,7)
10 yr	137m		Pu-24	1	Pu-240	(n,γ)
50 yr	Am-24		Pu-24		U-235	(fission)
100 yr	Am-24		Pu-24		Pu-239	(fission)
500 yr	Am-24	1	Am-24	1	U-234	(n,γ)
1000 yr	Am-24		Am-24			
2500 yr	Pu-240		Pu-24			
5000 yr	Pu-240 Pu		Pu-240, Pı			
10000 yr	Pu-240 Pu		Pu-240, Pı			

Table C.4: Results table for BWR fuel burned at 0% void.

R	esults Tab	le for	BWR, 35%	Void,	4	.5 w/o UO	X
		burne	ed 40 GWI	D/MTU			
	Uncertainty fo	or Key M	letrics				ic Mass tainties
	Heat		Activit	Activity		Actindes (+/- %)	
	W	+/- %	Ci	+/- %		u235	1.472
Charge	2.063E+00	N/A	5.817E-02	N/A		u236	0.608
Discharge	2.274E+08	N/A	2.405E+06	N/A		u237	2.390
1 yr	2.817E+06	N/A	1.128E+04	N/A		u238	0.065
5 yr	6.855E+05	0.674	2.087E+03	0.854		np237	0.593
10 yr	4.735E+05	0.631	1.344E+03	0.811		np239	11.086
50 yr	1.565E+05	0.982	5.912E+02	0.468		pu238	1.101
100 yr	4.907E+04	1.853	2.916E+02	0.475		pu239	0.867
500 yr	2.632E+03	3.436	8.375E+01	3.343		pu240	2.381
1000 yr	1.533E+03	2.897	4.816E+01	2.798		pu241	2.390
2500 yr	7.191E+02	2.478	2.125E+01	2.418		pu242	2.252
5000 yr	5.470E+02	2.539	1.581E+01	2.447		am241	2.372
10000 yr	4.023E+02	2.190	1.142E+01	2.093		am242m	2.071
	Inhalation H	lazard	Ingestion F	lazard		am243	11.086
	m³ air	+/- %	m³ water	+/- %		cm242	2.070
Charge	1.133E+13	N/A	5.380E+06	N/A		cm244	9.233
Discharge	6.788E+17	N/A	6.560E+12	N/A		cm245	8.431
1 yr	4.769E+17	N/A	7.444E+11	N/A		FP (-	⊦/- %)
5 yr	4.228E+17	2.369	3.535E+11	0.683		c 14	0.441
10 yr	4.171E+17	2.166	2.886E+11	0.698		se 79	0.363
50 yr	3.703E+17	2.562	1.472E+11	1.242		sr 90	0.381
100 yr	3.180E+17	2.935	8.316E+10	2.117		tc 99	1.609
500 yr	1.569E+17	3.217	2.951E+10	3.217		i129	0.408
1000 yr	9.474E+16	2.710	1.782E+10	2.710		cs137	0.374
2500 yr	4.713E+16	2.538	8.863E+09	2.536		ba137m	0.374
5000 yr	3.602E+16	2.555	6.779E+09	2.553		y90	0.381
10000 yr	2.594E+16	2.206	4.894E+09	2.204		cs134	1.063
Ma	in Contributor	s to Unc	ertainty in			1-group Cro	ss Sections
	Decay Heat an	d Radioa	activity			Causing L	Incertainty
	Heat		Activit	:y		Nuclide	Reaction
5 yr	Cm-244, C		Pu241	1		Am-243	(n,γ)
	Cm-244,Y						
10 yr	Ba-137		Pu-24			Pu-240	(n,γ)
50 yr	Am-24		Pu-24			U-235	(fission)
100 yr	Am-24	1	Am-24	1		Pu-239	(fission)
500 yr	Am-24		Am-24			U-234	(n,γ)
1000 yr	Am-24		Am-24				
2500 yr	Pu-240		Pu-240				
5000 yr	Pu-240 Pu	-239	Pu-240, Pu	ı-239			
10000 yr	Pu-240 Pu	-239	Pu-240, Pu	u-239			

Table C.5: Results table for BWR fuel burned at 35% void.

F	Results Tak		WR, 50% d 40 GWD	•	.5 w/o UOX		
				, <u> </u>		ic Mass	
	Uncertainty 1				Uncertainties		
	Heat		Activit			s (+/- %)	
	W	+/- %	Ci	+/- %	u235	1.757	
Charge	2.063E+00	N/A	5.817E-02	N/A	u236	0.731	
Discharge	2.303E+08	N/A	2.425E+06	N/A	u237	2.492	
1 yr	2.853E+06	N/A	1.153E+04	N/A	u238	0.069	
5 yr	7.015E+05	0.880	2.146E+03	1.025	np237	0.638	
10 yr	4.848E+05	0.858	1.379E+03	0.977	np239	12.685	
50 yr	1.582E+05	1.171	6.249E+02	0.548	pu238	1.147	
100 yr	4.979E+04	2.098	3.230E+02	0.556	pu239	0.874	
500 yr	3.074E+03	3.667	9.827E+01	3.574	pu240	2.637	
1000 yr	1.763E+03	3.111	5.561E+01	3.010	pu241	2.492	
2500 yr	7.953E+02	2.609	2.360E+01	2.560	pu242	2.437	
5000 yr	6.019E+02	2.678	1.747E+01	2.597	am241	2.474	
10000 yr	4.463E+02	2.290	1.276E+01	2.207	am242m	2.156	
	Inhalation	Hazard	Ingestion F	lazard	am243	12.685	
	m³ air	+/- %	m³ water	+/- %	cm242	2.155	
Charge	1.133E+13	N/A	5.380E+06	N/A	cm244	10.618	
Discharge	7.862E+17	N/A	6.652E+12	N/A	cm245	9.687	
1 yr	5.685E+17	N/A	7.698E+11	N/A	FP (·	+/- %)	
5 yr	5.087E+17	2.786	3.686E+11	0.868	c 14	0.483	
10 yr	5.013E+17	2.506	3.017E+11	0.881	se 79	0.408	
50 yr	4.415E+17	2.710	1.595E+11	1.439	sr 90	0.436	
100 yr	3.778E+17	3.100	9.405E+10	2.347	tc 99	1.864	
500 yr	1.831E+17	3.443	3.444E+10	3.443	i129	0.444	
1000 yr	1.087E+17	2.906	2.043E+10	2.906	cs137	0.413	
2500 yr	5.218E+16	2.669	9.813E+09	2.667	ba137m	0.413	
5000 yr	3.979E+16	2.690	7.488E+09	2.688	y90	0.436	
10000 yr	2.898E+16	2.302	5.467E+09	2.300	cs134	1.166	
M	ain Contributo	rs to Uncer	tainty in		1-group Cro	ss Sections	
	Decay Heat a	nd Radioad	tivity		Causing U	Incertainty	
	Heat	t	Activit	y	Nuclide	Reaction	
5 yr	Cm-244, C	Cs-134	Pu241		Am-243	(n,γ)	
10 yr	Cm-244,Y-90,	Ba-137m	Pu-24	1	Pu-240	(n,γ)	
50 yr	Am-24	41	Pu-24	1	U-235	(fission)	
100 yr	Am-24	11	Am-24	1	Pu-239	(fission)	
500 yr	Am-24	11	Am-24		U-234	(n,γ)	
1000 yr	Am-24	41	Am-24	1			
2500 yr	Pu-24	10	Pu-240				
5000 yr	Pu-240 P	u-239	Pu-240, Pu	u-239			
10000 yr	Pu-240 P	000	Pu-240, Pu	000		1	

Table C.6: Results table BWR fuel burned at 50% void.

ı	Results Ta		WR, 65% d 40 GWD		4.5 w/o U	OX	
	Uncertainty			/ IVI I O		topic Mass certainties	
	Hea		Activity	,		Actindes (+/- %)	
	W	+/- %	Ci	+/- %	u235	1.677	
Charge	2.063E+00	N/A	5.817E-02	N/A	u236	0.839	
Discharge	2.346E+08	N/A	2.449E+06	N/A	u237	2.479	
1 yr	2.907E+06	N/A	1.189E+04	N/A	u238	0.079	
5 yr	7.286E+05	1.223	2.243E+03	1.226	np237	0.697	
10 yr	5.046E+05	1.232	1.440E+03	1.175	np239	14.483	
50 yr	1.613E+05	1.385	6.824E+02	0.649	pu238	1.172	
100 yr	5.111E+04	2.288	3.761E+02	0.657	pu239	0.920	
500 yr	3.823E+03	3.706	1.229E+02	3.619	pu240	2.536	
1000 yr	2.155E+03	3.138	6.835E+01	3.040	pu241	2.479	
2500 yr	9.305E+02	2.390	2.777E+01	2.374	pu242	2.531	
5000 yr	7.006E+02	2.459	2.048E+01	2.421	am241	2.462	
10000 yr	5.253E+02	2.130	1.518E+01	2.086	am242m	2.154	
	Inhalation	Hazard	Ingestion H	azard	am243	14.483	
	m³ air	+/- %	m³ water	+/- %	cm242	2.153	
Charge	1.133E+13	N/A	5.380E+06	N/A	cm244	12.041	
Discharge	9.632E+17	N/A	6.791E+12	N/A	cm245	10.851	
1 yr	7.212E+17	N/A	8.098E+11	N/A	F	P (+/- %)	
5 yr	6.526E+17	3.237	3.941E+11	1.146	c 14	0.544	
10 yr	6.423E+17	2.862	3.242E+11	1.150	se 79	0.465	
50 yr	5.615E+17	2.730	1.803E+11	1.624	sr 90	0.496	
100 yr	4.784E+17	3.106	1.125E+11	2.489	tc 99	2.238	
500 yr	2.276E+17	3.482	4.281E+10	3.482	i129	0.502	
1000 yr	1.326E+17	2.906	2.494E+10	2.906	cs137	0.473	
2500 yr	6.124E+16	2.427	1.152E+10	2.425	ba137m	0.473	
5000 yr	4.663E+16	2.457	8.775E+09	2.455	y90	0.496	
10000 yr	3.450E+16	2.131	6.507E+09	2.129	cs134	1.305	
N	lain Contributo				5	Cross Sections	
	Decay Heat a	nd Radioac	tivity		Causir	ng Uncertainty	
	Hea	t	Activity	y	Nuclide	Reaction	
5 yr	Cm-244, (Cs-134	Pu241		Am-243	(n,γ)	
10 yr	Cm-244,Y-90	, Ba-137m	Pu-241		Pu-240	(n,γ)	
50 yr	Am-2	41	Pu-241		U-235	(fission)	
100 yr	Am-2	41	Am-24	1	Pu-239	(fission)	
500 yr	Am-2		Am-24		U-234	(n,γ)	
1000 yr	Am-2		Am-24				
2500 yr	Pu-240 P		Pu-240, Pu				
5000 yr	Pu-240 P		Pu-240, Pu	The state of the s			
10000 yr	Pu-240 P	u-239	Pu-240, Pu	ı - 239			

Table C.7: Results table for BWR fuel burned at 65% void.

Re	sults Table f	or PWR	, MOX fuel	burned	50 GWD/N	I TU
	Uncertainty	for Key M	etrics		Isotopic Mass	Uncertainties
	Heat		Activity	y		s (+/- %)
	W	+/- %	Ci	+/- %	u235	1.367
Charge	9.887E+05	N/A	3.926E+03	N/A	u236	1.037
Discharge	2.888E+08	N/A	2.990E+06	N/A	u237	1.929
1 yr	4.927E+06	N/A	2.543E+04	N/A	u238	0.092
5 yr	1.602E+06	11.670	7.824E+03	2.723	np237	0.790
10 yr	1.157E+06	11.020	6.267E+03	2.691	np239	24.978
50 yr	3.288E+05	4.158	3.765E+03	1.744	pu238	1.312
100 yr	1.255E+05	2.058	2.567E+03	1.325	pu239	1.384
500 yr	2.354E+04	2.743	7.581E+02	2.777	pu240	2.361
1000 yr	1.287E+04	2.803	4.055E+02	3.009	pu241	1.929
2500 yr	5.661E+03	3.832	1.679E+02	4.423	pu242	2.590
5000 yr	4.153E+03	4.149	1.212E+02	4.742	am241	1.838
10000 yr	2.909E+03	3.725	8.490E+01	4.263	am242m	1.213
	Inhalation	Hazard	Ingestion H	azard	am243	24.978
	m³ air	+/- %	m³ water	+/- %	cm242	1.213
Charge	9.004E+18	N/A	1.701E+12	N/A	cm244	17.637
Discharge	1.156E+19	N/A	1.021E+13	N/A	cm245	14.425
1 yr	1.015E+19	N/A	2.783E+12	N/A	FP (·	+/- %)
5 yr	9.341E+18	8.639	2.037E+12	7.608	c 14	0.837
10 yr	8.825E+18	7.588	1.852E+12	6.863	se 79	0.734
50 yr	6.207E+18	2.906	1.238E+12	2.758	sr 90	0.690
100 yr	4.600E+18	1.989	8.878E+11	1.944	tc 99	2.561
500 yr	1.410E+18	2.612	2.652E+11	2.611	i129	0.764
1000 yr	7.853E+17	2.697	1.477E+11	2.696	cs137	0.759
2500 yr	3.658E+17	3.576	6.884E+10	3.573	ba137m	0.759
5000 yr	2.717E+17	3.763	5.121E+10	3.761	y90	0.690
10000 yr	1.895E+17	3.386	3.591E+10	3.383	cs134	1.673
	Main Contribut	ors to Unce	ertainty in		1-group Cro	ss Sections
	Decay Heat	and Radioa	ctivity		Causing U	Incertainty
	Heat		Activity	y	Nuclide	Reaction
5 yr	Cm-24	14	Pu-241, Cm	า-244	Am-243	(n,γ)
10 yr	Cm-24	l 4	Pu-241, Cm	า-244	Pu-239	(fission)
50 yr	Cm-24	l 4	Pu-241, Cm	า-244	Pu-240	(fission)
100 yr	Pu-238, Am-24	1, Cm-244	Pu-238, Am-241	, Cm-244	Pu-242	(n,γ)
500 yr	Am-24	1	Am-24	1	Pu-240	(n,γ)
1000 yr	Am-241, A	m-243	Am-241, An	n-243		
2500 yr	Pu-240, Ar	m-243	Pu-240, Am	1-243		
5000 yr	Pu-240, Ar	m-243	Pu-240, Am	1-243		
10000 yr	Pu-240, Ar	m-243	Pu-240, Am	1-243		

Table C.8: Results table for clean MOX fuel.

Result	s Table for		MOX fuel wi GWD/MTU	th Impi	urities burn	ed 50	
	Uncertainty:	for Key M	etrics		Isotopic Mass	Uncertainties	
	Heat		Activity	y	Actinde	es (+/- %)	
	w	+/- %	Ci	+/- %	u235	1.354	
Charge	1.040E+06	N/A	5.643E+03	N/A	u236	1.059	
Discharge	2.977E+08	N/A	3.134E+06	N/A	u237	1.863	
1 yr	5.927E+06	N/A	6.168E+04	N/A	u238	0.096	
5 yr	1.796E+06	6.146	1.419E+04	2.353	np237	0.373	
10 yr	1.343E+06	5.588	1.232E+04	2.281	np239	23.074	
50 yr	4.637E+05	2.103	8.179E+03	1.316	pu238	0.791	
100 yr	2.200E+05	1.265	5.641E+03	1.009	pu239	1.354	
500 yr	3.481E+04	1.741	1.120E+03	1.788	pu240	2.299	
1000 yr	1.656E+04	2.095	5.249E+02	2.310	pu241	1.863	
2500 yr	6.460E+03	3.484	1.917E+02	4.053	pu242	2.393	
5000 yr	4.640E+03	3.893	1.349E+02	4.459	am241	1.183	
10000 yr	3.332E+03	3.457	9.670E+01	3.955	am242m	1.162	
	Inhalation Hazard		Ingestion H	azard	am 243 23.074		
	m³ air	+/- %	m³ water	+/- %	cm242	1.162	
Charge	1.196E+19	N/A	2.257E+12	N/A	cm244	16.130	
Discharge	3.657E+19	N/A	1.502E+13	N/A	cm245	13.127	
1 yr	2.528E+19	N/A	5.618E+12	N/A	FP (·	+/- %)	
5 yr	2.145E+19	3.866	4.323E+12	3.659	c 14	0.946	
10 yr	2.046E+19	3.401	4.049E+12	3.259	se 79	0.816	
50 yr	1.477E+19	1.565	2.856E+12	1.531	sr 90	0.764	
100 yr	1.057E+19	1.256	2.016E+12	1.244	tc 99	2.942	
500 yr	2.069E+18	1.681	3.895E+11	1.679	i129	0.828	
1000 yr	9.956E+17	2.074	1.874E+11	2.072	cs137	0.819	
2500 yr	4.101E+17	3.294	7.734E+10	3.288	ba137m	0.819	
5000 yr	2.976E+17	3.564	5.637E+10	3.557	у90	0.764	
10000 yr	2.104E+17	3.194	4.041E+10	3.189	cs134	1.682	
	Main Contributo Decay Heat a		-			oss Sections Incertainty	
	Heat		Activity	v	Nuclide	Reaction	
5 yr	Cm-24	4	Pu-241, Cm		Am-243	(n,γ)	
10 yr	Cm-244	4	Pu-241, Cm	n-244	Pu-239	(fission)	
50 yr	Cm-244	4	Pu-241, Cm		Pu-242	(n,γ)	
100 yr	Pu-238, Am		Pu-238, Am		Pu-241	(fission)	
500 yr	Am-24	1	Am-24	1	Pu-240	(n,γ)	
1000 yr	Am-241, An	า-243	Am-241, An	n-243			
2500 yr	Pu-240, Am	n-243	Pu-240, Am	n-243			
5000 yr	Pu-240, Am	n-243	Pu-240, Am	n-243			
10000 yr	Pu-240, Am	1-243	Pu-240, Am	1-243			

Table C.9: Results table for MOX fuel with impurities.

	Uncertainty fo					ed 48 GWD Isotopio Uncerta	Mass
	Heat		Activit	y		Actindes (+/- %)	
	W	+/- %	Ci	+/- %		u235	0.620
Charge	6.096E-02	N/A	2.160E+00	N/A		u236	1.052
Discharge	6.036E+04	N/A	2.249E+07	N/A		u237	2.067
1 yr	4.050E+03	N/A	8.538E+05	N/A		u238	0.015
5 yr	2.156E+03	0.696	6.427E+05	0.807		np237	1.609
10 yr	1.638E+03	0.779	5.280E+05	0.771		np239	3.878
50 yr	7.799E+02	1.020	1.825E+05	0.352		pu238	1.749
100 yr	4.090E+02	1.736	5.788E+04	0.384		pu239	1.065
500 yr	1.205E+02	2.990	3.737E+03	2.906		pu240	2.486
1000 yr	6.878E+01	2.616	2.177E+03	2.525		pu241	2.065
2500 yr	3.045E+01	2.577	1.018E+03	2.481		pu242	3.897
5000 yr	2.263E+01	2.672	7.679E+02	2.539		am241	2.063
10000 yr	1.625E+01	2.343	5.551E+02	2.212		am242m	2.121
	Inhalation I	Hazard	Ingestion F	lazard		am243	3.884
	m³ air	+/- %	m³ water	+/- %	T	cm242	2.121
Charge	1.188E+13	N/A	5.636E+06	N/A		cm244	4.337
Discharge	1.051E+18	N/A	1.654E+12	N/A		cm245	4.673
1 yr	8.750E+17	N/A	6.188E+11	N/A		FP (+	/- %)
5 yr	8.076E+17	2.334	4.527E+11	0.794		c 14	0.360
10 yr	7.801E+17	2.178	3.819E+11	0.845		se 79	0.087
50 yr	6.216E+17	2.233	2.036E+11	1.285		sr 90	0.120
100 yr	5.075E+17	2.476	1.216E+11	1.945		tc 99	0.076
500 yr	2.257E+17	2.824	4.246E+10	2.823		i129	0.174
1000 yr	1.352E+17	2.509	2.542E+10	2.508		cs137	0.025
2500 yr	6.736E+16	2.668	1.266E+10	2.665		ba137m	0.025
5000 yr	5.144E+16	2.711	9.671E+09	2.708		y90	0.120
10000 yr	3.693E+16	2.378	6.944E+09	2.376		cs134	0.318
	ain Contributor	s to Unce	rtainty in			k-effectiv	e Values
	Decay Heat an		-			BOL	1.4181
	Heat		Activit	y		EOL	0.9696
5 yr	Cm-24		Pu241				
10 yr	Pu-238, Cr		Pu-24				
50 yr	Pu-238, Ar		Pu-24				
100 yr	Am-24		Am-24	i i			
500 yr	Am-24		Am-24				
1000 yr	Pu-240, Ar		Pu-240, Ar				
2500 yr	Pu-240 Pu		Pu-240, Pt				
5000 yr	Pu-240 Pu		Pu-240, Pu	i			
10000 yr	Pu-240 Pu		Pu-240, Pu				

Table C.10: Results table for TRITON PWR model, 48 GWD/MTU

Results Table for FR, CR=0.25, burned 94.3 GWD/MTU							
	Uncertainty	Isotopic Mass Uncertainties					
	Heat Activity				Actindes (+/- %)		
	W	+/- %	Ci	+/- %	u235	1.490	
Charge	6.397E+04	N/A	5.520E+06	N/A	u236	1.795	
Discharge	3.748E+05	N/A	3.982E+07	N/A	u237	2.764	
1 yr	8.308E+04	N/A	6.625E+06	N/A	u238	0.205	
5 yr	5.399E+04	14.160	5.170E+06	5.200	np237	1.114	
10 yr	4.820E+04	13.142	4.290E+06	5.104	np239	10.752	
50 yr	2.555E+04	6.068	1.339E+06	3.610	pu238	3.371	
100 yr	1.677E+04	3.388	6.221E+05	2.753	pu239	1.468	
500 yr	5.375E+03	1.732	1.702E+05	1.708	pu240	0.906	
1000 yr	3.142E+03	1.581	1.014E+05	1.652	pu241	2.764	
2500 yr	1.639E+03	1.768	5.534E+04	1.978	pu242	0.548	
5000 yr	1.218E+03	1.867	4.150E+04	2.076	am241	1.341	
10000 yr	8.012E+02	1.817	2.728E+04	2.009	am242m	1.434	
	Inhalation	Hazard			am243	10.752	
	m³ air	+/- %	m³ water	+/- %	cm242	1.434	
Charge	7.719E+19	N/A	1.458E+13	N/A	cm244	13.025	
Discharge	1.044E+20	N/A	2.136E+13	N/A	cm245	4.344	
1 yr	8.267E+19	N/A	1.612E+13	N/A	FP (+/	- %)	
5 yr	7.580E+19	10.165	1.471E+13	9.930	c 14	1.655	
10 yr	6.983E+19	9.232	1.350E+13	9.048	se 79	1.137	
50 yr	4.339E+19	4.610	8.305E+12	4.559	sr 90	0.348	
100 yr	3.108E+19	3.422	5.895E+12	3.410	tc 99	0.319	
500 yr	1.033E+19	1.623	1.944E+12	1.623	i129	0.412	
1000 yr	6.330E+18	1.486	1.190E+12	1.485	cs137	0.094	
2500 yr	3.619E+18	1.664	6.805E+11	1.663	ba137m	0.094	
5000 yr	2.735E+18	1.733	5.143E+11	1.732	y90	0.348	
10000 yr	1.797E+18	1.698	3.379E+11	1.697	cs134	2.553	
N	lain Contributo	rs to Unce	rtainty in		k-effective Values		
	Decay Heat a	nd Radioa	ctivity		BOL	1.2856	
	Hea	t	Activity		EOL	1.1792	
5 yr	Cm-2	44	Cm-244, P	u-241			
10 yr	Cm-2	44	Cm-244, P	u-241			
50 yr	Cm-244, F	Pu-238	Cm-244, P	u-238			
100 yr	Pu-238, A	m-241	Pu-238, An				
500 yr	Am-2		Am-24				
1000 yr	Am-2		Am-24				
2500 yr	Am-243, F		Am-243, P				
5000 yr	Am-243, F		Am-243, P				
10000 yr	Am-243, Pu- 240		Am-243, Pu-2 240	239, Pu-			
<u> </u>							

Table C.11: Results table for fast reactor fuel of CR = 0.25.

Results Table for FR, CR=0.7, burned 78.4 GWD/MTU							
Uncertainty for Key Metrics					Isotopic Mass Uncertainties		
	Heat		Activity		Actindes		
	W	+/- %	Ci	+/- %	u235	2.587	
Charge	9.086E+03	N/A	1.138E+06	N/A	u236	4.793	
Discharge	1.300E+05	N/A	3.333E+07	N/A	u237	3.832	
1 yr	1.763E+04	N/A	2.142E+06	N/A	u238	0.230	
5 yr	1.208E+04	14.673	1.729E+06	4.552	np237	1.680	
10 yr	1.068E+04	13.774	1.434E+06	4.416	np239	15.475	
50 yr	6.002E+03	6.423	4.625E+05	2.845	pu238	4.824	
100 yr	4.051E+03	4.192	1.920E+05	2.680	pu239	2.406	
500 yr	1.502E+03	2.777	4.738E+04	2.710	pu240	1.992	
1000 yr	9.631E+02	2.588	3.085E+04	2.584	pu241	3.833	
2500 yr	5.749E+02	2.903	1.896E+04	2.933	pu242	1.150	
5000 yr	4.473E+02	3.009	1.482E+04	3.024	am241	2.369	
10000 yr	3.094E+02	2.993	1.024E+04	2.993	am242m	1.836	
	Inhalation	Hazard	Ingestion	Hazard	am243	15.475	
	m³ air	+/- %	m³ water	+/- %	cm242	1.837	
Charge	1.487E+19	N/A	2.809E+12	N/A	cm244	16.925	
Discharge	2.039E+19	N/A	5.674E+12	N/A	cm245	9.295	
1 yr	1.689E+19	N/A	3.633E+12	N/A	FP (+/	- %)	
5 yr	1.571E+19	11.560	3.293E+12	10.454	c 14	0.962	
10 yr	1.465E+19	10.447	3.025E+12	9.585	se 79	0.627	
50 yr	9.804E+18	5.515	1.946E+12	5.256	sr 90	0.180	
100 yr	7.387E+18	4.353	1.422E+12	4.273	tc 99	0.472	
500 yr	2.967E+18	2.627	5.580E+11	2.627	i129	0.460	
1000 yr	2.003E+18	2.559	3.765E+11	2.558	cs137	0.098	
2500 yr	1.293E+18	2.922	2.431E+11	2.920	ba137m	0.092	
5000 yr	1.019E+18	2.998	1.916E+11	2.996	y90	0.181	
10000 yr	7.050E+17	2.992	1.325E+11	2.991	cs134	2.250	
M	ain Contributo	ors to Unce	ertainty in		k-effective Values		
	Decay Heat and Radioa		ctivity		BOL	1.1779	
	Heat	t	Activity		EOL	1.1195	
5 yr	Cm-24		Cm-244, F				
10 yr	Cm-24	44	Cm-244, F				
50 yr	Cm-244, F	Pu-238	Cm-244, Pu-238, Pu- 241				
100 yr	Pu-238, A		Pu-238, A				
500 yr	Am-24		Am-2				
1000 yr	Pu-240, A		Pu-240, A				
2500 yr	Pu-239, P		Pu-239, P				
5000 yr	Pu-239, P		Pu-239, P				
10000 yr	Pu-239, P		Pu-239, P				

Table C.12: Results table for fast reactor fuel of CR = 0.70.

Resu	Its Table fo	or FR,	CR=1.05, k	ourned	67.7 GWD/M	ΓU	
	Uncertainty f	Isotopic Mass Uncertainties					
	Heat Activity			Actindes (+/- %)			
	W	+/- %	Ci	+/- %	u235	3.267	
Charge	1.632E+03	N/A	3.224E+05	N/A	u236	5.535	
Discharge	1.024E+05	N/A	3.537E+07	N/A	u237	4.247	
1 yr	6.572E+03	N/A	1.415E+06	N/A	u238	0.255	
5 yr	4.281E+03	6.259	1.149E+06	3.371	np237	2.828	
10 yr	3.682E+03	6.126	9.487E+05	3.213	np239	15.989	
50 yr	2.403E+03	3.672	3.106E+05	1.628	pu238	5.204	
100 yr	1.735E+03	3.634	1.156E+05	1.685	pu239	2.800	
500 yr	8.261E+02	3.638	2.591E+04	3.572	pu240	3.066	
1000 yr	5.835E+02	3.513	1.852E+04	3.496	pu241	4.246	
2500 yr	3.873E+02	3.989	1.253E+04	3.966	pu242	3.141	
5000 yr	3.117E+02	4.060	1.011E+04	4.025	am241	3.271	
10000 yr	2.240E+02	4.014	7.275E+03	3.977	am242m	1.972	
•	Inhalation	Hazard	Ingestion Hazard		am243	15.989	
	m³ air	+/- %	m³ water	+/- %	cm242	1.972	
Charge	3.353E+18	N/A	6.323E+11	N/A	cm244	19.875	
Discharge	5.606E+18	N/A	3.066E+12	N/A	cm245	11.948	
1 yr	4.657E+18	N/A	1.334E+12	N/A	FP (+/		
5 yr	4.466E+18	6.841	1.169E+12	4.951	c 14	0.708	
10 yr	4.339E+18	6.173	1.076E+12	4.714	se 79	0.544	
50 yr	3.603E+18	4.224	7.747E+11	3.707	sr 90	0.159	
100 yr	3.061E+18	3.878	6.055E+11	3.695	tc 99	0.554	
500 yr	1.681E+18	3.508	3.159E+11	3.507	i129	0.494	
1000 yr	1.247E+18	3.560	2.344E+11	3.559	cs137	0.099	
2500 yr	8.838E+17	4.036	1.661E+11	4.035	ba137m	0.099	
5000 yr	7.181E+17	4.073	1.349E+11	4.072	y90	0.158	
10000 yr	5.161E+17	4.029	9.696E+10	4.028	cs134	2.214	
Ma	in Contributor	s to Unce	rtainty in		k-effective Values		
	Decay Heat an	d Radioa	ctivity		BOL	1.0234	
	Heat		Activity		EOL	1.0452	
5 yr	Cm-24	14	Cm-244, P	u-241			
10 yr	Cm-24	14	Cm-244, P	u-241			
			Cm-244, Pu-	238, Pu-			
50 yr	Cm-244, P		241				
100 yr	Pu-238, Ar		Pu-238, Am-241				
500 yr	Pu-240, Ar		Pu-240,Am-241				
4000	Pu-238, Pu-2	240, Am-	Pu-239, Pu-240, Am-				
1000 yr	241	040	241	242			
2500 yr	Pu-239, P		Pu-239, P				
5000 yr	Pu-239, P		Pu-239, P	i			
10000 yr	Pu-239, Pu-240		Pu-239, Pu-240				

Pu-239, Pu-240 Pu-239, Pu-240 Table C.13: Results table for fast reactor fuel of CR = 1.05.

Result	s Table for	Recycled	FR, CR=0	.7, burn	ec	41.4 GWD	/MTU		
	Uncertainty for Key Metrics						Isotopic Mass Uncertainties		
	Heat		Activity			Actindes	(+/- %)		
	W	+/- %	Ci	+/- %		u235	2.819		
Charge	9.086E+03	N/A	1.138E+06	N/A		u236	5.757		
Discharge	1.331E+05	N/A	3.272E+07	N/A		u237	6.671		
1 yr	1.770E+04	N/A	2.020E+06	N/A		u238	0.260		
5 yr	1.220E+04	19.983	1.640E+06	8.875		np237	11.435		
10 yr	1.100E+04	18.698	1.356E+06	8.585		np239	46.662		
50 yr	6.348E+03	11.953	4.069E+05	6.594		pu238	12.869		
100 yr	4.433E+03	10.818	1.790E+05	8.118		pu239	4.924		
500 yr	1.765E+03	8.443	5.555E+04	8.215		pu240	5.482		
1000 yr	1.126E+03	7.500	3.599E+04	7.452		pu241	6.671		
2500 yr	6.560E+02	7.838	2.160E+04	7.966		pu242	6.429		
5000 yr	5.044E+02	7.979	1.669E+04	8.082		am241	7.244		
10000 yr	3.430E+02	7.500	1.134E+04	7.591		am242m	9.284		
	Inhalatio	n Hazard	Ingestion Hazard			am243	27.480		
	m³ air	+/- %	m³ water	+/- %		cm242	9.284		
Charge	1.487E+19	N/A	2.809E+12	N/A		cm244	19.265		
Discharge	2.265E+19	N/A	5.779E+12	N/A		cm245	15.308		
1 yr	1.866E+19	N/A	3.736E+12	N/A		FP (+/- %)			
5 yr	1.733E+19	16.097	3.442E+12	15.353		c 14	3.042		
10 yr	1.615E+19	15.176	3.188E+12	14.555		se 79	1.548		
50 yr	1.081E+19	12.097	2.092E+12	11.819		sr 90	0.410		
100 yr	8.227E+18	11.132	1.566E+12	11.042		tc 99	0.514		
500 yr	3.474E+18	7.866	6.533E+11	7.866		i129	0.854		
1000 yr	2.333E+18	7.262	4.385E+11	7.261		cs137	0.144		
2500 yr	1.475E+18	7.865	2.772E+11	7.861		ba137m	0.140		
5000 yr	1.150E+18	7.919	2.161E+11	7.915		y90	0.410		
10000 yr	7.816E+17	7.453	1.469E+11	7.449		cs134	2.966		
	Main Contribu	itors to Uncer	tainty in			k-effective Values			
	Decay Hea	t and Radioac	tivity			BOL	1.1731		
	He	eat	Activi	ity		EOL	1.1468		
5 yr	Cm-	-244	Cm-244, F	Pu-241					
10 yr	Cm-	-244	Cm-244, F	Pu-241					
F0	0 044	D., 000	Cm-244, Pu-						
50 yr		, Pu-238	241						
100 yr	 	Am-241	Pu-238, Am-241						
500 yr	Pu-240,	Am-241	Pu-240,Am-241 Pu-239, Pu-240, Am-						
1000 yr	Pu-239, Pu-2	240, Am-241	241						
2500 yr	Pu-239,	Pu-240	Pu-239, Pu-240						
5000 yr	Pu-239,	Pu-240	Pu-239, Pu-240						
10000 yr	 	Pu-240	Pu-239, Pu-240						

Table C.14: Results table for equilibrium recycled fast reactor fuel of CR = 0.70.

Results Ta	able for Re	cycled	FR, CR=0	.77, bu	rned 76.5 G		
	Uncertainty fo	Isotopic Mass Uncertainties					
	Heat Activity				Actindes (+/- %)		
	W	+/- %	Ci	+/- %	u235	0.809	
Charge	3.474E+04	N/A	1.806E+06	N/A	u236	2.034	
Discharge	3.871E+04	N/A	1.941E+06	N/A	u237		
1 yr	1.634E+04	N/A	1.288E+06	N/A	u238	0.950	
5 yr	9.725E+03	22.005	9.632E+05	15.819	np237	7.344	
10 yr	8.980E+03	20.959	7.979E+05	15.327	np239		
50 yr	5.699E+03	17.649	2.522E+05	13.526	pu238	18.616	
100 yr	4.192E+03	16.693	1.374E+05	15.399	pu239	1.336	
500 yr	1.616E+03	11.382	5.039E+04	11.044	pu240	7.463	
1000 yr	1.017E+03	9.549	3.194E+04	9.371	pu241	10.570	
2500 yr	5.961E+02	8.750	1.912E+04	8.745	pu242	18.769	
5000 yr	4.650E+02	8.556	1.498E+04	8.526	am241	10.163	
10000 yr	3.258E+02	7.269	1.050E+04	7.241	am242m	14.836	
	Inhalation I	Inhalation Hazard Inges		lazard	am243		
	m³ air	+/- %	m³ water	+/- %	cm242	8.856	
Charge	1.779E+19	N/A	3.348E+12	N/A	cm244	20.387	
Discharge	1.930E+19	N/A	3.631E+12	N/A	cm245	38.508	
1 yr	1.712E+19	N/A	3.230E+12	N/A	FP (+	-/- %)	
5 yr	1.575E+19	19.334	2.974E+12	19.369	c 14		
10 yr	1.482E+19	19.027	2.797E+12	19.062	se 79		
50 yr	1.040E+19	18.267	1.962E+12	18.306	sr 90		
100 yr	7.993E+18	16.936	1.506E+12	16.977	tc 99		
500 yr	3.182E+18	10.477	5.984E+11	10.479	i129		
1000 yr	2.109E+18	9.062	3.964E+11	9.061	cs137		
2500 yr	1.342E+18	8.907	2.522E+11	8.902	ba137m		
5000 yr	1.061E+18	8.624	1.995E+11	8.619	y90		
10000 yr	7.437E+17	7.322	1.398E+11	7.316	cs134		
Ma	ain Contributors	s to Unce	rtainty in		k-Effective Values		
	Decay Heat an	d Radioa	ctivity		вос	1.00638	
	Heat		Activity		EOC	0.99925	
5 yr	Pu-238, Cr	n-244		n-244, Pu-241 Conversion Ratio		Ratio at EOC	
10 yr		Cm-244, Pu-238		Cm-244, Pu-238, Pu- 241		695	
50 yr	Cm-244, P Am24		Am-241, Pu-238, Pu- 241				
100 yr	Pu-238, Ar	m-241	Pu-238, Ar	m-241			
500 yr	Pu-240, Ar	m-241	Pu-240,An	n-241			
1000 yr	Pu-240, Ar	m-241	Pu-240, Ar	m-241			
2500 yr	Pu-239, P	u-240	Pu-239, P	u-240			
5000 yr	Pu-239, P	u-240	Pu-239, P	u-240			
10000 yr	Pu-239, P	u-240	Pu-239, P	u-240			

Table C.15: Results table for REBUS equilibrium recycled fast reactor fuel.