

# Wind Turbine Blade Workshop

## SNL, DOE, NREL

24 Feb 04

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***Proof-of-Concept Manufacturing  
and Testing of Composite Wind  
Generator Blades Made by  
HCBMP (High Compression Bladder  
Molded Prepreg)***

**DE-FG36-03GO13140**

Alaska Applied Sciences, Inc. (AASI)

Prime subcontractor:

Quatro Composites, Poway, CA (QC)

# DE-FG36-03G013140

- Goal: Lower long-term COE
- Small wind turbines
- Low windspeed: Class 3
- 1 Oct 03 – 1 Oct 04
- Mfg proof-of-concept, not blade design or performance
- Lower cost, high quality blades
- \$ 118K = \$78K DOE + \$40K AASI
- New blade design: 12m diam;  
    Salter & Associates, via royalty
- Tooling design and fab: QC
- Blade fabrication: QC
- Field testing, San Geronio Pass, CA:  
    AASI windplant, short-term and long-term

# DE-FG36-03G013140

- Produce several sets 12m diam blades by HCBMP: QC prime sub
- Test at AASI Palm Springs windplant
  - Short term: dynamics, power curve, survival
  - Accelerated fatigue life test
  - Long term: survival, degradation
- Document tooling and production costs
- Develop cost model for HCBMP process
- Poster at Global Windpower 2004:  
Ken Gamble, QC principal, presenting

# AASI windplant, Palm Springs, CA

14 wind gen's, fixed speed, 50 kW drivetrain, downwind, free yaw, 26 m hub height





*Met tower*

Instrumented,  
unobstructed test  
site

AASI windplant,  
Palm Springs, CA

13m diam  
blade set installed

View to WNW



Met tower

***NOMAD datalogger,  
on wind generator  
control cabinet***

- Anem 1
- Anem 2
- Wind direction
- Wind slope
- kW (OSI precision xducer)
- State: on-line
- Tracking yaw error

# ***Preliminary cost model :***

**5.6 m LOA blade; 1000 per year**

- Tooling design and fab \$30 K
  - Aluminum mold to cure blade; 1 K blade life
  - Airspring press constrains and heats mold
- Amortization in first year = \$30 per blade



# ***Preliminary cost model :***

**5.6 m LOA blade; 1000 per year**

<i><b>Turbine diameter:</b></i>	<i><b>12m</b></i>	<i><b>13m</b></i>	<i><b>14m</b></i>
■ Carbon prepreg: 35 lbs @ \$12 / lb @ 12 m diam	\$ 420	\$ 500	\$ 650
■ Other material:			
■ LE mass	\$ 50	\$ 70	\$ 85
■ Paint, primer, supplies	\$ 60	\$ 70	\$ 85
■ Root box fill, SS washer plates	\$ 60	\$ 70	\$ 85
■ Prepreg cut, assemble, layup:			
■ 4 man-hours @ \$25 / hr (burdened) =	\$ 100	\$ 125	\$ 150
■ Demold, prep for paint:			
■ 2 man-hours @ \$25 / hr (" ) =	\$ 50	\$ 50	\$ 60
■ Prime and paint:			
■ 1 man-hour @ \$40 / hr (" ) =	\$ 40	\$ 40	\$ 50
■ Fill + drill root box, install washer plates:			
■ 2 man-hours @ \$25 / hr (" ) =	\$ 50	\$ 50	\$ 50
■ Balance:			
■ 1/2 man-hour @ \$25 / hr (" ) =	\$ 13	\$ 13	\$ 13
<b>TOTAL MATERIAL AND BURDENED LABOR</b>	<b>\$ 743</b>	<b>\$ 988</b>	<b>\$ 1,228</b>
<b>TOOLING AMORTIZATION</b>	<b>\$ 30</b>	<b>\$ 40</b>	<b>\$ 60</b>
<b>MARKUP @ 20% (on total M+L+TA)</b>	<b>\$ 150</b>	<b>\$ 198</b>	<b>\$ 250</b>
<b>SELLING PRICE</b>	<b>\$ 923</b>	<b>\$ 1,068</b>	<b>\$ 1,538</b>
<b>PRICE PER SQUARE METER</b>	<b>\$ 7.90</b>	<b>\$ 7.73</b>	<b>\$ 9.72</b>



## ***Tooling Design Team, with Tip Mold Sections, 20 Feb 04, at QC***

- Jason Nelson, QC, blade project manager (L)
- Ed Salter, S&A, blade shape design (C)
- Ken Gamble, QC, principal (R)

QC: Quatro Composites, Poway, CA

S&A: Salter and Associates, Oceanside, CA

***First article root section, thin-wall***

***Root box end: 2.25" x 7"***





***First article root section, thin-wall***



***First article  
root section  
mockup on  
turbine hub***

***Note 6-bolt  
pattern***

# *Blade Design Criteria*

- Low-cost, high-quality manufacturing
- Modest chord: mold and press cost
- Power curve: stall-regulated
- Startup in light winds
- LE dirt degradation relief
- NREL 822, 823 airfoils

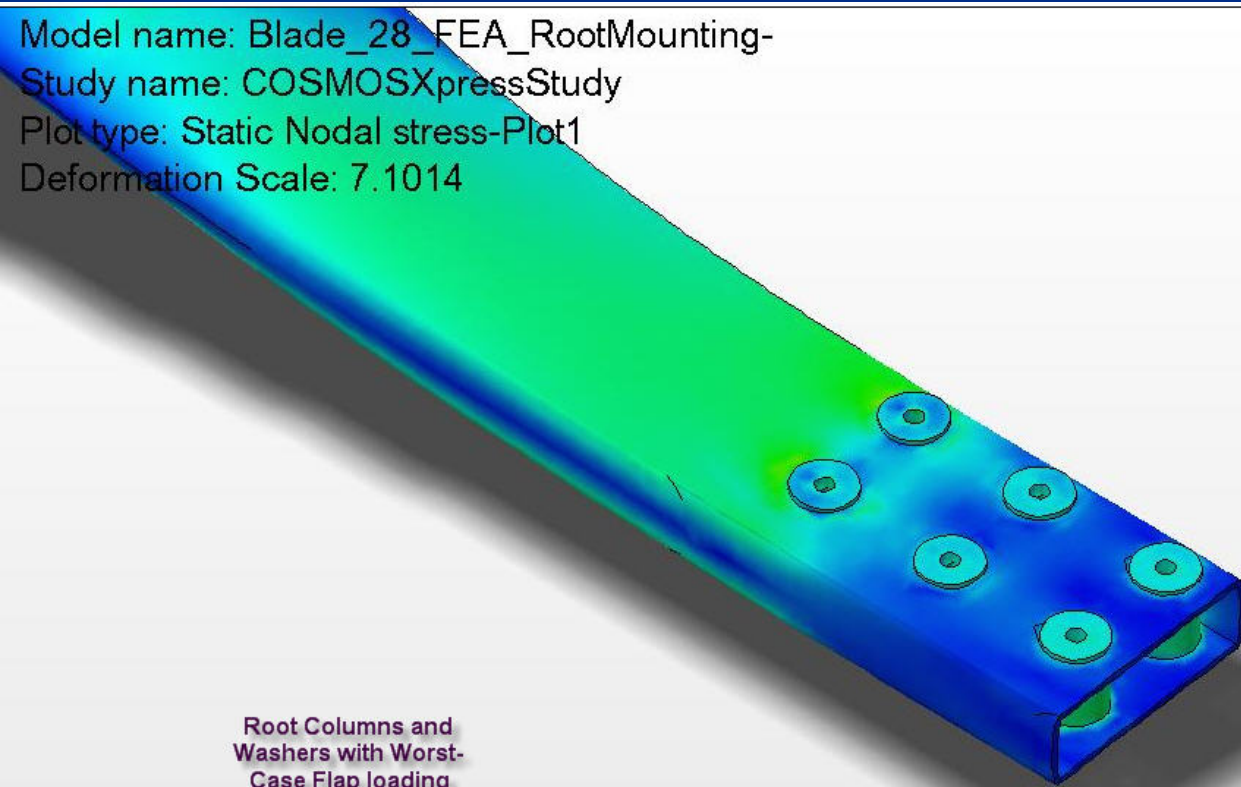
# *Blade Design*

- 12 m diameter, 130 rpm, AASI Palm Springs windplant
  - Fixed-speed
  - Downwind, 3-blade, free-yaw
  - Root box bolts to extant hub: 6 @ 1/2" bolts
  - 225" LOA
  - Chord 10" max, 9" for ~60% of span, 7" tip region
  - Preliminary wall thickness: .300" root, .065" tip region
  - 35 lbs prepreg, ~ 50-60 lbs total
- Salter & Associates: NREL 822, 823 airfoils
- Ideal flap stiffness
  - "soft" turbine
  - Centrifugally-stiffened
  - Avoid tower
- Flutter - free: stiffness, quarter-chord mass-balance

# *FEA analysis of root box stress COSMOS, in SolidWorks, by Salter & Associates*

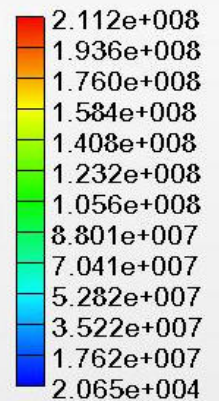
## *Case: Columns and washers*

Model name: Blade\_28\_FEA\_RootMounting-  
Study name: COSMOSXpressStudy  
Plot type: Static Nodal stress-Plot1  
Deformation Scale: 7.1014



Root Columns and  
Washers with Worst-  
Case Flap loading

von Mises (N/m<sup>2</sup>)

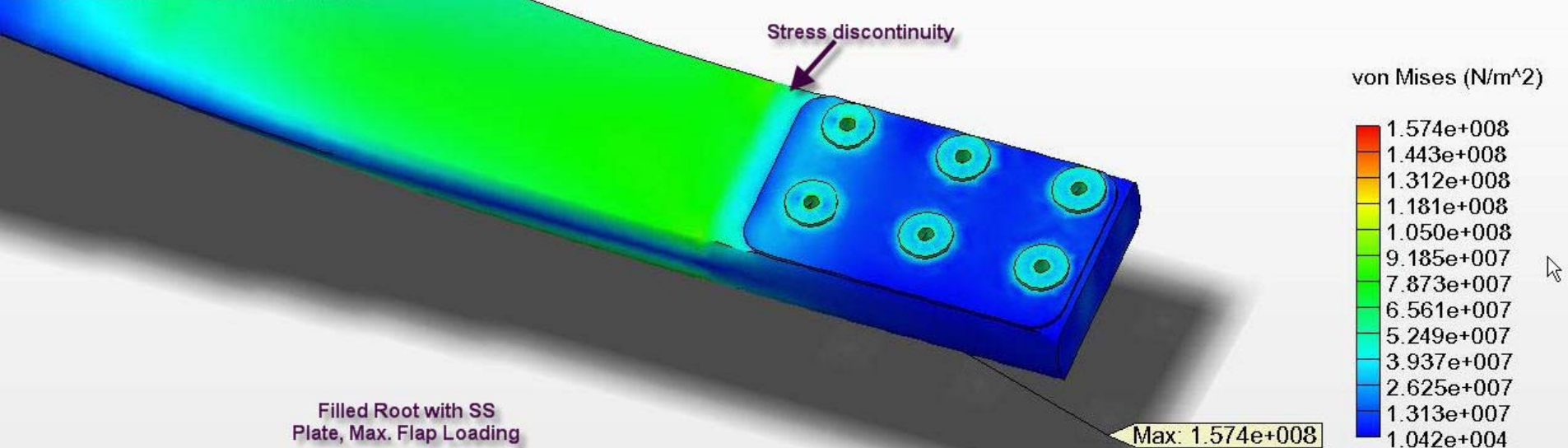




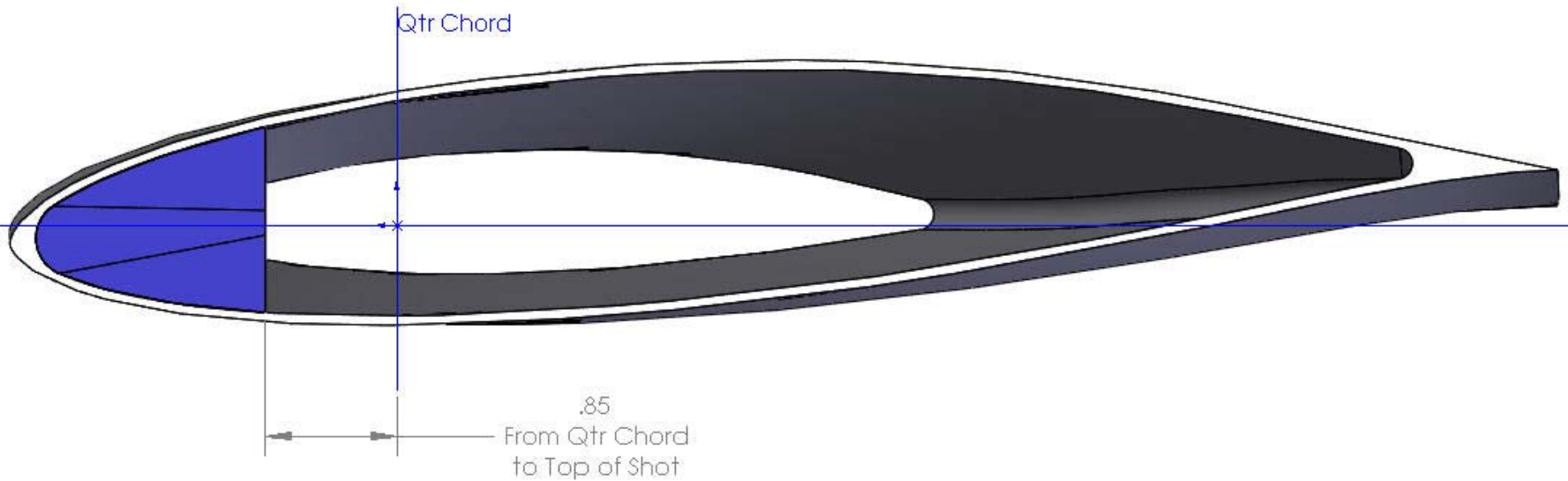
# *FEA analysis of root box stress COSMOS, in SolidWorks, by Salter & Associates*

## *Case: solid fill with washer plates*

Model name: Blade\_28\_FEA\_RootMounting-  
Study name: COSMOSXpressStudy  
Plot type: Static Nodal stress-Plot1  
Deformation Scale: 8.59484



***Tip region mass balance ahead of quarter-  
chord point***  
***Blue: Leading Edge D-Spar (LEDS),  
steel shot + epoxy***





***Flap and torsional  
stiffness test fixture***



# ***Torsional Stiffness Test***

# *Incentives for this project*

- Class 3 wind: higher ratio of turbine swept area to drive train power rating
- Blades larger fraction of total wind turbine system capital cost
- Emphasize low-cost, high-quality blade production methods
- Need high quality composite for long blade service life
- Requires proof-of-concept blade fabrication
  - + extensive field testing
- Capitalize on recent advances:
  - Solid modeling CAD – CAM software tools
  - CNC machining facility, lower cost
  - HCBMP process
  - Prepreg availability, cost: carbon and glass “tape”

# ***HC BMP advantages***

- No liquids
- Good process control
- Modest mfg skill level; die cut plies, layup
- Controlled high fiber : resin ratio
- High quality:
  - High compaction, minimum voids
  - Little variation in parts
  - Seamless blade
- Short production cycle time per part
- Easy to build entirely of "tape "
- Modular tooling: easy to modify, move, store
- Easy to vary lamination schedule
- High HDT: cure at 285 F
- Foam filling optional

# ***HC BMP limitations***

- Hollow, single-cavity part
- Limited structural enhancements
- Root attachment finished post-molding
- Foam filling required ?
- Limited size ? Blade length 10 – 15 m ?

# *Prepreg*

- Fiber pre-impregnated with catalyzed resin; dry, slightly sticky to touch (see sample)
- Activated at automated 250 F cure; freezer storage; room temp working
- Well-controlled fiber : resin ratio
- Wide variety material available:
  - Fiber-resin ratio
  - Tape, stitched, woven; tape lowest-cost
  - Carbon, glass, Kevlar
  - Roll width standard
- Assemble lamination schedule dry, at room temp:
  - Cut + stack tape: ply orientation
  - Vacuum consolidate, de-air
  - Die cut
  - No liquids
  - Minimum material waste



# *Blade quality*

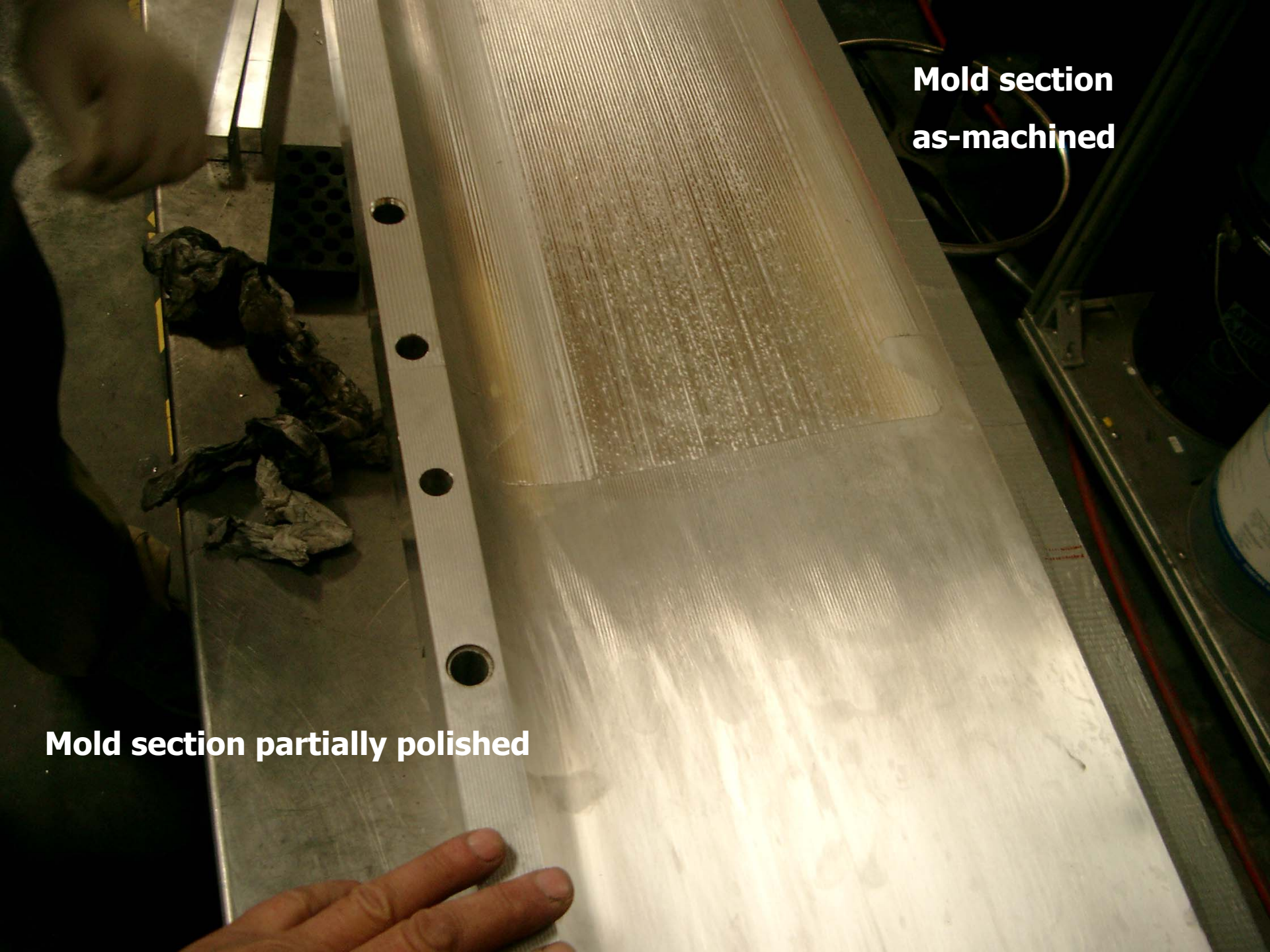
- Process control
- Uniform properties, weight (balancing)
- Freedom from defects, voids
  - Macro
  - Micro
- Infinite fatigue life
- SR+T expertise: adjunct to contract
  - SNL, NREL, NWTC
  - Blade design review
  - NDE, ultimate strength, failure modes
  - Test to failure; failure modes
- AASI windplant accelerated lifetime testing

**CNC machining mold section**



**Mold section  
as-machined**

**Mold section partially polished**





Mold  
Sanding  
and  
Polishing

*Mold sections showing dovetail joints*





***Blade Mold  
Sections, QC,  
20 Feb 04***

JiJoe, CNC  
Manager, QC (L)

Ken Gamble,  
Principal, QC (R)



**Prepreg cutting**



***Complete Mold Section Half  
(right)***

***Mold Segments (left)***

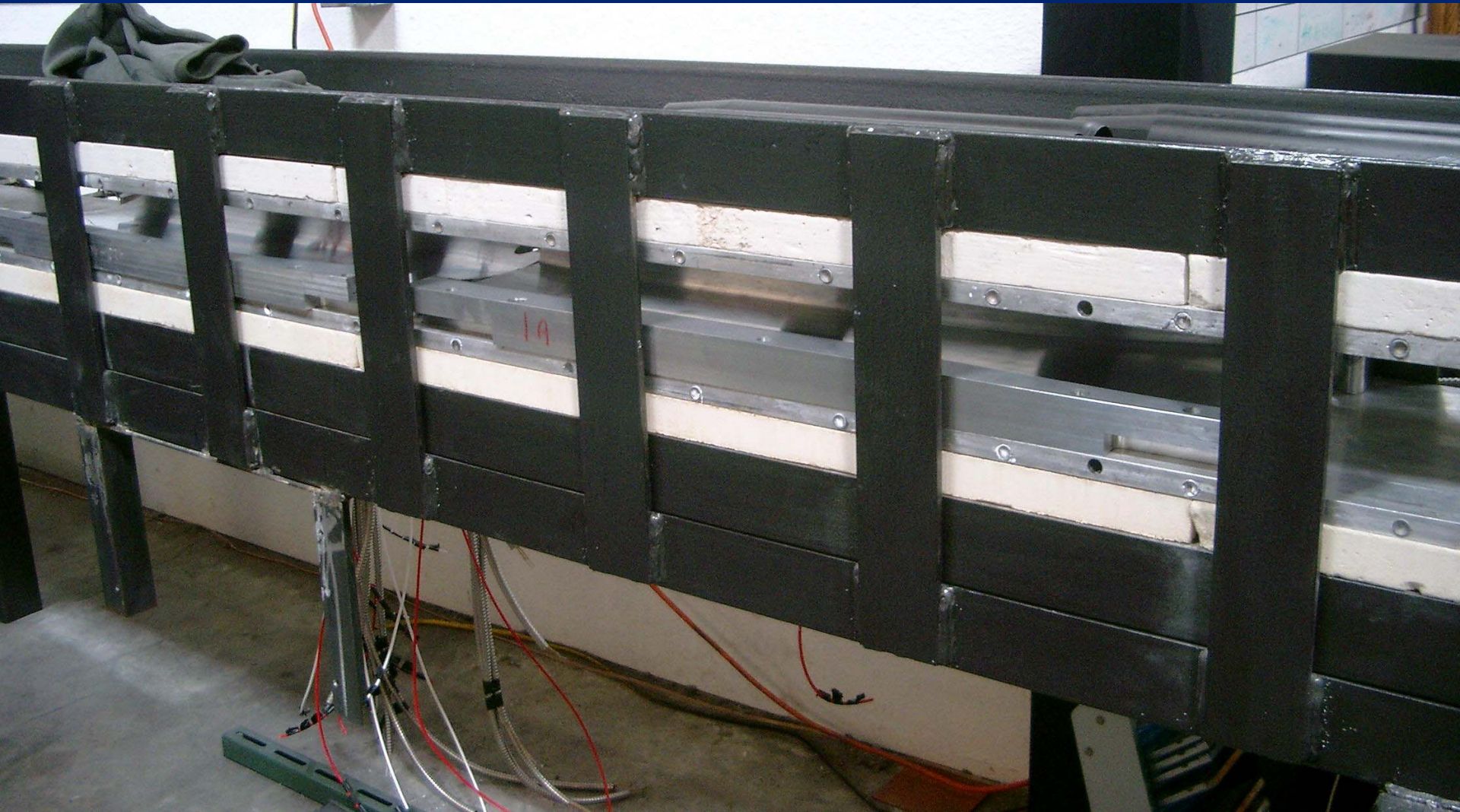
***QC, 20 Feb 04***



***Closed mold rolls into the  
airspring press, at rear***



# *Heated airspring press*



# *Tentative project schedule Rev 20 Feb 04*

- Tooling completed: mold, airspring press, at QC
- First article blade mold, prep for test: 23-26 Feb
- Properties test; root box fill + drill develop;  
    revise design (lam schedule and LE mass): 27-29 Feb
- Second article blade mold, prep for test: 2 Mar
- Properties test; approve design (lam schedule and LE mass)  
    for first blade set: 2-4 Mar
- Mold blade set 1; document time + material: 5-10 Mar
- Finish blade set 1: 10-13 Mar
- SR+T blade (SNL?): \_\_\_\_\_
- Install set 1 at Palm Springs test site: 14-16 Mar
- Test fly set 1; power curve; video; accel fatigue test: 16-30 Mar
- Revise design, if necessary: 1-5 Apr
- Build, install, test fly set 2: 6-15 Apr
- Build, install \_\_\_ other test sets: Apr-May
- Long-term monitoring and reporting to DOE (NREL, NWTC, SNL)

## Optional HCBMP concept

Test specimen: HCBMP with spanwise-sawed pultruded core  
GO798 airfoil; 19 cm chord



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