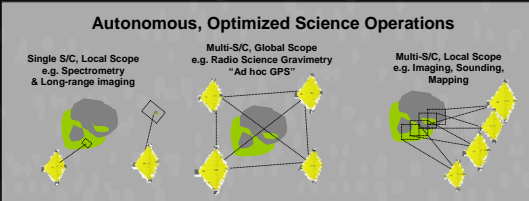


Science Specialists and Measurements		
Instrument	Measurements	Modes of Study
Magnetometer	magnetic field	Interior structure and origin
All spectrometers	compositional variations	
Visible Imager	Photogeology, stratigraphy	
Thermal/Radio Sounder	regolith particle properties	Physical and compositional characterization of regolith components
UV/Vis/IR	mineral assemblages	
Neutron/Gamma-ray	volatiles, organics	
X-ray/Gamma-ray	major, refractory elements	km to m scale physical/Dynamic
Imager/Ranger/Radio Science	Morphology, gravity, spin	
Visible Imager	stereo imaging, triangulation	

PAM Requirements and Capabilities

Requirements for structure and size of teams and subswarms are driven by large distances between dynamic small, irregular targets making detecting, locating, modeling the shape, and orbiting are challenges requiring large multi-specialist teams.

- 9 science specialist classes include imagers, ranging, and radio science, magnetometers, and a range of spectrometers.
- An additional class of Messenger/Leaders specializes in strategic planning and goal assessment as well as communication and navigation, as required. A Messengers will be required to return to Earth with data for every ten or more asteroids surveyed.
- Start with subswarm size of 50, 10 in each of 10 classes
- Anticipated attrition 90% per year for 5 years through asteroid belt, leaving less than one in each class, subswarms would need to combine periodically.
- Viewing requirements and capabilities of specialist teams translate into capability to explore an asteroid in 1 month.
- Propulsion system, distances between asteroids >1 km in size translate into requirement for 1 month travel between asteroids.
- Use of 10 to 20 subswarms should allow exploration of 50 to 100 asteroids per year, ~5 for each subswarm, and hundreds of asteroids over the course of the mission.



SMART deployable spacecraft

Uses SMART Sail, Subsystem Platform, Tethers
 Can be multiple tethered platforms
 Optimized for pre and post deployment ops
 Pre-deployment size much smaller than post

SMART deployable subsystems

SMART Space Frame
 Deployable Tetrahedral 3D Truss Structure
 Communication either thin wire or wireless
 Nodes (MEMS) spool/unspool nanotubules which form tethers, struts, fibers or act as attach points for subsystems

SMART spacecraft pre-deployment shell

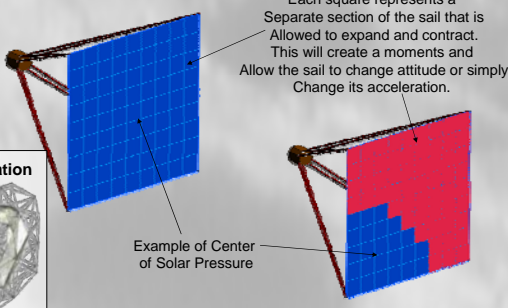
Moves SMART spacecraft to deployment position
 Totally autonomous
 Cubic Pre-deployment: forms cubic array
 Sheet Post-deployment: forms flat array

SMART Solar Sail characteristics

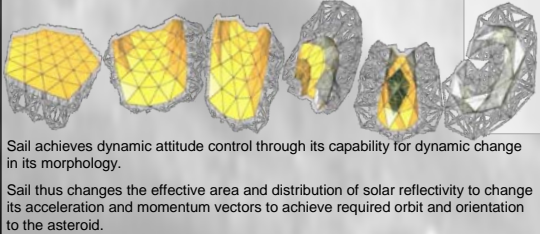
10 to 100 times linear stretch from multilayer dendritic polymers
 Polymers consist of layers of Slinky-like helical nanotubule chains
 Multilayer fabric has sufficient reflectivity when fully extended
 Self-configuring morphology for attitude control or adaptation
 Self-deploying surface and struts for attitude control or repair



PAM Attitude Control: Using SMART Solar Sail Self-Deployment



PAM Attitude Control: Using SMART Solar Sail Self-Configuration



Conclusions

- ANTS approach allows simultaneous survey of many interesting targets by teams of flexible instrument specialists able to operate at optimal locations at the target.
- Redundancy allows for planned attrition and the return of communications specialists, or messengers, without compromising ongoing prospecting.
- Reversibly deployable gossamer structures minimize weight and power requirements, and are adaptable and reconfigurable for environmental changes and swarm needs.
- The architecture is based on already anticipated incremental advances in technology.
- ANTS are suitable to low gravity space or surface environment applications, such as the surfaces of asteroids, the Moon, or Mars.

