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Bank: (Aviation Mechanic Airframe) Airman Knowledge Test Question Bank

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1. AMA021 AMA

The I beam wooden spar is routed to

- A) increase strength.
- B) obtain uniform strength.
- C) reduce weight.

2. AMA102 AMA

Glue deterioration in wood aircraft structure is indicated

- A) when a joint has separated and the glue surface shows only the imprint of the wood with no wood fibers clinging to the glue.
- B) when a joint has separated and the glue surface shows pieces of wood and/or wood fibers clinging to the glue.
- C) by any joint separation.

3. AMA102 AMA

Which of the following conditions will determine acceptance of wood with mineral streaks?

- A) Careful inspection fails to reveal any decay.
- B) They produce only a small effect on grain direction.
- C) Local irregularities do not exceed limitations specified for spiral and diagonal grain.

4. AMA102 AMA

Laminated wood is sometimes used in the construction of highly stressed aircraft components. This wood can be identified by its

- A) parallel grain construction.
- B) similarity to standard plywood construction.
- C) perpendicular grain construction.

5. AMA021 AMA

The cantilever wing uses

A) external	struts or wire bracing.	
B) no exterr	nal bracing.	
C) the skin	to carry most of the load to the wir	ng butt.
6.	AMA102	AMA
In cases of	elongated boltholes in a wood spa	ar or cracks in the vicinity of boltholes,
A) it is perm	issible to ream the hole, plug with	hardwood, and redrill.
B) the spar	may be reinforced by using hardw	ood reinforcing plates.
C) a new se	ction of spar should be spliced in	or the spar replaced entirely.
7.	AMA102	AMA
A) bearing s B) glue.	n of a well designed and properly surface of the wood fibers. ment plates.	prepared wood splice joint is provided by the
8.	AMA040	AMA
(2) A plain la Regarding t	apped seam is never permissible. he above statements, 1 and No. 2 are true. 1 is true.	abrics may be of the folded fell or French fell types.
9.	AMA040	AMA
of the follow 1. Aluminun 2. Resin imp	ring provide an acceptable protect oregnated cloth tape. part type metal primer.	tructure that come in contact with doped fabric, which ive coating?
10.	AMA040	AMA
The strengt A) bearing s	n classification of fabrics used in a strength.	ircraft covering is based on

B) shear strengt	h.	
C) tensile streng	th.	
11.	AMA040	AMA
Finishing tape (s	surface tape) is used for wha	t purpose?
A) To help preve	ent 'ripple formation' in cover	ing fabric.
B) To provide ac	Iditional anti-tear resistance	under reinforcement tape.
C) To provide ad	dditional wear resistance ove	er the edges of fabric forming structures.
12.	AMA040	AMA
_	factor(s) for the selection of e of aircraft is the	the correct weight of textile fabric to be used in
A) maximum wir	ng loading.	
B) speed of the	aircraft.	
C) speed of the	aircraft and the maximum wi	ng loading.
13.	AMA040	AMA
Before applying	a protective coating to any u	npainted clean aluminum, you should
A) wipe the surfa	ace with avgas or kerosene.	
B) remove any c	conversion coating film.	
C) avoid touchin	g the surface with bare hand	ls.
14.	AMA040	AMA
•	occur if unhydrated wash pr s later a finish topcoat, when	mer is applied to unpainted aluminum and then about the humidity is low?
A) Corrosion.		
B) A glossy, blus	sh-free finish.	
C) A dull finish d	lue to the topcoat 'sinking in'	to primer that is still too soft.
15.	AMA088	AMA
~	mbers are to be applied to a required for the registration	n aircraft with a letter height of 12 inches, what is the mark N1683C?
Note:		
$2/3 \times \text{height} = \text{ch}$	naracter width.	
1/6 x height = wi	dth for 1.	
1/4 x 2/3 height	= spacing.	
$1/6 \times height = st$	roke or line width.	
A) 52 inches.		

B) 48 inches.		
C) 57 inches.		
16.	AMA040	AMA
Which type of coapplication?	pating typically includes phosph	oric acid as one of its components at the time of
A) Wash primer.		
B) Epoxy primer		
C) Zinc chromate	e primer.	
17.	AMA040	AMA
Aluminum-pigme	ent in dope is used primarily to	
A) provide a silve	er color.	
B) aid in sealing	out moisture from the fabric.	
C) reflect ultravio	olet from the fabric.	
18.	AMA040	AMA
Which defect in a temperature?	aircraft finishes may be caused	by adverse humidity, drafts, or sudden changes in
A) Orange peel.		
B) Blushing.		
C) Pinholes.		
19.	AMA094	AMA
	to compute a bend allowance putral axis of the bend can be	problem and bend allowance tables are not
A) represented b	by the actual length of the requi	ed material for the bend.
B) found by addi	ing approximately one half of th	e stock thickness to the bend radius.
C) found by subt	tracting the stock thickness from	n the bend radius.
20.	AMA094	AMA
The sharpest be called the	nd that can be placed in a piece	e of metal without critically weakening the part is
A) bend allowan	ce.	
B) minimum radi	ius of bend.	
C) maximum rad	lius of bend.	
21.	AMA094	AMA
A piece of sheet	metal is bent to a certain radiu	s. The curvature of the bend is referred to as the

A) bend allow	vance.	
B) neutral line	e.	
C) bend radiu	JS.	
22.	AMA094	AMA
The purpose	of a joggle is to	
A) allow clea	rance for a sheet or an extrusion.	
B) increase o	bstruction for a sheet or an extrusion.	
C) decrease	the weight of the part and still retain th	e necessary strength.
23.	AMA094	AMA
(Refer to Airf Setback = .25	rame figure 6.) Determine the dimensi	ons of A, B, and C in the flat layout.
Bend allowar		
	B = 2.252; C = 2.004.	
•	B = 1.496; C = 1.248.	
•	; B = 2.504; C = 1.752.	
24.	AMA094	AMA
The sight line marked	e on a sheet metal flat layout to be ben	t in a cornice or box brake is measured and
A) one-half ra	adius from either bend tangent line.	
B) one radius	from either bend tangent line.	
C) one radius	s from the bend tangent line that is pla	ced under the brake.
25.	AMA094	AMA
On a sheet m	netal fitting layout with a single bend, a	llow for stretching by
A) adding the	e setback to each leg.	
B) subtracting	g the setback from one leg.	
C) subtracting	g the setback from both legs.	
26.	AMA094	AMA
(Refer to Airf	rame figure 4.) The length of flat A is	
A) 3.750 inch	es.	
B) 3.875 inch	es.	
C) 3.937 inch	nes.	
27.	AMA094	AMA

(Refer to Airframe A) 7.0 inches. B) 6.8 inches. C) 6.6 inches.	figure 5.) What is the flat layout dimension?	
28.	AMA016	AMA
You can distinguis A) filing the metal.	h between aluminum and aluminum alloy by	
B) testing with an	acetic acid solution.	
C) testing with a 1	0 percent solution of caustic soda.	
29.	AMA037	AMA
	a composite panel using the ring test/tapping me rength curing of the matrix. ne laminates.	thod, a dull thud may indicate
C) an area of too i	much matrix between fiber layers.	
30.	AMA037	AMA
One of the best was	ays to assure that a properly prepared batch of i	matrix resin has been achieved is
A) perform a chem	nical composition analysis.	
3) have mixed end	ough for a test sample.	
C) test the viscosi	ty of the resin immediately after mixing.	
31.	AMA004	AMA
•	materials installed in current standard category a	airplanes must
A) be fireproof.		
B) be at least flam		
o) meet the requir	rements prescribed in Part 43.	
32.	AMA078	AMA
f no scratches are surfaces should be	e visible after transparent plastic enclosure mate e	rials have been cleaned, their
A) polished with ru	ubbing compound applied with a damp cloth.	
3) buffed with a cl	ean, soft, dry cloth.	
C) covered with a	thin coat of wax.	
33.	AMA078	AMA

•	ansparent plastic enclosures which are retained nd self-locking nuts, the nuts should be	by bolts extending through the
•	firm fit, plus one full turn.	
. •	firm fit, then backed off one full turn.	
C) tightened to a		
-,g		
34.	AMA021	AMA
Which part(s) of a fuselage?	a semi monocoque fuselage prevent(s) tension a	nd compression from bending the
A) The fuselage o	covering.	
B) Longerons and	d stringers.	
C) Bulkheads and	d skin.	
35.	AMA016	AMA
When straighteni	ng members made of 2024-T4, you should	
A) straighten cold	-	
,	I and anneal to remove stress.	
,	he inside of the bend.	
o) apply fleat to t	The mode of the bend.	
36.	AMA016	AMA
Aircraft structural metal, are norma	units, such as spars, engine supports, etc., which	ch have been built up from sheet
A) repairable, usi	ng approved methods.	
B) repairable, exc	cept when subjected to compressive loads.	
C) not repairable,	but must be replaced when damaged or deterio	rated.
07	AMA 020	Λ N A Λ
37.	AMA029	AMA
	he included angle of a twist drill for soft metals?	
A) 118°.		
B) 90°.		
C) 65°.		
38.	AMA016	AMA
Shallow scratche	s in sheet metal may be repaired by	
A) burnishing.		
B) buffing.		
C) stop drilling.		
39.	AMA094	AMA

• .	•	o repair a section of damaged aluminum skin. If a double um allowable overlap will be
A) 1/2 inch.		
B) 3/4 inch.		
C) 13/16 inch.		
40.	AMA058	AMA
		pable of softening or flowing when reheated is described as a
A) thermoplas		
B) thermocure).	
C) thermoset.		
41.	AMA037	AMA
The strength a	and stiffness of a properly	constructed composite buildup depends primarily on
A) a 60 percer	nt matrix to 40 percent fibe	er ratio.
B) the orientat	ion of the plies to the load	I direction.
C) the ability of	of the fibers to transfer stre	ess to the matrix.
42.	AMA037	AMA
Sandwich pan type of constru	•	comb construction are used on modern aircraft because this
A) is lighter the	an single sheet skin of the	e same strength and is more corrosion resistant.
B) may be rep	aired by gluing replaceme	ent skin to the inner core material with thermoplastic resin.
C) has a high	strength to weight ratio.	
43.	AMA037	AMA
What is the maduring of		ne vacuum bag pressure system to absorb excess resin
A) Bleeder.		
B) Breather.		
C) Release.		
44.	AMA037	AMA
Which of the f	ollowing are generally cha	aracteristic of carbon/graphite fiber composites?
1. Flexibility.		
2. Stiffness.		
3. High compr	essive strength.	
4. Corrosive e	ffect in contact with alumi	num.

5. Ability to c A) 1 and 3. B) 2, 3, and 4 C) 1, 3, and 8		
45. A potted com	AMA037 npound repair on honeycom	AMA b can usually be made on damages less than
A) 4 inches i	n diameter.	
B) 2 inches i	n diameter.	
C) 1 inch in d	diameter.	
46.	AMA037	AMA
		g is generally accomplished by
	external heat.	
-	perature exposure.	
_	catalyst or curing agent to the	e resin.
4. applying p	ressure.	
A) 2 and 3.		
B) 1 and 4.	4	
C) 1, 3, and 4	4.	
47.	AMA037	AMA
-	ng puncture type damage of should be tapered to	of a metal faced laminated honeycomb panel, the edges of
A) two times	the thickness of the metal.	
B) 100 times	the thickness of the metal.	
C) whatever	is desired for a neat, clean	appearance.
48.	AMA078	AMA
What is the r	nost common method of ce	menting transparent plastics?
A) Heat meth	nod.	
B) Soak met	hod.	
C) Bevel met	thod.	
49.	AMA045	AMA
	g repairs to fiberglass, clea aning should be made using	ning of the area to be repaired is essential for a good bond.
	thyl ethyl ketone).	

B) soap, water, a	and a scrub brush.	
C) a thixotropic a	agent.	
50.	AMA045	AMA
Fiberglass lamin	ate damage not exceeding the f	rst layer or ply can be repaired by
A) filling with a p	utty consisting of a compatible r	esin and clean, short glass fibers.
B) sanding the d	amaged area until aerodynamic	smoothness is obtained.
C) trimming the r	ough edges and sealing with pa	int.
51.	AMA037	AMA
Fiberglass lamina A) cannot be rep		etely through one facing and into the core
B) requires the re	eplacement of the damaged cor	e and facing.
C) can be repaire	ed by using a typical metal facin	g patch.
52.	AMA037	AMA
	owing, when added to wet resing composite panels?	s, provide strength for the repair of damaged
1. Microballoons		
2. Flox.		
3. Chopped fiber	S.	
A) 2 and 3.		
B) 1 and 3.		
C) 1, 2, and 3.		
53.	AMA037	AMA
Which of these n water?	nethods may be used to inspect	fiberglass/honeycomb structures for entrapped
1. Acoustic emis	sion monitoring.	
2. X-ray.		
3. Backlighting.		
A) 1 and 2.		
B) 1 and 3.		
C) 2 and 3.		
54.	AMA017	AMA
Which rivets sho	uld be selected to join two shee	ts of .032-inch aluminum?
A) MS20425D-4-	•	

B) MS20470AD-	4-4.	
C) MS20455DD	-5-3.	
55.	AMA017	AMA
		ered when determining minimum rivet spacing?
A) Rivet diamete	-	
B) Rivet length.		
C) Type of mate	rial being riveted.	
56.	AMA029	AMA
Which procedure	e is correct when using a r	eamer to finish a drilled hole to the correct size?
A) Turn the rean remove from the	_	when enlarging the hole and in the opposite direction to
B) Turn the rean	ner only in the cutting direc	ction.
C) Apply conside finishing the cut.		mer when starting the cut and reduce the pressure when
57.	AMA016	AMA
		wo pieces of 0.0625-inch aluminum riveted together. All ne length of the rivets to be used will be
A) 5/32 inch.		
B) 3/16 inch.		
C) 5/16 inch.		
58.	AMA016	AMA
Rivet pitch is the	e distance between the	
A) centers of rive	ets in adjacent rows.	
B) centers of adj	acent rivets in the same ro	DW.
C) heads of rive	ts in the same row.	
59.	AMA020	AMA
(Refer to Airfram made by a 100°		ivets shown will accurately fit the conical depression
A) 1.		
B) 2.		
C) 3.		
60.	AMA017	AMA

	t to be chosen when making a structural repair the ch aluminum sheet, drilled with a No. 30 drill, is	hat involves the joining of 0.032-
A) 7/16 inch.		
3) 5/16 inch.		
C) 1/4 inch.		
61.	AMA094	AMA
What is the minim	num edge distance for aircraft rivets?	
A) Two times the	diameter of the rivet shank.	
•	diameter of the rivet head.	
•	e diameter of the rivet shank.	
62.	AMA016	AMA
What is the minim	num spacing for a single row of aircraft rivets?	
A) Two times the	diameter of the rivet shank.	
B) Three times the	e length of the rivet shank.	
C) Three times th	e diameter of the rivet shank.	
63.	AMA017	AMA
	ditions, type A rivets are not used because of th	eir
A) low strength ch	naracteristics.	
high alloy conte		
C) tendency towa	rd embrittlement when subjected to vibration.	
64.	AMA017	AMA
The dimensions o	f an MS20430AD-4-8 rivet are	
A) 1/8 inch in diar	neter and 1/4 inch long.	
3) 1/8 inch in diar	neter and 1/2 inch long.	
C) 4/16 inch in dia	ameter and 8/32 inch long.	
65.	AMA016	AMA
Refer to Airframe	figure 2.) Select the preferred drawing for prope	er countersinking.
` Α) All are accepta		G
B) 2.		
C) 1.		
66.	AMA017	AMA

	al repair is to be made using two piece re drilled for 3/32-inch rivets. The lengt	s of 0.040-inch aluminum riveted together. All h of the rivets to be used will be
A) 1/8 inch.		
B) 1/4 inch.		
C) 5/16 inch.		
67.	AMA017	AMA
MS20426AD	0-6-5 indicates a countersunk rivet which	ch has
A) a shank le	ength of 5/16 inch (excluding head).	
B) a shank le	ength of 5/32 inch (excluding head).	
C) an overall	l length of 5/16 inch.	
68.	AMA017	AMA
	rivets in the D and DD series that are removal from refrigeration	not driven within the prescribed time after heat
A) must be re	eheat treated before use.	
B) must be d	liscarded.	
C) may be re	eturned to refrigeration and used later	without reheat treatment.
69.	AMA017	AMA
A factor which	ch determines the minimum space bety	veen rivets is the
A) length of t	the rivets being used.	
B) diameter	of the rivets being used.	
C) thickness	of the material being riveted.	
70.	AMA020	AMA
Which rivet is	s used for riveting nickel steel alloys?	
A) 2024 alun	ninum.	
B) Mild steel	•	
C) Monel.		
71.	AMA017	AMA
A DD rivet is	heat treated before use to	
A) harden ar	nd increase strength.	
B) relieve int	ernal stresses.	
C) soften to	facilitate riveting.	
72.	AMA017	AMA

The length of equal to	a rivet to be used to join a sheet of .032-inch	and .064-inch aluminum alloy should be
A) two times t	he rivet diameter plus .064 inch.	
B) one and or	ne half times the rivet diameter plus .096 inch.	
C) three times	s the rivet diameter plus .096 inch.	
73.	AMA016	AMA
Joggles in ren	noved rivet shanks would indicate partial	
A) bearing fail		
B) torsion failu		
C) shear failui	re.	
74.	AMA011	AMA
A main differe application) is	nce between Lockbolt/ Huckbolt tension and sin the	shear fasteners (other than their
A) number of	locking collar grooves.	
B) shape of th		
C) method of	installation.	
75.	AMA011	AMA
One of the ma	ain advantages of Hi-Lok type fasteners over e	earlier generations is that
A) they can be	e removed and reused again.	
B) the squeez	ed on collar installation provides a more secu	re, tighter fit.
C) they can be	e installed with ordinary hand tools.	
76.	AMA011	AMA
The Dzus turn measured in	llock fastener consists of a stud, grommet, an	d receptacle. The stud diameter is
A) tenths of a	n inch.	
B) hundredths	s of an inch.	
C) sixteenths	of an inch.	
77.	AMA017	AMA
Hole filling fas primarily beca	steners (for example, MS20470 rivets) should use of the	not be used in composite structures
A) possibility (of causing delamination.	
B) increased բ	possibility of fretting corrosion in the fastener.	
C) difficulty in	forming a proper shop head.	

78.	AMA011	AMA
The Dzus turr measured in	nlock fastener consists of a stud, grommet, and rec	ceptacle. The stud length is
A) hundredths	s of an inch.	
B) tenths of a	n inch.	
C) sixteenths	of an inch.	
79.	AMA017	AMA
The general r	ule for finding the proper rivet diameter is	
A) three times	s the thickness of the materials to be joined.	
B) two times t	he rivet length.	
C) three times	s the thickness of the thickest sheet.	
80.	AMA094	AMA
What method	of repair is recommended for a steel tube longeror	n dented at a cluster?
A) Welded sp	lit sleeve.	
B) Welded ou	ter sleeve.	
C) Welded pa	tch plate.	
81.	AMA101	AMA
The oxyacety	lene flame for silver soldering should be	
A) oxidizing.		
B) neutral.		
C) carburizing	J.	
82.	AMA101	AMA
Why should a	carburizing flame be avoided when welding steel?	
A) It removes	the carbon content.	
B) It hardens	the surface.	
C) A cold weld	d will result.	
83.	AMA101	AMA
The oxyacety	lene flame used for aluminum welding should	
A) be neutral	and soft.	
B) be slightly	oxidizing.	
C) contain an	excess of acetylene and leave the tip at a relative	ly low speed.
84.	AMA101	AMA

	ering iron cannot be used effectively until after the	ne working face has been
A) fluxed.		
B) polished.		
C) tinned.		
85.	AMA101	AMA
Welding over braz	zed or soldered joints is	
A) not permitted.		
B) permissible for	mild steel.	
C) permissible for	most metals or alloys that are not heat treated.	
86.	AMA101	AMA
Oxygen and acety	lene cylinders are made of	
A) seamless alum	inum.	
B) steel.		
C) bronze.		
87.	AMA058	AMA
Annealing of alum	ninum	
A) increases the to	ensile strength.	
B) makes the mat	erial brittle.	
C) removes stress	ses caused by forming.	
88.	AMA101	AMA
Edge notching is of the cause it	generally recommended in butt welding above a	certain thickness of aluminum
A) helps hold the	metal in alignment during welding.	
B) aids in the rem	oval or penetration of oxides on the metal surfac	e.
C) aids in getting	full penetration of the metal and prevents local d	istortion.
89.	AMA101	AMA
Where should the	flux be applied when oxyacetylene welding alun	ninum?
A) Painted only or	n the surface to be welded.	
B) Painted on the	surface to be welded and applied to the welding	rod.
C) Applied only to	the welding rod.	
90.	AMA101	AMA
Why are aluminun	n plates 1/4 inch or more thick usually preheated	d before welding?
•	• •	-

A) Reduce	s internal stresses and assures	more complete penetration.
B) Reduce	s welding time.	
C) Prevent	s corrosion and ensures proper	distribution of flux.
91.	AMA101	AMA
Oxides formalum to		etals are hot. It is important, therefore, when welding
A) solvent.		
B) filler.		
C) flux.		
92.	AMA101	AMA
When a bu	tt welded joint is visually inspec	ted for penetration,
A) the pend	etration should be 25 to 50 perc	ent of the thickness of the base metal.
B) the pend	etration should be 100 percent o	of the thickness of the base metal.
C) look for	evidence of excessive heat in the	ne form of a very high bead.
93.	AMA101	AMA
Which state	ement best describes magnesiu	m welding?
A) Magnes	ium can be welded to other met	als.
B) Filler roo	d should be nickel steel.	
C) Filler ro	d should be the same compositi	on as base metal.
94.	AMA038	AMA
	_	ormally used to ensure that the correct amount of waged-type terminals on aircraft control cable?
A) Measure	e the finished length of the term	inal barrel and compare with the beginning length.
B) Use a te	erminal gauge to check the diam	neter of the swaged portion of the terminal.
C) Check t incomplete	•	n of the terminal for small cracks which indicate
95.	AMA038	AMA
	ctions issued by the swaging too se resultant swaged terminal str	ol manufacturer are followed when swaging a cable ength should be
A) the full r	ated strength of the cable.	
B) 80 perce	ent of the full rated strength of the	ne cable.
C) 70 perce	ent of the full rated strength of the	ne cable.
96.	AMA001	AMA

The purpose of th	ne vertical fin is to provide	
A) directional stat	oility.	
B) longitudinal sta	ability.	
C) lateral stability	•	
97.	AMA011	AMA
When used in clo	se proximity to magnetic comp	asses, cotter pins are made of what material?
A) Corrosion resis	sting steel.	
B) Anodized alum	ninum alloy.	
C) Cadmium plate	ed low carbon steel.	
98.	AMA081	AMA
•	s to a control surface require s	tatic rebalancing of the control surface. Generally, ed by
A) checking for ed	qual distribution of weight throu	ughout the control surface.
B) the behavior of	f the trailing edge when the su	rface is suspended from its hinge points.
C) suspending the weight distribution		ng edge in the streamline position and checking
99.	AMA001	AMA
Stability about the	e axis which runs parallel to the	e line of flight is referred to as
A) longitudinal sta	ability.	
B) lateral stability		
C) directional stat	oility.	
100.	AMA038	AMA
Fairleads should	never deflect the alignment of	a cable more than
A) 12°.		
B) 8°.		
C) 3°.		
101.	AMA081	AMA
If control cables a	are adjusted properly and the o	ontrol surfaces tend to vibrate, the probable cause is
A) worn attachme	ent fittings.	
B) oil can effects	on the control surfaces.	
C) excessive cab	le tension.	
102.	AMA081	AMA

a large all metal aircraft is used primarily to
AMA
craft primary control systems?
AMA
ane equipped with differential-type aileron ause
eamlined position) than down travel.
reamlined position) than up travel.
dograps (from full up to full down) than the
degrees (from full up to full down) than the
AMA
AMA
AMA
AMA
AMA
AMA nt controls is moved forward and to the right,
AMA AMA AMA Consition during a ground operational check of
AMA AMA AMA Consition during a ground operational check of
AMA AMA AMA Consition during a ground operational check of
AMA AMA AMA Cosition during a ground operational check of of the trim tab to move in which direction?
AMA AMA AMA Consistion during a ground operational check of of the trim tab to move in which direction? AMA Consider the elevator is in the DOWN position.
AMA AMA AMA AMA AMA AMA AMA AMA
AMA AMA AMA AMA AMA AMA AMA AMA

108.	AMA081	AMA
Differential cont	rol on an aileron system means t	hat
	vel is more than the up travel.	
•	is more than the down travel.	
, .	n one wing travels further up tha	n the aileron on the opposite wing to adjust for
109.	AMA082	AMA
If the control stic the right aileron		ed flight controls is moved rearward and to the left,
A) down and the	e elevator will move down.	
B) up and the el	evator will move down.	
C) down and the	e elevator will move up.	
110.	AMA081	AMA
With which syste	em is differential control associat	ed?
A) Trim.		
B) Aileron.		
C) Elevator.		
111.	AMA081	AMA
The correct dihe	edral angle can be determined by	r
A) measuring th	e angular setting of each wing at	the rear spar with a bubble protractor.
B) placing a straposition.	aightedge and bubble protractor a	across the spars while the airplane is in flying
C) using a dihed	dral board and bubble level along	the front spar of each wing.
112.	AMA081	AMA
Where would you aircraft?	ou find precise information to perf	form a symmetry alignment check for a particular
A) Aircraft Spec	ification or Type Certificate Data	Sheet.
B) Manufacture	's service bulletins.	
C) Aircraft servi	ce or maintenance manual.	
113.	AMA059	AMA
Where is fuselage	ge station No. 137 located?	
A) 137 centimet	ers aft of the nose or fixed refere	nce line.
B) 137 inches a	ft of the zero or fixed reference li	ne.
C) Aft of the eng	gine.	

114.	AMA100	AMA
Why is it generally necessity	essary to jack an aircraft indoors for weighing	J ?
A) So aircraft may be p	laced in a level position.	
B) So that air currents of	do not destabilize the scales.	
C) So weighing scales	may be calibrated to 0 pounds.	
115.	AMA001	AMA
As the angle of attack of	of an airfoil increases, the center of pressure	will
A) move toward the train	iling edge.	
B) remain stationary be angle of attack.	cause both lift and drag components increas	e proportionally to increased
C) move toward the lea	ding edge.	
116.	AMA050	AMA
An airplane which has	good longitudinal stability should have a mini	mum tendency to
A) roll.		
B) pitch.		
C) yaw.		
117.	AMA100	AMA
An airplane's center of	lift is usually located aft of its center of gravity	У
,	will have a tail heavy tendency.	
	will have a nose heavy tendency.	
C) to improve stability a	about the longitudinal axis.	
118.	AMA050	AMA
An airplane that has a t motion has	endency to gradually increase a pitching mo	ment that has been set into
A) poor longitudinal sta	bility.	
B) good lateral stability.		
C) poor lateral stability.		
119.	AMA001	AMA
The angle of incidence	of an airplane at rest	
A) affects the dihedral of	of the wings in flight.	
B) is the same as the a	ngle between the relative wind and the chord	of the wing.
C) does not change wh	en in flight.	

120.	AMA001	AMA
Movement of an airplar	ne along its lateral axis (roll) is also movemer	nt
•	longitudinal axis controlled by the elevator.	•
•	-	
•	lateral axis controlled by the ailerons.	
C) around or about the	longitudinal axis controlled by the ailerons.	
121.	AMA001	AMA
· – · · The primary purpose o		,,
A) provide added lift at	•	
, .	•	
B) stall the inboard por	•	
C) provide added lift at	nigh angles of attack.	
122.	AMA001	AMA
The chord of a wing is	measured from	
A) wingtip to wingtip.		
B) wing root to the wing	atin	
C) leading edge to trail		
c) leading edge to trail	ing eage.	
123.	AMA001	AMA
What physical factors a	are involved in the aspect ratio of airplane win	igs?
A) Thickness and chore	·	
B) Span and chord.		
C) Dihedral and angle	of attack.	
o) Billoural alla allgio		
124.	AMA093	AMA
If the vertical fin of a sin parallel to	ngle engine, propeller driven airplane is rigge	d properly, it will generally be
A) the longitudinal axis	but not the vertical axis.	
B) the vertical axis but	not the longitudinal axis.	
C) both the longitudina	I and vertical axes.	
125.	AMA081	AMA
Rigging and alignment avoided, the aircraft sh	checks should not be undertaken in the oper ould be positioned	; however, if this cannot be
A) obliquely into the wi	nd.	
B) facing any direction	since it makes no difference if the wind is ste	ady (not gusting).
C) with the nose into th	ne wind.	

126.	AMA091	AMA
What is the purpose	of the free wheeling unit in a helicopter drive sy	vstem?
A) It disconnects the	rotor whenever the engine stops or slows belo	w the equivalent of rotor RPM.
B) It releases the roto	or brake for starting.	
C) It relieves bending	stress on the rotor blades during starting.	
127.	AMA091	AMA
Which statement is c	orrect concerning torque effect on helicopters?	
A) Torque direction is	s the same as rotor blade rotation.	
B) As horsepower de	creases, torque increases.	
C) Torque direction is	s the opposite of rotor blade rotation.	
128.	AMA091	AMA
One purpose of the fi	reewheeling unit required between the engine a	and the helicopter transmission
A) automatically dise	ngage the rotor from the engine in case of an ϵ	engine failure.
B) disconnect the rote	or from the engine to relieve the starter load.	
C) permit practice of	autorotation landings.	
129.	AMA091	AMA
If a single rotor helico	ppter is in forward horizontal flight, the angle of	attack of the advancing blade is
A) more than the retr	eating blade.	
B) equal to the retrea	iting blade.	
C) less than the retre	ating blade.	
130.	AMA091	AMA
A helicopter in forwar	d flight, cruise configuration, changes direction	by
A) varying the pitch of	f the main rotor blades.	
B) changing rotor RP	M.	
C) tilting the main rot	or disk in the desired direction.	
131.	AMA091	AMA
The auxiliary (tail) rot of the following?	or of a helicopter permits the pilot to compensa	ate for and/or accomplish which
A) Attitude and airspe	eed.	
B) Lateral and yaw po	osition.	
C) Torque and directi	ional control.	

132. AMA088 AMA

Which statement is correct regarding an aircraft that is found to be unairworthy after an annual inspection, due to an item requiring a major repair (assuming approved data is used to accomplish the repair)?

- A) An appropriately rated mechanic may accomplish the repair, and an IA may approve the aircraft for return to service.
- B) An appropriately rated mechanic or repair station may repair the defect and approve the aircraft for return to service.
- C) Only the person who performed the annual inspection may approve the aircraft for return to service, after the major repair.

133. AMA019 AMA

Large airplanes and turbine-powered multiengine airplanes operated under Federal Aviation Regulation Part 91, General Operating and Flight Rules, must be inspected

- A) in accordance with an inspection program authorized under Federal Aviation Regulation Part 91, Subpart E.
- B) in accordance with a continuous airworthiness maintenance program (camp program) authorized under Federal Aviation Regulation Part 91, Subpart E.
- C) in accordance with the progressive inspection requirements of Federal Aviation Regulation Section 91.409(d).

134. AMA088 AMA

Which statement about Airworthiness Directives (AD's) is true?

- A) AD's are information alert bulletins issued by the airframe, powerplant, or component manufacturer.
- B) Compliance with an AD is not mandatory unless the aircraft affected is for hire.
- C) Compliance with an applicable AD is mandatory and must be recorded in the maintenance records.

135. AMA065 AMA

How can it be determined that all air has been purged from a master cylinder brake system?

- A) By operating a hydraulic unit and watching the system pressure gauge for smooth, full scale deflection.
- B) By noting whether the brake is firm or spongy.
- C) By noting the amount of fluid return to the master cylinder upon brake release.

136. AMA031 AMA

What is one effect a restricted compensator port of a master cylinder will have on a brake system?

- A) The brakes will operate normally.
- B) The reservoir will be filled by reverse flow.

C) The restriction v	will cause slow release of the	brakes.
137.	AMA063	AMA
	e effect if the piston return spruld become spongy.	ing broke in a brake master cylinder?
•	el would become excessive.	
C) The brakes wou	uld drag.	
138.	AMA032	AMA
	right brake on an aircraft is sp ne probable cause is	oongy when the brake pedal is depressed in a
•	aster cylinder piston is stickin	g.
B) air in the brake	•	
C) the hydraulic ma	aster cylinder piston return sp	oring is weak.
139.	AMA031	AMA
Many brake types adaptable to mech		echanically or hydraulically. Which type is not
A) Single disk spot	• •	
B) Single servo typ		
C) Expander tube	type.	
140.	AMA031	AMA
The purpose of an	orifice check valve is to	
A) relieve pressure	e to a sensitive component.	
B) restrict flow in o	one direction and allow free flo	ow in the other.
C) relieve pressure	e in one direction and prevent	: flow in the other direction.
141.	AMA029	AMA
A special bolt in a foot-pounds are re		uires a torque value of 440 inch-pounds. How many
A) 36.8.		
B) 38.		
C) 36.6.		
142.	AMA063	AMA
An O ring intended marked with	d for use in a hydraulic systen	n using MIL-H-5606 (mineral base) fluid will be
A) a blue stripe or	dot.	

B) one or more w	hite dots.	
C) a white and ye	ellow stripe.	
143.	AMA063	AMA
What device in a no demands are	•	nstant delivery pump allows circulation of the fluid when
A) Pressure relief	f valve.	
B) Shuttle valve.		
C) Pressure regu	lator.	
144.	AMA031	AMA
Lockout debooste	ers are primarily pressure r	educing valves that
A) allow full debo pressure chambe	-	fluid from the high pressure side entering the low
B) cannot allow for pressure chambe		without fluid from the high pressure side entering the low
C) must be bled s	separately after brake bleed	ding has been completed.
145.	AMA068	AMA
An electric motor	used to raise and lower a	landing gear would most likely be a
A) shunt field ser	ies wound motor.	
B) split field shun	t wound motor.	
C) split field serie	s wound motor.	
146.	AMA068	AMA
A landing gear po	sition and warning system	will provide a warning in the cockpit when the throttle is
A) retarded and g	gear is not down and locked	d.
B) advanced and	gear is down and locked.	
C) retarded and g	gear is down and locked.	
147.	AMA068	AMA
What should be o	checked when a shock stru	t bottoms during a landing?
A) Air pressure.		
B) Packing seals	for correct installation.	
C) Fluid level.		
148.	AMA068	AMA
What is the functi	on of a cam incorporated in	n a nose gear shock strut?
A) Provides an in	ternal shimmy damper.	

B) Straightens	the nosewheel.	
C) Provides ste	eering of aircraft during ground operation	on.
149.	AMA063	AMA
The purpose o	f a sequence valve in a hydraulic retra	ctable landing gear system is to
A) prevent hea	vy landing gear from falling too rapidly	upon extension.
B) provide a m emergency sou	•	ce of hydraulic power and connecting the
C) ensure oper	ration of the landing gear and gear doc	ors in the proper order.
150.	AMA031	AMA
Debooster cylii	nders are used in brake systems prima	rily to
A) reduce brak	e pressure and maintain static pressur	e.
B) relieve exce	essive fluid and ensure a positive relea	se.
C) reduce the p	pressure to the brake and increase the	volume of fluid flow.
151.	AMA068	AMA
	an out of tolerance toe in condition of nent or twisted components consists of	nain landing gear wheels determined not to be
A) shimming th	ne axle in the oleo trunnion.	
B) inserting, re the scissor tord		shers or spacers at the center pivotal point of
C) placing shin	ns or spacers behind the bearing of the	e out of tolerance wheel or wheels.
152.	AMA097	AMA
Aircraft tire pre	ssure should be checked	
A) using only a	a push on stick-type gauge having 1-po	und increments.
B) at least onc	e a week or more often.	
C) as soon as	possible after each flight.	
153.	AMA097	AMA
Overinflated ai	rcraft tires may cause damage to the	
A) brake lining	S.	
B) wheel hub.		
C) wheel flang	e.	
154.	AMA031	AMA
The purpose o	f a relief valve in a brake system is to	
A) reduce pres	sure for brake application.	

B) prevent the tire from	m skidding.	
C) compensate for the	ermal expansion.	
155.	AMA097	AMA
	alled in some aircraft wheels will	
A) indicate tire tread s		
B) prevent overinflation	•	
	elevated temperature.	
156.	AMA031	AMA
Debooster valves are	used in brake systems primarily to	
A) ensure rapid applic	cation and release of the brakes.	
B) reduce brake press	sure and maintain static pressure.	
C) reduce the pressur	e and release the brakes rapidly.	
157.	AMA097	AMA
The primary purpose	for balancing aircraft wheel assemblies is to	
A) prevent heavy spot	ts and reduce vibration.	
B) distribute the aircra	aft weight properly.	
C) reduce excessive v	wear and turbulence.	
158.	AMA063	AMA
The pressure source f	for power brakes is	
A) the main hydraulic	system.	
B) the power brake re		
C) a master cylinder.		
159.	AMA031	AMA
Internal leakage in a b	orake master cylinder unit can cause	
A) slow release of bra	ikes.	
B) the pedal to slowly	creep down while pedal pressure is applied.	
C) fading brakes.		
160.	AMA031	AMA
Which statement is tru	ue with respect to an aircraft equipped with hys?	draulically operated multiple disk
• •	num or maximum disk clearance checks requi	red due to the use of self

B) Do not set parking I	brake when brakes are hot.	
C) No parking brake p	rovisions are possible for this type of brake a	ssembly.
161.	AMA068	AMA
When an empty shock strut completely at lea	strut is filled with fluid, care should be taken st two times to	to extend and compress the
A) thoroughly lubricate	e the piston rod.	
B) force out any exces	ss fluid.	
C) ensure proper pack	king ring seating and removal of air bubbles.	
162.	AMA032	AMA
In brake service work, A) withdrawing air only	the term 'bleeding brakes' is the process of y from the system.	
	om the system for the purpose of removing ai	r that has entered the system.
	ounts of fluid in reservoir.	·
163.	AMA068	AMA
When servicing an air/	oil shock strut with MIL-5606 the strut should	be
A) collapsed and fluid	added at the filler opening.	
B) fully extended and	fluid added at the filler opening.	
C) partially extended a	and fluid added at the filler opening.	
164.	AMA068	AMA
In shock struts, chevro	on seals are used to	
A) absorb bottoming e		
B) prevent oil from esc	caping.	
C) serve as a bearing	. •	
165.	AMA068	AMA
	pleo shock struts serve to	
A) lock the struts in the		
•	il as the struts are compressed.	
•	mount of air in the struts.	
166.	AMA097	AMA
	wait after a flight before checking tire pressure	5 :
,	hours in hot weather).	
ען או ופמאני אווטעוא (4	hours in hot weather).	

C) At least 4 hours (5 h	ours in hot weather).	
167. Excessive wear in the o	AMA097 center of the tread of an aircraft tire is an indic	AMA
A) incorrect camber. B) excessive toe out. C) overinflation.		
168.	AMA097	AMA
Why do tire and wheel before removing the wh	manufacturers often recommend that the tires	s on split rim wheels be deflated
A) To relieve the strain	on the wheel retaining nut and axle threads.	
B) As a safety precaution or weakened.	on in case the bolts that hold the wheel halve	s together have been damaged
C) To remove the static	load imposed upon the wheel bearings by the	ne inflated tire.
169.	AMA097	AMA
Exposure to and/or sto	rage near which of the following is considered	d harmful to aircraft tires?
1. Low humidity.		
2. Fuel.		
3. Oil.		
4. Ozone.		
5. Helium.		
Electrical equipment		
7. Hydraulic fluid.		
8. Solvents.		
A) 2, 3, 4, 5, 6, 7, 8.		
B) 1, 2, 3, 5, 7, 8.		
C) 2, 3, 4, 6, 7, 8.		
170.	AMA064	AMA
Two types of hydraulic	fluids currently being used in civil aircraft are	
A) mineral base, and pl	hosphate ester base.	
B) mixed mineral base	and phosphate ester base.	
C) petroleum base and	mixed mineral base.	
171.	AMA064	AMA

(1) Materials which a polyurethane and ep	are Skydrol compatible or resistant inclu boxy paints.	ide most common aircraft metals and
(2) Skydrol hydraulid	c fluid is compatible with nylon and natu	ral fibers.
Regarding the above	e statements,	
A) neither No. 1 nor		
B) both No. 1 and N	o. 2 are true.	
C) only No. 1 is true		
172.	AMA064	AMA
How can the proper	hydraulic fluid to be used in an airplane	be determined?
A) Refer to the aircra	aft parts manual.	
B) Consult the aircra	aft Type Certificate Data Sheet.	
C) Consult the aircra	aft manufacturer's service manual.	
173.	AMA064	AMA
The internal resistar	nce of a fluid which tends to prevent it fro	om flowing is called
A) volatility.		
B) viscosity.		
C) acidity.		
174.	AMA064	AMA
Which of the following	ng is adversely affected by atmospheric	humidity if left unprotected?
1. MIL-H-5606 hydra	aulic fluid.	
2. Skydrol hydraulic	fluid.	
3. None of the above	e.	
A) 1 and 2.		
B) 3.		
C) 2.		
175.	AMA064	AMA
Where can informati aircraft materials?	ion be obtained about the compatibility of	of fire resistant hydraulic fluid with
A) Fluid manufacture	er`s technical bulletins.	
B) Aircraft manufact	urer`s specifications.	
C) AC 43.13-1A.		
176.	AMA063	AMA
The unit which caus	es one hydraulic operation to follow and	other in a definite order is called a

A) selector valve.		
B) sequence valv	e.	
C) shuttle valve.		
177.	AMA063	AMA
•	ne, but closed if the fluid fl I to as a	ice which is designed to remain open to allow a normal ow increases above an established rate. This device is
178.	AMA063	AMA
	tion of the flap overload value	
	•	airspeeds which would impose excessive structural loads.
B) cause the flap	segments located on oppose aircraft will not become	osite sides of the aircraft centerline to extend and retract e aerodynamically unbalanced to the extent that it
C) boost normal s relatively large fla		ps in order to overcome the air loads acting on the
179.	AMA065	AMA
Unloading valves A) dampen out pr B) relieve the pur C) relieve system	ressure surges. np pressure.	ne driven hydraulic pumps to
180.	AMA063	AMA
	cooling units are required	in some aircraft hydraulic systems because of
	and high rates of fluid flogenerated from braking.	w.
181. If hydraulic fluid is of	AMA065 s released when the air va	AMA alve core of the accumulator is depressed, it is evidence
B) a leaking chec	umulator air pressure. k valve. ohragm or leaking seals.	
, 1	9 9 1 1 1 1 1	

A) lower pressure than B) higher pressure thar	AMA063 nal relief valves are set to open at a the system relief valve. n the system relief valve. the system pressure regulator.	AMA
183. What is the main purpo A) Prevent tank collaps B) Prevent hydraulic pu C) Prevent hydraulic flu	ump cavitation.	AMA ystem?
 Dampen pressure su Supplement the syst Store power for limite 	AMA063 mulators serve which of the following function urges. em pump when demand is beyond the pumped operation of components if the pump is not supply of fluid to the pump.	's capacity.
A) easily replacing hyding A) quickly connecting a contaminates entering	AMA063 lings in hydraulic systems provide a means of raulic lines in areas where leaks are commonand disconnecting hydraulic lines and eliminatine system. and disconnecting hydraulic lines without loss	te the possibility of
	·	AMA
187.	AMA063	AMA

-		
188.	AMA003	AMA
Teflon hose that has one comperature should A) not be straightened B) not be reinstalled one compension of the compens	nce removed.	ed to high pressure or
189. A hydraulic motor con A) linear motion. B) rotary motion. C) angular motion.	AMA063 verts fluid pressure to	AMA
A) They are considera B) There is no fire haz	AMA063 e of piston type hydraulic motors over electricably quieter in operation. Eard if the motor is stalled. Storily over a wider temperature range.	AMA c motors?
1. Minimum maintenaı 2. Lightweight.	AMA063 apply to aircraft hydraulic systems? nce requirements. perating efficiency (20 percent loss due to flui	AMA id friction).
	AMA063 essure regulator in a hydraulic system is to perating pressure within a predetermined rang	AMA ge and to unload the pump.

B) regulate the amo	ount of fluid flow to the actua	ting cylinders within the system.
C) prevent failure of	f components or rupture of h	ydraulic lines under excessive pressure.
193.	AMA063	AMA
~ .	or valve is one of the most c of fluid into and out of a conn	ommonly used in hydraulic systems to provide for ected actuating unit?
A) Four port, closed	d center valve.	
B) Three port, four	way valve.	
C) Two port, open o	center valve.	
194.	AMA079	AMA
Pneumatic systems	utilize	
A) return lines.		
B) relief valves.		
C) diluter valves.		
195.	AMA079	AMA
An aircraft pneumat compressor, also re	-	es an engine driven multistage reciprocating
A) an oil separator.		
B) a surge chambei	r.	
C) a moisture sepai	rator.	
196.	AMA063	AMA
		directs pressurized fluid to one end of an actuating o the reservoir from the other end.
197.	AMA063	AMA
		sed to direct the flow of fluid is the
A) check valve.	•	
, B) orifice check valv	ve.	
C) selector valve.		
198.	AMA011	AMA
(Refer to Airframe f	igure 11.) Which fitting is an	AN flared tube fitting?
` A) 1.		-

B) 2.			
C) 3.			
199.	AMA065	AMA	
•	the heel of a bend are not per oviding they are less than what	missible, they are acceptable in the remainder of a percent of the tube diameter?	
A) 5.			
B) 10.			
C) 20.			
200.	AMA063	AMA	
The installation of	a new metal hydraulic line sho	ould be made with	
A) a straight tube	to withstand the shocks and vi	bration to which it will be subjected.	
B) a straight tube leakage.	to permit proper alignment of t	he fitting and thereby reduce fluid loss through	
C) enough bends vibration.	to allow the tube to expand an	d contract with temperature changes and to absorb)
201.	AMA065	AMA	
•	accumulator has been installed e gauge will not show a hydrau	and air chamber charged, the main system lic pressure reading until	
A) at least one se accumulator.	lector valve has been actuated	to allow fluid to flow into the fluid side of the	
B) the air pressure	e has become equal to the fluid	f pressure.	
C) the fluid side o	f the accumulator has been ch	arged.	
202.	AMA063	AMA	
Which seals are u	used with petroleum base hydra	aulic fluids?	
A) Polyester.	·		
B) Butyl rubber.			
C) Buna-N.			
203.	AMA063	AMA	
		o, it is found that the handle cannot be moved in the kely cause is an incorrectly installed	е
A) hand pump inp	ort check valve.		
	orifice check valve.		
C) hand pump ou	tport check valve.		

204.	AMA065	AMA
Hydraulic fluid filterir	ng elements constructed of porc	us paper are normally
A) cleaned and reus	ed.	
B) discarded at regu	lar intervals and replaced with r	new filtering elements.
C) not approved for	use in certificated aircraft.	
205.	AMA065	AMA
To prevent external seal is the	and internal leakage in aircraft h	nydraulic units, the most commonly used type of
A) O ring seal.		
B) gasket seal.		
C) chevron seal.		
206.	AMA065	AMA
	h a system normally serviced wi	
A) Methyl ethyl keto		•
B) Naphtha or varso		
C) Lacquer thinner of		
207.	AMA065	AMA
Which must be done pressure regulator?	e before adjusting the relief valve	e of a main hydraulic system incorporating a
A) Eliminate the acti	on of the unloading valve.	
B) Adjust all other sy	stem relief valves which have a	lower pressure setting.
C) Manually unseat	all system check valves to allow	unrestricted flow in both directions.
208.	AMA063	AMA
Severe kickback of t indicate which of the		ump handle during the normal intake stroke will
A) The hand pump in	nlet check valve is sticking oper	ı.
B) The main system	relief valve is set too high.	
C) The hand pump o	outlet check valve is sticking ope	en.
209.	AMA065	AMA
The main system provalve should be adju	· · · · · · · · · · · · · · · · · · ·	ydraulic system equipped with a power control
•	ontrol valve held in the CLOSED	position.
B) while one or more	e actuating units are in operation	٦.
C) with the power co	ontrol valve in the OPEN position	า.

210. AMA063	AMA
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How is the air in a hydraulic accumulator prevented from entering the fluid system?

- A) By forcing the oil/air mixture through a centrifugal separating chamber that prevents the air from leaving the accumulator.
- B) By physically separating the air chamber from the oil chamber with a flexible or movable separator.
- C) By including a valve that automatically closes when the fluid level lowers to a preset amount.

211. AMA065 AMA

If it is necessary to adjust several pressure regulating valves in a hydraulic system, what particular sequence, if any, should be followed?

- A) Units most distant from the hydraulic pump should be adjusted first.
- B) Units with the highest pressure settings are adjusted first.
- C) Units are independent of each other, and therefore, no particular sequence is necessary.

212. AMA065 AMA

Which is true regarding the ground check of a flap operating mechanism which has just been installed?

- A) If the time required to operate the mechanism increases with successive operations, it indicates the air is being worked out of the system.
- B) If the time required to operate the mechanism decreases with successive operations, it indicates the air is being worked out of the system.
- C) All hydraulic lines and components should be checked for leaks by applying soapy water to all connections.

213. AMA065 AMA

A hydraulic system operational check during ground runup of an aircraft indicates that the wing flaps cannot be lowered using the main hydraulic system, but can be lowered by using the emergency hand pump. Which is the most likely cause?

- A) The flap selector valve has a severe internal leak.
- B) The pressure accumulator is not supplying pressure to the system.
- C) The fluid level in the reservoir is low.

214. AMA063 AMA

If two actuating cylinders which have the same cross sectional area but different lengths of stroke are connected to the same source of hydraulic pressure, they will exert

- A) different amounts of force but will move at the same rate of speed.
- B) equal amounts of force but will move at different rates of speed.
- C) equal amounts of force and will move at the same rate of speed.

215.	AMA065	AMA
	e air pressure charge in the accumul n still has hydraulic pressure?	lator be determined if the engine is inoperative,
A) Read it dire	ectly from the main system pressure	gauge with all actuators inoperative.
	stem pressure with the emergency pe air side of the accumulator.	oump and then read the pressure on a gauge
C) Operate a h goes toward z	•	essure at which a rapid pressure drop begins as it
216.	AMA063	AMA
-	that when the hydraulic pump is run ed, no hydraulic pressure is availabl	nning, the pressure is normal. However, when the e. This is an indication of a
A) leaking sele	ector valve.	
B) low accumu	ulator fluid preload.	
C) leaking acc	umulator air valve.	
217.	AMA065	AMA
What type of p	•	components to be installed in a system
A) AN packing	s made of natural rubber.	
B) Packing ma	aterials made for ester base fluids.	
C) AN packing	s made of neoprene.	
218.	AMA064	AMA
manufacturer's	•	the type fluid specified in the aircraft truction plate affixed to the reservoir or unit.
	above statements,	
A) only No. 1 i	s true.	
B) only No. 2 i	s true.	
C) both No. 1	and No. 2 are true.	
219.	AMA002	AMA
When Refriger	rant 12 is passed over an open flam	e, it

A) changes to methane gas.

C) changes to phosgene gas.

B) is broken down into its basic chemical elements.

220.	AMA005	AMA
What is the condition on A) Low pressure liquid. B) High pressure liquid		a vapor cycle cooling system?
C) High pressure vapor	r.	
221.	AMA005	AMA
	t possibly be damaged if liquid refrigerant is instem when the pressure is too high or the outs	
222.	AMA080	AMA
A) controlling the air inf B) inflating door seals a	an aircraft in flight is maintained at the selecter flow rate. and recirculating conditioned cabin air. at which air leaves the cabin.	ed altitude by
223.	AMA035	AMA
•	used for on a combustion heater?	
•	n air for ground blower.	
B) Carries heat to the p C) Provides air required		
224.	AMA006	AMA
essentially the same te A) The system is function B) The expansion valve	eycle cooling system, if the two lines connected imperature, what does this indicate? oning normally. The is not metering freon properly. The pumping too much refrigerant.	ed to the expansion valve are
225.	AMA006	AMA
The point at which freo from a liquid to a gas is	n flowing through a vapor cycle cooling systes the	m absorbs heat and changes
A) condenser.		
B) evaporator.		
C) expansion valve.		

226.	AMA080	AMA
What compone airplane altitude		m prevents the cabin altitude from becoming higher than
A) Cabin rate of	f descent control.	
B) Negative pre	ssure relief valve.	
C) Positive pres	ssure relief valve.	
227.	AMA006	AMA
The function of	the evaporator in a freon c	ooling system is to
A) liquefy freon	in the line between the cor	mpressor and the condenser.
B) lower the ten	nperature of the cabin air.	
C) transfer heat	from the freon gas to amb	ient air.
228.	AMA068	AMA
On some cabin	pressurization systems, pr	essurization on the ground is restricted by the
A) cabin pressu	re regulator.	
B) negative pre	ssure-relief valve.	
C) main landing	gear operated switch.	
229.	AMA080	AMA
How is the cabi	n pressure of a pressurized	d aircraft usually controlled?
A) By a pressur	e sensitive switch that caus	ses the pressurization pump to turn on or off as required.
B) By an autom set.	atic outflow valve that dum	ps all the pressure in excess of the amount for which it is
C) By a pressur	e sensitive valve that contr	rols the output pressure of the pressurization pump.
230.	AMA006	AMA
from a gas to a		vapor cycle cooling system gives up heat and changes
A) condenser.		
B) evaporator.	ali va	
C) expansion va	aive.	
231.	AMA080	AMA
The altitude cor	ntroller maintains cabin altit	ude by modulation of the
A) safety and o	utflow valves.	
B) safety valve.		
C) outflow valve	9.	

232.	AMA074	AMA
What type of o	xygen system uses the rebreathe	er bag-type mask?
A) Diluter dem	and.	
B) Continuous	flow.	
C) Demand.		
233.	AMA080	AMA
A pressurization	on controller uses	
A) bleed air pro	essure, outside air temperature, a	and cabin rate of climb.
B) barometric _l	pressure, cabin altitude, and cab	in rate of change.
C) cabin rate c	of climb, bleed air volume, and ca	bin pressure.
234.	AMA080	AMA
Which best de	scribes cabin differential pressure	e?
A) Difference b	petween cabin flight altitude press	sure and Mean Sea Level pressure.
B) Difference b	petween the ambient and internal	air pressure.
C) Difference b	petween cabin pressure controlle	r setting and actual cabin pressure.
235.	AMA002	AMA
•	nme figure 13.) Determine what uve in a freon refrigeration system.	nit is located immediately downstream of the
A) Condenser.		
B) Compresso	r.	
C) Evaporator	coils.	
236.	AMA035	AMA
The operation	of an aircraft combustion heater	is usually controlled by a thermostat circuit which
A) alternately t	urns the fuel on and off, a proces	ss known as cycling.
B) meters the a	amount of fuel continuously enter	ring the heater and therefore regulates the heater's
C) regulates th	e voltage applied to the heater's	ignition transformer.
237.	AMA080	AMA
One purpose o	of a jet pump in a pressurization a	and air conditioning system is to
A) produce a h	igh pressure for operation of the	outflow valve.
B) provide for a	augmentation of airflow in some	areas of the aircraft.
C) assist in the	e circulation of freon.	

238.	AMA006	AMA
The function of an expa	ansion valve in a freon cooling system is to a	ct as a metering device and to
A) reduce the pressure	of the gaseous freon.	
B) increase the pressur	re of the liquid freon.	
C) reduce the pressure	of the liquid freon.	
239.	AMA080	AMA
The cabin pressurization	on modes of operation are	
A) isobaric, differential,	and maximum differential.	
B) differential, unpressu	urized, and isobaric.	
C) ambient, unpressuri	zed, and isobaric.	
240.	AMA012	AMA
•	good practice concerning the inspection of he around the engine exhaust as a heat source	•
A) Supplement physica	ll inspections with periodic operational carbor	n monoxide detection tests.
B) All exhaust system or by the magnetic particles	components should be removed periodically, e inspection method.	and their condition determined
C) All exhaust system operiod.	components should be removed and replaced	d at each 100-hour inspection
241.	AMA073	AMA
How often should stand	dard weight high pressure oxygen cylinders b	e hydrostatically tested?
A) Every 5 years.		
B) Every 4 years.		
C) Every 3 years.		
242.	AMA005	AMA
The evacuation of a va	por-cycle cooling system removes any water	that may be present by
A) drawing out the liqui	d.	
B) raising the boiling po	pint of the water and drawing out the vapor.	
C) lowering the boiling	point of the water and drawing out the vapor.	
243.	AMA005	AMA
When servicing an air o	conditioning system that has lost all of its frec	n, it is necessary to
A) check oil and add as	s necessary, evacuate the system, relieve va	cuum, and add freon.
B) check oil and add as	s necessary, evacuate the system, and add fi	eon.
C) check oil and add as	s necessary, and add freon.	

244.	AMA005	AMA
	a vapor cycle cooling system n. What is indicated?	n after evacuation, the low pressure gauge fails to come
A) Blockage in	the system.	
B) The expansi	on valve failed to close.	
C) The compres	ssor is not engaging.	
245.	AMA006	AMA
Frost or ice buil	dup on a vapor cycle cooling	system evaporator would most likely be caused by
A) the mixing va	alve sticking closed.	
B) moisture in t	he evaporator.	
C) inadequate a	airflow through the evaporato	r.
246.	AMA005	AMA
take a freon cha	arge?	stem would most likely be at fault if a system would not
A) Expansion vB) Condenser.	alve.	
C) Receiver dry	/er	
o, 110001701 u.,		
247.	AMA006	AMA
The purpose of	a subcooler in a vapor cycle	cooling system is to
A) augment the	cooling capacity during perio	ods of peak demand.
B) aid in quick o	cooling a hot aircraft interior.	
C) cool the freo	n to prevent premature vapor	rization.
248.	AMA002	AMA
	a freon air conditioning syster son for the slow rate discharg	m, it is important to release the charge at a slow rate.
A) Prevent the	large amount of freon from co	ontaminating the surrounding atmosphere.
B) Prevent exce	essive loss of refrigerant oil.	
C) Prevent con	densation from forming and c	ontaminating the system.
249.	AMA074	AMA
The main cause	e of contamination in gaseous	s oxygen systems is
A) moisture.		
B) dust and oth	er airborne particulates.	
C) other atmosp	oheric gases.	

250.	AMA074	AMA
High pressure cylinders	s containing oxygen for aviation use can be id	dentified by their
•	words 'BREATHING OXYGEN' stenciled in 1	•
, •		
	words 'AVIATOR'S BREATHING OXYGEN'	
C) green color and the	words 'AVIATOR'S BREATHING OXYGEN'	stenciled in 1-inch white letters.
251.	AMA074	AMA
If a high pressure oxyg the	en cylinder is to be installed in an airplane, it	must meet the specifications of
A) aircraft manufacture	r or the cylinder manufacturer.	
B) Department of Trans	•	
, ·	tion Safety Board or the Standards of Compr	assad Gas Cylindars
C) National Hansporta	tion Safety Board of the Standards of Compr	essed das Cyllinders.
252.	AMA072	AMA
What controls the amou	unt of oxygen delivered to a mask in a contin	uous flow oxygen system?
A) Calibrated orifice.		
•	ralua.	
B) Pressure reducing v	aive.	
C) Pilot's regulator.		
253.	AMA072	AMA
	low metering aneroid assembly found in oxyg	
to	iow metering anerold assembly found in oxyg	en diluter demand regulators is
A) regulate airflow in re	elation to oxygen flow when operating in emer	gency or diluter demand
positions.	7.5	,
B) regulate airflow in re	elation to cabin altitude when in diluter deman	d position.
C) automatically put the	e regulator in emergency position if the dema	nd valve diaphragm ruptures.
-,, , ,		3 1/10
254.	AMA074	AMA
(1) Oxygen used in aird	craft systems is at least 99.5 percent pure and	d is practically water free.
. , , , ,	craft systems is 99.5 percent pure and is hosp	•
Regarding the above s	·	nai quanty.
0	ialements,	
A) only No. 1 is true.		
B) both No. 1 and No. 2	2 are true.	
C) neither No. 1 nor No	o. 2 is true.	
255.	AMA072	AMA
	ystem, which of the following are vented to bl	ow out plugs in the fuselage
skin?		
A) Pressure relief valve	es.	

B) Filler shutoff valves.		
C) Pressure reducer va	alves.	
050	4144.074	
256.	AMA074	AMA
	etermine the serviceability of an oxygen cylind	ier?
A) Pressure test with m		
B) Pressure test with n	G	
C) Pressure test with w	vater.	
257.	AMA074	AMA
Oxygen systems in unp	oressurized aircraft are generally of the	
A) continuous flow and	pressure demand types.	
B) pressure demand ty	pe only.	
C) portable bottle type	only.	
258.	AMA074	AMA
• .	e oxygen cylinder is serviced, it must be the co	orrect type and have been
	d within the proper time interval.	
,	tional Transportation Safety Board.	
C) inspected by a certi	ficated airframe mechanic.	
259.	AMA073	AMA
When an aircraft's oxy	gen system has developed a leak, the lines a	nd fittings should be
A) removed and replac	ed.	
B) inspected using a sp	pecial oxygen system dye penetrant.	
C) bubble tested with a	a special soap solution manufactured specific	ally for this purpose.
260.	AMA085	AMA
- oo. A radar altimeter indica		,
A) flight level (pressure		
B) altitude above sea le	•	
C) altitude above grour		
o, aa. a a g. a		
261.	AMA085	AMA
A radar altimeter deter	mines altitude by	
A) transmitting a signal	l and receiving back a reflected signal.	
B) receiving signals tra	nsmitted from ground radar stations.	
C) means of transpond	ler interrogation.	

262.	AMA014	AMA
A Bourdon tube instrum	nent may be used to indicate	
1. pressure.		
2. temperature.		
3. position.		
A) 1 and 2.		
B) 1.		
C) 2 and 3.		
263.	AMA063	AMA
The operating mechanis	sm of most hydraulic pressure gauges is	
A) a Bourdon tube.	,	
B) an airtight diaphragn	n.	
, , ,	s filled with an inert gas to which suitable arr	ns, levers, and gears are
264.	AMA013	AMA
(1) Aircraft instruments	are color-coded to direct attention to operation	onal ranges and limitations.
• •	range markings are not specified by Title 14 ndardized by aircraft manufacturers.	of the Code of Federal
Regarding the above st	atements,	
A) only No. 1 is true.		
B) only No. 2 is true.		
C) both No. 1 and No. 2	2 are true.	
265.	AMA036	AMA
Magnetic compass bow	ls are filled with a liquid to	
A) retard precession of	the float.	
B) reduce deviation error	ors.	
C) dampen the oscillation	on of the float.	
266.	AMA054	AMA
The function of a symbol	ol generator (SG) in an EFIS is to	
A) display alphanumeric	c data and representations of aircraft instrum	ents.
B) allow the pilot to sele	ect the appropriate system configuration for the	ne current flight situation.
C) receive and process appropriate display.	input signals from aircraft and engine sensor	rs and send the data to the

267.	AMA014	AMA
Data transmit	ted between components in an E	FIS are converted into
A) digital sign	als.	
B) analog sigi	nals.	
C) carrier wav	ve signals.	
268.	AMA041	AMA
Fuel flow tran	smitters are designed to transmit	data
A) mechanica	ally.	
B) electrically		
C) utilizing flu	id power.	
269.	AMA010	AMA
	n of an angle of attack indicating sere the airstream flows in a directi	system is based on detection of differential pressure on
A) not paralle	I to the true angle of attack of the	aircraft.
B) parallel to	the angle of attack of the aircraft.	
C) parallel to	the longitudinal axis of the aircraf	t.
270.	AMA008	AMA
What does a operating?	reciprocating engine manifold pre	ssure gauge indicate when the engine is not
A) Zero press	sure.	
B) The differe	ential between the manifold pressu	ure and the atmospheric pressure.
C) The existing	ng atmospheric pressure.	
271.	AMA076	AMA
The requirem are contained		instrument static systems required by Section 91.411
A) Type Certi	ficate Data Sheets.	
B) AC 43.13-	1A.	
C) Part 43, ap	ppendix E.	
272.	AMA090	AMA
Who is author	rized to repair an aircraft instrume	ent?
1. A certified	mechanic with an airframe rating.	
2. A certificate	ed repairman with an airframe rati	ing.
3. A certificate	ed repair station approved for that	t class instrument.

4. A certificated airfram	ne repair station.	
A) 1, 2, 3, and 4.		
B) 3 and 4.		
C) 3.		
273.	AMA096	AMA
system integrity check	loss permitted during an unpressurized aircra is	iit iiistiuiileiit static piessure
A) 50 feet in 1 minute.		
B) 200 feet in 1 minute.		
C) 100 feet in 1 minute		
274.	AMA096	AMA
pressurized cabin durin	if the instrument static pressure line becomes ng cruising flight?	disconnected inside a
A) The altimeter and ai	rspeed indicator will both read low.	
B) The altimeter and ai	rspeed indicator will both read high.	
C) The altimeter will rea	ad low and the airspeed indicator will read hig	gh.
275.	AMA013	AMA
What marking color is ι	used to indicate if a cover glass has slipped?	
A) Red.		
B) White.		
C) Yellow.		
276.	AMA013	AMA
The green arc on an ai	rcraft temperature gauge indicates	
A) the instrument is not	t calibrated.	
B) the desirable temper	rature range.	
C) a low, unsafe tempe	erature range.	
277.	AMA014	AMA
	ectional gyro is used to	
A) represent the nose of	•	
B) align the instrument		
C) represent the wings		
278.	AMA013	AMA

The method of mounting A) instrument manufact B) design of the instrunct C) design of the instrunct	ment case.	els depends on the
279.	AMA013	AMA
Aircraft instrument pand A) all vibration.	els are generally shock mounted to absorb	
B) low frequency, high C) high frequency, high		
280.	AMA013	AMA
An aircraft instrument p A) act as a restraint str B) provide current retui C) aid in the panel insta	rn paths.	icture to
281.	AMA013	AMA
vvnere may a person in engine instrument? 1. Engine manufacture 2. Aircraft flight manual 3. Instrument manufact 4. Aircraft maintenance A) 2 or 4. B) 1 or 4. C) 2 or 3.	I. turer's specifications.	ie the required markings on an
282. A certificated mechanic A) minor repairs to inst B) 100-hour inspection C) instrument overhaul	ruments. s of instruments.	AMA
283.	AMA096	AMA
-	d aircraft's static pressure system is leak che n 91.411, what aircraft instrument may be us	

 Vertical speed indica Cabin altimeter. Altimeter. 	ator.	
4. Cabin rate-of-change 5. Airspeed indicator. A) 1 or 5.	e indicator.	
B) 2 or 4. C) 3.		
284.	AMA096	AMA
When performing the sind static pressure. B) positive pressure. C) negative pressure.	tatic system leakage check required by Section	on 91.411, the technician utilizes
285.	AMA090	AMA
•	d you use if you find a vacuum operated instiglass with a slippage mark. Iment.	rument glass loose?
286.	AMA071	AMA
On modern large aircra autopilot functions?	ft, what electronic device typically monitors fl	ight parameters and performs
A) Flight management (B) Transponder.	computer.	
C) Control/display unit.		
287.	AMA071	AMA
A) continually transmit B) monitor aircraft spee	e of an aircraft transponder is to heading, speed, and rate of climb/decent etc. ed, heading, altitude, and attitude whenever the tion signal from a ground station and automa	ne autopilot system is engaged.
288.	AMA071	AMA
atmosphere at	eliminate radio interference by dissipating sta	atic electricity into the
A) low current levels.		

B) high voltage lev		
C) high current lev	els.	
289.	AMA071	AMA
Part of the ADF sy	stem used on aircraft includes	
A) RMI indicator ar	ntenna.	
B) marker beacon	antenna.	
C) sense and loop	antennas.	
290.	AMA023	AMA
An aircraft antenna	a installation must be grounded	
A) to the airframe.		
B) to the engine.		
C) to the radio rack	ζ.	
291.	AMA086	AMA
The preferred loca	tion of an ELT is	
A) where it is readi	ily accessible to the pilot or a meml	per of the flightcrew while the aircraft is in flight
B) as far aft as pos	ssible.	
C) as far aft as pos	ssible, but forward of the vertical fir	
292.	AMA087	AMA
When must the rac	dio station license be displayed in a	n aircraft equipped with a two-way radio?
A) When the aircra	Ift is operated outside the U.S	
B) When the aircra	Ift is returned to service.	
C) When the aircra	aft is certified for IFR flight.	
293.	AMA044	AMA
How may the batte	ery replacement date be verified for	an emergency locator transmitter (ELT)?
A) By removing the the useful life rema	_	a measured load to determine if 50 percent of
B) By observing the	e battery replacement date marked	on the outside of the transmitter.
C) By activating the	e transmitter and measuring the sig	gnal strength.
294.	AMA025	AMA
In which control ele	ement of an autopilot system is an	attitude indicator?
A) Command.	•	
B) Sensing.		
-		

C) Input.		
295.	AMA025	AMA
What component A) Servo. B) Controller. C) Gyro.	of an autopilot system ap	oplies torque to the control surfaces of an aircraft?
296.	AMA025	AMA
Which channel of A) Elevator. B) Aileron. C) Rudder.	an autopilot detects cha	nges in pitch attitude of an aircraft?
297.	AMA025	AMA
What component	is the sensing device in a	an electromechanical autopilot system?
A) Servo.		
B) Gyro.		
C) Controller.		
298.	AMA025	AMA
In an autopilot, wh	nich signal nullifies the in	put signal to the ailerons?
A) Displacement s	signal.	
B) Course signal.		
C) Followup signa	al.	
299.	AMA026	AMA
•	lly checking an autopilot s , the autopilot should be	system on the ground, after the aircraft's main power has engaged
A) only after the g	lyros come up to speed a	and the amplifier warms up.
B) whenever the	•	
C) for only a few r	minutes at a time.	
300.	AMA025	AMA
Dutch roll, a comb counteracted with		g oscillation that affects many sweptwing aircraft, is
A) a flight director	system.	
B) an aileron dam	per system.	

C) a yaw damp	per system.	
301.	AMA023	AMA
When an anter	nna is installed, it should be fastened	
A) to the prima	ry structure at the approximate intersection of the	e three aircraft axes.
B) with a reinfo	orcing doubler on each side of the aircraft skin.	
C) so that load	s imposed are transmitted to the aircraft structure	e.
302.	AMA023	AMA
Doublers are u	sed when antennas are installed to	
A) eliminate an	ntenna vibration.	
B) prevent oil o	canning of the skin.	
C) reinstate the	e structural strength of the aircraft skin.	
303.	AMA023	AMA
What characte evaluated?	ristics of the installation of a rigid antenna on a ve	ertical stabilizer should be
A) Polarization	and impedance.	
B) Impedance	and interference.	
C) Flutter and	vibration.	
304.	AMA023	AMA
	me figure 15.) What is the approximate drag load	
-	feet installed on an aircraft with a speed of 225 M	MPH?
A) 2.069 pound		
B) 2.073 pound		
C) 2.080 pound	ds.	
305.	AMA085	AMA
A DME antenn	a should be located in a position on the aircraft tl	hat will
A) not be blank	ked by the wing when the aircraft is banked.	
B) permit interr	ruptions in DME operation.	
C) eliminate the	e possibility of the DME locking on a station.	
306.	AMA023	AMA
When installing	g a DME antenna, it should be aligned with the	
A) null position		
B) angle of inci	idence.	

C) centerline on	the airplane.	
307. (Refer to Airfram A) 2. B) 3. C) 4.	AMA023 e figure 16.) Which of the a	AMA Intennas shown is a typical glideslope antenna?
308. How much cleara A) 3 inches with t B) No set minimu	the seat unoccupied. um as long as the equipmer	AMA s required when installing radio equipment under a seat? nt receives adequate cooling and damage protection. ted to maximum downward seat spring deflection.
	independent systems. stem.	AMA ain lateral stability when jettisoning fuel?
310. The primary purp A) lower landing B) balanced fuel C) reduced fire h	weight. load.	AMA ison system is to quickly achieve a
A) boost pumps. B) gravity.	AMA054 verboard in most fuel jettiso ngine driven fuel pumps.	AMA n systems by
A) fuel control pa B) lower wing su	AMA055 fueling systems instruction nel access door. rface adjacent to the acces d connection point.	AMA al procedures are normally placarded on the s door.

313.	AMA052	AMA
A) All outside ele B) Fuel to be use	owing precautions is most important ctrical sources must be disconnected and must be appropriately identified. witches must be in OFF position.	ted from the aircraft.
314. What is the maxi A) 7 PSI. B) 5 PSI. C) 3 PSI.	AMA052 mum vapor pressure allowable for	AMA an aircraft fuel?
315.	AMA054	AMA
fuel filler cover fo A) The word 'Avo B) The word 'Fue	t 23, what minimum required marking reciprocating engine-powered airgust and the minimum fuel grade. But and usable fuel capacity. But and the total fuel capacity.	ings must be placed at or near each appropriate planes?
316.	AMA003	AMA
What minimum routility category ai		on or near each appropriate fuel filler cover on
A) The word 'Avo	as' and the minimum fuel grade, a	nd the total fuel tank capacity.
B) The word 'Avo tank capacity.	as' and the minimum fuel grade or	designation for the engines, and the usable fuel
C) The word 'Avo	gas' and the minimum fuel grade.	
317.	AMA054	AMA
The primary purp	ose of a fuel tank sump is to provi	de a
	m of maintaining the design minim	
	vater and dirt accumulations in the	
C) reserve supply	y of fuel to enable the aircraft to lar	nd safely in the event of fuel exhaustion.
318.	AMA052	AMA
The vapor pressu	ure of aviation gasoline is	
•	vapor pressure of automotive gas	
, •	e vapor pressure of automotive ga	soline.
C) approximately	20 PSI at 100 °F.	

319.	AMA054	AMA
` '	aircraft pressure refueling system, person to fuel or defuel any or all	a pressure refueling receptacle and control panel fuel tanks of an aircraft.
(2) Because of aircraft.	the fuel tank area, there are more	advantages to a pressure fueling system in light
Regarding the	above statements,	
A) only No. 1 is	s true.	
B) only No. 2 is	s true.	
C) both No. 1 a	and No. 2 are true.	
320.	AMA054	AMA
What type of fu	uel booster pump requires a pressu	ure relief valve?
A) Concentric.		
B) Sliding vane	€.	
C) Centrifugal.		
321.	AMA055	AMA
Why are centri	fugal type boost pumps used in fue	el systems of aircraft operating at high altitude?
A) Because the	ey are positive displacement pump	s.
B) To supply fu	uel under pressure to engine driver	າ pumps.
C) To permit co	ooling air to circulate around the m	otor.
322.	AMA054	AMA
Flapper valves	are used in fuel tanks to	
A) reduce pres	ssure.	
B) prevent a ne	egative pressure.	
C) act as chec	k valves.	
323.	AMA054	AMA
Fuel boost pun	nps are operated	
A) to provide a	positive flow of fuel to the engine.	
B) primarily for	fuel transfer.	
C) automatical	ly from fuel pressure.	
324.	AMA052	AMA
How may the a	antiknock characteristics of a fuel b	e improved?
A) By adding a	knock inhibitor.	
B) By adding a	knock enhancer.	

C) By adding a	fungicide agent.	
325.	AMA052	AMA
	_	tank which is known to be uncontaminated with dirt or nk sumps and system strainers
	nated except for the strainer c ring 100-hour or annual inspe	heck before the first flight of the day and the fuel tank ctions.
B) are still nece	essary due to the possibility of	contamination from other sources.
•	ply reduced since contamination modern aircraft fuel systems	on from other sources is relatively unlikely and of little .
326.	AMA054	AMA
-	ns must be observed if a gravore than one tank at a time?	ity feed fuel system is permitted to supply fuel to an
A) The tank airs	spaces must be interconnecte	d.
B) The fuel out	let ports of each tank must hav	ve the same cross sectional area.
C) Each tank mempty.	nust have a valve in its outlet the	hat automatically shuts off the line when the tank is
327.	AMA052	AMA
(1) If aviation gaincreased fuel f	•	fuel lines may become filled with vapor and cause
(2) A measure	of a gasoline's tendency to va	por lock is obtained from the Reid vapor pressure test
Regarding the	above statements,	
A) only No. 2 is	s true.	
B) both No. 1 a	nd No. 2 are true.	
C) neither No.	1 nor No. 2 is true.	
328.	AMA054	AMA
What is the pur	pose of a float operated trans	mitter installed in a fuel tank?
A) It sends an e	electric signal to the fuel quant	ity indicator.
B) It senses the	e total amount of fuel density.	
C) It senses the	e dielectric qualities of fuel and	d air in the tank.
329.	AMA052	AMA
How does temp	perature affect fuel weight?	
A) Cold fuel is I	heavier per gallon.	
B) Warm fuel is	s heavier per gallon.	

C) Temperature has no effect.

330.	AMA054	AMA
A drip gauge may be u	sed to measure	
A) the amount of fuel ir		
B) system leakage with	n the system shut down.	
C) fuel pump diaphragi	m leakage.	
331.	AMA054	AMA
The probe of a capacit	ance type fuel level gauge is essentially a	
A) float actuated variab	ole capacitor.	
B) capacitor with fuel a	nd air acting as one plate.	
C) capacitor with fuel a	and air acting as a dielectric.	
332.	AMA041	AMA
An electrical type fuel o	quantity indicating system consists of an indic	cator in the cockpit and a
A) float operated transi	mitter installed in the tank.	
B) float resting on the s	surface of the tank.	
C) float operated receive	ver installed in the tank.	
333.	AMA054	AMA
A capacitance type fue	el quantity indicating system measures fuel in	
A) pounds.		
B) pounds per hour.		
C) gallons.		
334.	AMA041	AMA
One advantage of elec	trical and electronic fuel quantity indicating sy	ystems is that
A) the indicators are ca	alibrated in gallons; therefore, no conversion i	is necessary.
B) only one transmitter	and one indicator are needed regardless of t	the number of tanks.
C) several fuel tank lev	vels can be read on one indicator.	
335.	AMA054	AMA
A probe or a series of μ	probes is used in what kind of fuel quantity in	dicating system?
A) Selsyn.		
B) Capacitor.		
C) Synchro.		
336.	AMA054	AMA

fuel remaining is equal A) The total unusable for the sound in the so	uantity indicator be calibrated to read during le to the unusable fuel supply? uel quantity. ble fuel quantity and the unusable fuel quanti	
C) Zero.		
337. Fuel system componen A) drain off static charg B) prevent stray curren C) retard galvanic corro	ts.	AMA
338. Which procedure must A) Defuel all the tanks a B) Defuel the inboard w C) Defuel the outboard	ving tanks first.	AMA otback wings?
A) calls for jettisoning o B) reduces contaminati	AMA056 system operation in multiengine aircraft of fuel overboard to correct lateral instability. ion and/or fire hazards during fueling or defue o maintain a balanced fuel load condition.	AMA eling operations.
340. What is one purpose of A) To maintain atmosph B) To decrease fuel var C) To decrease tank int	heric pressure. por pressure.	AMA
A) It traps any small am B) It provides a drain fo	AMA054 rainer located at the lowest point in the fuel synount of water that may be present in the fuel or residual fuel. I micro organisms that may be present in the	system.
342. What is the recommend	AMA055 ded practice for cleaning a fuel tank before we	AMA elding?

A) Purge the tank with a B) Flush the inside of the C) Steam clean the tan	ne tank with clean water.		
343.	AMA052	AMA	
If it is necessary to ente	er an aircraft's fuel tank, wh	ich procedure should be avoided?	
A) Continue purging the	e tank during the entire wo	k period.	
B) Station an assistant	outside the fuel tank acces	s to perform rescue operations if required.	
C) Conduct the defueling	ng and tank purging operat	on in an air conditioned building.	
344.	AMA055	AMA	
Which of the following r	may be used for the repair	of fuel leaks on most integral fuel tanks?	
A) Welding and resealing	ng.		
B) Brazing and resealing	ng.		
C) Riveting and resealing	ng.		
345.	AMA055	AMA	
What method would be from the aircraft?	used to check for internal	eakage of a fuel valve without removing the valve	
A) Place the valve in the see if fuel flows to the s	-	rainer bowl, and with boost pump on, watch to	
B) Remove fuel cap(s),	turn boost pump(s) on, an	d watch for bubbling in the tanks.	
C) Apply regulated air pathrough the valve.	pressure on the downstrear	n side of the fuel pump and listen for air passing	
346.	AMA055	AMA	
When moving the mixtuRPM should	ure control on a normally op	erating engine into the idle cutoff position, engine	
A) slightly increase before	ore the engine starts to die		
B) slightly decrease and	d then drop rapidly.		
C) remain the same un	til the cutoff is effected, the	n drop rapidly.	
347.	AMA054	AMA	
(1) A fuel pressure relie	ef valve is required on an a	rcraft positive displacement fuel pump.	
(2) A fuel pressure relie	ef valve is required on an a	rcraft centrifugal fuel boost pump.	
Regarding the above st	tatements,		
A) only No. 1 is true.	A) only No. 1 is true.		
B) only No. 2 is true.			
C) both No. 1 and No. 2 are true.			

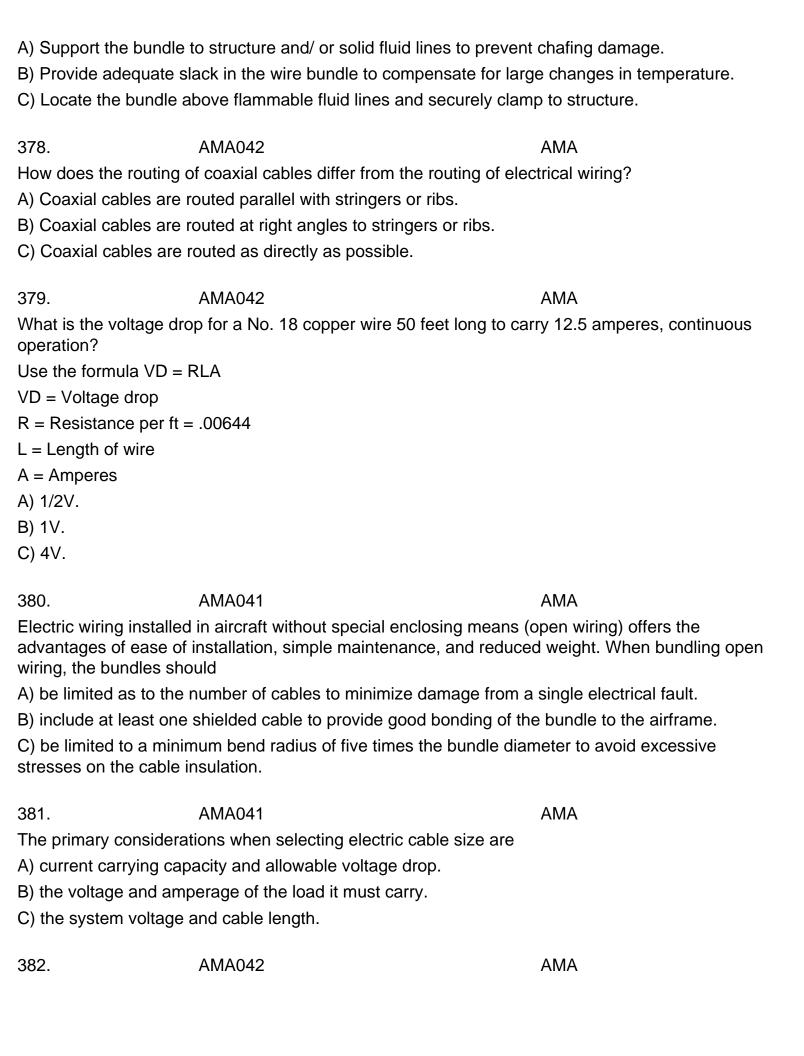
348.	AMA053	AMA
(1) A fuel heater can us	se engine bleed air as a source of heat.	
(2) A fuel heater can us	se engine lubricating oil as a source of heat.	
Regarding the above s	-	
A) only No. 1 is true.		
B) both No. 1 and No.	2 are true.	
C) neither No. 1 nor No		
349.	AMA053	AMA
(1) The function of a fu	el heater is to protect the engine fuel system	from ice formation.
(2) An aircraft fuel heat	ter cannot be used to thaw ice in the fuel scre	een.
Regarding the above s	tatements,	
A) only No. 1 is true.		
B) only No. 2 is true.		
C) both No. 1 and No.	2 are true.	
350.	AMA054	AMA
	g switch contacts close and warning light is to	
-	y of fuel has passed through it.	
B) the fuel flow stops.	, or racinial passed imeaginin	
,	ops below specified limits.	
c) the raci process are	opo bolow oposmod mimo.	
351.	AMA054	AMA
	on turbine powered aircraft to determine where er of forming ice crystals?	n the condition of the fuel is
A) Fuel pressure warni	ng.	
B) Fuel pressure gauge	e.	
C) Fuel temperature in	dicator.	
352.	AMA054	AMA
What is the purpose of	flapper type check valves in integral fuel tan	ks?
A) To allow defueling c		
•	flowing away from the boost pumps.	
•	driven pumps to draw fuel directly from the to	ank if the boost pump fails.
353.	AMA055	AMA
	justed to change the fuel pressure warning lir	
That arm would bo day	passes to origings the raoi procedure mairing in	

A) Fuel flowmeter by B) Pressure sensitive	e mechanism.	
C) Fuel pressure relie	ef valve.	
354.	AMA055	AMA
Which of the following aircraft fuel system?	g would be most useful to locate and troublesh	noot an internal fuel leak in an
A) Aircraft structure r	epair manual.	
B) Illustrated parts m	anual.	
C) A fuel system sch	ematic.	
355.	AMA041	AMA
CSD driven generato	rs are usually cooled by	
A) oil spray.		
B) an integral fan.		
C) both ram air and a	an integral fan.	
356.	AMA041	AMA
Integrated drive gene	erators (IDG) employ a type of high output ac o	generator that utilizes
A) brushes and slip r	ings to carry generated dc exciter current to th	e rotating field.
B) battery current to	excite the field.	
C) a brushless syster	m to produce current.	
357.	AMA041	AMA
When necessary duri	ing operation, CSD disconnect is usually acco	mplished by
A) a switch in the coo	ckpit.	
B) circuit breaker acti		
C) a shear section in	the input shaft.	
358.	AMA041	AMA
A voltage regulator c	ontrols generator voltage by changing the	
A) resistance in the g	enerator output circuit.	
B) current in the gene	erator output circuit.	
C) resistance of the o	generator field circuit.	
359.	AMA041	AMA
The voltage in an ac	transformer secondary that contains twice as	many loops as the primary will be
A) greater and the an	nperage less than in the primary.	

	perage greater than in the primary. Age greater than in the primary.	
A) Install a switch inder B) Use shielded electric	AMA015 must be accomplished when installing an ant bendent of the position light switch. cal cable to assure fail safe operation. ision light to the aircraft position light switch.	AMA icollision light?
361. How are generators rat A) Watts at rated voltag B) Amperes at rated vo C) The impedance at ra	ge. oltage.	AMA
• •	AMA041 ter output voltage is controlled by the d the speed of the motor.	AMA
normally provide currer A) a stepdown transfor	Itage dropping resistor.	-
A) that ac electrical mo B) greater ease in step	AMA041 g ac electrical power in aircraft is tors can be reversed while dc motors cannot ping the voltage up or down. Itage is 1.41 times the maximum instantaneous	
365. What is a method used A) Flash the fields.	AMA054 for restoring generator field residual magnet	AMA ism?

B) Reseat the brush	es.	
C) Energize the arm	ature.	
366.	AMA042	AMA
	n electrical circuit, if an ohmme e value of resistance is read,	eter is properly connected across a circuit
A) the component ha	as continuity and is open.	
B) either the compor	nent or the circuit is shorted.	
C) the component ha	as continuity and is not open.	
367.	AMA041	AMA
If it is necessary to u should	se an electrical connector who	ere it may be exposed to moisture, the mechanic
A) coat the connecto	or with grease.	
B) use a special moi	sture proof type.	
C) spray the connec	tor with varnish or zinc chroma	ate.
368.	AMA042	AMA
When using the volta	age drop method of checking o	circuit resistance, the
A) input voltage mus	t be maintained at a constant	value.
B) output voltage mu	ust be maintained at a constan	t value.
C) input voltage mus	st be varied.	
369.	AMA041	AMA
A certain switch is de indicates the numbe	•	ole throw switch (SPDT). The throw of a switch
A) circuits each pole	can complete through the swi	tch.
B) terminals at which	n current can enter or leave the	e switch.
C) places at which the time open or close a		unger, etc.) will come to rest and at the same
370.	AMA042	AMA
What is an important	t factor in selecting aircraft fus	es?
A) The current excee	eds a predetermined value.	
B) The voltage rating	g should be lower than the ma	ximum circuit voltage.
C) Capacity matches	s the needs of the circuit.	
371.	AMA042	AMA
What is the advantag	ge of a circuit breaker when co	empared to a fuse?

A) Never need:	s replacing.	
B) Always elim	inates the need of a switch.	
C) Resettable	and reusable.	
372.	AMA042	AMA
A circuit breake	er is installed in an aircraft electri	cal system primarily to protect the
A) circuit and s	hould be located as close to the	source as possible.
B) circuit and s	hould be located as close to the	unit as possible.
C) electrical un	it in the circuit and should be loc	ated as close to the source as possible.
373.	AMA041	AMA
Aircraft fuse ca	pacity is rated in	
A) volts.		
B) ohms.		
C) amperes.		
374.	AMA041	AMA
	ust be operated only in an emerg stem frequently employ	ency or whose inadvertent activation could
A) guarded swi	itches.	
B) push-pull-ty	pe circuit breakers only (no switc	hes).
C) spring-loade	ed to off toggle or rocker switches	S.
375.	AMA041	AMA
	cables must pass through holes ected from chafing by	in bulkheads, formers, ribs, firewalls, etc., the wires
A) wrapping wi	th electrical tape.	
B) using a suita	able grommet.	
C) wrapping wi	th plastic.	
376.	AMA041	AMA
If a wire is instagiven the wire?		vith some moving parts, what protection should be
A) Wrap with s	oft wire solder into a shield.	
B) Wrap with fr	iction tape.	
C) Pass throug	ıh conduit.	
377.	AMA041	AMA
Which of the fo	ollowing should be accomplished	in the installation of aircraft wiring?



does not limit	<u> </u>	erator or alternator lead, and the regulator system nerator or alternator can deliver, the ammeter can be ernator rating?
383.	AMA042	AMA
Bonding conr	nections should be tested for	
A) resistance	value.	
B) amperage	value.	
C) reactance.		
384.	AMA054	AMA
The poles of	a generator are laminated to	
A) reduce flux	k losses.	
B) increase fl	ux concentration.	
C) reduce ed	dy current losses.	
385.	AMA054	AMA
If any one ge	nerator in a 24-volt dc system sho	ws low voltage, the most likely cause is
A) an out of a	djustment voltage regulator.	
B) shorted or	grounded wiring.	
C) a defective	e reverse current cutout relay.	
386.	AMA054	AMA
_		supply power for a single load, their controls include share the load equally. The equalizer circuit
A) increasing	the output of the low generator to	equal the output of the high generator.
B) decreasing	g the output of the high generator t	o equal the output of the low generator.
C) increasing they are equa		nd decreasing the output of the high generator until
387.	AMA030	AMA
(Refer to Airfr	rame figure 18.) Which of the batte	ries are connected together incorrectly?
A) 1.		
B) 2.		
C) 3.		

388.	AMA009	AMA
What is the color and	d orientation of the position lights for navigation	on on civil airplanes?
A) Left side - green,	right side - red, rear aft - white.	·
B) Left side - red, rig	ht side - green, rear aft - white.	
C) Left side - white,	right side - green, rear aft - red.	
200	AMAQ44	Λ λ Λ
• •	AMA041 wire terminals used for most aircraft application abilities, are designed primarily	AMA ons, in addition to providing good
	failure due to terminal disconnection.	
	d and rapid circuit connection and disconnecti	on.
•	nnection to the circuit.	
390.	AMA041	AMA
What does a rectifie		/ (IVI) (
	urrent into alternating current.	
-	ing current into direct current.	
C) Reduces voltage.	_	
o) reduced venage.		
391.	AMA042	AMA
Electric circuits are p	protected from overheating by means of	
A) thermocouples.		
B) shunts.		
C) fuses.		
392.	AMA039	AMA
	advantage of the series wound dc motor?	AIVIA
A) High starting torq	-	
B) Suitable for const		
C) Low starting torqu	•	
O) Low starting torqu	do.	
393.	AMA039	AMA
A series wound dc e	electric motor will normally require	
A) more current at h	igh RPM than at low RPM.	
B) approximately the	e same current throughout its operating range	of speed.
C) more current at lo	ow RPM than at high RPM.	

394.	AMA041	AMA
The method most of	ten used in overcoming the effect of armature	reaction is through the use of
A) interpoles.		
B) shaded poles.		
C) drum wound arma	atures in combination with a negatively connection	cted series field.
395.	AMA041	AMA
<u>-</u>	ction boxes located in a fire zone are usually	constructed of
A) asbestos.		
B) cadmium plated s	steel.	
C) stainless steel.		
396.	AMA041	AMA
The pin section of ar	n AN/MS connector is normally installed on	
A) the power supply	side of a circuit.	
B) the ground side o	f a circuit.	
C) either side of a ci	rcuit (makes no difference).	
397.	AMA054	AMA
The most common n the	nethod of regulating the voltage output of a co	mpound dc generator is to vary
A) current flowing the	rough the shunt field coils.	
B) total effective field	d strength by changing the reluctance of the m	agnetic circuit.
C) resistance of the	series field circuit.	
398.	AMA054	AMA
In a generator, what of the brushes within	eliminates any possible sparking to the brush the holder?	guides caused by the movement
A) The brush pigtail.		
B) Brush spring tens	sion.	
C) Undercutting the	mica on the commutator.	
399.	AMA041	AMA
A voltage regulator o	controls generator output by	
A) introducing a resis	stance in generator-to-battery lead in the ever	nt of overload.
B) shorting out field	coil in the event of overload.	
C) varying current flo	ow to generator field coil.	

400.	AMA054	AMA
Which of the fo	llowing is not one of the purp	oses of interpoles in a generator?
A) Reduce field	l strength.	
•	rmature reaction.	
C) Reduce arci	ng at the brushes.	
404	*****	
401.	AMA054	AMA
-		n the commutator bars of a dc generator undercut?
•	width of the mica.	
, .	ce the width of the mica.	
C) Equal to the	width of the mica.	
402.	AMA054	AMA
The pole pieces	s or shoes used in a dc gene	rator are a part of the
A) armature as:	_	·
B) field assemb	•	
C) brush assen		
,	•	
403.	AMA054	AMA
		ints fail to open after the generator output has dropped
	otential, current will flow thro	
•	•	nunt field opposite the normal direction.
•	nt field opposite the normal di	
C) opposite the	normal direction and through	n the shunt field in the normal direction.
404.	AMA054	AMA
To test generat	or or motor armature winding	s for opens,
A) place armatulight.	ure in a growler and connect	a 110V test light on adjacent segments; light should
B) check adjace	ent segments on commutator	with an ohmmeter on the high resistance scale.
C) use a 12/24	V test light between the arma	ture core segments and the shaft.
405.	AMA054	AMA
The commutato	or of a generator	
A) changes dire armature.	ect current produced in the ar	mature into alternating current as it is taken from the
B) changes alte armature.	ernating current produced in t	he armature into direct current as it is taken from the
C) reverses the	current in the field coils at the	e proper time in order to produce direct current.

406.	AMA042	AMA		
When approved, splices may be used to repair manufactured harnesses or installed wiring. The maximum number of splices permitted between any two connectors is				
A) one.				
B) two.				
C) three.				
407.	AMA041	AMA		
How should the splices	be arranged if several are to be located in a	n electrical wire bundle?		
A) Staggered along the	length of the bundle.			
B) Grouped together to	facilitate inspection.			
C) Enclosed in a condu	iit.			
408.	AMA043	AMA		
What is the most accura	ate type of frequency measuring instrument?			
A) Integrated circuit chi	p having a clock circuit.			
B) Electrodynamomete	rs using electromagnetic fields.			
C) Electromagnets usin	ng one permanent magnet.			
409.	AMA041	AMA		
For general electrical us	se in aircraft, the acceptable method of attac	hing a terminal to a wire is by		
A) crimping.				
B) soldering.				
C) crimping and solderi	ng.			
410.	AMA042	AMA		
Which of the following fuse for an aircraft install	actors must be taken into consideration when	n determining the wire size to		
1. Mechanical strength.				
2. Allowable power loss	S.			
3. Ease of installation.				
4. Resistance of curren	t return path through the aircraft structure.			
5. Permissible voltage of	drop.			
6. Current carrying capa	ability of the conductor.			
7. Type of load (continu	uous or intermittent).			
A) 2, 5, 6, 7.				
B) 1, 2, 4, 5.				

C) 2, 4, 6, 7.		
411.	AMA041	AMA
The most common methodomector is by	hod of attaching a pin or socket to an individu	ual wire in an MS electrical
A) crimping.		
B) soldering.		
C) crimping and solderi	ng.	
412.	AMA068	AMA
`	ety switch.	
413.	AMA070	AMA
Major adjustments on e accomplished outside the Adjustment procedure s	equipment such as regulators, contactors, and he airplane on test benches with necessary is should be as outlined by	d inverters are best
A) the equipment manu	facturer.	
B) the FAA.		
C) aircraft technical ord	ers.	
414.	AMA042	AMA
One purpose of a growl A) an out of round common B) a broken field lead. C) a shorted armature.	ler test is to determine the presence of mutator.	
415.	AMA041	AMA
A) Measure the input vo B) The output voltage w	ed if a transformer winding has some of its to oltage with an ohmmeter. vill be high. get hot in normal operation.	irns shorted together?

 AMA

416.

An antiskid system is A) a hydraulic system.

AMA031

B) an electrohy	•	
C) an electrical	system.	
417.	AMA032	AMA
Antiskid braking	g systems are generally armed by	
A) a centrifugal	switch.	
B) a switch in the	ne cockpit.	
C) the rotation (of the wheels above a certain speed	l.
418.	AMA041	AMA
In an antiskid s	ystem, wheel skid is detected by	
A) an electrical	sensor.	
B) a discriminat	tor.	
C) a sudden ris	e in brake pressure.	
419.	AMA031	AMA
(1) An antiskid	system is designed to apply enough	force to operate just below the skid point.
2) A warning la failure.	mp lights in the cockpit when the an	tiskid system is turned off or if there is a system
Regarding the	above statements,	
A) only No. 1 is	true.	
B) only No. 2 is	true.	
C) both No. 1 a	nd No. 2 are true.	
420.	AMA031	AMA
The purpose of	antiskid generators is to	
A) monitor hydr	aulic pressure applied to brakes.	
B) indicate whe	n a tire skid occurs.	
C) measure wh	eel rotational speed and any speed	changes.
421.	AMA010	AMA
The angle of at	tack detector operates from differen	tial pressure when the airstream
A) is parallel to	the longitudinal axis of the aircraft.	
B) is not paralle	el to the true angle of attack of the a	rcraft.
C) is parallel to	the angle of attack of the aircraft.	
422.	AMA069	AMA
(Refer to Airfrar	me figure 20.) What will illuminate th	e amber indicator light?

A) Closing the	nosewheel gear full retract switc	ch.
B) Retarding o	ne throttle and closing the left w	heel gear locked down switch.
C) Closing the	nose, left and right wheel gear f	ull retract switches.
423.	AMA018	AMA
	rpose of a takeoff warning systenset prior to takeoff. The system	em is to alert the crew that a monitored flight control is activated by
A) an 80 knot a	airspeed sensor.	
B) an ignition s	ystem switch not set for takeoff.	
C) a thrust leve	er.	
424.	AMA010	AMA
(1) A dc selsyn movement or p		cal method of indicating a remote mechanical
(2) A synchro to an		trical system used for transmitting information from
Regarding the	above statements,	
A) only No. 1 is	s true.	
B) only No. 2 is	s true.	
C) both No. 1 a	and No. 2 are true.	
425.	AMA068	AMA
Where is the la	nding gear safety switch usually	/ located?
A) On the main	gear shock strut.	
B) On the landi	ng gear drag brace.	
C) On the pilot'	s control pedestal.	
426.	AMA068	AMA
What landing g	ear warning device(s) is/are inc	orporated on retractable landing gear aircraft?
A) A visual indi	cator showing gear position.	
B) A light which	n comes on when the gear is ful	ly down and locked.
C) A horn or ot	her aural device and a red warn	ing light.
427.	AMA010	AMA
The rotor in an	autosyn remote indicating syste	em uses
A) an electrom	agnet.	
B) a permanen	t magnet.	
C) neither an e	lectromagnet nor a permanent i	nagnet.

428.	AMA010	AMA
The rotor in a magnes	yn remote indicating system uses	
A) a permanent magn	et.	
B) an electromagnet.		
C) an electromagnet a	and a permanent magnet.	
429.	AMA041	AMA
Microswitches are use	ed primarily as limit switches to	
A) limit generator outp	ut.	
B) control electrical un	its automatically.	
C) prevent overchargir	ng of a battery.	
420	AMA069	0.840
430. Which renair would re		AMA
•	quire a landing gear retraction test?	
A) Landing gear safety Bod warning light be		
B) Red warning light b		
C) Gear downlock mic	SIOSWILCH.	
431.	AMA041	AMA
What is used as a tem	perature sensing element in an electrically he	eated windshield?
A) Thermocouple.		
B) Thermistor.		
C) Thermometer.		
432.	AMA024	AMA
_		
combustion heaters?	y employed to control the temperature of an a	and leng system using surface
A) Thermo cycling swi	tches.	
B) Thermostats in the		
C) Heater fuel shutoff	•	
-,		
433.	AMA024	AMA
What is the purpose o	f the oil separator in the pneumatic deicing sy	stem?
A) To protect the deice	er boots from oil deterioration.	
B) To remove oil from	air exhausted from the deicer boots.	
C) To prevent an accu	imulation of oil in the vacuum system.	
40.4	A N A A O O A	A B 4 A
434.	AMA024	AMA

•	s in deicer boots alternatel	
•	on of deicer boot tubes ke on of deicer boot tubes do	eps disturbance of the airflow to a minimum. es not disturb airflow.
C) Alternate inflati	on of deicer boot tubes re	lieves the load on the air pump.
435.	AMA024	AMA
What controls the A) Shuttle valve.	inflation sequence in a pr	eumatic deicer boot system?
B) Vacuum pump.		
C) Distributor valv	e.	
436.	AMA024	AMA
	cing system is off?	n of the air pump to hold the deicing boots deflated when
B) Pressure regula		
C) Suction relief v	alve.	
437.	AMA024	AMA
intake ducts. Whe	protected against airframents on is this type of anti-ice synhile the aircraft is in flight.	
,	•	ns to remove ice as it accumulates.
,		untered or expected to occur.
438.	AMA034	AMA
Why should a che	mical rain repellant not be	used on a dry windshield?
A) It will etch the o	glass.	
B) It will restrict vis	sibility.	
C) It will cause gla	ass crazing.	
439.	AMA079	AMA
What is the princip	oal characteristic of a wind	Ishield pneumatic rain removal system?
A) An air blast spr from clinging to th	-	t evenly over the windshield that prevents raindrops
B) An air blast fori	ms a barrier that prevents	raindrops from striking the windshield surface.
, ·	nin removal system is simp matic system pressure.	oly a mechanical windshield wiper system that is

440.	AMA076	AMA
What is one che	ck for proper operation of a pito	t/static tube heater after replacement?
A) Ammeter rea	ding.	
B) Voltmeter rea	ading.	
C) Continuity ch	eck of system.	
441.	AMA046	AMA
A contaminated	carbon monoxide portable test	unit would be returned to service by
A) heating the ir	ndicating element to 300 °F to re	activate the chemical.
B) installing a ne	ew indicating element.	
C) evacuating th	ne indicating element with CO2.	
442.	AMA095	AMA
What occurs wh	en a visual smoke detector is a	ctivated?
A) A warning be	II within the indicator alarms aut	omatically.
B) A lamp withir	the indicator illuminates autom	atically.
C) The test lamp	o illuminates and an alarm is pro	vided automatically.
443.	AMA095	AMA
Smoke detection	n instruments are classified by t	neir method of
A) construction.		
B) maintenance		
C) detection.		
444.	AMA046	AMA
When used in fi	re detection systems having a si	ngle indicator light, thermal switches are wired in
A) parallel with e	each other and in series with the	light.
B) series with ea	ach other and the light.	
C) series with ea	ach other and parallel with the li	ght.
445.	AMA047	AMA
The thermal swi	tches of a bimetallic thermal swi	tch type fire detection (single-loop)system are heat
sensitive units th	nat complete circuits at a certain	temperature. They are connected in
A) parallel with e	each other, and in parallel with t	ne indicator lights.
B) parallel with e	each other, but in series with the	indicator lights.
C) series with ea	ach other, but in parallel with the	e indicator lights.
446.	AMA048	AMA

Which fire extinguishing A) Carbon dioxide. B) Bromotrifluorometha C) Bromochloromethan	,	
447.	AMA048	AMA
A) water, carbon dioxid	uishing agents for aircraft interior fires are e, dry chemical, and halogenated hydrocarbo methyl bromide, and chlorobromomethane.	ons.
C) water, carbon tetrac	hloride, carbon dioxide, and dry chemical.	
448.	AMA048	AMA
In some fire extinguishi indicated by the absend	ng systems, evidence that the system has be	een intentionally discharged is
A) red disk on the side B) green disk on the sid	•	
C) yellow disk on the si	•	
449.	AMA047	AMA
Maintenance of fire det	ection systems includes the	
A) repair of damaged se	ensing elements.	
B) removal of excessive	e loop or element material.	
C) replacement of dame	aged sensing elements.	
450.	AMA072	AMA
Which of the following a system?	are fire precautions which must be observed	when working on an oxygen
1. Display 'No Smoking	' placards.	
2. Provide adequate fire		
	kygen servicing equipment free from oil or gre	ease.
•	aft radio or electrical systems.	
A) 1, 3, and 4.	·	
B) 1, 2, and 4.		
C) 1, 2, 3, and 4.		
451.	AMA048	AMA
The proper fire extingui A) water.	shing agent to use on an aircraft brake fire is	

B) carbon dioxid	de.	
C) dry powder o	chemical.	
452.	AMA046	AMA
A fire extinguish	ner container can be checked	I to determine its charge by
A) attaching a re	emote pressure gauge.	
B) weighing the	container and its contents.	
C) a hydrostatic	test.	
453.	AMA048	AMA
In reference to a	aircraft fire extinguishing syst	ems,
(1) during remo	val or installation, the termina	als of discharge cartridges should be grounded or
		the electrical system, the system should be checked at the terminal connections.
Regarding the a	above statements,	
A) only No. 2 is	true.	
B) both No. 1 ar	nd No. 2 are true.	
C) neither No. 1	nor No. 2 is true.	
454.	AMA048	AMA
•	,	at pressure is acceptable for a fire extinguisher when Rounded to the nearest whole number.)
A) 215 to 302 P	SIG.	
B) 214 to 301 P	SIG.	
C) 215 to 301 P	SIG.	