

**Global Positioning System (GPS)
Standard Positioning Service (SPS)
Performance Analysis Report**

Submitted To

**Federal Aviation Administration
GPS Product Team
AND 730
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Washington, DC 20024**

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Submitted by

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EXECUTIVE SUMMARY

The GPS Product Team (AND 730) has tasked the Navigation Branch (ACT 360) at the William J. Hughes Technical Center to document Global Positioning System (GPS) Standard Positioning Service (SPS) performance in quarterly GPS Performance Analysis (PAN) Reports. The report contains the analysis performed on data collected at the following NSTB and Wide Area Augmentation System (WAAS) Reference Station locations: Atlantic City, Columbus, Denver, Grand Forks, Green Bay, Greenwood, Prescott, Anchorage (WAAS), Billings (WAAS), Chicago (WAAS), Kansas City (WAAS), Salt Lake City (WAAS), Miami (WAAS) and Atlanta (WAAS). This analysis verifies the GPS SPS performance as compared to the performance parameters stated in the SPS Specification Annex A.

This report, Report #37, includes data collected from 1 January through 31 March 2002. The next quarterly report will be issued 31 July 2002.

Analysis of this data includes the following categories: Coverage performance, Service Availability Performance, Position Performance, Range Performance, Solar Storm Effects on GPS SPS performance, GPS/GLONASS performance and WAAS performance.

Coverage performance was based on Position Dilution of Precision (PDOP). Utilizing the weekly almanac posted on the US Coast Guard navigation web site, the coverage for every 5° grid point between 180W to 180E and 80S and 80N was calculated for every minute over a 24-hour period for each of the weeks covered in the reporting period. For this reporting period, the coverage based on PDOP less than six for the CONUS was 99.583% or better.

Availability was verified by reviewing the "Notice: Advisory to Navstar Users" (NANU) reports issued between 1 January and 31 March 2002 and by calculating the satellite availability from the data obtained from the fourteen sites. A total of eighteen outages were reported in the NANU's. Fifteen of the outages were scheduled and three were unscheduled. The quarterly availabilities for Salt Lake City was 99.986%; while Atlantic City, Columbus, Denver, Grand Forks, Green Bay, Greenwood, Prescott, Anchorage, Billings, Chicago, Atlanta, Kansas City and Miami were all 100%. Each of these availabilities is within the SPS value of 99.85%. These availability percentages were calculated using DOP data collected at one-second intervals.

The statistics on the days of significant solar activity met all GPS Standard Positioning Service (SPS) specifications.

Position accuracies were verified by calculating the 95% and 99.99% values of horizontal and vertical errors. Range performance was verified for each satellite using the data collected from the NSTB Atlantic City site. The data was collected in one-second samples. All of the satellites met the range error specifications. The maximum range error recorded was 28.452 meters on Satellite PRN 20. The SPS specification states that the range error should never exceed 150 meters. The maximum range rate error recorded was 0.971435 Meters/second on Satellite PRN 13. The SPS specification states that the range rate error should never exceed 2 meters/second. The maximum range acceleration error recorded was 9.452 Millimeters/second² on Satellite PRN 20. The SPS specification states that the range acceleration error should never exceed 19 Millimeters/second².

A GLONASS/GPS performance section was added to the PAN report. In April 1999, ACT-360 was tasked to monitor, analyze and characterize GLONASS and GPS/GLONASS system performance. The objective of this task is to evaluate the ability of GLONASS to provide navigation by itself and with SPS GPS and to assess the incremental benefit to WAAS obtained from using GLONASS. A GPS/GLONASS receiver was used in the NSTB laboratory at the FAA Technical Center. The GPS/GLONASS performance (from an Ashtech GG24) was compared against GPS-only performance. The 95% horizontal error and vertical error for the GPS/GLONASS solution were 6.576 Meters and 11.034 Meters, respectively.

From the analysis performed on data collected between 1 January and 31 March 2002, the GPS performance met all SPS requirements that were evaluated.

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1.0 Introduction

1.1 Objective of GPS SPS Performance Analysis Report

In 1993, the FAA began monitoring and analyzing Global Positioning System (GPS) Standard Positioning Service (SPS) performance data. At present, the FAA has approved GPS for IFR and is developing Wide Area Augmentation System (WAAS) and Local Area Augmentation (LAAS), both of which are GPS augmentation systems. In order to ensure the safe and effective use of GPS and its augmentation systems within the NAS, it is critical that characteristics of GPS performance as well as specific causes for service outages be monitored and understood. To accomplish this objective, GPS SPS performance data is documented in a quarterly GPS Analysis report. This report contains data collected at the following National Satellite Test Bed (NSTB) and WAAS reference station locations:

- Atlantic City, NJ
- Columbus, NE
- Denver, CO
- Grand Forks, ND
- Elko, NV
- Green Bay, WI
- Greenwood, MS
- Prescott, AZ
- Billings, MT
- Anchorage, AK
- Chicago, IL
- Kansas City, KS
- Salt Lake City, UT
- Miami, FL
- Atlanta, GA

(Future reports will include all WAAS sites but a database that can handle all that data needs to be developed. ACT-360 is in the process of setting up an Oracle database for this purpose.)

The analysis of the data is divided into the four performance categories stated in the Standard Positioning Service Performance Specification (SPS) Annex A (June 2, 1995). These categories are:

- Coverage Performance
- Satellite Availability Performance
- Service Reliability Standard
- Positioning, Ranging and Timing Accuracy Standard.

The results were then compared to the performance parameters stated in the SPS.

1.2 Summary of Performance Requirements and Metrics

Table 1-1 lists the performance parameters from the SPS and identifies those parameters verified in this report.

Appendix E Table 1.2 contains the performance parameters evaluated for the WAAS in this report.

1.3 Report Overview

Section 2 of this report summarizes the results obtained from the coverage calculation program called SPS_CoverageArea developed by ACT-360. The SPS_CoverageArea program uses the GPS satellite almanacs to compute each satellite position as a function of time for a selected day of the week. This program establishes a 5-degree grid between 180 degrees east and 180 degrees west, and from 80 degrees north and 80 degrees south. The program then computes the PDOP at each grid point (1485 total grid points) every minute for the entire day and stores the results. After the PDOP's have been saved the 99.99% index

of 1-minute PDOP at each grid point is determined and plotted as contour lines (Figure 2-1). The program also saves the number of satellites used in PDOP calculation at each grid point for analysis.

Section 3 summarizes the GPS availability performance by providing the "Notice: Advisory to Navstar Users" (NANU) messages to calculate the total time of forecasted and actual satellite outages. This section also includes the maximum and minimum of the PDOP, HDOP and VDOP for each of the thirteen NSTB/WAAS sites.

Section 4 summarizes service reliability performance. It will be reported at the end of the first year of this analysis because the SPS standard is based a measurement interval of one year. Data for the quarter is provided for completeness.

Section 5 provides the position and repeatable accuracies based on data collected on a daily basis at one-second intervals. This section also provides the statistics on the range error, range error rate and range acceleration error for each satellite. The overall average, maximum, minimum and standard deviations of the range rates and accelerations are tabulated for each satellite.

In Section 6, the data collected during solar storms is analyzed to determine the effects, if any, of GPS SPS performance.

Section 7 provides the analysis on GPS/GLONASS performance. A GPS/GLONASS receiver was used in the NSTB laboratory at the FAA Technical Center.

Appendix A provides a summary of all the results as compared to the SPS specification.






Appendix B provides the geomagnetic data used for Section 6.

Appendix C provides a PAN Problem Report. The SPS specification was met in all instances this quarter.

Appendix D provides a glossary of terms used in this PAN report. This glossary was obtained directly from the GPS SPS specification document.

Table 1-1 SPS Performance Requirements

Coverage Standard	Conditions and Constraints	Evaluated in This Report
≥ 99.9% global average	<ul style="list-style-type: none"> Probability of 4 or more satellites in view over any 24 hour interval, averaged over the globe 4 satellites must provide PDOP of 6 or less 5° mask angle with no obscura Standard is predicated on 24 operational satellites, as the constellation is defined in the almanac 	✓
≥ 96.9% at worst-case point	<ul style="list-style-type: none"> Probability of 4 or more satellites in view over any 24 hour interval, for the worst-case point on the globe 4 satellites must provide PDOP of 6 or less 5° mask angle with no obscura Standard is predicated on 24 operational satellites, as the constellation is defined in the almanac 	✓
Satellite Availability Standard	Conditions and Constraints	
≥ 99.85% global average	<ul style="list-style-type: none"> Conditioned on coverage standard Standard based on a typical 24 hour interval, averaged over the globe Typical 24 hour interval defined using averaging period of 30 days 	✓
≥ 99.16% single point average	<ul style="list-style-type: none"> Conditioned on coverage standard Standard based on a typical 24 hour interval, for the worst-case point on the globe Typical 24 hour interval defined using averaging period of 30 days 	✓
≥ 95.87% global average on worst-case day	<ul style="list-style-type: none"> Conditioned on coverage standard Standard represents a worst-case 24 hour interval, averaged over the globe 	✓
≥ 83.92% at worst-case point on worst-case day	<ul style="list-style-type: none"> Conditioned on coverage standard Standard based on a worst-case 24 hour interval, for the worst-case point on the globe 	✓
Service Availability Standard	Conditions and Constraints	
≥ 99.97% global average	<ul style="list-style-type: none"> Conditioned on coverage and service availability standards 500 meter NTE predictable horizontal error reliability threshold Standard based on a measurement interval of one year; average of daily values over the globe Standard predicated on a maximum of 18 hours of major service failure behavior over the sample interval 	✓

<p>≥ 99.79% single point average</p>	<ul style="list-style-type: none"> • Conditioned on coverage and service availability standards • 500 meter Not-to-Exceed (NTE) predictable horizontal error reliability threshold • Standard based on a measurement interval of one year; average of daily values from the worst-case point on the globe • Standard based on a maximum of 18 hours of major service failure behavior over the sample interval 	
<p>Accuracy Standard</p>	<p>Conditions and Constraints</p>	
<p>Predictable Accuracy ≤ 100 m horz. error 95% of time ≤ 156 m vert. error 95% of time ≤ 300 m horz. error 99.99% of time ≤ 500 m vert. error 99.99% of time</p>	<ul style="list-style-type: none"> • Conditioned on coverage, service availability and service reliability standards • Standard based on a measurement interval of 24 hours, for any point on the globe 	
<p>Repeatable Accuracy ≤ 141 m horz. error 95% of time ≤ 221 m vert. error 95% of time</p>	<ul style="list-style-type: none"> • Conditioned on coverage, service availability and service reliability standards • Standard based on a measurement interval of 24 hours, for any point on the globe 	
<p>Relative Accuracy ≤ 1.0 m horz. error 95% of time ≤ 1.5 m vert. error 95% of time</p>	<ul style="list-style-type: none"> • Conditioned on coverage, service availability and service reliability standards • Standard based on a measurement interval of 24 hours, for any point on the globe • Standard presumes that the receivers base their position solutions on the same satellites, with position solutions computed at approximately the same time 	<p>Future Reports</p>
<p>Time Transfer Accuracy ≤ 340 nanoseconds time transfer error 95% of time</p>	<ul style="list-style-type: none"> • Conditioned on coverage, service availability and service reliability standards • Standard based upon SPS receiver time as computed using the output of the position solution • Standard based on a measurement interval of 24 hours, for any point on the globe • Standard is defined with respect to Universal Coordinated Time, as it is maintained by the United States Naval Observatory 	
<p>Range Domain Accuracy ≤ 150 m NTE range error ≤ 2 m/s NTE range rate error ≤ 8 mm/s² range acceleration error 95% of time ≤ 19 mm/s² NTE range acceleration error</p>	<ul style="list-style-type: none"> • Conditioned on satellite indicating healthy status • Standard based on a measurement interval of 24 hours, for any point on the globe • Standard restricted to range domain errors allocated to space/control segments • Standards are not constellation values -- each satellite is required to meet the standards • Assessment requires minimum of four hours of data over the 24 hour period for a satellite in order to evaluate that satellite against the standard 	

2.0 Coverage Performance

Coverage: The percentage of time over a specified time interval that a sufficient number of satellites are above a specified mask angle and provide an acceptable position solution geometry at any point on or near the Earth.

Dilution of Precision (DOP): A Root Mean Square (RMS) measure of the effects that any given position solution geometry has on position errors. Geometry effects may be assessed in the local horizontal (HDOP), local vertical (VDOP), three-dimensional position (PDOP), or time (TDOP) for example.

Coverage Standard	Conditions and Constraints
≥ 99.9% global average	<ul style="list-style-type: none"> • Probability of 4 or more satellites in view over any 24 hour interval, averaged over the globe • 4 satellites must provide PDOP of 6 or less • 5° mask angle with no obscura • Standard is predicated on 24 operational satellites, as the constellation is defined in the almanac
≥ 96.9% at worst-case point	<ul style="list-style-type: none"> • Probability of 4 or more satellites in view over any 24 hour interval, for the worst-case point on the globe • 4 satellites must provide PDOP of 6 or less • 5° mask angle with no obscura • Standard is predicated on 24 operational satellites, as the constellation is defined in the almanac

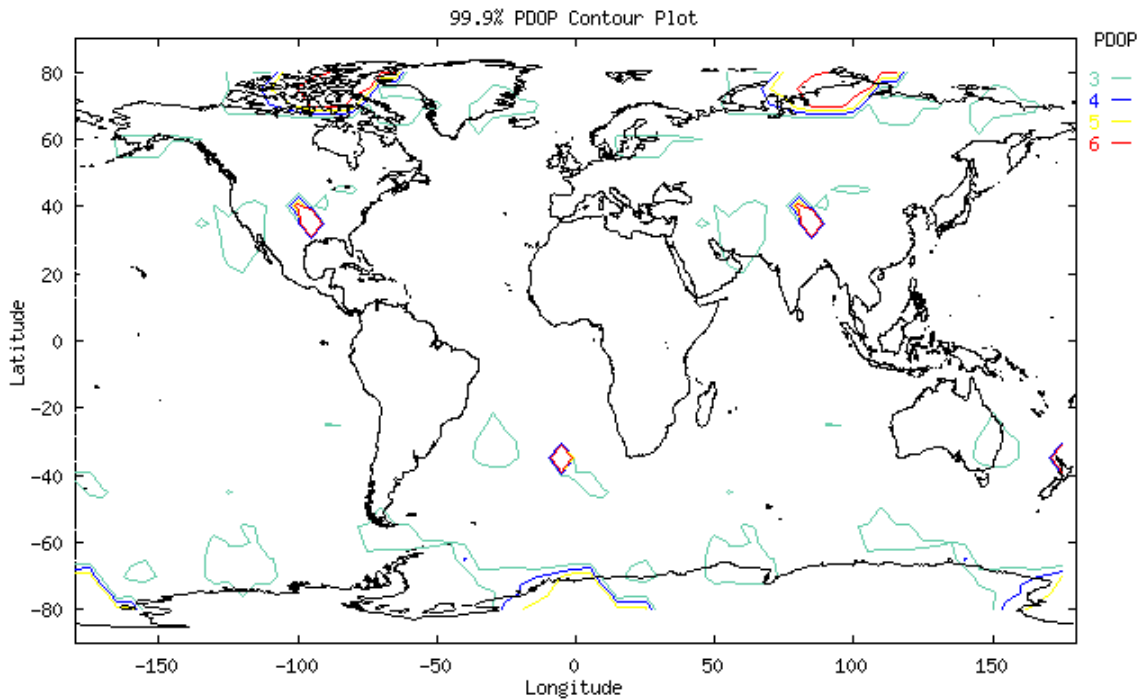
Almanacs for GPS weeks 97-109 used for this coverage portion of the report were obtained from the Coast Guard web site (www.navcen.uscg.mil). Using these almanacs, an SPS coverage area program developed by ACT-360 was used to calculate the PDOP at every 5° point between longitudes of 180W to 180E and 80S and 80N at one-minute intervals. This gives a total of 1440 samples for each of the 2376 grid points in the coverage area. Table 2-1 provides the global averages and worst-case availability over a 24-hour period for each week. Table 2-1 also gives the global 99.9% PDOP value for each of the thirteen GPS Weeks. The PDOP was 3.19503 or better 99.9% for each of the 24-hour intervals.

The GPS coverage performance evaluated met the specifications stated in the SPS.

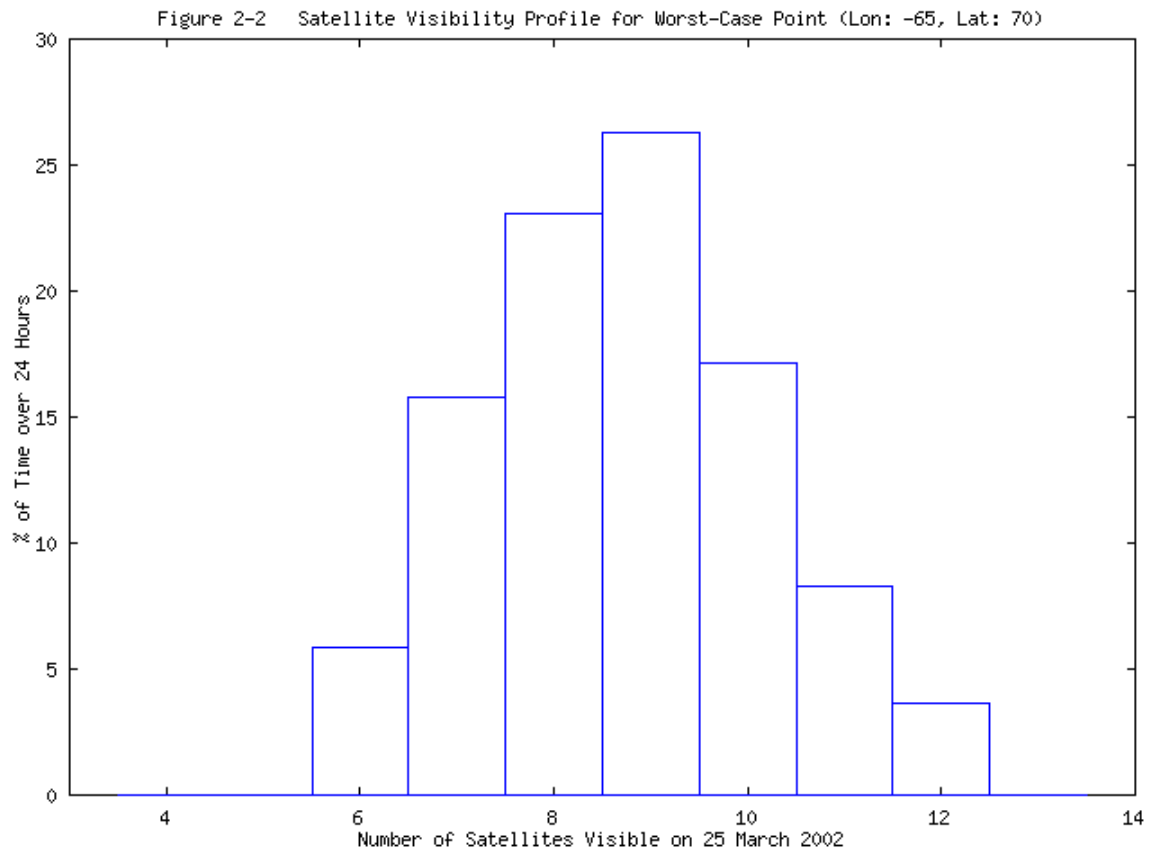
Table 2-1 Coverage Statistics

GPS Week	Global 99.9% PDOP Value*	Global Average* (Spec: $\geq 99.9\%$)	Worst-Case Point (Spec: $\geq 96.9\%$)
123	3.02986	100%	99.931%
124	3.08791	100%	100%
125	3.19503	100%	100%
126	3.18854	100%	100%
127	3.18816	100%	99.792%
128	3.13980	100%	99.931%
129	3.11303	100%	100%
130	3.09211	100%	100%
131	3.07164	100%	100%
132	3.04833	100%	100%
133	3.05545	100%	100%
134	3.09082	100%	99.931%
135	3.12436	99.999%	99.653%

Figure 2-1 SPS Coverage (24-Hour Period: 25 March 2002)



Developed by FAA William J. Hughes Technical Center



3.0 Service Availability Performance

Service Availability: Given coverage, the percentage of time over a specified time interval that a sufficient number of satellites are transmitting a usable ranging signal within view of any point on or near the Earth.

3.1 Satellite Outages from NANU Reports

Satellite availability performance was analyzed based on published "Notice: Advisory to Navstar Users" messages (NANU's). During this reporting period, 1 January through 31 March 2002, there were a total of seventeen reported outages. Twelve of these outages were maintenance activities and were reported in advance. Five were unscheduled outages. A complete listing of outage NANU's for the reporting period is provided in Table 3-1. A complete listing of the forecasted outage NANU's for the reporting period can be found in Table 3-2. Canceled outage NANU's are provided in Table 3-3.

Table 3-1 NANUs Affecting Satellite Availability									
NANU #	PRN	Type	Start Date	Start Time	End Date	End Time	Total	Total	Total
							Unscheduled	Scheduled	
2003	23	U	7-Jan	19:00	N/A	N/A			
2006	24	U	31-Dec	18:03	7-Jan	15:19	165.27		165.27
2008	11	S	7-Jan	5:21	7-Jan	7:49		2.47	2.47
2011	21	S	10-Jan	6:15	10-Jan	12:18		6.05	6.05
2013	23	U	7-Jan	19:00	14-Jan	22:06	171.10		171.10
2014	26	S	15-Jan	5:39	15-Jan	12:26		6.78	6.78
2015	8	S	16-Jan	14:59	16-Jan	23:03		7.07	7.07
2019	20	S	29-Jan	5:20	29-Jan	7:06		1.77	1.77
2021	18	S	31-Jan	19:01	31-Jan	21:15		2.23	2.23
2023	3	S	5-Feb	1:16	5-Feb	7:47		6.52	6.52
2024	4	S	7-Feb	8:18	7-Feb	14:57		6.65	6.65
2026	5	S	11-Feb	16:16	11-Feb	21:15		4.98	4.98
2028	10	S	14-Feb	13:18	14-Feb	16:31		3.22	3.22
2031	23	S	21-Feb	22:40	22-Feb	4:07		5.45	5.45
2032	6	S	26-Feb	20:22	27-Feb	2:40		6.30	6.30
2033	18	S	28-Feb	0:57	28-Feb	5:07		4.17	4.17
2037	24	U	9-Mar	18:22	N/A	N/A			
2038	24	U	9-Mar	18:22	9-Mar	18:38	0.27		0.27
2039	22	S	18-Mar	22:01	19-Mar	8:47		10.77	10.77
2043	18	S	25-Mar	22:59	26-Mar	3:24		4.42	4.42
Total Actual Unscheduled and Scheduled Downtime and Total Actual Downtime							336.64	78.85	415.49
Type:	S = Scheduled		U = Unscheduled						

Table 3-2 NANUs Forecasted to Affect Satellite Availability								
NANU #	PRN	Type	Start Date	Start Time	End Date	End Time	Total	Comments
2002	21	F	10-Jan	6:00	10-Jan	18:00	12	See NANU 2011
2005	11	F	8-Jan	4:30	8-Jan	16:30	12	See NANU 2008
2010	26	F	15-Jan	5:15	15-Jan	17:15	12	See NANU 2014
2012	8	F	16-Jan	14:45	17-Jan	2:45	12	See NANU 2015
2016	20	F	29-Jan	5:00	29-Jan	17:00	12	See NANU 2019
2017	18	F	31-Jan	18:00	1-Feb	6:00	12	See NANU 2021
2018	3	F	5-Feb	1:00	5-Feb	13:00	12	See NANU 2023
2020	4	F	7-Feb	8:00	7-Feb	20:00	12	See NANU 2024
2022	5	F	11-Feb	16:00	12-Feb	4:00	12	See NANU 2026
2025	10	F	14-Feb	12:30	15-Feb	0:30	12	See NANU 2028
2027	23	F	21-Feb	22:15	22-Feb	10:15	12	See NANU 2031
2029	6	F	26-Feb	19:45	27-Feb	7:45	12	See NANU 2032
2030	18	F	28-Feb	0:00	28-Feb	12:00	12	See NANU 2033
2034	22	F	18-Mar	21:00	19-Mar	9:00	12	See NANU 2039
2040	18	F	25-Mar	22:00	26-Mar	10:00	12	See NANU 2043
2041	30	F	26-Mar	10:30	26-Mar	22:30	12	See NANU 2042
Total Forecast Downtime							180	

Table 3-3 NANUs Canceled					
NANU#	PRN	Type	Start Date	Start Time	Comments
2042	30	C	26-Mar	10:30	See NANU 2041

Satellite Reliability, Maintainability, and Availability (RMA) data is being collected based on published "Notice: Advisory to Navstar Users" messages (NANU's). This data has been summarized in Table 3-4. The "Total Satellite Observed MTTR" was calculated by taking the average downtime of all satellite outage occurrences. Schedule downtime was forecasted in advance via NANU's. All other downtime reported via NANU was considered unscheduled. The "Percent Operational" was calculated based on the ratio of total actual operating hours to total available operating hours for every satellite.

Table 3-4 GPS Block II/IIA Satellite RMA Data		
Satellite Reliability/Maintainability/Availability (RMA) Parameter	1 Jan - 31 Mar, 2002	1 October, 1999- 31 Mar, 2002
Total Forecast Downtime (hrs):	180.00	2408.25
Total Actual Downtime (hrs):	415.49	4104.47
Total Actual Scheduled Downtime (hrs):	78.85	1318.76
Total Actual Unscheduled Downtime (hrs):	336.64	2785.71
Total Satellite Observed MTTR (hrs):	23.08	21.83
Scheduled Satellite Observed MTTR (hrs):	5.26	9.03
Unscheduled Satellite Observed MTTR (hrs):	112.21	66.33
# Total Satellite Outages:	18	188
# Scheduled Satellite Outages:	15	146
# Unscheduled Satellite Outages:	3	42
Percent Operational -- Scheduled Downtime:	99.87	99.78
Percent Operational -- All Downtime:	99.94	99.33

3.2 Service Availability

Service Availability Standard	Conditions and Constraints
≥ 99.85% global average	<ul style="list-style-type: none"> • Conditioned on coverage standard • Standard based on a typical 24 hour interval, averaged over the globe • Typical 24 hour interval defined using averaging period of 30 days
≥ 99.16% single point average	<ul style="list-style-type: none"> • Conditioned on coverage standard • Standard based on a typical 24 hour interval, for the worst-case point on the globe • Typical 24 hour interval defined using averaging period of 30 days
≥ 95.87% global average on worst-case day	<ul style="list-style-type: none"> • Conditioned on coverage standard • Standard represents a worst-case 24 hour interval, averaged over the globe
≥ 83.92% at worst-case point on worst-case day	<ul style="list-style-type: none"> • Conditioned on coverage standard • Standard based on a worst-case 24 hour interval, for the worst-case point on the globe

To verify availability, the data collected from receivers at the nine NSTB/WAAS sites was reduced to calculate DOP information and reported in Tables 3-5 to 3-7. The data was collected at one-second intervals between 1 January and 31 March 2002.

Table 3-5 PDOP Statistics

NSTB/WAAS Site	Min PDOP	Max PDOP	VDOP at Max PDOP	Mean PDOP	99.99% PDOP	99.99% VDOP	Number of Samples
Atlantic City	1.206	5.469	4.547	1.855	3.827	3.342	7775714
Columbus	1.180	6.000	5.283	1.794	5.582	5.031	7774276
Denver	1.180	5.993	5.132	1.791	5.534	4.846	7597273
Grand Forks	1.167	5.605	5.280	1.769	5.211	4.898	7745735
Green Bay	1.128	5.940	5.124	1.804	5.478	5.071	7462989
Greenwood	1.264	6.000	5.475	1.855	4.675	4.188	7451187
Prescott	1.349	5.997	5.115	2.128	5.406	5.159	7200532
Billings	1.099	4.743	4.028	1.758	4.661	3.989	7307633
Anchorage	1.211	5.999	5.454	1.783	4.896	4.298	7715890
Chicago	1.247	5.296	4.815	1.786	4.973	4.513	7529430
Kansas City	1.203	6.000	5.740	1.798	5.654	5.029	7573127
Salt Lake City	1.172	6.991	6.258	1.769	6.342	5.688	7731398
Miami	1.208	5.323	4.898	1.809	4.260	3.782	7319221
Atlanta	1.248	5.365	5.056	1.856	4.414	3.944	7556438

Tables 3-6 and 3-7 show the statistics related to maximum PDOP and PDOP greater than six, respectively. Table 3-6 shows the PDOP statistics for the worst-case point on the worst-case day. NOTE: Global in this report refers to the nine sites used. Although future reports will have all WAAS sites, a true global availability cannot be determined since there aren't reference stations around the world.

Whenever the PDOP goes above six and an SPS requirement is not met, an investigation is performed to determine what caused the PDOP to go above six. The following is a list of programs/procedures used during times of high PDOP:

- Notice of Advisory to Navstar Users (NANU's) messages are used to verify that satellite outages did occur. (See Section 3.1 for more details about NANU's for this quarter.)
- A satellite outage detection program developed by ACT-360 verifies satellite outages that are not verified through a NANU. For example, a satellite outage can occur for just a few seconds during an upload. This satellite detection program monitors all the receivers and keeps track of what satellites the receiver should be tracking versus what satellites the receiver is actually tracking. At least six receivers need to be tracking the satellite prior to the outage and no receiver can be tracking the satellite for the program to detect an outage. This program is also being enhanced so that false locks and late ephemeris problems can also be detected. This program will also output flags from the receivers so that problems with the receiver or TRS software, if any, can be tracked more easily.
- Data from co-located receivers is analyzed for times that the PDOP goes above six. This helps in determining whether the problem is due to the environment.

The instance of worst performance where the PDOP went above six is reported in Table 3-6. The column labeled "NANU/SOD" reports whether the outage was detected via a NANU or the Satellite Outage Detection (SOD) program along with the Satellite PRN number that had the outage.

Table 3-6 Maximum PDOP Statistics

Site	GPS Week/ Day	Max PDOP	Number of Seconds of Whole Day PDOP > 6	NANU/SOD, Satellite PRN Number	Number of Samples	Availability on days when PDOP > 6
Salt Lake City	1147_6	6.982	367	None	86391	99.575
Salt Lake City	1148_0	6.989	358	None	86390	99.585
Salt Lake City	1148_1	6.991	347	2013, PRN23	86352	99.598
Worst-Case Point on Worst-Case Day = 99.575% (SPS Spec. \geq 83.92%)						

Global Average on Worst-Case Day = 99.967% (SPS Spec. \geq 95.87%)

Table 3-7 PDOP > 6 Statistics

NSTB/WAAS Site	Total Number of Seconds of PDOP Monitoring	Total Seconds with PDOP > 6	Overall % Availability
Atlantic City	7775714	0	100%
Columbus	7774276	0	100%
Denver	7597273	0	100%
Grand Forks	7745735	0	100%
Green Bay	7462989	0	100%
Greenwood	7451187	0	100%
Prescott	7200532	0	100%
Billings	7307633	0	100%
Anchorage	7715890	0	100%
Chicago	7529430	0	100%
Kansas City	7573127	0	100%
Salt Lake City	7731398	1072	99.986%
Miami	7319221	0	100%
Atlanta	7556438	0	100%
Worst Single Point Average = 99.986% (SPS Spec. \geq 99.16%)			

Global Average over Reporting Period = 99.999% (SPS Spec. > 99.85%)

4.0 Service Reliability Standard

Service Reliability: Given coverage and service availability, the percentage of time over a specified time interval that the instantaneous predictable horizontal error is maintained within a specified threshold at any point on or near the Earth.

Service Reliability Standard	Conditions and Constraints
≥ 99.97% global average	<ul style="list-style-type: none"> • Conditioned on coverage and service availability standards • 500 meter NTE predictable horizontal error reliability threshold • Standard based on a measurement interval of one year; average of daily values over the globe • Standard predicated on a maximum of 18 hours of major service failure behavior over the sample interval
≥ 99.79% single point average	<ul style="list-style-type: none"> • Conditioned on coverage and service availability standards • 500 meter Not-to-Exceed (NTE) predictable horizontal error reliability threshold • Standard based on a measurement interval of one year; average of daily values from the worst-case point on the globe • Standard based on a maximum of 18 hours of major service failure behavior over the sample interval

Table 4-1 has the 99.99% horizontal errors reported by a receiver at each of the fourteen NSTB/WAAS sites. This will be evaluated against the SPS specification at the end of the year.

Table 4-1 Service Reliability Based on Horizontal Error

NSTB/WAAS Site	Number of Samples This Quarter	Maximum Horizontal Error (Meters)
Atlantic City	7775714	16.6
Columbus	7774276	11.1
Denver	7597273	33.3
Grand Forks	7745735	16.3
Green Bay	7462989	18.8
Greenwood	7451187	11.0
Prescott	7200532	12.0
Billings	7307633	12.0
Anchorage	7715890	30.7
Chicago	7529430	13.3
Kansas City	7573127	11.6
Salt Lake City	7731398	12.0
Miami	7319221	13.8
Atlanta	7556438	10.8

5.0 Accuracy Characteristics

Accuracy: Given coverage, service availability and service reliability, the percentage of time over a specified time interval that the difference between the measured and expected user position or time is within a specified threshold at any point on or near the Earth.

Accuracy Standard	Conditions and Constraints
Predictable Accuracy ≤ 100 meters horizontal error 95% of time ≤ 156 meters vertical error 95% of time ≤ 300 meters horizontal error 99.99% of time ≤ 500 meters vertical error 99.99% of time	<ul style="list-style-type: none"> • Conditioned on coverage, service availability and service reliability standards • Standard based on a measurement interval of 24 hours, for any point on the globe
Repeatable Accuracy ≤ 141 meters horizontal error 95% of time ≤ 221 meters vertical error 95% of time	<ul style="list-style-type: none"> • Conditioned on coverage, service availability and service reliability standards • Standard based on a measurement interval of 24 hours, for any point on the globe
Relative Accuracy ≤ 1.0 meters horizontal error 95% of time ≤ 1.5 meters vertical error 95% of time	<ul style="list-style-type: none"> • Conditioned on coverage, service availability and service reliability standards • Standard based on a measurement interval of 24 hours, for any point on the globe • Standard presumes that the receivers base their position solutions on the same satellites, with position solutions computed at approximately the same time
Time Transfer Accuracy ≤ 340 nanoseconds time transfer error 95% of time	<ul style="list-style-type: none"> • Conditioned on coverage, service availability and service reliability standards • Standard based upon SPS receiver time as computed using the output of the position solution • Standard based on a measurement interval of 24 hours, for any point on the globe • Standard is defined with respect to Universal Coordinated Time, as it is maintained by the United States Naval Observatory
Range Domain Accuracy ≤ 150 meters NTE range error ≤ 2 meters/second NTE range rate error ≤ 8 millimeters/second ² range acceleration error 95% of time ≤ 19 millimeters/second ² NTE range acceleration error	<ul style="list-style-type: none"> • Conditioned on satellite indicating healthy status • Standard based on a measurement interval of 24 hours, for any point on the globe • Standard restricted to range domain errors allocated to space/control segments • Standards are not constellation values -- each satellite is required to meet the standards • Assessment requires minimum of four hours of data over the 24 hour period for a satellite in order to evaluate that satellite against the standard

5.1 Position Accuracies

The data used for this section was collected for every second between 1 January through 31 March 2002 at the NSTB and WAAS selected locations.

Table 5-1 provides the 95% and 99.99% horizontal and vertical error accuracies for the quarter.

Table 5-1 Horizontal & Vertical Accuracy Statistics for the Quarter

NSTB Site	95% Horizontal (Meters)	95% Vertical (Meters)	99.99% Horizontal (Meters)	99.99% Vertical (Meters)
Atlantic City	5.896	11.209	15.160	20.123
Columbus	5.459	10.312	10.278	17.245
Denver	5.279	11.101	10.021	20.182
Grand Forks	6.282	9.099	13.878	15.062
Green Bay	6.027	9.709	12.926	17.141
Greenwood	5.259	11.351	9.630	22.609
Prescott	5.007	11.918	10.761	21.503
Billings	6.264	9.673	11.946	17.829
Anchorage	7.226	7.860	14.051	18.702
Chicago	5.758	9.842	11.976	16.014
Kansas City	5.518	10.165	10.758	17.878
Salt Lake City	5.418	11.062	10.665	21.323
Miami	6.040	13.574	12.344	22.243
Atlanta	5.522	11.194	9.893	19.758

Figures 5-1 and 5-2 are the combined histograms of the vertical and horizontal errors for all fourteen NSTB and WAAS sites from 1 January to 31 March 2002.

Figure 5-1 Combined Vertical Error Histogram

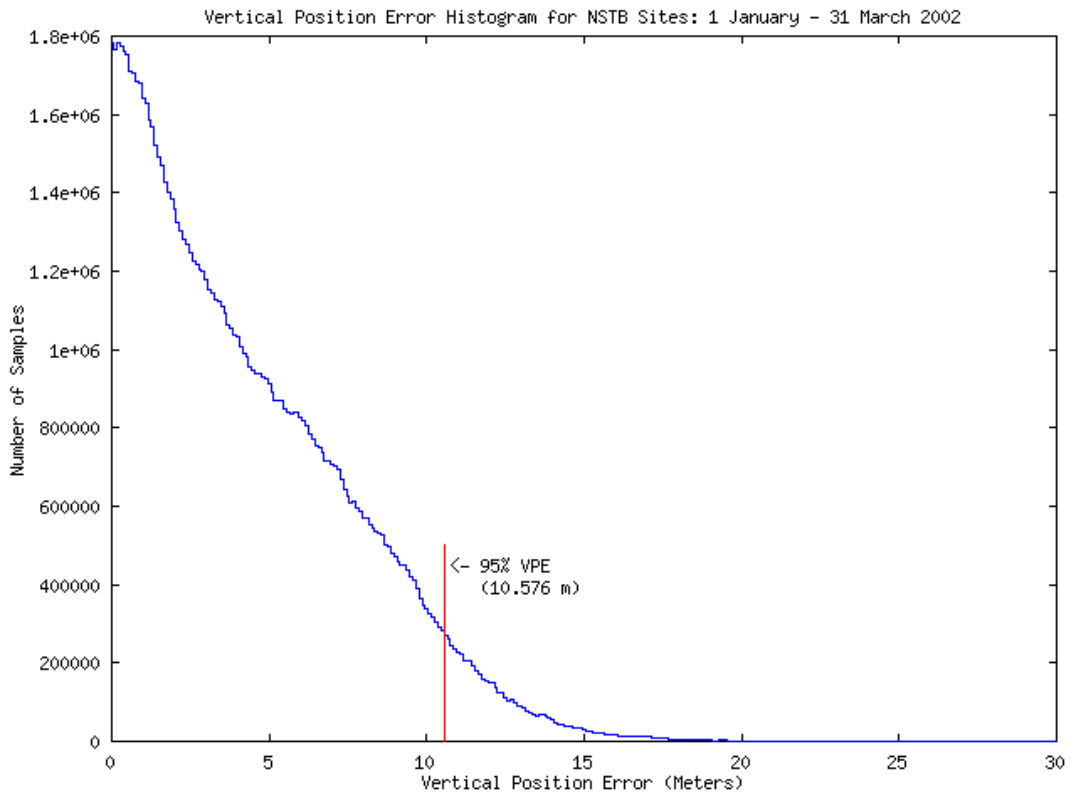
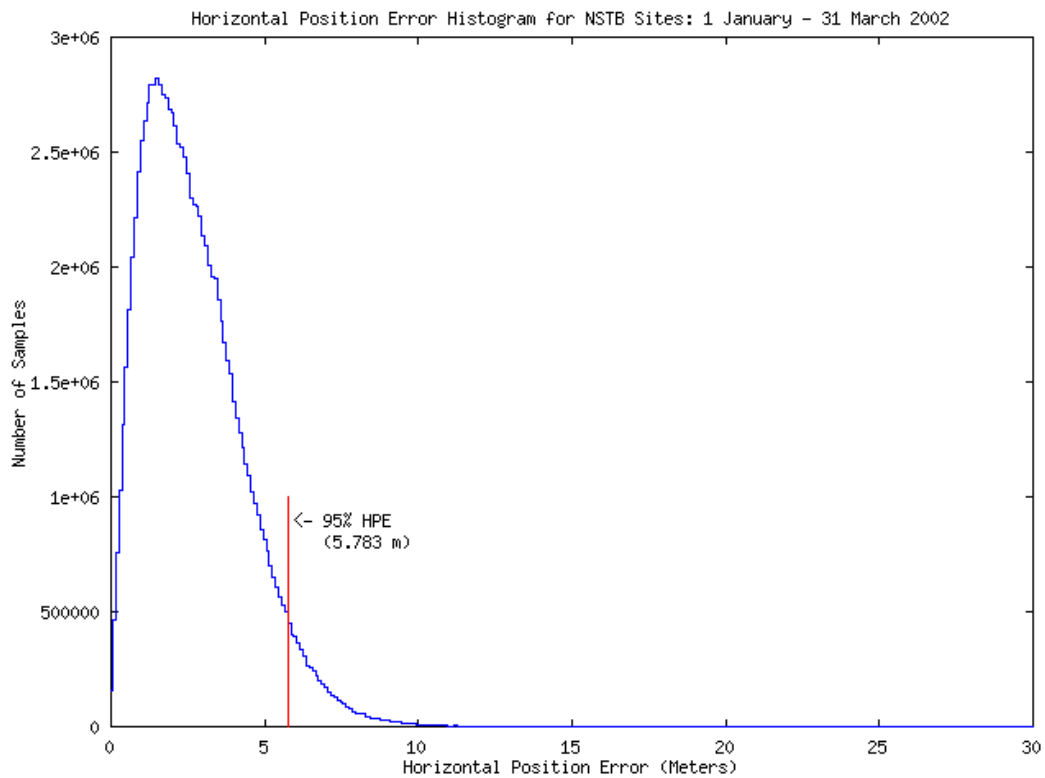


Figure 5-2 Combined Horizontal Error Histogram



5.2 Repeatable Accuracy

Table 5-2 provides the repeatability statistics, which met all of the evaluated requirements stated in the SPS.

Table 5-2 Repeatability Statistics

NSTB Site	95% Horizontal (m)	95% Vertical (m)
Atlantic City	1.569	4.029
Columbus	1.567	4.200
Denver	1.727	4.715
Grand Forks	1.815	5.300
Green Bay	1.919	5.251
Greenwood	1.573	4.925
Prescott	1.604	4.292
Billings	1.870	5.037
Anchorage	2.449	4.364
Chicago	1.714	4.189
Kansas City	1.426	3.957
Salt Lake City	1.571	4.494
Miami	1.655	4.200
Atlanta	1.733	4.392

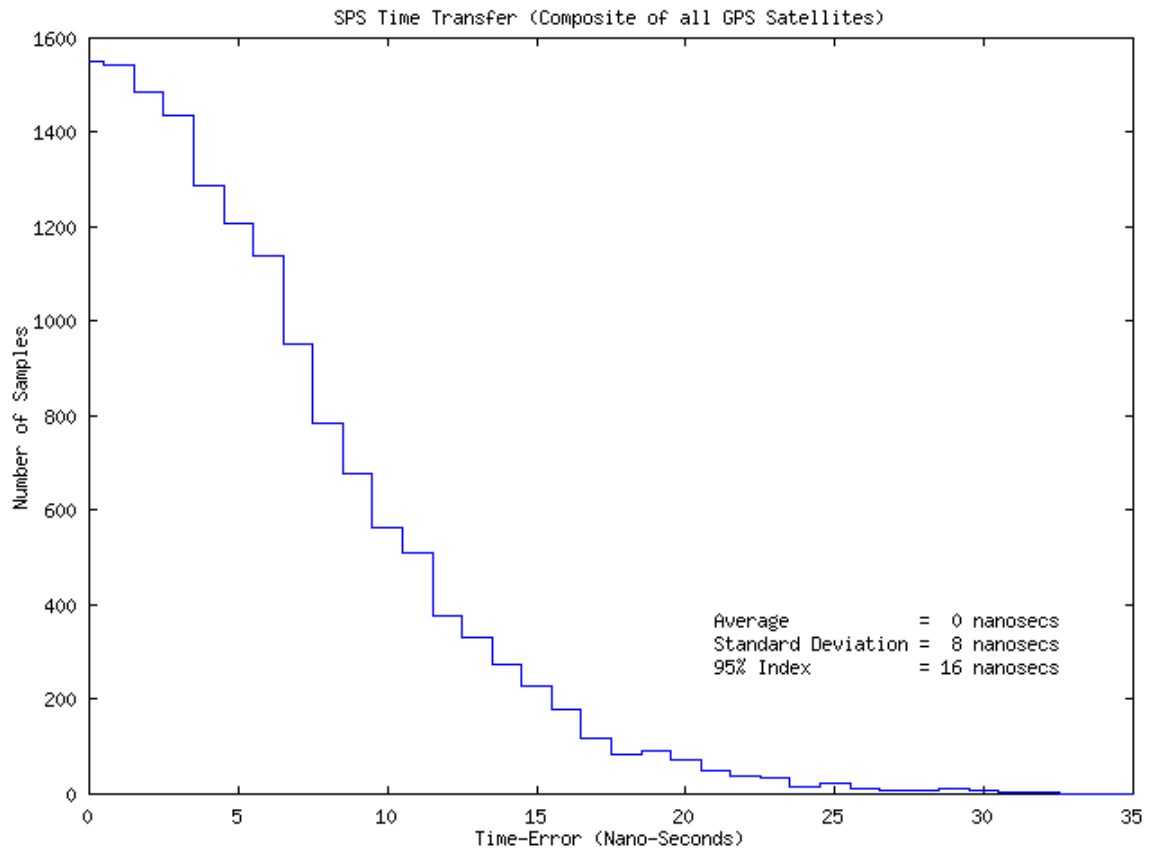
5.3 Relative Accuracy

To be included in future reports.

5.4 Time Transfer Accuracy

The GPS time error data between 1 January and 31 March 2002 was down loaded from USNO Internet site. The USNO data file contains the time difference between the USNO master clock and GPS system time for each GPS satellites during the time period. Over 10,000 samples of GPS time error are contained in the USNO data file. In order to evaluate the GPS time transfer error, the data file was used to create a histogram (Fig 5-3) to represent the distribution of GPS time error. The histogram was created by taking the absolute value of time difference between the USNO master clock and GPS system time, then creating data bins with one nanosecond precision. The number of samples in each bin was then plotted to form the histogram in Fig 5-3. The mean, standard deviation, and 95% index are within the requirements of GPS SPS time error.

Figure 5-3 Time Transfer Errors



5.5 Range Domain Accuracy

Tables 5-3 through 5-5 provide the statistical data for the range error, range rate error and the range acceleration error for each satellite. This data was collected between 1 January and 31 March 2002. The Millennium at Anderson was used to collect range measurement. Future PAN reports will contain statistics from all WAAS sites.

A weighted average filter was used for the calculation of the range rate error and the range acceleration error. All Range Domain SPS specifications were met.

Table 5-3 Range Error Statistics (meters)

PRN	Range Error Mean	Range Error RMS	1s	95% Range Error	Max Range Error (SPS Spec. \leq 150 m)	Samples
1	-3.962	5.181	3.339	9.634	15.573	2219787
2	-3.827	5.494	3.942	10.828	17.780	2268682
3	-3.005	4.899	3.869	9.886	18.395	1902098
4	-4.275	5.589	3.600	10.119	14.962	2153961
5	-0.533	2.711	2.658	5.418	16.321	2518474
6	-0.972	2.852	2.681	5.831	11.854	2404861
7	-4.679	5.893	3.583	10.180	15.589	2240755
8	-4.300	5.738	3.799	9.969	18.947	2163005
9	-1.318	3.278	3.001	6.534	12.513	2228896
10	-2.132	4.119	3.524	7.938	14.391	2093923
11	-4.351	5.785	3.812	10.877	17.797	2124430
13	-4.873	5.766	3.082	10.043	18.458	2436668
14	-0.290	2.459	2.441	4.899	11.619	2210290
15	-1.288	3.770	3.543	7.384	19.430	1957726
17	-1.727	3.844	3.434	7.757	11.993	1836350
18	-1.494	3.245	2.881	6.307	11.183	2116485
20	-2.636	4.126	3.175	7.677	28.452	2501480
21	-1.051	3.265	3.092	6.653	14.831	1954118
22	-1.264	3.459	3.220	6.983	13.010	1978277
23	-1.243	3.591	3.369	7.117	11.056	2062283
24	-3.902	5.323	3.620	9.945	14.018	2098916
25	-0.114	2.656	2.654	5.365	9.921	2237666
26	-2.408	4.132	3.359	7.897	14.867	1832979
27	-6.228	7.456	4.099	12.637	19.041	1811229
28	-5.092	6.304	3.716	11.170	19.356	1999786
29	-0.971	3.000	2.838	6.249	11.259	2227081
30	-0.686	2.451	2.353	4.966	21.076	2461048
31	-4.339	6.006	4.154	11.212	17.490	1877835

Table 5-4 Range Rate Error Statistics (meters/second)

PRN	Range Rate Error Mean	Range Rate Error RMS	Range Rate Error 1s	95% Range Rate Error	Max Range Rate Error (SPS Spec. ≤ 2 m)	Samples
1	0.00003	0.00275	0.00274	0.0047	0.893348	2219787
2	0.00011	0.00316	0.00316	0.00461	0.237918	2268682
3	-0.00011	0.00386	0.00386	0.00474	0.465037	1902098
4	-0.00015	0.00242	0.00241	0.00424	0.147574	2153961
5	0.00007	0.00653	0.00653	0.00402	0.646728	2518474
6	-0.00007	0.00419	0.00419	0.00417	0.495731	2404861
7	0.00004	0.00238	0.00238	0.00413	0.144026	2240755
8	0.00028	0.00256	0.00254	0.00439	0.153678	2163005
9	0.00009	0.00405	0.00405	0.00428	0.484189	2228896
10	-0.00021	0.00287	0.00286	0.00420	0.161983	2093923
11	-0.00019	0.00501	0.00501	0.00481	0.485909	2124430
13	-0.00006	0.00583	0.00583	0.00480	0.971435	2436668
14	0.00007	0.00193	0.00193	0.00364	0.146641	2210290
15	-0.00011	0.00337	0.00337	0.00446	0.379788	1957726
17	-0.00015	0.00328	0.00327	0.00405	0.343642	1836350
18	-0.00013	0.00223	0.00222	0.00409	0.03694	2116485
20	-0.00035	0.00840	0.00840	0.00461	0.68308	2501480
21	0.00006	0.00307	0.00307	0.00418	0.239722	1954118
22	0.00007	0.00279	0.00279	0.00418	0.24731	1978277
23	-0.00021	0.00251	0.00250	0.00404	0.211268	2062283
24	-0.00008	0.00247	0.00247	0.00408	0.234369	2098916
25	0.00015	0.00276	0.00276	0.00391	0.180027	2237666
26	-0.00026	0.00359	0.00358	0.00418	0.508695	1832979
27	0.00035	0.00347	0.00346	0.00491	0.223539	1811229
28	0.00023	0.00280	0.00279	0.00447	0.196144	1999786
29	0.00008	0.00226	0.00226	0.00375	0.221652	2227081
30	0.00012	0.00709	0.00709	0.00403	0.945772	2461048
31	-0.00020	0.00405	0.00405	0.00503	0.868046	1877835

Table 5-5 Range Acceleration Error Statistics (meters/second²)

PRN	Range Acceleration Error Mean	Range Acceleration Error RMS	Range Acceleration 1s	% ≤ 0.008 (SPS Spec. 95% of Time)	Max Range Acceleration Error (SPS Spec. ≤ 0.019 m/s ²)	Samples
1	0.00000	0.00002	0.00002	100%	0.004595	2219787
2	0.00000	0.00002	0.00002	100%	0.002393	2268682
3	0.00000	0.00003	0.00003	100%	0.004654	1902098
4	0.00000	0.00002	0.00002	100%	0.001461	2153961
5	0.00000	0.00006	0.00006	100%	0.006468	2518474
6	0.00000	0.00004	0.00004	100%	0.004962	2404861
7	0.00000	0.00002	0.00002	100%	0.00144	2240755
8	0.00000	0.00002	0.00002	100%	0.001542	2163005
9	0.00000	0.00004	0.00004	100%	0.004849	2228896
10	0.00000	0.00002	0.00002	100%	0.001603	2093923
11	0.00000	0.00004	0.00004	100%	0.004842	2124430
13	0.00000	0.00005	0.00005	100%	0.006026	2436668
14	0.00000	0.00001	0.00001	100%	0.001473	2210290
15	0.00000	0.00003	0.00003	100%	0.003803	1957726
17	0.00000	0.00003	0.00003	100%	0.003397	1836350
18	0.00000	0.00001	0.00001	100%	0.000361	2116485
20	0.00000	0.00008	0.00008	100%	0.006872	2501480
21	0.00000	0.00003	0.00003	100%	0.002425	1954118
22	0.00000	0.00002	0.00002	100%	0.002482	1978277
23	0.00000	0.00002	0.00002	100%	0.002113	2062283
24	0.00000	0.00002	0.00002	100%	0.002321	2098916
25	0.00000	0.00002	0.00002	100%	0.00181	2237666
26	0.00000	0.00003	0.00003	100%	0.005078	1832979
27	0.00000	0.00003	0.00003	100%	0.002195	1811229
28	0.00000	0.00002	0.00002	100%	0.001938	1999786
29	0.00000	0.00002	0.00002	100%	0.002201	2227081
30	0.00000	0.00007	0.00007	99.999%	0.009452	2461048
31	0.00000	0.00003	0.000073	99.999%	0.008687	1877835

Figures 5-4, 5-5 and 5-6 are graphical representations of the distributions of the maximum range error, range rate error and range acceleration error for all satellites. None of the range errors for any of the satellites exceeded the 150-meter SPS requirement. The highest maximum range error occurred on satellite 20 with an error of 28.452 meters. Satellite 25 had the lowest maximum range error of 9.921 meters.

Figure 5-4 Distribution of Daily Max Range Errors

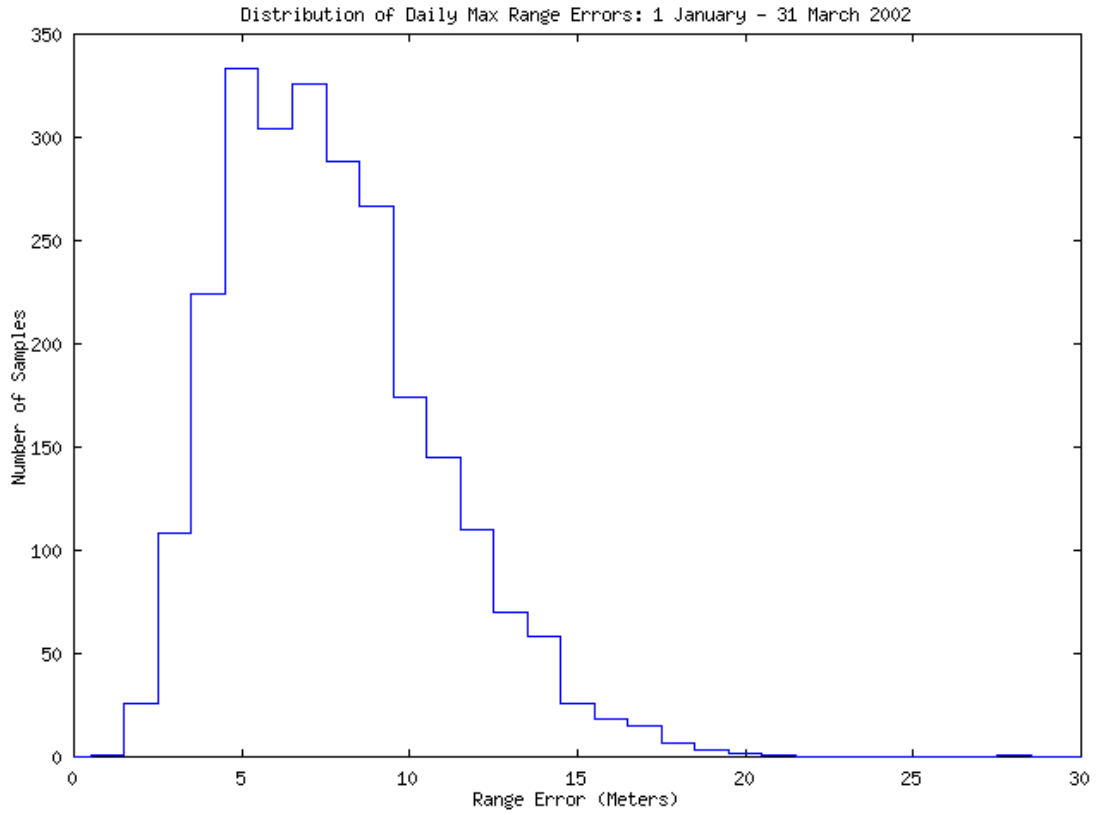


Figure 5-5: Distribution of Daily Max Range Rate Errors

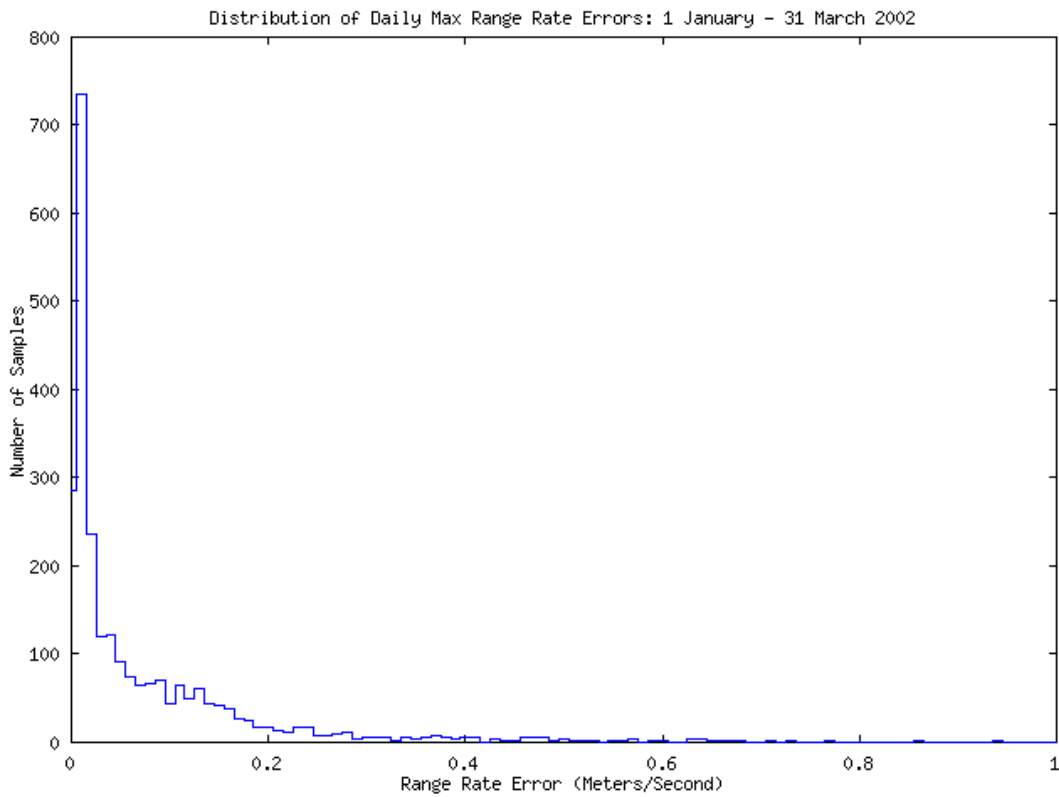
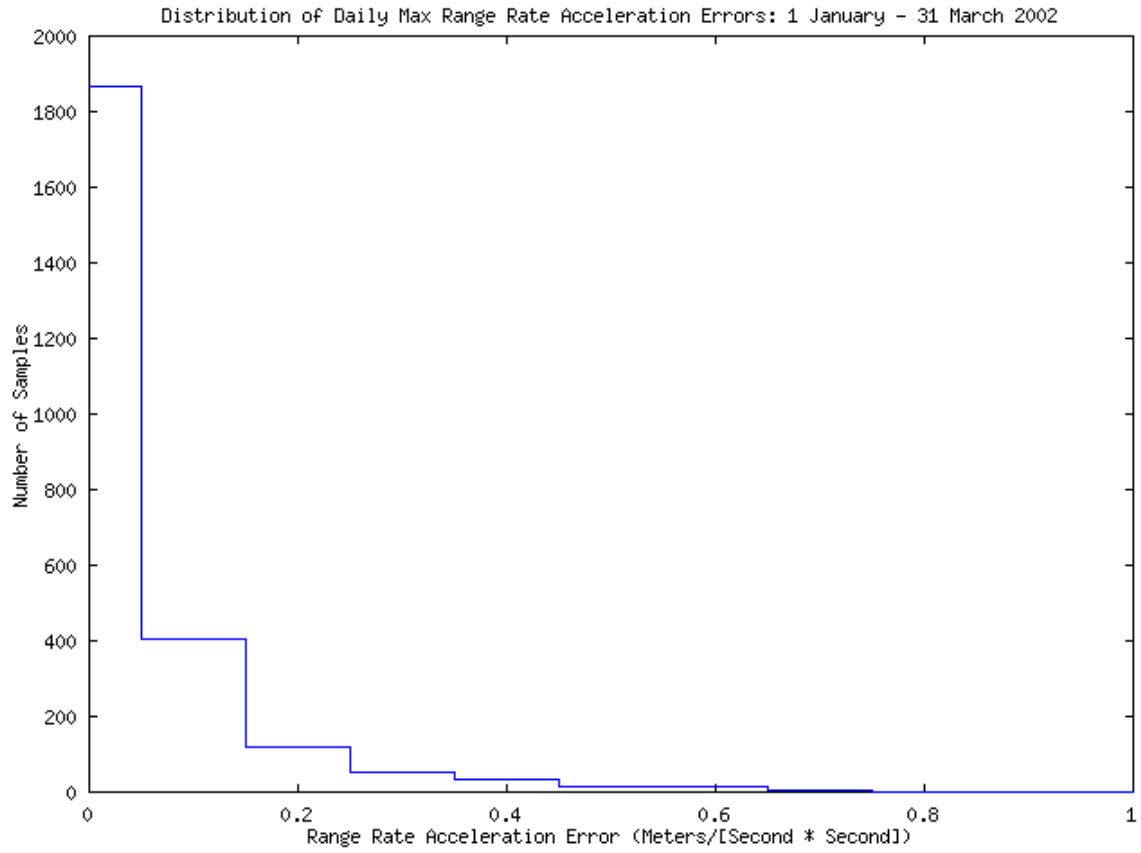
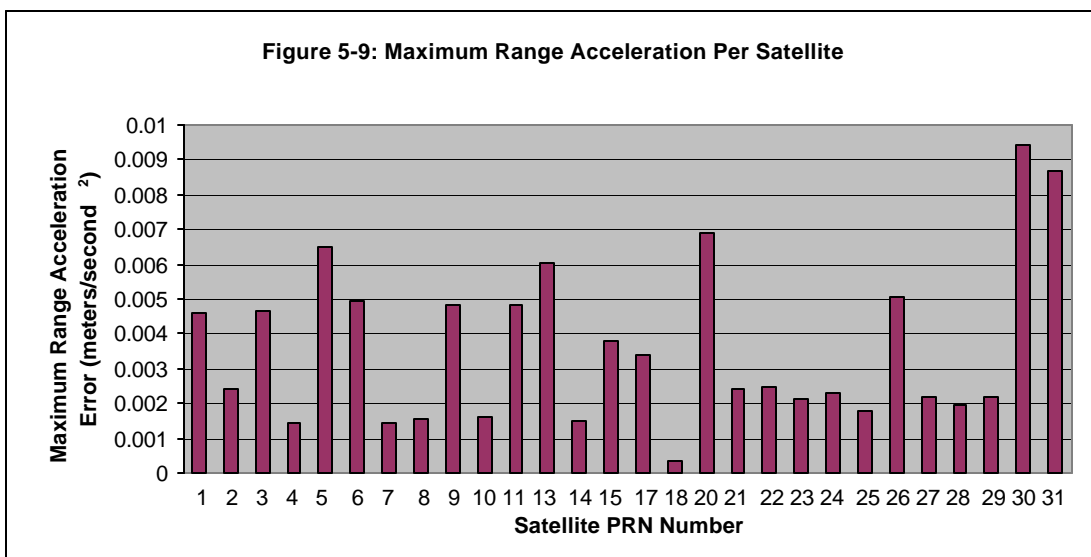
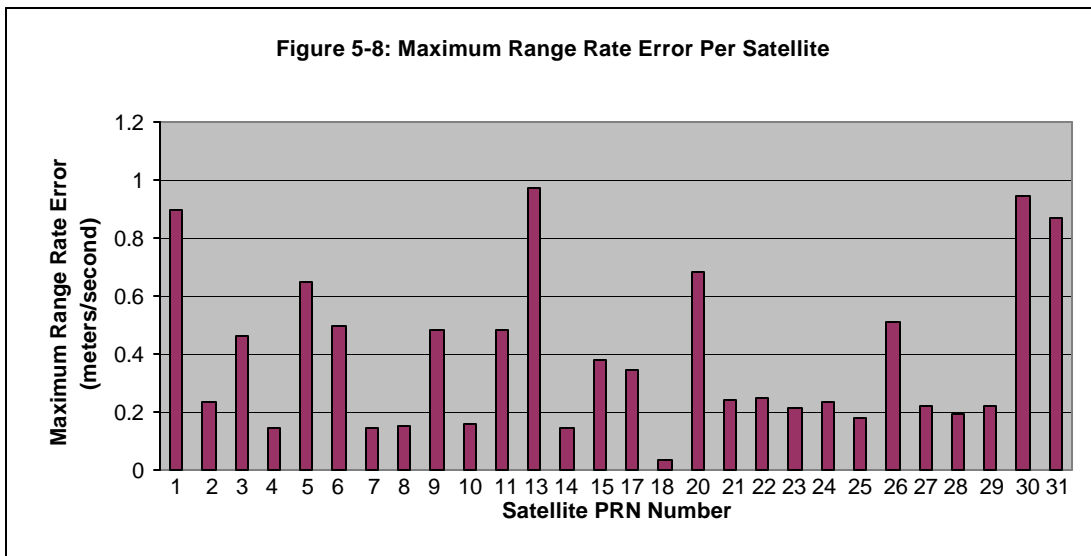
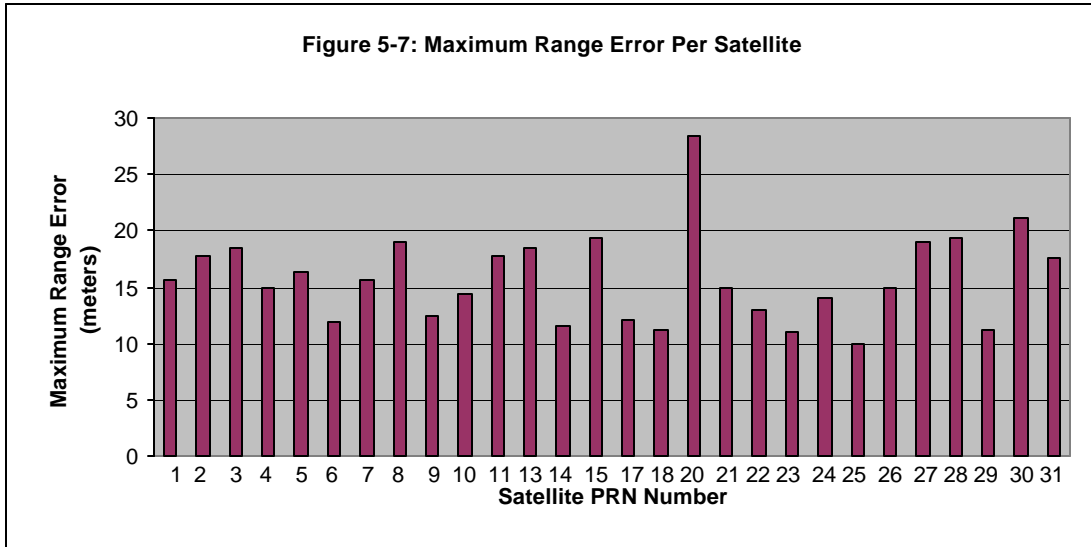


Figure 5-6: Distribution of Daily Max Acceleration Rate Errors





6.0 Solar Storms

Solar storm activity is being monitored in order to assess the possible impact on GPS SPS performance. Solar activity is reported by the Space Environment Center (SEC), a division of the National Oceanic and Atmospheric Administration (NOAA). When storm activity is indicated, ionospheric delays of the GPS signal, satellite outages, position accuracy and availability will be analyzed.

The following article was taken from the SEC web site <http://sec.noaa.gov>. It briefly explains some of the ideas behind the association of the aurora with geomagnetic activity and a bit about how the 'K-index' or 'K-factor' works.

The aurora is caused by the interaction of high-energy particles (usually electrons) with neutral atoms in the earth's upper atmosphere. These high-energy particles can 'excite' (by collisions) valence electrons that are bound to the neutral atom. The 'excited' electron can then 'de-excite' and return back to its initial, lower energy state, but in the process it releases a photon (a light particle). The combined effect of many photons being released from many atoms results in the aurora display that you see.

The details of how high energy particles are generated during geomagnetic storms constitute an entire discipline of space science in its own right. The basic idea, however, is that the Earth's magnetic field (let us say the 'geomagnetic field') is responding to an outwardly propagating disturbance from the Sun. As the geomagnetic field adjusts to this disturbance, various components of the Earth's field change form, releasing magnetic energy and thereby accelerating charged particles to high energies. These particles, being charged, are forced to stream along the geomagnetic field lines. Some end up in the upper part of the earth's neutral atmosphere and the auroral mechanism begins.

An instrument called a magnetometer may also measure the disturbance of the geomagnetic field. At NOAA's operations center magnetometer data is received from dozens of observatories in one-minute intervals. The data is received at or near to 'real-time' and allows NOAA to keep track of the current state of the geomagnetic conditions. In order to reduce the amount of data NOAA converts the magnetometer data into three-hourly indices, which give a quantitative, but less detailed measure of the level of geomagnetic activity. The K-index scale has a range from 0 to 9 and is directly related to the maximum amount of fluctuation (relative to a quiet day) in the geomagnetic field over a three-hour interval.

The K-index is therefore updated every three hours. The K-index is also necessarily tied to a specific geomagnetic observatory. For locations where there are no observatories, one can only estimate what the local K-index would be by looking at data from the nearest observatory, but this would be subject to some errors from time to time because geomagnetic activity is not always spatially homogenous.

Another item of interest is that the location of the aurora usually changes geomagnetic latitude as the intensity of the geomagnetic storm changes. The location of the aurora often takes on an 'oval-like' shape and is appropriately called the auroral oval.

Figures 6-1 through 6-3 show the K-index for three time periods with significant solar activity. Although there were other days with increased solar activity, these time periods were selected as examples. (See Appendix B for the actual geomagnetic data for this reporting period.)

Figure 6-1 K-Index for 23-25 March 2002

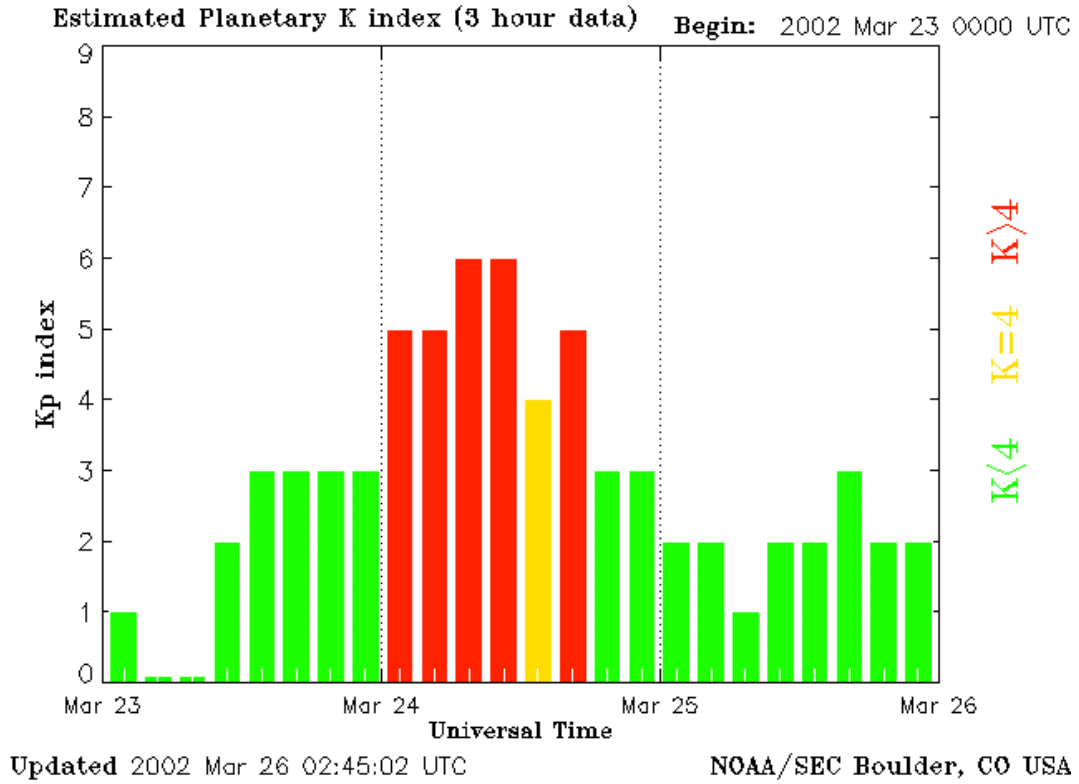


Figure 6-2 K-Index for 18-20 March 2002

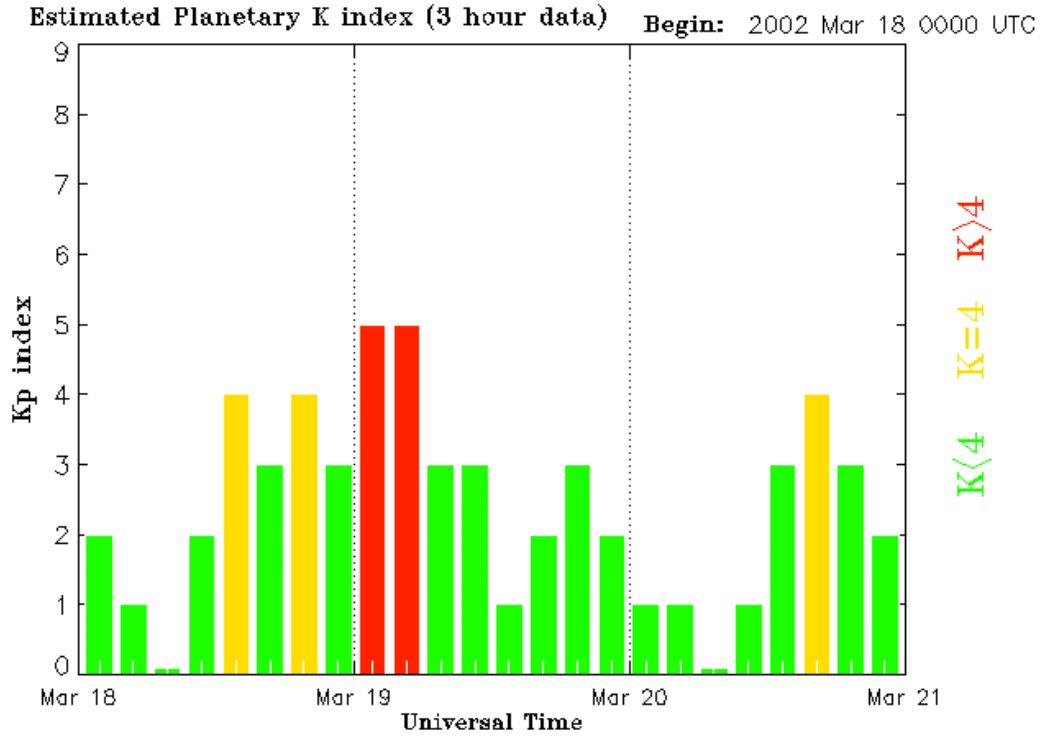
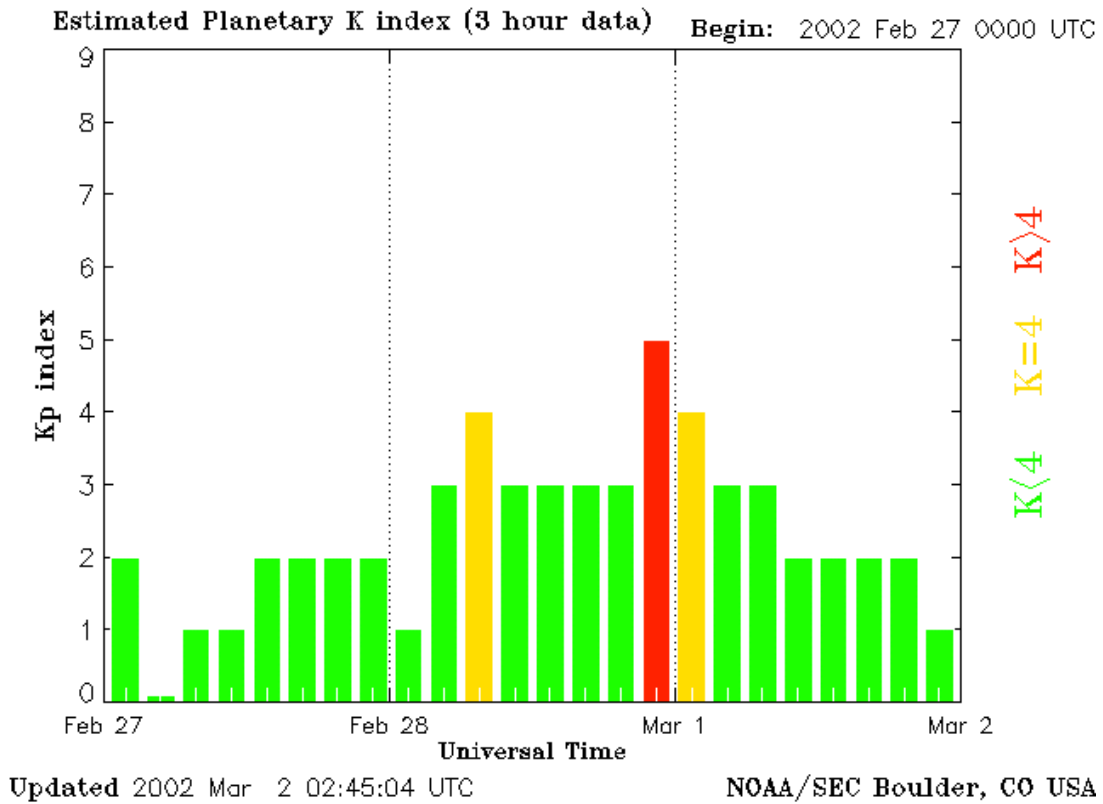


Figure 6-3 K-Index for 27 February – 1 March 2002



Tables 6-1 and 6-2 below show the PDOP and position accuracy information, respectively, for the days corresponding to Figure 6-1. The GPS SPS performance met the availability requirements during all storms that occurred during this quarter.

Table 6-1 PDOP Statistics for 24 March 2002

NSTB/WAAS Site	Min PDOP	Max PDOP	Mean PDOP	99.99% PDOP	99.99% VDOP
Atlantic City	1.212	3.808	1.798	3.803	3.335
Columbus	1.256	3.099	1.737	3.099	2.653
Denver	1.215	2.881	1.767	2.878	2.418
Grand Forks	1.246	4.069	1.769	4.057	3.570
Green Bay	1.171	3.636	1.749	3.636	3.185
Greenwood	1.309	3.385	1.747	3.384	2.836
Prescott	1.401	5.070	2.094	5.070	4.767
Billings	1.121	2.840	1.718	2.840	2.403
Anchorage	1.257	4.870	1.758	4.867	4.653
Chicago	1.266	2.971	1.697	2.969	2.466
Kansas City	1.276	2.283	1.709	2.281	1.988
Salt Lake City	1.184	3.137	1.736	3.133	2.661
Miami	1.208	3.339	1.760	3.339	3.064
Atlanta	1.313	3.374	1.769	3.374	2.932

Table 6-2 Horizontal & Vertical Accuracy Statistics for 24 March 2002

NSTB Site	95% Horizontal (Meters)	95% Vertical (Meters)	99.99% Horizontal (Meters)	99.99% Vertical (Meters)
Atlantic City	4.090	8.671	4.786	11.206
Columbus	5.502	9.091	8.322	11.230
Denver	5.680	9.375	7.284	13.677
Grand Forks	7.631	8.205	10.058	16.092
Green Bay	5.365	8.330	8.114	13.289
Greenwood	4.872	9.716	5.747	12.096
Prescott	4.957	11.284	6.885	13.098
Billings	7.175	8.221	9.473	12.540
Anchorage	10.043	8.733	11.998	15.333
Chicago	5.341	7.787	6.265	10.572
Kansas City	4.989	8.958	6.240	11.198
Salt Lake City	6.474	9.735	7.490	13.142
Miami	5.962	12.826	7.112	18.388
Atlanta	5.219	10.177	6.555	17.183

7.0 GLONASS/GPS Performance

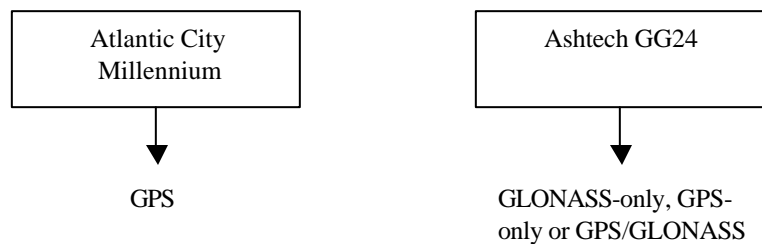
7.1 Introduction

In April 1999, ACT-360 was tasked to monitor, analyze and characterize GLONASS and GPS/GLONASS system performance. The objective of this task is to evaluate the ability of GLONASS to provide navigation by itself and with SPS GPS and to assess the incremental benefit to WAAS obtained from using GLONASS.

7.2 Approach

The GPS, GLONASS and blended data will be collected daily at one-second intervals. Since ACT-360 already collects the GPS data from the NSTB reference station sites, existing techniques and software programs will be used for the GLONASS and blended data collection and analysis. Initially, GPS/GLONASS receivers will be placed only at one site, Atlantic City. The Ashtech GG24 provides the three solutions but only one at a time. Therefore we have the Ashtech permanently outputting a blended solution.

Figure 7-1 Receivers with Corresponding Solutions



Analysis will include the comparison of the different solutions obtained from the Ashtech GG24 and the NSTB Millennium receiver. The GPS/GLONASS receiver solutions will be compared to the Millennium GPS-only and GPS/WAAS-corrected solutions.

The following table summarizes the performance data that will be reported on a quarterly basis.

Performance	GPS	GLONASS	GPS+GLONASS
Coverage	X	X	X
Service Availability	X	X	X
Position Accuracy	X	X	X
Range Accuracy	X	X	X
Time Accuracy	X	X	X
Satellite Visibility	X	X	X
Ionospheric Effects	X	X	X

7.3 Quarter Results

For this quarter, data collected from the Atlantic City Ashtech GG24 Glonass/GPS receiver and the Millennium GPS receiver will be analyzed and compared.

Tables 7-1 and 7-2 provide PDOP and Position Accuracy statistics for the two receivers from 1 January through 31 March 2002. The statistics are cumulative.

Table 7-1 PDOP Statistics for Ashtech GG24 & Atlantic City

Receiver	Solution	Maximum PDOP	Minimum PDOP	Mean PDOP	95% PDOP	Number of Samples
Ashtech GG24	GPS/GLONASS	5.808	1.000	1.699	2.302	7542135
Millenium	GPS Only Atlantic City	5.469	1.206	1.855	2.470	7775714

Table 7-2 Position Accuracy Statistics for Ashtech GG24 & Atlantic City

Receiver	Solution	95% Horizontal (m)	95% Vertical (m)	99.99% Horizontal (m)	99.99% Vertical (m)	Number of Samples
Ashtech GG24	GPS/GLONASS	6.576	11.034	27.216	46.701	7542135
Millenium	GPS Only Atlantic City	5.896	11.209	15.160	20.123	7775714

Figures 7-2 and 7-3 show the Horizontal and Vertical Error histograms for the GG24 GLONASS/GPS solution.

Figure 7-2 Horizontal Position Error Histogram for GPS/GLONASS

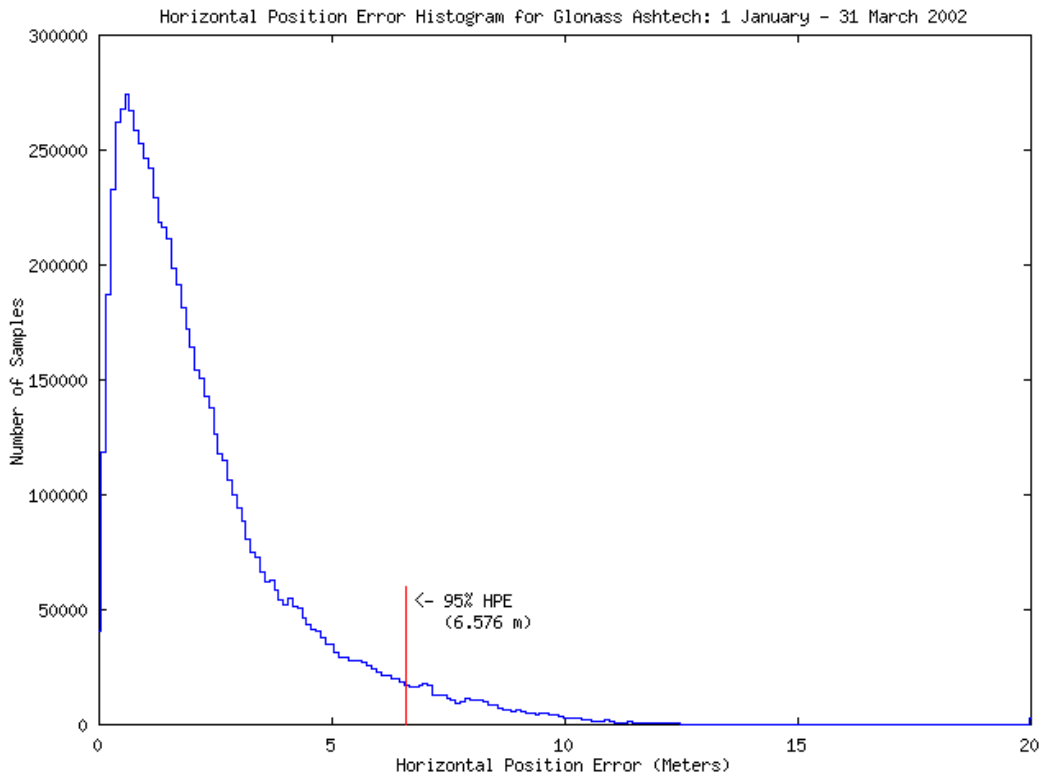


Figure 7-3 Vertical Position Error Histogram for GPS/GLONASS

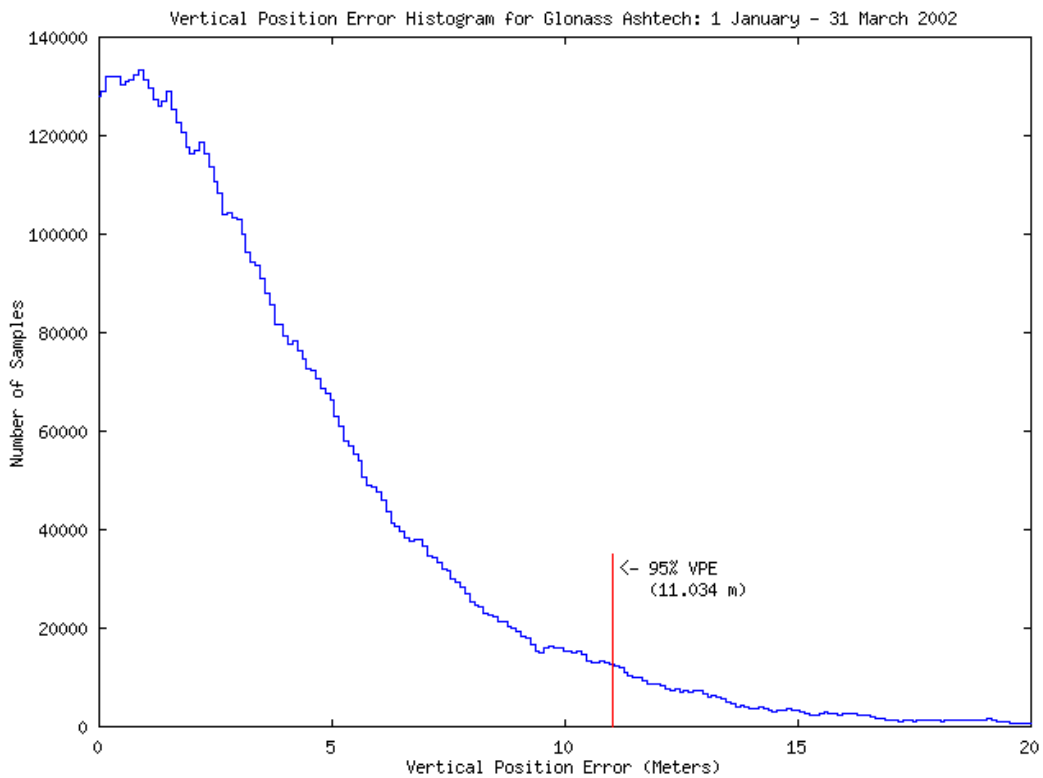
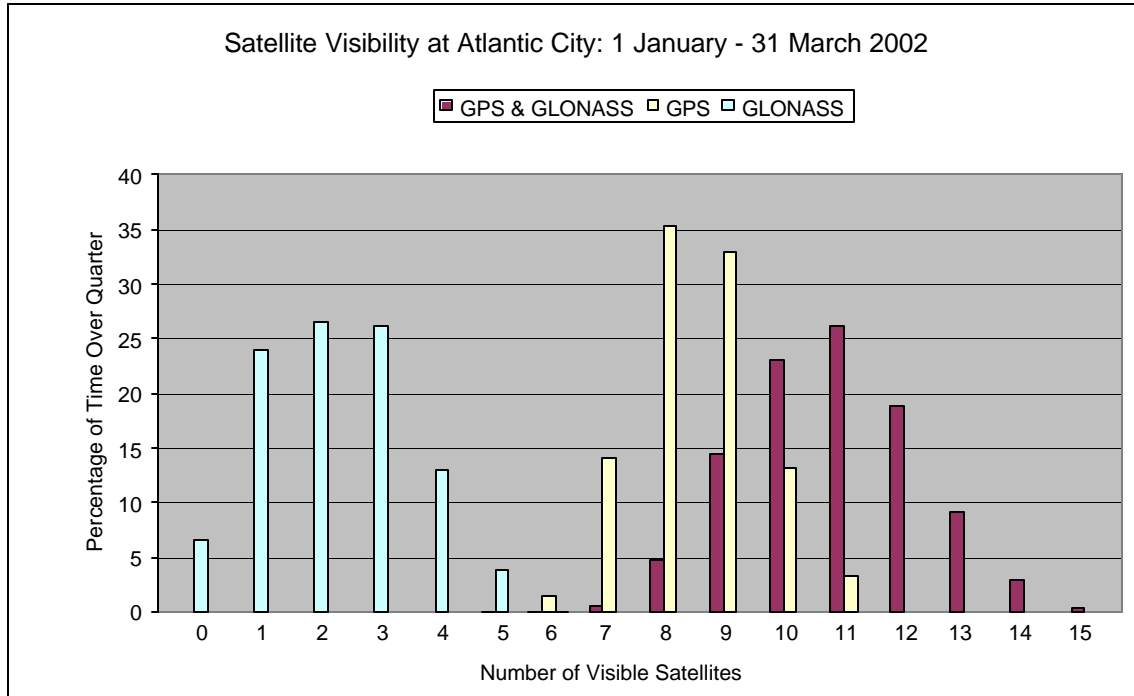


Figure 7-4 Glonass and GPS Satellite Visibility



APPENDICES A – D

Appendix A Performance Summary

<i>Conditions and Constraints</i>	<i>Coverage Standard</i>	<i>Measured Performance</i>
<ul style="list-style-type: none"> • Probability of 4 or more satellites in view over any 24 hour interval, averaged over the globe • 4 satellites must provide PDOP of 6 or less • 5° mask angle with no obscura • Standard is predicated on 24 operational satellites, as the constellation is defined in the almanac 	≥ 99.9% global average	100%
<ul style="list-style-type: none"> • Probability of 4 or more satellites in view over any 24 hour interval, for the worst-case point on the globe • 4 satellites must provide PDOP of 6 or less • 5° mask angle with no obscura • Standard is predicated on 24 operational satellites, as the constellation is defined in the almanac 	≥ 96.9% at worst-case point	99.653% Availability 99.9% PDOP was 3.12436
<i>Conditions and Constraints</i>	<i>Satellite Availability Standard</i>	<i>Measured Performance</i>
<ul style="list-style-type: none"> • Conditioned on coverage standard • Standard based on a typical 24 hour interval, averaged over the globe • Typical 24 hour interval defined using averaging period of 30 days 	≥ 99.85% global average	99.999%
<ul style="list-style-type: none"> • Conditioned on coverage standard • Standard based on a typical 24 hour interval, for the worst-case point on the globe • Typical 24 hour interval defined using averaging period of 30 days 	≥ 99.16% single point average	99.986%
<ul style="list-style-type: none"> • Conditioned on coverage standard • Standard represents a worst-case 24 hour interval, averaged over the globe 	≥ 95.87% global average on worst-case day	99.967%
<ul style="list-style-type: none"> • Conditioned on coverage standard • Standard based on a worst-case 24 hour interval, for the worst-case point on the globe 	≥ 83.92% at worst-case point on worst-case day	99.575%
<i>Conditions and Constraints</i>	<i>Service Reliability Standard</i>	<i>Measured Performance</i>
<ul style="list-style-type: none"> • Conditioned on coverage and service availability standards • 500 meter NTE predictable horizontal error reliability threshold • Standard based on a measurement interval of one year; average of daily values over the globe • Standard predicated on a maximum of 18 hours of major service failure behavior over the sample interval 	≥ 99.97% global average	100%

<ul style="list-style-type: none"> • Conditioned on coverage and service availability standards • 500 meter Not-to-Exceed (NTE) predictable horizontal error reliability threshold • Standard based on a measurement interval of one year; average of daily values from the worst-case point on the globe • Standard based on a maximum of 18 hours of major service failure behavior over the sample interval 	≥ 99.79% single point average	100%
<i>Conditions and Constraints</i>	<i>Accuracy Standard</i>	<i>Measured Performance</i>
<ul style="list-style-type: none"> • Conditioned on coverage, service availability and service reliability standards • Standard based on a measurement interval of 24 hours, for any point on the globe 	<u>Predictable Accuracy</u> ≤ 100 m horz. error 95% of time ≤ 156 m vert. error 95% of time ≤ 300 m horz. error 99.99% of time ≤ 500 m vert. error 99.99% of time	≤7.226m HE 95% ≤15.160m HE 99.99% ≤13.574m VE 95% ≤22.609m VE 99.99%
<ul style="list-style-type: none"> • Conditioned on coverage, service availability and service reliability standards • Standard based on a measurement interval of 24 hours, for any point on the globe 	<u>Repeatable Accuracy</u> ≤ 141 m horz. error 95% of time ≤ 221 m vert. error 95% of time	≤2.449m HE 95% ≤5.300m VE 95%
<ul style="list-style-type: none"> • Conditioned on coverage, service availability and service reliability standards • Standard based on a measurement interval of 24 hours, for any point on the globe • Standard presumes that the receivers base their position solutions on the same satellites, with position solutions computed at approximately the same time 	<u>Relative Accuracy</u> ≤ 1.0 m horz. error 95% of time ≤ 1.5 m vert. error 95% of time	Future Reports
<ul style="list-style-type: none"> • Conditioned on coverage, service availability and service reliability standards • Standard based upon SPS receiver time as computed using the output of the position solution • Standard based on a measurement interval of 24 hours, for any point on the globe • Standard is defined with respect to Universal Coordinated Time, as it is maintained by the United States Naval Observatory 	<u>Time Transfer Accuracy</u> ≤ 340 nanoseconds time transfer error 95% of time	≤16 ns 95% of the time
<ul style="list-style-type: none"> • Conditioned on satellite indicating healthy status • Standard based on a measurement interval of 24 hours, for any point on the globe • Standard restricted to range domain errors allocated to space/control segments • Standards are not constellation values -- each 	<u>Range Domain Accuracy</u> ≤ 150 m NTE range error ≤ 2 m/s NTE range rate error ≤ 19 mm/s ² NTE range	28.452m NTE Range Error 0.971435m/s NTE Rate Error 9.452mm/s ² NTE Accl. Error

<p>satellite is required to meet the standards</p> <ul style="list-style-type: none"> Assessment requires minimum of four hours of data over the 24 hour period for a satellite in order to evaluate that satellite against the standard 	<p>acceleration error $\leq 8 \text{ mm/s}^2$ range acceleration error 95% of time</p>	<p>$\leq 8 \text{ mm/s}^2$ 99.9999% of the time</p>
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Appendix B Geomagnetic Data

Product: Daily Geomagnetic Data quar_DGD.txt
 Issued: 1820 UT 03 Apr 2002

 # Prepared by the U.S. Dept. of Commerce, NOAA, Space Environment Center.
 # Please send comment and suggestions to sec@sec.noaa.gov
 #
 # Current Quarter Daily Geomagnetic Data
 #

Date	Middle Latitude - Fredericksburg -								High Latitude ---- College ----								Estimated --- Planetary ---										
	A	K-indices							A	K-indices							A	K-indices									
2002 01 01	4	2	1	1	1	2	0	2	1	9	2	2	2	4	3	1	1	1	7	2	2	2	2	2	3	2	
2002 01 02	7	1	1	1	1	3	3	2	2	-1	-1	-1	-1	-1	-1	-1	-1	-1	7	1	1	1	1	3	3	2	2
2002 01 03	6	1	1	1	1	2	3	2	2	-1	-1	-1	-1	-1	-1	-1	-1	-1	3	1	1	1	1	1	1	2	1
2002 01 04	4	0	1	1	1	1	2	3	1	-1	-1	-1	-1	-1	-1	-1	-1	-1	3	1	1	0	1	1	1	2	1
2002 01 05	2	2	1	0	0	0	1	1	0	-1	-1	-1	-1	-1	-1	-1	-1	-1	3	2	0	0	1	1	1	2	1
2002 01 06	3	1	0	0	0	2	2	2	1	-1	-1	-1	-1	-1	-1	-1	-1	-1	4	1	0	0	1	1	2	2	2
2002 01 07	8	0	1	1	3	1	3	3	2	-1	-1	-1	-1	-1	-1	-1	-1	-1	7	1	1	0	3	2	3	3	2
2002 01 08	6	2	1	3	1	3	1	0	0	-1	-1	-1	-1	-1	-1	-1	-1	-1	7	2	1	3	2	2	2	2	1
2002 01 09	2	0	0	0	0	1	2	2	0	-1	0	0	0	0	1	1	0	-1	3	1	0	0	0	1	2	2	0
2002 01 10	16	1	2	1	2	4	5	3	3	45	0	1	3	4	6	7	6	3	17	2	1	2	3	4	5	4	3
2002 01 11	12	3	3	3	3	2	3	2	1	37	4	4	5	6	4	5	4	3	21	4	4	4	4	4	4	3	2
2002 01 12	12	3	3	3	3	2	3	2	1	30	2	3	5	6	4	5	2	2	15	3	3	4	4	3	3	3	2
2002 01 13	10	1	2	3	2	3	2	2	3	25	2	2	5	5	5	4	2	2	11	2	2	4	3	3	2	2	3
2002 01 14	7	2	2	1	1	3	2	1	2	11	1	1	2	3	4	2	2	3	8	3	2	1	2	3	3	2	2
2002 01 15	4	1	1	1	2	2	1	1	1	11	1	1	1	4	4	3	1	1	6	1	2	2	2	2	2	2	2
2002 01 16	3	1	0	2	0	1	1	1	0	4	3	1	2	2	0	0	0	0	4	1	1	3	1	1	2	1	1
2002 01 17	6	3	1	0	2	2	2	1	1	8	0	0	1	4	2	4	1	0	6	3	1	1	2	2	3	1	2
2002 01 18	5	2	1	1	1	3	2	1	0	10	1	1	0	3	1	2	5	0	5	2	1	0	3	2	2	1	1
2002 01 19	11	0	2	2	3	3	4	2	2	27	0	1	1	2	6	6	3	4	11	1	1	2	2	4	4	3	3
2002 01 20	6	2	3	1	1	1	1	2	1	8	1	1	0	4	1	3	2	1	7	3	2	1	2	2	3	2	2
2002 01 21	7	2	2	2	2	1	2	2	2	14	2	2	2	5	4	2	2	1	9	3	2	2	3	3	2	3	2
2002 01 22	5	1	1	2	1	2	2	1	1	7	0	0	2	3	4	1	1	0	6	1	1	3	2	2	2	2	2
2002 01 23	4	2	0	1	2	1	1	2	1	10	1	0	1	4	3	2	3	2	6	2	0	1	2	2	3	3	2
2002 01 24	3	1	1	0	0	2	1	1	1	1	1	1	0	0	0	1	0	0	4	1	2	0	0	1	2	2	2
2002 01 25	6	0	1	1	1	3	2	3	1	12	0	0	1	0	3	4	5	1	8	1	0	1	1	3	3	4	2
2002 01 26	6	1	1	2	2	2	2	2	1	6	1	1	2	3	1	2	2	0	7	2	2	2	2	2	2	2	2
2002 01 27	4	1	1	1	1	1	2	1	1	5	1	2	1	2	3	1	1	0	7	3	3	1	2	2	2	2	2
2002 01 28	6	2	0	1	2	2	2	2	2	9	1	0	1	4	4	1	1	1	6	2	1	1	2	3	2	2	2
2002 01 29	2	1	1	1	1	0	0	0	0	2	0	1	0	2	1	0	0	0	4	2	1	1	2	1	1	1	1
2002 01 30	1	0	0	0	0	0	1	1	1	-1	0	0	1	0	2	2	0	-1	3	1	0	1	0	1	1	1	2
2002 01 31	6	2	1	0	0	2	2	2	3	-1	1	1	0	0	-1	-1	-1	-1	5	2	1	0	0	1	2	1	3
2002 02 01	11	3	2	2	2	3	3	3	1	-1	-1	-1	-1	-1	-1	-1	1	4	11	3	2	2	2	3	4	2	2
2002 02 02	14	2	3	4	3	3	3	2	2	-1	1	5	-1	-1	-1	-1	-1	-1	18	3	3	5	4	4	3	3	2
2002 02 03	6	1	2	0	1	3	3	2	0	-1	-1	-1	-1	-1	-1	-1	-1	-1	5	1	2	0	2	2	2	2	0

2002 02 04	8	1	1	0	0	3	3	3	3	-1	-1	-1	-1	-1	-1	1	2	6	1	1	0	0	2	3	3	3	
2002 02 05	15	3	2	2	2	3	4	4	3	-1	2	3	2	3	4	-1	4	3	16	4	3	2	2	3	4	4	3
2002 02 06	13	2	3	3	2	3	3	3	3	-1	3	4	-1	5	-1	4	4	4	16	3	3	4	4	4	3	3	2
2002 02 07	10	2	2	2	2	3	3	1	3	25	3	3	4	6	5	2	1	2	14	2	3	4	3	4	2	3	3
2002 02 08	9	3	1	2	2	3	3	2	1	-1	3	-1	-1	-1	-1	3	-1	10	3	1	1	2	3	3	3	2	
2002 02 09	6	2	2	2	2	1	1	2	2	9	1	2	2	3	3	3	1	2	9	2	3	2	3	2	2	2	3
2002 02 10	7	2	2	3	2	2	1	1	2	12	2	2	3	4	4	1	1	1	8	3	2	3	2	2	2	2	2
2002 02 11	8	2	1	2	3	3	2	1	2	26	1	0	2	6	5	5	3	2	13	2	2	3	4	3	3	3	3
2002 02 12	5	3	1	2	1	0	1	1	2	6	3	2	2	2	0	1	1	2	6	3	2	2	1	1	2	2	2
2002 02 13	11	2	1	3	3	3	3	2	2	15	1	1	3	4	5	3	2	1	10	2	1	3	3	3	3	2	1
2002 02 14	2	0	0	2	0	1	1	1	0	1	0	0	1	1	0	1	0	1	4	1	0	2	1	1	2	2	1
2002 02 15	3	0	0	0	0	1	3	1	1	4	0	0	0	0	0	4	1	0	4	1	0	0	1	1	2	2	1
2002 02 16	2	1	0	0	0	1	2	1	1	5	0	0	0	0	4	2	1	1	5	2	0	0	1	1	2	2	2
2002 02 17	6	2	2	3	1	1	1	1	1	9	1	1	4	1	3	3	0	1	8	1	3	4	2	2	2	2	2
2002 02 18	6	1	2	1	0	2	2	2	3	7	0	0	3	2	0	1	3	3	9	1	2	2	2	1	3	3	4
2002 02 19	4	2	1	1	1	0	1	2	1	2	1	0	1	0	0	1	2	1	4	2	1	1	1	1	2	2	1
2002 02 20	6	1	0	2	1	2	2	2	3	11	0	0	4	3	4	2	2	2	8	2	0	3	2	3	2	2	3
2002 02 21	6	1	2	1	1	2	1	2	3	9	0	1	2	2	3	2	4	2	6	1	1	2	1	2	2	3	3
2002 02 22	6	1	2	1	0	2	1	3	2	4	1	1	0	0	3	0	2	1	5	2	2	0	1	1	2	3	2
2002 02 23	3	2	2	0	0	1	1	0	1	2	1	2	0	0	1	0	0	0	4	2	1	1	0	1	2	2	1
2002 02 24	4	1	1	1	1	2	1	1	1	5	0	0	0	2	4	0	1	0	4	2	1	1	1	2	1	1	1
2002 02 25	4	1	0	2	1	2	2	1	1	12	0	0	1	5	4	3	1	1	7	1	0	2	2	3	3	2	2
2002 02 26	8	2	2	2	2	2	2	3	2	11	1	1	4	4	3	1	1	1	8	2	2	3	2	2	2	2	3
2002 02 27	4	2	0	1	1	2	1	1	1	5	3	1	0	2	2	1	1	1	4	2	0	1	1	2	2	2	2
2002 02 28	15	1	4	3	2	3	2	3	4	19	1	2	3	4	5	2	3	4	17	1	3	4	3	3	3	3	5
2002 03 01	9	5	2	2	1	2	1	0	0	9	4	1	3	2	2	1	0	2	11	4	3	3	2	2	2	2	1
2002 03 02	4	1	1	1	1	1	2	1	2	3	1	1	0	0	1	2	1	1	5	1	1	1	1	1	3	3	2
2002 03 03	10	2	1	2	1	3	2	2	4	15	1	0	4	3	4	4	2	3	10	1	1	3	2	3	3	3	4
2002 03 04	9	3	0	1	1	2	4	2	2	9	3	0	1	2	2	3	3	2	9	2	2	1	2	2	3	3	2
2002 03 05	10	2	3	2	2	3	3	2	2	26	2	4	4	3	4	5	5	2	15	3	4	2	3	3	4	3	3
2002 03 06	16	2	3	1	2	3	4	5	2	39	3	2	3	5	5	6	6	3	15	2	3	3	3	3	4	4	3
2002 03 07	5	2	1	2	2	2	0	0	2	-1	1	1	3	-1	1	2	2	5	10	2	1	3	3	4	2	2	2
2002 03 08	4	1	1	2	2	1	1	1	1	10	1	1	3	5	1	0	0	0	4	1	1	2	3	0	1	1	1
2002 03 09	3	1	0	0	1	1	2	2	1	12	0	0	0	4	3	4	1	1	5	1	0	0	2	3	3	2	1
2002 03 10	6	0	1	0	3	3	1	1	2	9	0	1	0	4	4	2	1	2	9	1	1	0	4	4	2	2	3
2002 03 11	8	2	4	1	2	2	1	2	1	10	3	2	1	3	4	1	2	1	9	3	3	2	2	3	2	3	2
2002 03 12	8	2	2	3	2	2	2	2	2	17	2	0	3	5	5	3	2	1	10	2	1	2	3	3	3	3	3
2002 03 13	6	2	0	2	1	3	2	1	1	-1	1	1	2	3	2	-1	0	0	5	2	1	2	2	2	2	2	1
2002 03 14	2	1	0	0	0	2	1	0	0	1	1	0	0	0	1	0	0	0	4	1	0	0	1	1	2	2	1
2002 03 15	3	0	1	1	0	2	1	1	2	2	0	0	0	0	0	1	2	1	6	1	0	1	1	2	3	3	2
2002 03 16	5	1	1	1	0	3	3	1	0	8	1	0	0	0	4	4	1	0	5	1	0	0	1	2	3	2	1
2002 03 17	2	0	0	0	0	0	1	2	1	0	0	0	0	0	0	0	0	0	4	1	0	0	1	2	2	2	1
2002 03 18	9	1	1	0	1	4	2	3	3	12	1	1	0	3	3	4	2	4	12	2	1	0	2	4	3	4	3
2002 03 19	15	5	4	3	2	1	1	2	2	14	3	5	4	2	1	1	1	2	17	5	5	3	3	1	2	3	2
2002 03 20	7	1	1	0	0	3	4	2	1	15	0	0	0	0	3	5	2	5	7	1	1	0	1	3	4	3	2
2002 03 21	5	1	1	2	1	2	1	2	2	8	2	1	4	3	0	1	2	1	8	2	1	3	3	1	2	3	2
2002 03 22	7	2	3	2	3	2	0	1	0	9	1	2	2	5	0	1	1	1	8	2	3	3	3	1	2	2	1
2002 03 23	8	1	0	0	2	3	2	2	3	16	0	0	0	2	4	5	4	3	9	1	0	0	2	3	3	3	3
2002 03 24	29	4	4	5	5	4	4	3	3	68	4	5	7	6	6	7	3	3	47	5	5	6	6	4	5	3	3
2002 03 25	7	2	2	1	1	2	3	2	1	-1	2	1	1	1	2	-1	2	1	5	2	2	1	2	2	3	2	2
2002 03 26	10	2	3	2	3	3	1	1	3	15	3	3	1	4	3	2	2	4	11	3	2	1	3	3	3	2	3
2002 03 27	3	1	2	2	0	0	1	1	1	3	2	1	1	0	0	1	1	1	5	2	1	2	2	2	2	2	1
2002 03 28	2	0	0	0	1	0	2	1	0	5	0	0	0	2	3	3	0	0	6	1	0	1	2	3	3	3	1
2002 03 29	6	0	0	0	1	2	2	1	4	-1	1	0	0	-1	-1	-1	-1	7	7	1	0	0	2	2	3	2	3
2002 03 30	18	2	4	3	3	4	4	3	2	-1	-1	-1	-1	-1	-1	-1	-1	17	2	4	4	3	4	3	3	2	
2002 03 31	12	2	3	3	2	3	3	3	2	-1	-1	-1	-1	-1	-1	-1	-1	14	2	3	3	3	3	3	4	3	

Appendix C Performance Analysis (PAN) Problem Report

Background:

In 1993, the FAA began monitoring and analyzing Global Positioning System (GPS) Standard Positioning Service (SPS) performance data. At present, the FAA has approved GPS for IFR and is developing WAAS and LAAS, both of which are GPS augmentation systems. In order to ensure the safe and effective use of GPS and its augmentation systems within the NAS, it is critical that characteristics of GPS performance as well as specific causes for service outages be monitored and understood. To accomplish this objective, GPS SPS performance data is documented in a quarterly GPS Performance Analysis (PAN) report. The PAN report contains data collected at various National Satellite Test Bed (NSTB) and Wide Area Augmentation System (WAAS) reference station locations. This PAN Problem Report will be issued only when the performance data fails to meet the GPS Standard Positioning Service (SPS) Signal Specification.

Problem Description:

GPS did not fail SPS specification in any instances during this quarter.

The terms and definitions discussed below are taken from the Standard Positioning Service Performance Specification (SPS) (June 2, 1995). An understanding of these terms and definitions is a necessary prerequisite to full understanding of the Signal Specification.

General Terms and Definitions

Block I and Block II Satellites. The Block I is a GPS concept validation satellite; it does not have all of the design features and capabilities of the production model GPS satellite, the Block II. The FOC 24 satellite constellation is defined to consist entirely of Block II/IIA satellites. For the purposes of this Signal Specification, the Block II satellite and a slightly modified version of the Block II known as the Block IIA provide an identical service.

Dilution of Precision (DOP). The magnifying effect on GPS position error induced by mapping GPS ranging errors into position through the position solution. The DOP may be represented in any user local coordinate desired. Examples are HDOP for local horizontal, VDOP for local vertical, PDOP for all three coordinates, and TDOP for time.

Geometric Range. The difference between the estimated locations of a GPS satellite and an SPS receiver.

Major Service Failure. A condition over a time interval during which one or more SPS performance standards are not met and the civil community was not warned in advance.

Minimum SPS Receiver Capabilities. Minimum standards for signal reception and processing capabilities that are incorporated into the design of an SPS receiver. This ensures consistent performance with the SPS performance standards.

Navigation Data. Data provided to the SPS receiver via each satellite's ranging signal, containing the ranging signal time of transmission, the transmitting satellite's orbital elements, an almanac containing abbreviated orbital element information to support satellite selection, ranging measurement correction information, and status flags.

Navigation Message. Message structure designed to carry navigation data.

Operational Satellite. A GPS satellite that is capable of, but may or may not be, transmitting a usable ranging signal. For the purposes of the SPS, any satellite contained within the transmitted navigation message almanac is considered to be an operational satellite.

Position Solution. The use of ranging signal measurements and navigation data from at least four satellites to solve for three position coordinates and a time offset.

Selective Availability. Protection technique employed by the DOD to deny full system accuracy to unauthorized users.

Service Disruption. A condition over a time interval during which one or more SPS performance standards are not supported, but the civil community was warned in advance.

SPS Performance Envelope. The range of variation in specified aspects of SPS performance.

SPS Performance Standard. A quantifiable minimum level for a specified aspect of GPS SPS performance.

Standard Positioning Service (SPS). Three-dimensional position and time determination capability provided to a user equipped with a minimum capability GPS SPS receiver in accordance with GPS national policy and the performance specifications.

SPS Ranging Signal Measurement. The difference between the ranging signal time of reception (as defined by the receiver's clock) and the time of transmission contained within the satellite's navigation data (as defined by the satellite's clock) multiplied by the speed of light. Also known as the *pseudo range*.

SPS Signal, or SPS Ranging Signal. An electromagnetic signal originating from an operational satellite. The SPS ranging signal consists of a Pseudo Random Noise (PRN) Coarse/Acquisition (C/A) code, a timing reference and sufficient data to support the position solution generation process.

Usable SPS Ranging Signal. An SPS ranging signal that can be received, processed and used in a position solution by a receiver with minimum SPS receiver capabilities.

Performance Parameter Definitions

The definitions provided below establish the basis for correct interpretation of the GPS SPS performance standards. The GPS performance parameters contained in the SPS are defined differently than other radio navigation systems in the Federal Radio Navigation Plan. For a more comprehensive treatment of these definitions and their implications on system use, refer to Annex B of the SPS.

Coverage. The percentage of time over a specified time interval that a sufficient number of satellites are above a specified mask angle and provide an acceptable position solution geometry at any point on or near the Earth. The term "near the Earth" means on or within approximately 200 kilometers of the Earth's surface.

Positioning Accuracy. Given reliable service, the percentage of time over a specified time interval that the difference between the measured and expected user position or time is within a specified tolerance at any point on or near the Earth. This general accuracy definition is further refined through the more specific definitions of four different aspects of positioning accuracy:

- **Predictable Accuracy.** Given reliable service, the percentage of time over a specified time interval that the difference between a position measurement and a surveyed benchmark is within a specified tolerance at any point on or near the Earth.
- **Repeatable Accuracy.** Given reliable service, the percentage of time over a specified time interval that the difference between a position measurement taken at one time and a position measurement taken at another time at the same location is within a specified tolerance at any point on or near the Earth.
- **Relative Accuracy.** Given reliable service, the percentage of time over a specified time interval that the difference between two receivers' position estimates taken at the same time is within a specified tolerance at any point on or near the Earth.
- **Time Transfer Accuracy.** Given reliable service, the percentage of time over a specified time interval that the difference between a Universal Coordinated Time (commonly referred to as UTC) time estimate from the position solution and UTC as it is managed by the United States Naval Observatory (USNO) is within a specified tolerance.

Range Domain Accuracy. Range domain accuracy deals with the performance of each satellite's SPS ranging signal. Range domain accuracy is defined in terms of three different aspects:

- **Range Error.** Given reliable service, the percentage of time over a specified time interval that the difference between an SPS ranging signal measurement and the "true" range between the satellite and an SPS user is within a specified tolerance at any point on or near the Earth.

- **Range Rate Error.** Given reliable service, the percentage of time over a specified time interval that the instantaneous rate-of-change of range error is within a specified tolerance at any point on or near the Earth.
- **Range Acceleration Error.** Given reliable service, the percentage of time over a specified time interval that the instantaneous rate-of-change of range rate error is within a specified tolerance at any point on or near the Earth.

Service Availability. Given coverage, the percentage of time over a specified time interval that a sufficient number of satellites are transmitting a usable ranging signal within view of any point on or near the Earth.

Service Reliability. Given service availability, the percentage of time over a specified time interval that the instantaneous predictable horizontal error is maintained within a specified reliability threshold at any point on or near the Earth. Note that service reliability does not take into consideration the reliability characteristics of the SPS receiver or possible signal interference. Service reliability may be used to measure the total number of major failure hours experienced by the satellite constellation over a specified time interval.