# RADAR OBSERVATIONS OF ASTEROID 1999 JM8 

Lance A. M. Benner ${ }^{1 *}$, Steven J. Ostro ${ }^{1}$, Michael C. Nolan ${ }^{2}$, Jean-Luc MARGOT ${ }^{3}$, Jon D. GIorgini ${ }^{1}$, R. Scott Hudson ${ }^{4}$, RAYMOND F. Jurgens ${ }^{1}$, MARTIN A. SLADE ${ }^{1}$, ELLEN S. HOWELL², DONALD B. CAMPBELL ${ }^{5}$, DONALD K. YEOMANS ${ }^{1}$<br>${ }^{1}$ Jet Propulsion Laboratory, California Institute of Technology, Pasadena, CA 91109-8099<br>${ }^{2}$ Arecibo Observatory, National Astronomy and Ionosphere Center, Arecibo, PR 00612<br>${ }^{3}$ Division of Geological and Planetary Sciences, California Institute of Technology, Pasadena, CA 91125<br>${ }^{4}$ School of Electrical Engineering and Computer Science, Washington StateUniversity, Pullman, WA 99164-2752<br>${ }^{5}$ National Astronomy and Ionosphere Center, Space Sciences Building, Cornell University, Ithaca, NY 14853<br>Submitted to Meteoritics and Planetary Science<br>November 2001<br>Revised: January 2002<br>29 manuscript pages (including cover sheets, tables, and figure captions)<br>7 tables<br>6 figures

*Correspondence author:
Dr. Lance A. M. Benner email: lance@reason.jpl.nasa.gov
Mail Stop 300-233
Jet Propulsion Laboratory
phone: 818-354-7412
fax: 818-354-9476


#### Abstract

We report results of delay-Doppler observations of 1999 JM8 with the Goldstone 8560 MHz ( 3.5 cm ) and Arecibo $2380 \mathrm{MHz}(13 \mathrm{~cm})$ radars over 18 days in July-August 1999. The images place thousands of pixels on the asteroid and achieve range resolutions as fine as $15 \mathrm{~m} /$ pixel. The images reveal an asymmetric, irregularly shaped object with a typical overall dimension within $20 \%$ of 7 km . If we assume that $1999 \mathrm{JM8}$ 's effective diameter is 7 km , then the absolute magnitude, 15.15 , and the average Goldstone radar cross section, $2.49 \mathbf{k m}^{\wedge} 2$, correspond to optical and radar albedos of $\mathbf{0 . 0 2}$ and 0.06 , establishing that 1999 JM8 is a dark object at optical and radar wavelengths. The asteroid is in a non-principal axis spin state that, although not yet well determined, has a dominant periodicity of about 7 days. However, images obtained between July 31 and August 9 show apparent regular rotation of features from day to day, suggesting that the rotation state isn't far from principal axis rotation. 1999 JM8 has regions of pronounced topographic relief, prominent facets several kilometers in extent, numerous crater-like features between $\sim \mathbf{1 0 0} \mathbf{m}$ and 1.5 $\mathbf{k m}$ in diameter, and features whose structural nature is peculiar. Arecibo images provide the strongest evidence to date for a circular polarization ratio feature on any asteroid. Combined optical and radar observations from April 1990-December 2000 permit computation of planetary close approach times to within $\pm \mathbf{1 0}$ days over the interval from 293 to at least 2907, one of the longest spans for any Potentially Hazardous Asteroid. Integration of the orbit into the past and future shows close approaches to Earth, Mars, Ceres, and Vesta, but the probability of the object impacting Earth is zero for at least the next nine centuries.


## INTRODUCTION

1999 JM8 was discovered by LINEAR on May 13, 1999, fortuitously more than two months prior to an encounter within 0.057 AU (22 lunar distances) of Earth in July 1999, when it reached visual magnitude 14. It originally had been discovered at Palomar in April, 1990 by E. F. Helin and designated 1990 HD1, but was lost. M. Hicks, B. Buratti, and M. Hanner (pers. comm.) obtained photometric colors, vis-IR spectroscopy, and thermal infrared radiometry suggesting that 1999 JM8 is a C- or X-type (i.e., E-, M-, or P-type) object; subsequent vis-IR observations obtained by one of us (E. S. H.) at McDonald Observatory yielded a spectrum more consistent with an EMP-type object. Photometry obtained by L. Sarounova, P. Pravec, Y.

Krugly, V. Shevchenko, S. Mottola, F. Lahulla, and M. Hicks (P. Pravec, pers. comm.) between July 3.0 and 21.9 indicated that 1999 JM8 is a very slow rotator. Pravec et al. estimated a synodic rotation period of $5.7 \pm 0.2$ days. The slow rotation period indicated that the echoes would be very strong and that observations on many days would be necessary to obtain thorough coverage in rotation phase. Consequently, prompt communication of the slow rotation period was invaluable for planning the radar observations. Table 1 summarizes the asteroid's opticallydetermined physical properties.

## OVERVIEW OF THE RADAR EXPERIMENT

The asteroid's close approach, large size, and extremely slow rotation provided an outstanding radar opportunity and we observed 1999 JM8 at Goldstone and Arecibo on 18 days between July 18 and August 9, 1999. Orbit solution JPL \#15, used for the initial radar detection
at Goldstone on July 18, was very good due to the 9 -year arc of optical astrometry. Due to problems with the delay-Doppler data acquisition system, on July 18-19 we obtained only CW (continuous wave; i.e. Doppler-only) echoes. On July 20 we started with 2 CW transmit-receive cycles (runs), measured a Doppler correction, and then estimated the range with coarseresolution 10 and $11 \mu \mathrm{~s}$ ( 1500 m and 1650 m resolution) setups. We completed that 70 minute set of observations (track) with two imaging runs that resolved the target into about twenty $150-\mathrm{m}$ range cells. We updated the orbit solutions several more times during the experiment; after July 20 range drift due to the ephemeris was imperceptible. Table 2 summarizes the Goldstone and Arecibo observations.

After July 20, our strategy during the Goldstone tracks was to do one or two CW runs to verify that we had echoes and then devote the rest of the track to the finest resolution imaging permitted by the signal-to-noise ratio. For the final Goldstone tracks on August 7 and 8, the asteroid was $\sim 50 \%$ farther away than at the closest approach, so we used a coarser resolution of $0.25 \mu \mathrm{~s} x 0.075 \mathrm{~Hz}$. However, this still placed thousands of pixels on the target.

At Arecibo we used two imaging data acquisition systems: the Caltech Baseband Recorder (CBR), which was designed for observing pulsars and was made available to us by Stuart Anderson, and a new system (the Portable Fast Sampler, or PFS) that was then under development. We used both systems on each day, but the analyses reported here utilize the CBR data exclusively due to the modestly stronger signal-to-noise ratio (SNR) and the CBR's dualpolarization capability. The range resolution on all days was $0.1 \mu \mathrm{~s}$ and the frequency resolution varied from 0.010 Hz to 0.006 Hz depending on the date (Table 2).

## ASTROMETRY \& ORBIT REFINEMENT

Table 3 lists 1999 JM8 radar astrometry and best-fit residuals for a post-experiment orbit solution and Table 4 lists the estimated orbital elements and their uncertainties. Combined optical and radar observations from April 1990-December 2000 permit reliable computation of the orbit over the interval from 293 to at least 2907, one of the longest spans for any near-Earth asteroid. Here "reliable" means that the 3- $\sigma$ uncertainty in the epochs of close approaches is less than 10 days.

Table 5 shows approaches within 0.1 AU of Earth, Mars, Ceres, and Vesta and within 1.0 AU of Jupiter during that interval. There are multiple encounters with Earth, Mars, Ceres, Vesta, and Jupiter, but in the next millennium all the encounters are with Earth, none closer than that in 1999. The impact probability through 2907 is effectively zero.

## DISC-INTEGRATED PROPERTIES

Our methods of radar data reduction and analysis follow those described in detail by Ostro et al. (1992, 1996). In Doppler-only observations, echoes were received simultaneously in the opposite (OC) and same (SC) senses of circular polarization as the transmission. $\sigma_{\mathrm{OC}}$ is the OC radar cross section; uncertainties in $\sigma_{\mathrm{OC}}$ are dominated by systematic pointing and calibration errors that are typically between 20 and $50 \%$. The circular polarization ratio $\mathrm{SC} / \mathrm{OC}$ is a gauge
of near-surface roughness at spatial scales near an order of magnitude of the radar wavelength ( 12.6 cm at Arecibo and 3.53 cm at Goldstone). For SC/OC, systematic effects cancel and most remaining statistical errors propagate from receiver thermal noise.

Table 6 summarizes 1999 JM8's 3.5 cm disc-integrated properties and Fig. 1 shows a collage of CW spectra obtained at Goldstone on each day. The 3.5 cm cross section varies significantly from day-to-day and we obtain an average of $2.5 \mathrm{~km}^{2}$, to which we assign an uncertainty of $35 \%$. During the course of the experiment the bandwidths increased from about 1.5 Hz to 3.4 Hz , suggesting a more equatorial view on later dates and/or an irregular shape. 1999 JM8's average $\mathrm{SC} / \mathrm{OC}=0.19 \pm 0.01$ is less than the median of $\sim 0.28$ estimated for all radar-detected NEAs, so the the object's near-surface roughness is somewhat less than average.

## DELAY-DOPPLER IMAGES

Figure 2 shows a chronological sequence of OC delay-Doppler images obtained between July 20 and August 9. Each frame has the same range and radial velocity dimensions. Most of the images are sums of all the highest resolution runs on each day. Due to the well-known northsouth ambiguity inherent with delay-Doppler images, we do not know in which hemisphere individual features occur.

Table 7 gives the asteroid's visible range extents and bandwidths measured from the images on each day. The maximum extents are 5.3 km on August 6 and 3.6 Hz (when adjusted to Goldstone's frequency of 8560 MHz ) on August 5. The Arecibo images generally show larger
range extents than the Goldstone images, perhaps due to the greater sensitivity at Arecibo and/or due to differences in the asteroid's orientation. On August 1, the only day of overlap between the two observatories, the visible range extent in the Arecibo image is about 0.3 km deeper. The images have an average visible range extent of 3.6 km . Based on our experience with shape inversions of other objects, the visible range extent typically is about one-half of the true range extent; if so, then 1999 JM8's true range extent is about 7 km .

Prominent features in the images appear to rotate by about $50^{\circ}$ from day-to-day between July 31 and August 9, suggesting that 1999 JM8 had an apparent rotation period of about one week and that the subradar latitude was within a few tens of degrees of zero during that interval. Images obtained on July 24, August 1, and August 8 show very similar surface features, suggesting that the rotation phases were nearly the same and offering additional support for an apparent rotation period of about seven days. We searched for evidence of rotation among images obtained on the same day and found it on several days (e.g., on August 5 and 8 , the longest Arecibo and Goldstone tracks; Fig. 3) at rates consistent with those seen from day to day.

The rotation period evident in the delay-Doppler images is somewhat longer than the 5.7-d estimate obtained by Pravec et al. The difference could be due to the effects of the large solar phase angles between $85-120^{\circ}$ on dates when the lightcurves were obtained (1999 July 3-21) and the asteroid's irregular shape, sky motion, and the rotation state (which is discussed below). Given that we can see rotation directly in the delay-Doppler images, in the analysis below we adopt a rotation period of 7 days as our nominal rotation period estimate. A more precise estimate of the spin state will require inversion of the delay-Doppler images and lightcurves.

Because the July 31-August 9 image sequence is strikingly similar to what we'd expect for a normal rotator viewed close to its equatorial plane, we adjusted the images to the same delay and Doppler scales and aligned them by hand to construct an estimate of 1999 JM8's pole-on silhouette (Fig. 4). The silhouette's elongation is about 1.15, a value that ranks near the lower end of the distribution of radar-derived NEA elongations, which have a mean and rms dispersion of $1.6 \pm 0.4$ (Ostro et al. 2001).

## SPIN STATE

Is 1999 JM8's rotation principal axis (PA) or non-principal axis (NPA)? Let us pretend that it is principal axis and then examine the images to see if this assumption is valid. The apparent rotation vector $\mathbf{W}_{\text {app }}$ is the vector sum of the intrinsic rotation $\mathbf{W}_{\text {int }}$ and the contribution due to sky motion $\mathbf{W}_{\text {sky }}$. Figure 5 shows that the angular rate of motion varied from a minimum of about $2^{\circ}$ per day to a maximum of about $7^{\circ}$ per day. Between July 20-28, the appearance of the asteroid changed substantially from day to day (it is difficult to identify the same features on adjacent days), indicating considerable apparent daily rotation. Between July 31 and August 9 we observe about $50^{\circ}$ of rotation per day, so $W_{\text {int }}$ dominates over $W_{\text {sky }}$ and $W_{\text {app }} \sim W_{\text {int }}$.

Delay-Doppler images obtained on July 24 and August 1, days in which the sky motion was about $4^{\circ}$ and $7^{\circ}$, have very similar orientations but their bandwidths differ by about a factor of two:

$$
B_{\mathrm{jul} 24} / B_{\mathrm{aug} 1}=B_{1} / B_{2}=(1.7 \pm 0.15 \mathrm{~Hz}) /(3.3 \pm 0.15 \mathrm{~Hz})=0.52 \pm 0.05
$$

If the spin is PA , this bandwidth change was due to the change in subradar latitude $\delta$ from July 24 to August 1. Expressing the apparent rotation period $P$ in hours and the diameter $D$ in km gives the 8560 MHz bandwidth $B$ in $\mathrm{Hz}: B=100 D \cos \delta / P$, so the ratio of the bandwidths $B_{1} / B_{2}=\cos \delta_{1} / \cos \delta_{2}=0.52$ could be caused by a change in $\cos \delta$ due to about $50^{\circ}$ of sky motion between the two days. Furthermore, $\left|\delta_{2}\right| \geq 0^{\circ}$ and $\left|\cos \delta_{2}\right| \leq 1$, so $\left|\cos \delta_{1}\right| \leq 0.52$ and $\left|\delta_{1}\right| \geq 59^{\circ}$.

However, visual inspection of features in the July 24 and August 1 images suggests that the orientations of 1999 JM8 on those days differ by less than $10^{\circ}$ of latitude. To quantify the latitude difference $\left|\delta_{1}-\delta_{2}\right|$, we measured the locations in range relative to the leading edge of several features that are visible in images on both July 24 and August 1. The features differ in range by $\sim 0.5 \mu \mathrm{~s}$ to $\sim 1 \mu \mathrm{~s}$, which is 1-2 range pixels at the July 24 resolution, indicating that the displacement in latitude is small. We quantified the displacement further by computing the latitude difference that is implied by $0.15 \mathrm{~km}(1 \mu \mathrm{~s})$ of displacement on a sphere $\sim 7 \mathrm{~km}$ in diameter. We are convinced that $10^{\circ}$ is a conservative number unless the subradar latitude on July 24 was within a few degrees of the pole, which is impossible because that would imply an object larger by a factor of several than the one observed.

Let us adopt the upper limit on the change in $\left|\delta_{1}-\delta_{2}\right|$ of $10^{\circ}$ and explore its implications:

First

$$
\cos \delta_{1}=(0.52) \cos \delta_{2}
$$

Then

$$
\cos \left(\delta_{2}+10^{\circ}\right)=(0.52) \cos \delta_{2}
$$

After applying a trigonometry identity and some algebra we obtain:
So

$$
\begin{aligned}
\sin \delta_{2} / \cos \delta_{2}=\tan \delta_{2} & =\left(\cos 10^{\circ}-0.52\right) / \sin 10^{\circ} \\
\delta_{2} & =69.5^{\circ}
\end{aligned}
$$

Therefore: $\delta_{1}=\delta_{2}+10^{\circ}=79.5^{\circ}$

That is, principal axis rotation requires that the subradar latitude had an absolute value of at least $79.5^{\circ}$ on July 24. If $\left|\delta_{1}\right| \geq 79.5^{\circ}$ on July 24 , then the $\sim 7 \mathrm{~d}$ rotation period evident in the images and the bandwidth of 1.7 Hz constrains the diameter:

$$
\begin{aligned}
& D \geq(1.7 \mathrm{~Hz})(\sim 7 \mathrm{~d})(24 \mathrm{~h} / \mathrm{d}) /\left(100 \cos 79.5^{\circ}\right) \\
& D \geq 15.7 \mathrm{~km}
\end{aligned}
$$

However, the visible range extents, which presumably show about one-half of the true range extent, average only 3.6 km (Table 7), which is inconsistent with $D \geq 15.7 \mathrm{~km}$. Thus, the assumption of principal axis rotation leads to a contradiction and we are forced to conclude that 1999 JM8 is a non-principal axis rotator.

We also conducted a search for principal axis spin states in which sky motion was explicitly included. We searched for spin state/diameter combinations that match the observed bandwidths and produce similar longitudes on July 24, August 1, and August 8, days when the orientation of 1999 JM8 is very similar. The search covered the entire sky at $5^{\circ}$ intervals, rotation periods between 5.0 and 18.0 days at intervals of 0.1 days, diameters between $3.5-10.0 \mathrm{~km}$ in 0.1 km increments, and the search assumed that 1999 JM8 is spherical.

We found that principal axis spin states fit the observations only if the absolute values of the
subradar latitudes are about $55-63^{\circ}$ on July 24 , about $15-25^{\circ}$ on August 1 , and $5-15^{\circ}$ (and on the opposite side of the equator relative to the other two days) on August 8. That is, in order for principal axis rotation to fit the observations, the subradar latitudes on July 24 and August 1 must differ by $30-40^{\circ}$. However, that contradicts the striking similarities seen in the images, which indicate a latitude difference on those days of much less than $30^{\circ}$.

Stated more succinctly, the July 24 and August 1 images clearly show the same side of 1999 JM8 but have bandwidths that differ by nearly a factor of two. Given that $W_{\text {int }}$ dominates over $W_{\text {sky }}$, the position of the spin vector in the asteroid had to be different on the two days; that is, the spin must be NPA. On the other hand, the apparent regular rotation of features in images between July 31-August 1 suggests that $W_{\text {int }}$ has a period of about one week and that the rotation state isn't far from principal axis rotation. PA rotation would be admissible if the latitude difference between July 24 and August 1 was $30-40^{\circ}$, so the upper bound of $10^{\circ}$ on the difference in latitudes suggests that the spin axis moved by at least $20^{\circ}$ in seven days. Refined estimates of the spin state will require shape inversion, which is beyond the scope of this paper.

Several other slowly rotating asteroids are suspected of being NPA rotators. 4179 Toutatis is in a well-defined NPA state (Hudson and Ostro 1995) and NPA rotation is strongly suspected for 253 Mathilde (Mottola et al. 1995), 288 Glauke (Harris et al. 1999), 3288 Seleucus (Harris et al. 1999), 4486 Mithra (Ostro et al. 2000), and 1999 GU3 (Pravec et al. 2000).

## SURFACE FEATURES

The sequence of daily images from July 31-August 9 show a clear progression of familiar prominent features. 1999 JM8 has an irregular, asymmetric shape characterized by regions of pronounced topographic relief, prominent facets several kilometers in extent, at least one large concavity, and numerous smaller concavities. On July 23, 24, August 1, 2, 8, and 9, the leading edge is rounded, suggesting a spheroidal shape at those aspects. In contrast, on July 20, 27, 28, August 5, and 6, the leading edge is more angular and the July 20 image is almost triangular.

On several days the leading edges show relative topographic relief of up to several hundreds of meters. For example, the July 27 leading edge shows a pronounced "peak" extending $\sim 400 \mathrm{~m}$ toward the radar. The July 28 leading edge has a "valley" that is $100-300 \mathrm{~m}$ more distant in range than the two adjacent "hills."

There is a large, nearly flat feature evident on the leading edges of the July 31 and August 1 images. On July 31 the feature extends across the middle of the leading edge and on August 1 it is on the right (receding side), with a range extent of at least 2 km . The August 4-6 images show a prominent, nearly flat region on the leading edge that, as seen in the August 6 image, has a range extent of at least 5 km .

A prominent relatively dark feature, apparently a $2-\mathrm{km}$-diameter concavity, is near the center of the trailing edge on July 24 , August 1, 2, 8 , and 9 . It is the largest concavity evident in the images.

Other circular to ellipsoidal and relatively dark features are probably impact craters; they have diameters ranging from about 100 meters to about 1 km . One of the smallest, near the center of the July 28 image, is surrounded by a relatively bright annulus that is reminiscent of the relatively bright ejecta deposits seen near impact craters in Arecibo delay-Doppler images of the Moon (Thompson et al. 1981) and Venus (Campbell et al. 1990). Two large, kilometer-sized crater-like structures are particularly prominent on July 24, 31, Aug. 1, 2, and 8.

## POLARIZATION SIGNATURE

Figure 6 shows daily sums of SC, OC, and SC/OC images obtained at Arecibo. The images show only those pixels in which the echo power in both polarizations exceeds 3-standard deviations. Each ratio image shows a region of relatively low SC/OC $\sim 0.1$ at the echo's leading edge and a general pattern of increasing SC/OC (to $\geq 0.5$ ) as a function of increasing range toward the trailing edge. This pattern is similar to that seen in SC/OC images from Toutatis (Ostro et al. 1999). Low SC/OC at the leading edge reveals a smooth, specularly reflecting surface that preferentially returns OC echo near normal incidence. We also investigated thresholds of 5 and 10 standard deviations and found that although the number of points decreases with each increase in the threshold, the patterns in the distribution of SC/OC do not change significantly.

Near the trailing edge of the August 2 image is an ellipsoidal region of about 200 pixels with lower SC/OC than its surroundings. We filtered the image with a $10 \times 10$ pixel boxcar and found
that $\mathrm{SC} / \mathrm{OC}=0.08 \pm 0.01$ within the region and $0.24 \pm 0.01$ at more positive and negative Doppler frequencies in the same span of range gates. The ellipsoidal structure is within an oval region that is relatively bright in both the SC and OC images. Its origin is not clear, and due to the north-south ambiguity, it is possible that there are contributions to the SC/OC difference from both hemispheres. One plausible explanation is that this may be a crater wall oriented at a low incidence angle that gives more specular reflections than adjacent regions. This may be evidence for a polarization ratio feature, which, if true, would be the first observed on an asteroid. There are also suggestions of narrow regions of lower SC/OC on August 1,2,3, and 5 adjacent to arcuate features that may be crater rims.

## DISCUSSION

If, as we suspect, 1999 JM8's effective diameter is $\sim 7 \mathrm{~km}$, then the absolute magnitude of 15.15 (Table 1) corresponds to a very low optical geometric albedo $p_{\mathrm{v}}=0.02$. This albedo and the optical spectrum strongly suggests that 1999 JM8 is a P-class object. The average Goldstone radar cross section, $2.49 \mathrm{~km}^{2}$, corresponds to a radar albedo of 0.06 , an estimate that overlaps the radar albedos for C-, S-, and BFGP-type main-belt and near-Earth asteroids.

How did the NPA rotation originate? Perhaps 1999 JM8 is a collisional fragment that was excited into NPA rotation during its dispersal from a larger progenitor, either directly into an NPA rotation state (Giblin and Farinella 1997, Asphaug and Scheeres 1999) or due to gravitational interactions with other fragments and/or the parent body (Scheeres et al. 2000). Alternatively, perhaps the NPA rotation was caused by an impact into the asteroid or by
gravitational torques during one or more very close passes by Earth or another planet (Scheeres et al. 2000). The presence of at least three kilometer-sized concavities that appear to be impact craters is consistent with the hypothesis that the NPA rotation was induced by impacts, but the concavities do not rule out the other mechanisms. Perhaps a combination of these mechanisms is responsible.

Another viable explanation is that the NPA rotation could be the result of (or was modified by) outgassing if 1999 JM8 was once a comet. There is a precedent for this conjecture: comet Halley, which is known to be an NPA rotator (Belton 1990; Belton et al. 1991). 1999 JM8's timescale for damping to principal axis rotation (Harris 1994) exceeds the age of the solar system, so if the NPA rotation was caused by cometary outgassing, it could still be in that state after cometary activity ceased. 1999 JM8's optical albedo is also consistent with the value of ~0.04 estimated for comet Halley (Delamere et al. 1986, Sagdeev et al. 1986), although some comets have albedos as large as $\sim 0.1$.

However, no cometary activity was seen during the 1999 apparition despite extensive spectrophotometric observing campaigns. The most reliable cometary radar albedo available is the estimate of $\sim 0.04$ for IRAS-Araki-Alcock (Harmon et al. 1989), a result that is comparable to our estimate for 1999 JM8. However, the nominal radar albedo of 1999 JM8 is also consistent with those estimated for primitive B-, F-, G-, and P-type main-belt asteroids (Magri et al. 1999). The orbit of $1999 \mathrm{JM} 8\left(a=2.72 \mathrm{AU}, e=0.644, i=13.7^{\circ}\right)$ has a Tisserand criterion $=2.988$ that is consistent with an origin as a Jupiter-family comet. Still, many asteroids have comparable Tisserand values, so the Tisserand criterion is not compelling evidence for a cometary origin
(Valsecchi et al. 1995). Applying the Bottke et al. (2001) dynamical analysis to 1999 JM8, W. F. Bottke (pers. comm.) estimated a probability of about $8 \%$ that 1999 JM8 is a Jupiter-family comet. Thus, although the evidence favors an origin as a primitive, outer main-belt asteroid, an origin as a comet nucleus cannot be excluded.

Given our images, it seems likely that inversion of the delay-Doppler images can improve constraints on the asteroid's shape significantly and define its spin state, following the example of 4179 Toutatis (Hudson and Ostro 1995). The next radar opportunity is in 2008 when 1999 JM8 will approach within 0.315 AU of Earth. Estimated SNRs during that apparition could approach a few hundred per day and be adequate to refine the spin state.

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TABLE 1. Optically-derived physical properties

| Property | Value | Reference |  |
| :--- | :---: | :---: | :---: |
|  |  |  |  |
| $H$ (mag) | 15.15 | 0.10 | 3 |
| $G$ | -0.09 | 0.02 | 3 |
| Period (d) | 5.7 | 0.2 | 3 |
| $\Delta m$ (mag) | 0.7 | 3 |  |
| Taxonomy | C or EMP | 1 |  |
|  | EMP | 2 |  |

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1. M. Hicks et al., pers. comm.
2. E. S. Howell, pers. comm.
3. P. Pravec et al., pers. comm.

TABLE 2. Observations.



GOLDSTONE AUGUST 8

| 0.25 | x | 0.075 | 63.0 | 3.5 | 0.087 | 0.93 | 26 | 50 | $115342-142242$ |
| ---: | ---: | ---: | :--- | :--- | :--- | :--- | :--- | ---: | ---: |
| 0.25 | x | 0.075 |  |  |  |  | 26 | 53 | $142939-170807$ |
| 0.25 | x | 0.075 |  |  |  |  | 26 | 6 | $171246-172906$ |
| 0.25 | x | 0.075 |  |  |  |  | 26 | 36 | $173238-191931$ |

## ARECIBO AUGUST 1



## ARECIBO AUGUST 2

| CW | 74.9 | 28.4 | 0.061 | 0.34 | 23 | 1 | 123810-123903 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $0.1 \times 0.0094$ |  |  |  |  | 23 | 28 | 130204-135658 |
| ARECIBO AUGUST 3 |  |  |  |  |  |  |  |
| CW | 72.0 | 23.2 | 0.064 | 0.36 | 23 | 2 | 121929-122235 |
| $0.1 \times 0.0089$ |  |  |  |  | 23 | 31 | 124312-134900 |
| ARECIBO AUGUST 4 |  |  |  |  |  |  |  |
| $0.1 \times 0.0083$ | 69.5 | 18.3 | 0.068 | 0.04 | 23 | 5 | 132600-133504 |
| ARECIBO AUGUST 5 |  |  |  |  |  |  |  |
| CW | 67.6 | 14.1 | 0.072 | 0.26 | 26 | 1 | 115436-115540 |
| $0.1 \times 0.0078$ |  |  |  |  | 26 | 35 | 115814-131950 |
| ARECIBO AUGUST 6 |  |  |  |  |  |  |  |
| $0.1 \times 0.0074$ | 65.9 | 10.2 | 0.076 | 0.08 | 26 | 13 | 123350-130414 |
| ARECIBO AUGUST 9 |  |  |  |  |  |  |  |
| $0.1 \times 0.0060$ | 62.1 | 1.3 | 0.091 | 0.14 | 26 | 25 | 104635-115923 |

Right ascension, declination, and delta are given at the midepoch of each day's observations. Motion indicates the plane-ofsky motion during each track. OSOD refers to the orbit solution computed using the JPL On-Site Orbit Determination software. Runs are the numberof transmit-receive cycles with each setup. Start and stop refer to the UTC epochs at the beginning and end of reception of echoes.

TABLE 3. Radar astrometry.

| UTC Epoch |  |  |  | $\begin{aligned} & \text { OSOD } \\ & \text { Soln } \end{aligned}$ | Correction | Measurement | +/- | resid. |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1999 | 07 | 18 | 21:30:00 | 15 | +1.7 Hz | 580912.4 Hz | 0.4 Hz | 0.175 |
| 1999 | 07 | 20 | 02:40:00 | 17 | $-0.5 \mathrm{~Hz}$ | 547278.4 Hz | 0.3 Hz | 0.125 |
| 1999 | 07 | 20 | 03:00:00 | 17 | -3132.5 $\mu \mathrm{s}$ | 92.6270694 s | $10.0 \mu \mathrm{~s}$ | -2.215 |
| 1999 | 07 | 21 | 17:40:00 | 19 | +0.8 Hz | 528308.3 Hz | 0.4 Hz | 0.439 |
| 1999 | 07 | 24 | 17:00:00 | 21 | $+0.5 \mathrm{~Hz}$ | 420950.7 Hz | 0.3 Hz | 0.163 |
| 1999 | 07 | 24 | 19:00:00 | 21 | -4 $\mu \mathrm{s}$ | 69.426792 s | $5.0 \mu \mathrm{~s}$ | 2.538 |
| 1999 | 07 | 27 | 01:30:00 | 21 | $+0.5 \mathrm{~Hz}$ | 264385.7 Hz | 0.3 Hz | -0.272 |
| 1999 | 07 | 27 | 02:30:00 | 21 | -20 $\mu \mathrm{s}$ | 61.40113 s | $5.0 \mu \mathrm{~s}$ | -1.574 |

Astrometry corresponds to echoes from 1999 JM8's estimated center of mass. The reference point for Goldstone is the intersection of the altitude and azimuth axes of the 70-meter antenna, DSS-14. Residuals are the remaining difference when the best-fit prediction of solution \#34 is subtracted from the actual measurements. The range equivalent of 1 usec is 150 meters and the radial velocity equivalents of 1 Hz are $17.6 \mathrm{~mm} / \mathrm{s}$ at Goldstone's transmitter frequency of 8560 MHz .

TABLE 4. Orbit.

| Quantity | Value | Uncertainty |
| :---: | :---: | :---: |
| Epoch | 2451911.5 JD (= 2001 | Jan 02.0) |
| Eccentricity (e) | 0.64440537378 | $\pm 0.0000000086$ |
| Perihelion distance (q) | 0.96733323427 AU | $\pm 0.0000000156 \mathrm{AU}$ |
| Perihelion date ( $T_{\mathrm{p}}$ ) | $\begin{aligned} & 2451383.7257358508 \text { JD } \\ & \text { (1999 Jul } 24.22574 \text { ) } \end{aligned}$ | $\pm 0.0000052119 \mathrm{~d}$ |
| Long. asc. node ( $\Omega$ ) | $134.00962375234^{\circ}$ | $\pm 0.0000052995^{\circ}$ |
| Arg. of perihelion ( $\omega$ ) | $165.98932252768^{\circ}$ | $\pm 0.0000095614^{\circ}$ |
| Inclination (i) | $13.7134598550^{\circ}$ | $\pm 0.0000029247^{\circ}$ |
| Semimajor axis (a) | 2.72032579502 AU | $\pm 0.0000000472 \mathrm{AU}$ |
| Period | $\begin{aligned} & 1638.81453203005 \mathrm{~d} \\ & (4.48675103853089 \mathrm{y}) \end{aligned}$ | $\pm 0.00004263 \mathrm{~d}$ |
| Mean anomaly | $115.93669166356^{\circ}$ | $\pm 0.0000030143^{\circ}$ |

1999 JM8's heliocentric orbital elements (OSOD solution \#34) and formal 1-standard deviation uncertainties, estimated using our delay-Doppler radar astrometry (Table 3) and currently available optical astrometry (403 angular measurements from April 29, 1990 to December 31, 2000). The mean post-fit radar residuals are: time-delay $-0.417 \mu \mathrm{~s} \pm 2.57 \mu \mathrm{~s}$, and Doppler frequency, +0.125 Hz $\pm .255 \mathrm{~Hz}$. Mean post-fit optical residuals are RA, -0.02" $\pm$ $0.54 "$, and declination, $-0.03 " \pm 0.63^{\prime \prime}$. The r.m.s. of residuals normalized by the assigned measurement uncertainty are (0.368, 0.732 , 0.589) for delay data, Doppler data, and total data set (including optical), respectively. Elements are in the coordinate frame of the JPL planetary ephemeris DE405 (ICRF93/ J2000, a quasar-based radio frame, generally within 0.01 arcseconds of the optical FK5/J2000 frame). Angular elements are referred to the ecliptic and mean equinox of J2000.

TABLE 5. Close approaches.

|  | Date | Body | Close-Approach nominal min AU AU |  | Distan <br> max <br> AU | e <br> $V_{\text {rel }}$ <br> km/s | $\begin{gathered} \Delta T \\ \min \end{gathered}$ | Nsigs |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 293 | Sep 19.76 | Jupiter | 0.9235 | 0.9009 | 0.9460 | 5.7 | 5079 | 3.4E6 |
| 412 | May 24.48 | Jupiter | 0.9636 | 0.9523 | 0.9748 | 6.0 | 3783 | 1.5E6 |
| 507 | May 4.44 | Jupiter | 0.9576 | 0.9370 | 0.9779 | 5.6 | 4290 | 7.4E5 |
| 584 | May 11.34 | Earth | 0.0599 | 0.0437 | 0.0812 | 13.3 | 6516 | 5.9E4 |
| 625 | Dec 22.81 | Jupiter | 0.9933 | 0.9837 | 0.0029 | 6.0 | 2621 | 1.7 E 6 |
| 756 | Aug 1.34 | Jupiter | 0.9963 | 0.9788 | 0.0136 | 5.7 | 2641 | 5.9E5 |
| 811 | May 11.51 | Earth | 0.0591 | 0.0340 | 0.0881 | 13.1 | 6987 | 2.2E5 |
| 1024 | Nov 22.16 | Mars | 0.0591 | 0.0591 | 0.0621 | 15.1 | 2211 | 3.3E5 |
| 1060 | May 10.83 | Earth | 0.0867 | 0.0864 | 0.0887 | 13.9 | 2974 | 9.2E5 |
| 1078 | Jun 7.22 | Mars | 0.0510 | 0.0384 | 0.0641 | 14.4 | 1507 | 6.0E4 |
| 1091 | Aug 18.89 | Earth | 0.0794 | 0.0759 | 0.0833 | 14.4 | 1039 | 3.5E5 |
| 1105 | Aug 28.53 | Ceres | 0.0883 | 0.0881 | 0.0885 | 14.1 | 58 | 3.2E4 |
| 1123 | Jul 1.11 | Vesta | 0.0814 | 0.0770 | 0.0857 | 12.6 | 353 | 5.9E4 |
| 1194 | Aug 24.21 | Earth | 0.0462 | 0.0454 | 0.0470 | 15.0 | 214 | 1.7E5 |
| 1256 | Mar 23.79 | Mars | 0.0692 | 0.0658 | 0.0728 | 16.1 | 462 | 1.6E5 |
| 1261 | Jun 18.77 | Vesta | 0.0771 | 0.0757 | 0.0785 | 13.7 | 275 | 1.7E5 |
| 1269 | Jun 4.02 | Mars | 0.0701 | 0.0690 | 0.0712 | 14.5 | 319 | 2.0E5 |
| 1318 | May 3.59 | Mars | 0.0919 | 0.0914 | 0.0926 | 14.4 | 565 | 4.4E5 |
| 1412 | Aug 25.85 | Earth | 0.0459 | 0.0455 | 0.0463 | 15.1 | 54 | 1.3E4 |
| 1474 | Aug 21.81 | Earth | 0.0546 | 0.0545 | 0.0547 | 14.0 | 22 | 1.9E5 |
| 1692 | Aug 9.95 | Earth | 0.0963 | 0.0963 | 0.0964 | 13.4 | 15 | 7.6E5 |
| 1981 | Aug 20.29 | Earth | 0.0665 | 0.0665 | 0.0665 | 13.8 | 6 | 5.6E5 |
| 1990 | Aug 8.41 | Earth | 0.0335 | 0.0335 | 0.0335 | 12.7 | 2 | 6.5E5 |
| 1999 | Jul 30.40 | Earth | 0.0568 | 0.0568 | 0.0568 | 12.3 | 0 | 5.4E6 |
| 2137 | Aug 1.53 | Earth | 0.0764 | 0.0764 | 0.0764 | 13.3 | 6 | 1.5E6 |
| 2573 | Aug 11.86 | Earth | 0.0852 | 0.0852 | 0.0852 | 14.0 | 5 | 8.6E5 |
| 2791 | Aug 15.18 | Earth | 0.0947 | 0.0947 | 0.0947 | 14.0 | 15 | 4.3E5 |
| 2831 | Jul 26.37 | Earth | 0.0715 | 0.0715 | 0.0715 | 14.0 | 12 | 2.4E5 |
| 2907 | Jul 20.53 | Earth | 0.0911 | 0.0911 | 0.0911 | 14.2 | 3 | 6.9E4 |

Note: Close approaches within 0.1 AU of the given body except for Jupiter, which is indicated for approaches less than 1.0 AU , are listed along with nominal, 3-sigma minimum and maximum distances. $V_{\text {rel }}$ is the relative velocity at the nominal close approach, $\Delta T$ is the 3-sigma uncertainty in the epoch of close approach, and Nsigs is the number of standard deviations required for the uncertainty ellipse to intersect the close-approach body.

TABLE 6. Disc-Integrated 3.5-cm Radar Properties.

| Date | runs | FFTs | OC SNR | B <br> (Hz) | $\begin{gathered} \sigma_{\mathrm{OC}} \\ \left(\mathrm{~km}^{2}\right) \end{gathered}$ | $\begin{gathered} \mathrm{SC} / \mathrm{OC} \\ (+/-0.01) \end{gathered}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Jul 18 | 7 | 1270 | 940 | 1.5 | 2.96 | 0.19 |
| Jul 19 | 10 | 1576 | 540 | 1.3 | 2.46 | 0.21 |
| Jul 20 | 2 | 154 | 500 | 2.0 | 2.97 | 0.19 |
| Jul 21 | 1 | 140 | 830 | 1.7 | 3.54 | 0.22 |
| Jul 23 | 1 | 132 | 580 | 1.6 | 2.26 | 0.16 |
| Jul 24 | 2 | 162 | 1100 | 2.3 | 2.85 | 0.15 |
| Jul 27 | 2 | 216 | 520 | 2.6 | 1.08 | 0.18 |
| Jul 28 | 1 | 80 | 5700 | 2.9 | 3.02 | 0.17 |
| Jul 31 | 2 | 180 | 1400 | 3.3 | 1.94 | 0.22 |
| Aug 1 | 2 | 192 | 880 | 3.4 | 1.81 | 0.23 |
| Experiment average |  |  |  |  | 2.49 | 0.19 |

The bandwidths were estimated using CW spectra with a resolution of $0.122 \mathrm{~Hz} . \quad B$ is the echo bandwdith. The cross sections and SC/OC were estimated using a frequency resolution of 1.95 Hz in order to have enough fast-Fourier transform (FFTs) to approach Gaussian noise statistics.

TABLE 7. Delay-Doppler dispersions.

| Date |  | RESOLUTION |  |  | DELAY EXTENT | (km) | BANDWIDTH <br> (Hz) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | $\mu \mathrm{s}$ | x | Hz |  |  |  |
| GOLDSTONE (3.5 cm) |  |  |  |  |  |  |  |
| Jul |  | CW |  | 0.061 | no data |  | 1.3 |
| Jul | 20 | 1.0 | x | 0.1 | 2.85 |  | 1.8 |
| Jul | 21 | 1.0 | x | 0.1 | 2.70 |  | 1.6 |
| Jul | 23 | 0.5 | x | 0.075 | 2.55 |  | 1.43 |
| Jul | 24 | 0.5 | x | 0.075 | 4.28 |  | 1.65 |
| Jul | 27 | 0.25 | x | 0.05 | 3.26 |  | 2.40 |
| Jul | 28 | 0.25 | x | 0.05 | 3.90 |  | 2.65 |
| Jul | 31 | 0.125 | x | 0.05 | 3.45 |  | 3.35 |
| Aug | 1 | 0.125 | x | 0.05 | 4.16 |  | 3.05 |
| Aug | 7 | 0.25 | x | 0.075 | 2.66 |  | 3.15 |
| Aug | 8 | 0.25 | x | 0.075 | 3.30 |  | 3.38 |
| ARECIBO (12.6 cm) |  |  |  |  |  |  |  |
| Aug | 1 | 0.1 | x | 0.0098 | 4.50 |  | 3.31 |
| Aug | 2 | 0.1 | x | 0.0094 | 4.13 |  | 3.17 |
| Aug | 3 | 0.1 | x | 0.0089 | 3.71 |  | 3.17 |
| Aug | 4 | 0.1 | x | 0.0083 | 3.47 |  | 3.24 |
| Aug | 5 | 0.1 | x | 0.0078 | 4.32 |  | 3.63 |
| Aug | 6 | 0.1 | x | 0.0074 | 5.27 |  | 3.20 |
| Aug | 9 | 0.1 | x | 0.0060 | 3.96 |  | 3.34 |

Estimated dispersions include pixels with echo power above the 2-sigma level, except on July 27,28 , and 31 , when contiguous pixels with SNRs > 1.0 were used. July 24 and August 1 estimates include the distant arc of pixels at the trailing edge that Arecibo images indicate are real. Arecibo bandwidths have been multiplied by 8560/2380 (the ratio of the Goldstone and Arecibo transmitter frequencies) to facilitate comparison with 3.5 cm results.

## FIGURE CAPTIONS

Fig. 1. Weighted sums of Goldstone echo power spectra grouped by observation date between July 18 (top left) to August 1 (bottom right). The spectra have been smoothed to a resolution of 0.5 Hz .

Fig. 2. Sequence of OC delay-Doppler images obtained at Goldstone and Arecibo. In each image range increases from top to bottom and Doppler frequency increases from right to left, so rotation is clockwise. The height in each frame is 6.0 km ( 40 usec ). The images are shown with a Doppler extent of 3.7 Hz when adjusted to a frequency of 8560 MHz in order to facilitate direct comparison between Goldstone and Arecibo images. The images have logarithmic contrast stretches in order to take advantage of the dynamic range. The collage shows one image per day, where each frame is the sum of all the highest resolution images on a given day. On August 1 we imaged 1999 JM8 at both telescopes; the image shown was obtained at Arecibo.

Fig. 3. Difference images obtained on August 5 (left) and 8 (right), the longest tracks at Arecibo and Goldstone. Range increases from top to bottom and Doppler frequency increases from right to left, so rotation is clockwise. The August 5 (Arecibo) image shows the difference between the first (white) and last (black) runs on that day, which were obtained 1.3 hours apart. The August 8 (Goldstone) image shows the difference between the first 17 runs (white) and the last 14 runs (black), which were obtained 7.5 hours apart. Several degrees of rotation are evident in the August 8 image.

Fig. 4. Silhouette of 1999 JM8 using delay-Doppler images obtained between August 1-9 assuming principal axis rotation. Images were cut out from hardcopies and aligned by eye.

Fig. 5. Top: Right ascension and declination of 1999 JM8 at the mid epoch of each track. Observations at Goldstone and Arecibo are indicated with circles and crosses. Middle: Angular rate of sky motion as a function of day-of-year (DOY). Bottom: Angular separation of 1999 JM8 as a function of DOY relative to the first Goldstone track on DOY 199 (July 18).

Fig. 6. Arecibo SC (left), OC (middle), and SC/OC (right) images. The delay-Doppler extents and orientations are the same as in Fig. 2. SC/OC is plotted by adopting a detection threshold per pixel of three standard deviations in both SC and OC images. All other pixels are mapped to white. The color stretch is saturated at $\mathrm{SC} / \mathrm{OC}=0.5$ (black) in order to emphasize the most interesting regions of the dynamical range. An arrow points to a region with relatively low SC/OC on August 2

