DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION

E9CE Revision 19 CONTINENTAL **TSIO**-360-A, -AB, -B, -C, -CB, -D, -DB, -E, -F, -G, -GB, -H, -HB, -EB, -FB, -JB, -KB, -LB, -BB, -MB, -NB, -PB, -RB, -SB **LTSIO**-360-E, -EB, -KB, -RB November 21, 2007

TYPE CERTIFICATE DATA SHEET NO. E9CE

Engines of models described herein conforming with this data sheet (which is part of type certificate No. E9CE) and other approved data on file with the Federal Aviation Administration meet the minimum standards for use in certificated aircraft in accordance with pertinent aircraft data sheets and applicable portions of the Federal Aviation Regulations provided they are installed, operated and maintained as prescribed by the approved manufacturer's manuals and other approved instructions.

Type Certificate Holder	Teledyne Continental Motors P.O. Box 90 Mobile, Alabama 36601		
Model	<u>TSIO-360-A, -AB, -B, -BB</u>	<u>TSIO-360-C, -CB</u>	<u>TSIO-360-D, -DB</u>
Туре	бНОА		
Rating, ICAO or ARDC, standard atmosphere Max. continuous hp, r.p.m., in. Hg at Critical altitude, ft. Sea level pressure altitude Takeoff hp, 5 min., rpm, full throttle at sea level pressure altitude	: 195-2800-32.0-20, 000 210-2800-32.0 210-2800-32.0	208-2800-37-20, 000 225-2800-37 225-2800-37	218-2800-36-20,000 225-2800-36 225-2800-36
Fuel (aviation gasoline, min. grade)	100 or 100LL		
Lubricating oil	TCM Spec. MHS No. 24		
Bore and stroke, in.	4.438 x 3.875		
Displacement, cu. in.	360		
Compression ratio	7.5:1		
Weight (dry), lb.	301 for -A, -AB 297 for -B, -BB	317	300

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Model	<u>TSIO-360-A, -AB, -B, -BB</u>	<u>TSIO-360-C, -CB</u>	<u>TSIO-360-D, -DB</u>
C.G. Location (basic engine) Fwd of front face accessory case, in.	7.63		
Below crankshaft centerline, in. Below crankshaft centerline	1.02		
toward 1-3-5 side, in.	.20		
Propeller shaft	ARP-502, Type I Flange 4 7/8" O.D. with six 1/2" bolt holes in 4" diameter circle		
Fuel injection	TCM Injector P/N 635798A3	 P/N 639289A11	 P/N 639289A12
Ignition, dual magnetos	Two TCM or Bendix S6LN-25 or two TCM S6LSC-25	Two TCM or Bendix S6LN-25 or TCM S6LSC-25 (R.H.) & S6LSC-25T (L.H.)	
Timing, °BTC	20		
Spark Plugs	(See NOTE 12)		
Oil sump capacity, qt.	10; 7 usable at 25° noseup and nosedown attitudes		
Notes	1 thru 7, 9 thru 12	1 thru 12	1 thru 7, 9 thru 12
Model	<u>TSIO-360-E, EB</u>	LTSIO-360-E, -EB	<u>TSIO-360-F, -FB</u>
Туре	6HOA		
Rating, ICAO or ARDC, standard atmosphere Max. continuous hp, r.p.m., in. Hg a Critical altitude, ft. Sea level pressure altitude Takeoff hp, 5 min., rpm, full throttle at sea level pressure altitude	t: 215-2575-40-12,000 200-2575-40 200-2575-40	215-2575-40-12,000 200-2575-40 200-2575-40	200-2575-41-12,000 200-2575-41 200-2575-41
Fuel (aviation gasoline, min. grade)	100 or 100LL		
Lubricating oil	TCM Spec. MHS No.24		
Bore and stroke, in.	4.438 x 3.875		
Displacement, cu. in.	360		
Compression ratio	7.5:1		
Weight (dry), lb.	352	352	359

Model	<u>TSIO-360-E, EB</u>	<u>LTSIO-360-E, -EB</u>	<u>TSIO-360-F, -FB</u>
C.G. Location (basic engine) Fwd of front face accessory			
case, in. Below crankshaft centerline, in. Below crankshaft centerline	6.746 1.555	6.746 1.555	6.714 1.478
toward 1-3-5 side, in.	.476	.476	.757
Propeller shaft	ARP-502, Type I Flange 4 7/8" O.D. with six 1/2" bolt holes in 4" diameter circle		
Fuel injection	TCM Injector P/N 639289A17	 P/N 639289A18	P/N 639289A17
Ignition, dual magnetos	Two TCM or Bendix S6LN-25 or two TCM S6LSC-25 or two TCM S6LSC-25P Note 17	Two TCM or Bendix S6RN-25 or two TCM S6RSC-25 or two TCM S6RSC-25P Note 17	Two TCM or Bendix S6LN-25 or two TCM S6LSC-25 or two TCM S6LSC-25P
Timing, °BTC	20		
Spark Plugs	See NOTE 12		
Oil Sump capacity, qt.	8; 5 useable at 25° noseup and 5 usable at 18° nosedown		
Notes	1 thru 6, 9, 10, 12, 13 and 17		1 thru 6, 9, 10, 12, and 14
Model	<u>TSIO-360-G, -GB</u>	<u>TSIO-360-H, -HB</u>	<u>TSIO-360-JB</u>
Туре	6HOA		
Rating, ICAO or ARDC, standard atmosphere Max. continuous hp, r.p.m., in. Hg at: Critical altitude, ft.	210-2700-40-15,000	195-2800-34.5-20,000	210-2800-37.0-
20,000 Sea level pressure altitude Takeoff hp, 5 min., rpm, full throttle at sea level pressure altitude	210-2700-40 210-2700-40	210-2800-34.5 210-2800-34.5	225-2800-37.0 225-2800-37.0
Fuel (aviation gasoline, min. grade)	100 or 100LL		
Lubricating oil	TCM Spec. MHS No. 24		
Bore and stroke, in.	4.438 x 3.875		
Displacement, cu. in.	360		
Compression ratio	7.5:1		

Model	<u>TSIO-360-G, -GB</u>	<u>TSIO-360-H, -HB</u>	<u>TSIO-360-JB</u>
	350	313	313
Weight (dry), lb.	350	515	515
C.G. Location (basic engine) Fwd of front face accessory			
case, in. Below crankshaft centerline, in.	7.35 1.64	7.63 1.02	
Below crankshaft centerline			
toward 1-3-5 side, in.	.965	.20	
Propeller shaft	ARP-502, Type I Flange 4 7/8" O.D. with six		
	1/2" bolt holes in 4"		
	diameter circle		
Fuel injection	TCM Injector P/N 639289A15	 P/N 639289A16	 P/N639289A
Ignition, dual magnetos	Two TCM or Bendix S6LN-25 or two TCM	Two TCM or Bendix S6LN-25 or TCM	
	S6LSC-25 or two TCM S6LSC-25P Note 17	S6LSC-25 (R.H.) & S6LSC-25T (L.H.)	
Timing, °BTC	20		
Spark plugs	See NOTE 12		
Oil Sump capacity, qt.	8; 5 useable at 25° noseup and 5 usable		
	at 18° nosedown		
Notes	1 thru 6, 9 thru 12, 15	1 thru 7, 9 thru 12	
	and 17		
Model	<u>TSIO-360-KB</u>	LTSIO-360-KB	<u>TSIO-360-LB</u>
Туре	6HOA		
Rating, ICAO or ARDC,			
standard atmosphere Max. continuous hp, r.p.m., in. Hg at:			
Critical altitude, ft.	220-2800-40.0-14,500		210-2700-40-15,000
Sea level pressure altitude Takeoff hp, 5 min., rpm, full	220-2800-40.0 220-2800-40.0		210-2700-40 210-2700-40
throttle at sea level	220-2000-40.0		210-2700-40
pressure altitude			
Fuel (aviation gasoline, min. grade)	100 or 100LL		
Lubricating oil	TCM Spec MHS #24		
Bore and stroke, in.	4.438 x 3.875		
Displacement, cu. in.	360		

Model (cont'd)	TSIO-360-KB	LTSIO-360-KB	<u>TSIO-360-LB</u>
Compression ratio	7.5:1		
Weight (dry), lb.	359	359	354
C.G. Location (basic engine) Fwd of front face accessory case, in. Below crankshaft centerline, in. Below crankshaft centerline	6.746 1.555	6.746 1.555	7.35 1.64 .965
toward 1-3-5 side, in.	.476	.476	
Propeller shaft	ARP-502 Type I Flange 4 7/8" O.D. with six 1/2" bolt holes in 4" diameter circle.		
Fuel injection	TCM Injector P/N 639289A31	P/N 639289A32	P/N 639289A41
Ignition, dual magnetos	Two TCM or Bendix S6LN-25 or two TCM S6LSC-25 or two TCM S6LSC-25P Note 17	Two TCM or Bendix S6RN-25 or two TCM S6RSC-25 or two TCM S6RSC-25P Note 17	Two TCM or Bendix S6LN-25 or two TCM S6LSC-25 or two TCM S6LSC -25P Note 17
Timing, °BTC	20		
Spark Plugs	See NOTE 12		
Oil sump capacity, qt.	10; 7 useable at 25° noseup and nosedown attitudes		8; 5 usable at 26° noseup and 5 useable at 18° nosedown
Notes	1 thru 6, 9, 10, 12, 13 and 17		1 thru 6, 9 thru 12, 15 and 17
Model	<u>TSIO-360-MB</u>	<u>TSIO-360-NB</u>	<u>TSIO-360-PB</u>
Туре	6HOA		
Rating, ICAO or ARDC, standard atmosphere Max. continuous hp, r.p.m., in. Hg at: Critical altitude, ft. Sea level pressure altitude Takeoff hp, 5 min., rpm, full throttle at sea level pressure altitude	210-2700-36-22,000 210-2700-36 210-2700-36	210-2700-38-15,000 210-2700-38 210-2700-38	200-2600-37-18,000 200-2600-37 200-2600-37
Fuel (aviation gasoline, min. grade)	100 or 100LL		
Lubricating oil	TCM Spec. MHS No. 24		
Bore and stroke, in. 4.438 x 3.875			

Model (cont'd)	<u>TSIO-360-MB</u>	<u>TSIO-360-NB</u>	<u>TSIO-360-PB</u>
Displacement, cu. in.	360		
Compression ratio	7.5:1		
Weight (dry), lb.	354		
C.G. Location (basic engine) Fwd of front face accessory case, in. Below crankshaft centerline, in. Below crankshaft centerline	7.35 1.64		
toward 1-3-5 side, in.	.965		
Propeller shaft	ARP-502, Type I Flange 4 7/8" O.D. with six 1/2" bolt holes in 4" diameter circle.	;	
Fuel injection	TCM Injector P/N 639289A41		
Ignition, dual magnetos	Two TCM or Bendix S6LN-25P or two TCM S6LSC-25P Note 17		
Timing, °BTC	20		
Spark Plugs	See NOTE 12		
Oil sump capacity, qt.	8; 5 usable at 26° noseup and 5 usable at 18° nose down		
Notes	1 thru 6, 9, 10, 12, 17 and 18	1 thru 6, 9 thru 12, 15 and 17	1 thru 6, 9, 10, 12 17 and 18
Model	TSIO-360-RB	LTSIO-360-RB	TSIO-360-SB
Туре	6HOA		
Rating, ICAO or ARDC, standard atmosphere Max. continuous hp, rpm, in. Hg. at critical altitude, ft. Sea level pressure altitude Tkeoff hp, 5 min., rpm, full throttle at sea level pressure altitude	220-2600-38-19,000 220-2600-38 220-2600-38	220-2600-38-19,000 220-2600-38 220-2600-38	220-2600-39-19,000 220 2600-39 220-2600-39
Fuel (aviation gasoline, Min. grade)	100 or 100LL		
Lubricatin oil	TCM Spec. MHS 24		
Bore and stroke, in.	4.438 X 3.875		

Model	TSIO-360-RB	LTSIO-360-RB	TSIO-360-SB
Displacement, cu. in.	360		
Compression ratio	7.5:1		
Weight	357		354
CG Location (basic engine) Fwd of front face accessory case, in.	5.86		7.35
Below crankshaft centerline, in. Below crankshaft centerline toward 1-3-5 side, in.	1.26 0.56		1.64 0.965
Propeller shaft	ARP-502, Type In Flange 4 7/8" O.D. with six 1/2" bolt holes in 4" diameter circle		
Fuel Injection	Precision Airmotive RSA-5AD2		TCM Injector 639289A125
Ignition, dual magnetos	Two Unison Slick 6324 Note 19	Two Unison Slick 6320 Note 19	Two Unison Slick 6324 Note 19
Timing, °BTC	22	22	20
Spark plugs	See NOTE 12		
Oil sump capacity, qt.	8: 5 useable at 26° noseup and 5 useable at 18° nosedown		
Notes	1 thru 6, 9, 10, 12, 18, and 19		

"- - " indicates same as "preceding model."

"—" indicates "does not apply"

"E" designates both the Models TSIO-360-E and LTSIO-360-E unless otherwise specified.

"EB" designates both the Model & TSIO-360-EB and LTSIO-360-EB unless otherwise specified.

Certification basis Federal Aviation Regulations Part 33 effective February 1, 1965, as amended by 33-1.

Application for Type Certificate dated March 22, 1966.

Type Certificate No. E9CE issued October 11, 1966, for Models TSIO-360-A and TSIO-360-B; TSIO-360-C added January 19, 1972;

TSIO-360-D added June 7, 1973; TSIO-360-E added June 14, 1974; LTSIO-360-E added June 14, 1974; TSIO-360-F added June 24, 1976; TSIO-360-G added April 5, 1977; TSIO-360-H added September 1, 1977. TSIO-360-EB and LTSIO-360-EB added November 17, 1977; TSIO-360-FB added November 17, 1977; TSIO-360-CB added March 13, 1978; TSIO-360-DB added March 13, 1978; TSIO-360-GB added March 13, 1978; TSIO-360-HB added March 13, 1978; TSIO-360-JB added September 19, 1979; TSIO-360-AB added March 25, 1980; TSIO-360-KB and LTSIO-360-KB added July 30, 1980; TSIO-360-LB added March 9, 1982; TSIO-360-BB added July 29, 1982, TSIO-360-MB added October 11, 1985, TSIO-360-NB added December 10, 1986, TSIO-360-PB added October 21, 1996, LTSIO-360-RB added October 21, 1996 and TSIO-360-SB added February 27, 1997.

Production Basis Production Certificate No. 7, except Model -D, and/or Production Certificate No. 508.

NOTE 1.	- J	0°F 0°F (Not require -KB, -LB,	d on the L/TISO-360-E, -EB, -F, -FB, -MB, -NB, -PB, -RB, -SB.)
	Turbocharger inlet temperature (TIT): Max -CB, -D, -DB, -F, -FB, -G, -GB, -H, -HB, - continuous TIT is 1725°F for TSIO-360-E,	0°F imum continuous IB, -MB, -NB, -F -EB, -KB, -LB ar ated at 1700°F for	TIT is 1650°F for TSIO-360-A, -AB, -B, -BB, -C, PB, -RB, -SB and LTSIO-360-RB. Maximum nd LTSIO-360-E, -EB, -KB. The TSIO-360-MB, -PB, r a maximum of 60 seconds to determine peak TIT. The
NOTE 2.	Outlet to vapor return line plus 3.5 TSIO-360-C, -D, -E, -F, -G, -H, -EB, -FB, 1 -LB, -MB, -NB, -PB. Inlet to injection pump, minminus 2.5 max. plus 8 p	p.s.i.g. 5 p.s.i.g. (at T.O. LTSIO-360-E and p.s.i.g. p.s.i.g. p.s.i.g at 80 lb./ re into Pump ph	us 6.5 psi
NOTE 3.	Oil pressure limits:		
	2-4-6 side - Normal 30-60 p.s.i.g.; Except 3 and -SB. Idle 10 p.s.i.g. Max. allowable (cold oil) 100 p	•	EB, -F, -FB, -G, -GB, -KB, -LB, -MB, -NB, -PB, -RB
NOTE 4.	-		C, -CB, -D, -DB, -E, -EB, -F, -FB, -G, -GB, -H, -HB, 3, -NB, -PB, -RB, -SB
	Induction air max temperature T.O. and max continuous power 75 percent power Exhaust system max back pressure 1.5 in. from cylinder flange, in. Hg abs. 4 Turbine outlet, above ambient 2 in. Hg	250°F 250°F	280°F 280°F 280°F or -E, -EB, -F, -FB, -G, -GB, -KB, -LB, -MB, -NB, -PB,
NOTE 5			

NOTE 5. The following accessory drive or mounting provisions are available:

<u>Original</u> <u>Accessory</u>	Direction of Rotation *	Speed Ratio to Crankshaft	<u>Maximum</u> <u>Continuous</u> <u>Torque</u> (in. lb.)	Static torque (in. lb.)	<u>Maximum</u> Overhang Moment (in. lb.)
** Propeller governor	С	1:1	29	825	50
*** Vacuum Pump	CC	1.545:1	100	800	50

NOTE 5: Continued

			Maximum		
			<u>Continuous</u>		Maximum
<u>Original</u>	Direction of	Speed Ratio to	<u>Torque</u>	Static torque	Overhang Moment
Accessory	Rotation *	<u>Crankshaft</u>	<u>(in. lb.)</u>	<u>(in. lb.)</u>	<u>(in. lb.)</u>
Optional Kit - Freon compressor drive	С	1.545:1	110	400	
Optional					
**** (1-3-5 side)	С	1.316:1	27	800	8.3
***** (2-4-6 side)	CC	1.316:1	27	800	8.3
# Tachometer	CC	0.495:1			
## Generator (Alternator)	CC	2.035:1	60	600	50
Belt Drive Alternator					
TSIO-360-RB	CC	3.276:1	100	800	50
LTSIO-360-RB	С	3.276:1	100	800	50
Oil cooler					65

Starter: TCM P/N 627841 (Delco-Remy P/N 1108234), TCM 634592(Prestolite MCL-6501),

TCM 634433 (Prestolite MHJ 4002), TCM 646275 (24 volt), and TCM 646238 (12 volt) eligible.

* C - Clockwise viewing drive pad; CC - Counterclockwise.

** AND 20010 pad - Modified (speed increased).

*** AND 20000 pad - Modified (speed increased), -B only.

**** AND 20000 pad - Modified (no oil provision; accessory clearances limited), -A and -C, -AB.

***** AND 20000 pad - Modified (opposite rotation), -A, -C, and -AB.

Mechanical drive for TSIO-360-E, -EB, -F, -FB, -G, -GB, -KB, -LB only, rotation reversed for LTSIO-360-E, -EB, and -KB. The above values are the same for the LTSIO-360-E, -EB, -KB and -RB except the direction of rotation is reversed from that indicated.

L/TSIO-360-RB has optional gear driven alternator.

- - - - No data presented.

NOTE 6. The Model TSIO-360-B is similar to the TSIO-360-A except for accessory drive provisions.

The Model TSIO-360-C is similar to the TSIO-360-A except for rating, crankshaft dampers provision for cabin pressurization.

The Model TSIO-360-D is similar to the TSIO-360-C except for cabin pressurization.

The Model TSIO-360-E is similar to the TSIO-360-C except for rating, engine mounted turbocharger, freon compressor drive and cabin pressurization.

The Model LTSIO-360-E is similar to the TSIO-360-E except for direction of rotation.

The Model TSIO-360-F is similar to the TSIO-360-E except for rating and exhaust system configuration.

The Model TSIO-360-G is similar to the TSIO-360-E except for rating and exhaust system configuration.

The Model TSIO-360-H is similar to the TSIO-360-G except for rating and cabin pressurization.

The Model TSIO-360-EB is similar to the TSIO-360-E except for modified crankshaft.

The Model LTSIO-360-EB is similar to the LTSIO-360-E except for modified crankshaft.

The Model TSIO-360-FB is similar to the TSIO-360-F except for modified crankshaft.

The Model TSIO-360-CB is similar to the TSIO-360-C except for modified crankshaft.

The Model TSIO-360-DB is similar to the TSIO-360-D except for modified crankshaft.

The Model TSIO-360-GB is similar to the TSIO-360-G except for modified crankshaft.

The Model TSIO-360-HB is similar to the TSIO-360-H except for modified crankshaft.

The Model TSIO-360-JB is similar to the TSIO-360-HB except for the increased rating. The Model TSIO-360-AB is similar to the TSIO-360-A except for the modified crankshaft.

The Model TSIO-360-KB is similar to the TSIO-360-EB except for the increased rating.

The Model LTSIO-360-KB is similar to the LTSIO-360-EB except for the increased rating.

The Model TSIO-360-LB is similar to the TSIO-360-GB except for the larger throttle body, larger induction elbows, relocated overboost valve, magneto pressurization and an exhaust muffler. The Model TSIO-360-BB is similar to the TSIO-360-B except for modified crankshaft and different dampers. The Model TSIO-360-MB is similar to the TSIO-360-LB except for intercooler, wastegate controller and cluster manifold. The Model TSIO-360-NB is similar to the TSIO-360-LB except for intercooler. The Model TSIO-360-PB is similar to the TSIO-360-LB except for no intercooler. The Model TSIO-360-PB is similar to the TSIO-360-MB except for no intercooler. The Model TSIO-360-RB is similar to the TSIO-360-KB except the manifold pressure is controlled with a wastegate controller and an intercooler has been installed in the induction system. The Model LTSIO-360-RB is similar to the TSIO-360-RB except for direction of rotation. The Model TSIO-360-SB is similar to the TSIO-360-MB except for RPM and rating.

- NOTE 7. These engines are eligible for installation of TCM Eq. No. 6001 oil filter and adapter.
- NOTE 8. The Model TSIO-360-C and -CB supply approximately 4.5 lbs. of air per minute through a .5 inch diameter sonic venturi for cabin pressurization.
- NOTE 9. These engines are eligible for pusher or tractor operation.
- NOTE 10. Models TSIO-360-A and -B incorporate crankshafts with two 6th order dampers.
 Models TSIO-360-C, -AB, -CB, -D, -DB, -E, -EB, -F, -FB, -G, -GB, -H, -HB, -JB, KB, -LB, -MB, -NB, -PB, -BB, -RB, and -SB incorporate crankshafts with one 4 1/2 and one 6th order damper.
- NOTE 11. Required equipment:
 - (a) Unless otherwise substantiated by the installer, the turbocharger oil return line must be equipped with an oil air separator having a capacity of one pint maximum and capable of separating an air flow of 2 c.f.m. and oil flow of 15 lb./min. at 240°F.
 - (b) A turbocharger and exhaust system in accordance with or equivalent to TCM drawing 633737 for the -A and -AB, 633738 for the -B, 633737 for the -C, -CB, -D, and -DB.
 - (c) The -H engine was calibrated utilizing an AiResearch turbosupercharger Model TA0401, Cessna P/N C295001-0301, and TCM exhaust system outline Drawing 642374. This turbosupercharger has been found compatible with the -H, -HB, and -JB engines; however, the turbosupercharger and exhaust system are not included in this Type Certificate and must be found acceptable in accordance with FAR 23.909 at installation.
- NOTE 12. The following spark plugs are approved for these engines:

Engine Models	<u>Spark Plugs</u>	
TSIO-360-A, -B, -AB,	AC	SR83P, HSB83P, S86R, SR86, HSR86, SR93, HSR93
-BB, -C, -CB, -D, -DB,		
-E, -EB, -F, -FB, -G,	Auto Lite	PH26, PH260
-GB, -H, -HB, -JB,		
-KB, -LB, -MB, -NB,	Champion	REM38W, RHM38W, REM38P, RHM38P, REM38E, RHM38E
-PB, -RB, -SB		
	Red Seal	SE270, SE270P, SJ270, SJ270P
	~	
	Smith Industries	RSE23-3R, RSH23-3R
Also for the	0.4111.4	and down Hange Hangen
TSIO-360-A, -B,	Smith Industries	SR86, S86R, HSR86, HSR83P
-AB,-BB		

NOTE 13. The TSIO-360-E, -EB, -KB and LTSIO-360-E, -EB, -KB are equipped with engine mounted Rajay turbosupercharger Model 325E10-1 utilizing TCM exhaust system 640964. Compliance with FAR 23.909(c), effective with Amendments 23-7 has been shown.

- NOTE 14. The TSIO-360-F, -FB is equipped with engine mounted Rajay turbosupercharger Model 325E10-1 utilizing TCM exhaust system 641911. Compliance with FAR 23.909(c) effective with Amendment 23-7 has been shown.
- NOTE 15. The TSIO-360-G, -GB, -LB and -NB are equipped with engine mounted Rajay turbosupercharger Model 325E10-1 and the following TCM exhaust systems: TSIO-360-G, -GB1, -GB2, -GB3 (Exhaust System TCM 642220); TSIO-360-GB4B, -LB (Exhaust System TCM 646051); TSIO-360-NB (Exhaust System TCM 649202). Compliance with FAR 23.909(c) effective with amendment 23-7 has been shown.
- NOTE 16. Those engines which are designated with the suffix letter "B" (i.e., TSIO-360-EB) are interchangeable with those engines of the same model letter without the suffix letter (i.e., TSIO-360-E).

Those engines which are designated with the suffix letter (i.e., TSIO-360-E) are non-interchangeable with those engines which are designated with the suffix letter "B" (i.e., TSIO-360-EB).

- NOTE 17. These engines are eligible for installation of two Slick Electro Model 6214 magnetos (6210 for LTSIO) or two Unison Slick model 6314 (6310 for LTSIO) equipped with an appropriate harness, resulting in a weight change of -1 LB, or two Slick Electro Model 6224 magnetos (6220 for LTSIO) or two Unison Slick model 6324 (6320 for LTSIO) equipped with an appropriate pressurization manifold.
- NOTE 18. The TSIO-360-MB, -PB, RB, SB and LTSIO-360-RB are equipped with engine mounted AiResearch turbosupercharger Model TAO4 utilizing TCM exhaust systems.
- NOTE 19. These engines are eligible for installation of two TCM Model S6LSC-25P magnetos (S6RSC-25P for LTSIO) equipped with an appropriate harness and pressurization manifold, resulting in a weight change of +1.4 Lbs.

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